

CR 151456

INTERIM REPORT

18 May 1977

HIGH PERFORMANCE

N_2O_4 /AMINE ELEMENTS - "BLOWAPART"

{NASA-CR-151456} HIGH PERFORMANCE
N2O4/AMINE ELEMENTS: BLOWAPART Interim
Report (Aerojet Liquid Rocket Co.) 255 p
HC A12/MF A01 CSCL 21D

N77-29316

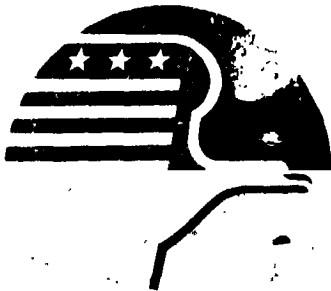
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Contract NAS 9-14186

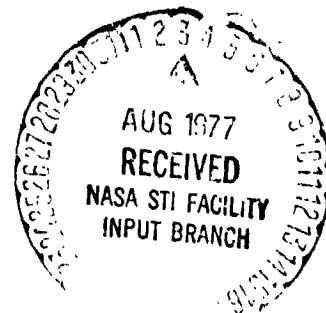
Report 14186-3.1.1

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FOREWORD

This interim report describes the analytical and experimental work conducted to identify the onset of reactive stream separation and develop techniques to predict its occurrence. The activity was performed by Aerojet Liquid Rocket Company on Contract NAS 9-14186, under the direction of Merlyn Lausten, NASA/JSC Project Manager. Aerojet personnel included L. B. Bassham, Program Manager, D. L. Kors, Project Manager and B. R. Lawver, Project Engineer.

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I INTRODUCTION AND SUMMARY

A. Introduction

The objective of this program was to develop an understanding of the mechanisms controlling hypergolic propellant reactive stream separation (RSS) and with that understanding develop design criteria which will allow the design of high performing injectors free from both steady state RSS and cyclic propellant stream separation ("pops"). This objective was accomplished through test and analysis of single element injectors using principally N_2O_4/MMH propellants. Injectors and test conditions were representative of the Space Shuttle Orbit Maneuvering Engine and Space Tug applications. The program consisted of four primary tasks:

Task I - Review of Existing Models and Experimental Data

The Task I objectives were to:

- Critique all existing models relating to RSS.
- Review and summarize all associated RSS data.
- Formulate updated RSS model based on existing data.

Task II - Detailed Program Plan Preparation

The Task II objective was the preparation of a detailed program plan, i.e., the formulation of a detailed method of approach to be taken for model testing and verification in subsequent tasks.

Task III - Definition of Governing Mechanisms

The Task III objectives were to define the mechanisms governing RSS, establish limits for RSS and "pops", and to define appropriate models and included:

I Introduction and Summary (cont.)

- Design, fabrication and testing of two single element unlike doublet injectors.
- Incorporation of high speed photographic techniques for visual propellant stream characterization.
- Correlation of propellant stream behavior (RSS) with various independent test variables (data analysis and evaluation).

Task IV - Verification of Governing Mechanisms

The Task IV objectives were to:

- Establish operating limits for RSS (and ("pops") for other injector types.
- Verify mechanisms governing RSS and the appropriate physical models resulting from the Task III work.

The Task IV effort included the design, fabrication and testing of triplet and self-atomizing platelet injectors. The term "self-atomizing" as used here means that a single propellant provides the atomization, either by self impingement or by impingement with a surface.

B. Summary

1. Task I - Model and Data Review

Results of the Task I model review effort indicated that neither of the two existing blowpart models (JPL model and

I Introduction and Summary (cont.)

ALRC model) adequately described RSS nor correlated the existing RSS experimental data. New correlations were developed using hydrazine data that showed that the chamber pressure, orifice diameter, and propellant temperature exhibit controlling influences over blowpart. Four regimes of reactive stream impingement were defined: (1) mixing, (2) popping (cyclic blowpart), (3) low pressure separation ($P_c \leq P_{\text{sat}_{\text{ox}}}$), and (4) high pressure separation ($P_c > 300$ psia).

New model concepts of reactive impingement were formulated on the basis of known hypergolic reaction mechanisms to account for all four modes. Hypergolic reaction was hypothesized to occur either as, high enthalpy gas phase oxidation reactions, low enthalpy surface reactions, fuel monopropellant gas phase reaction, or liquid phase reaction. However, subsequently true liquid phase reactions were deemed unlikely due to the reported immisibility of Amine fuels and N_2O_4 and the observed lack of liquid phase mixing produced by impinging streams of water and dyed water. Also, the fuel monopropellant gas phase reactions hypothesis was subsequently shown to be difficult to correlate with the observed RSS regimes. The mixing regime was postulated to be controlled by gas phase reaction in that mixing occurs under conditions of low gas phase concentrations. Popping was postulated as a transition from the low enthalpy surface reaction to the high enthalpy gas-phase reactions as observed for droplet ignition. The low pressure mode of RSS was found to be due to flash vaporization of N_2O_4 . Originally, the high pressure mode of RSS was postulated to be caused by recirculation gas heating of the fuel stream to its decomposition temperature ahead of impingement. However, this concept was subsequently revised on the basis of Task III and IV results.

2. Task II - Program Plan

The Task II Program Plan consisted of preparing a detailed work plan which included all aspects of program planning,

I Introduction and Summary (cont.)

i.e., program approach, hardware requirements, test plan and analytical requirements. This work was documented in Report 14186-DRL-1.

3. Task III - Definition of Governing Mechanisms

During the Task III testing the high pressure (300-1000 psia, $2-7 \times 10^6 \text{ N/m}^2$) mode of RSS was explored to define controlling mechanisms and verify the Task I model concepts. Also regions of RSS for the unlike doublet injector element were mapped. High speed (800 FR/sec) color movies were used to observe and define operating modes. Tests were run with two unlike doublet elements of differing impingement lengths to verify the high pressure RSS model. Both the experimental results and Task I analytical calculations showed recirculation gas heating to be inadequate to cause RSS by monopropellant decomposition. Therefore, it was concluded that other mechanisms are operative.

The high speed color movies showed that RSS occurs gradually as the chamber pressure is increased rather than in a sudden step change; at least with MMH and A-50 fuels. It was also observed that the severity of RSS increases with increasing chamber pressure, fuel velocity, and fuel temperature.

Exploratory impingement point temperature measurements were made using a 0.020 inch (.051 cm) diameter thermocouple probe to try to develop a better understanding of the reaction mechanisms. The probe results showed that the impingement point temperature increased with chamber pressure and injection velocity. The data showed an apparent temperature discontinuity at a chamber pressure of about 400 psia, ($2.8 \times 10^6 \text{ N/m}^2$) corresponding to the MMH saturation temperature. The temperature jump was interpreted as a change in reaction mechanism from low enthalpy surface reactions to high enthalpy gas phase reactions.

I Introduction and Summary (cont.)

It was concluded that surface reactions play an important role in RSS and that both the photographic and the temperature probing technique should be improved during the Task IV testing to develop a clearer understanding of RSS.

4. Task IV - Verification of Governing Mechanisms

The Task IV test results showed that the RSS is best correlated with a gas phase/surface reaction model. Orifice hydraulics were found to significantly influence RSS. For example, unlike doublet injectors using rounded or contoured orifice inlets were found to suppress RSS at low injection velocities as compared to sharp edged inlet orifices. Injector design correlation equations which define the onset of RSS were developed for a wide range of injector elements using both EDM and platelet orifices.

It was concluded that both gas phase and surface reactions control RSS. The rate of the surface reaction appears to be mixing limited rather than kinetically limited, as the propellants can react as fast as they can be put together, i.e., the reaction time is much shorter than the mean contact time. Also, based on the Task IV analysis it appears that increasing the interfacial surface area before impingement increases the surface reaction and hence RSS. For example, the data show a Reynolds number influence on RSS and also show that unlike impingement of self-atomized or pre-atomized streams promotes RSS.

Several correlation equations were evaluated during the program to relate RSS to injector design and operating parameters. The best correlation to date relates RSS to chamber pressure and a dimensionless propellant reactivity parameter, R , where R is defined as the product of fuel Reynolds number and the

I Introduction and Summary (cont.)

square root of the product of the fuel and oxidizer vapor pressures divided by the chamber pressure:

$$R = Re \times \sqrt{P_{VF} \times P_{VO}} / P_c$$

(Complete nomenclature is contained in Appendix A)

Using this parameter the onset of RSS can be characterized by a minimum chamber pressure for RSS by the power form equation

$$P_{cRSS} = a R^b$$

where a and b are functions of injector design and, perhaps, propellant combination, as will be discussed in Section III. The adequacy of the correlation can best be judged by referring to the data plots and model curves as shown in Appendix B.

Other correlation equations are currently being evaluated in order to more completely characterize the variables which influence RSS.

II RESULTS AND CONCLUSIONS

A. Results

The most significant result of this program was the identification of the gas phase/surface reaction mechanism which controls RSS. Injector design criteria which defined a critical chamber pressure for those operating conditions above which RSS occurs were developed from data correlations based on this mechanism. As defined here a surface reaction is produced by the diffusion of the surrounding vapor (N_2O_4) to the liquid (fuel) surface causing an on-going reaction which results in heat input and vaporization of the liquid to sustain the reaction

II Results and Conclusions (cont.)

Significant effort was devoted to the development of meaningful RSS correlations during the Task IV effort. The final correlations are based on data from a parallel RSS study (Ref. 25), an OMS subscale study (Ref. 26) and the Zung data (Ref. 27) as well as the data from this study.

A wide variety of RSS correlating parameters were investigated and are tabulated in Appendix C, with the best correlation being provided by plotting the chamber pressure versus a dimensionless propellant reactivity parameter, R , as shown in Figure 1. Additional data showing the model correlation are contained in Appendix B. The reactivity parameter is the product of the fuel orifice Reynolds number and the square root of the propellant vapor pressure products divided by the chamber pressure. As described in Section VA, it is believed that the interfacial surface area at the impingement point controls RSS and that propellant stream properties are an important consideration. For example, increasing fuel stream Reynolds number apparently increases the amount of surface reaction. Curves defining RSS operating regimes are summarized in Figure 2 for a wide variety of injector elements and for MMH, A-50, and N_2H_4 fuels.

The degree of gas phase reaction depends on the amount of propellant vapor available, which depends on the propellant temperatures. Hence, increasing the propellant temperature increases the amount of gas phase reaction. The chamber pressure influences the gas-phase reaction through the propellant vapor concentrations which increase with pressure.

It was found that the rounded inlet long L/D orifices used on this study and the Rocketdyne study (Ref. 25) suppresses RSS at low values of the Reactivity parameter, as compared to sharp edged short L/D orifices. The chamber pressure at which RSS occurs is seen to be strongly dependent on the specific element type suggesting a strong influence of hydraulics on the inter-facial surface

II Results and Conclusions (cont.)

area at impingement. In general the pre-atomized type of element exhibits the earliest onset of RSS, viz, the like-on-like doublet and platelet elements.

It was found that both the MH and A-50 rounded inlet data can be correlated with the same curve. The influence of the N_2H_4 fuel is not conclusive since data with the rounded inlet injector is not available.

B. Conclusions

The most significant conclusions are:

1. RSS occurs in two modes depending on chamber pressure: (1) surface reaction mode - when fuel saturation temperature is less than impingement zone temperature and (2) gas phase reaction mode - when fuel saturation temperature is greater than impingement zone temperature.
2. The onset of RSS is controlled primarily by the operating chamber pressure and secondarily by the propellant reactivity and the amount of interfacial surface area at impingement.
3. RSS does not occur discontinuously but rather gradually with severity increasing with chamber pressure.
4. The propellant reactivity is controlled by the propellant temperature/vapor pressure relationship.
5. Increasing propellant temperature increases reactivity and promotes RSS.
6. The amount of interfacial surface area at impingement is controlled by the injector hydraulics (i.e., orifice configuration and Reynolds number).

II Results and Conclusions (cont.)

7. Unlike impingement of self-atomized or pre-atomized propellant streams promote RSS through increased interfacial surface area.

8. Contouring the orifice inlets of unlike impinging streams can suppress RSS by reducing interfacial surface area.

9. The mechanisms controlling pops were not verified.

10. RSS can be related to chamber pressure and a propellant reactivity parameter, R. The minimum chamber pressure for RSS can be specified by a power-form expression relating P_c to R with coefficients and exponents a function of the injector design.

11. Further refinement of the RSS correlation equations is desirable in order to more adequately account for propellant stream influences on RSS.

III APPLICATION OF RESULTS

Injector design criteria were derived from the RSS data correlations by fitting curves of the form:

$$P_{c_{RSS}} = a R^b$$

where:

P_{sep}	=	chamber pressure above which RSS will occur
a	=	experimentally determined constant
b	=	experimentally determined constant
R	=	$Re_f \sqrt{\frac{P_{vox} P_{vf}}{P_c}}$

III Application of Results (cont.)

Re_f = Fuel stream Reynolds Number

P_{ox} = N_2O_4 vapor pressure at the injection temperature

P_f = fuel vapor pressure at the injection temperature

The experimentally determined constants are summarized in Table I. The chamber pressure above which severe RSS occurs is found by selecting the applicable values of a and b from Table I and calculating R from the propellant temperatures, orifice geometry, and injection velocity. For example, given a like-on-like injector using MMH and the following design and operation conditions:

d_f = 0.025 in. (.064 cm)
 V_f = 65 ft/sec (19.8 in/sec)
 P_c = 125 psia (8.6×10^5 N/m²)
 T_f = 70°F (294°K)
 T_{ox} = 70°F (294°K)

The fuel Reynolds number is:

$$Re_f = \rho_f V_f d_f / \mu_f$$

where,

$$\rho_f = 54.5 \text{ lb/ft}^3 @ 70^\circ\text{F} \quad (872.5 \text{ Kg/m}^3 \text{ at } 294^\circ\text{K})$$

$$\mu_f = 5.5 \times 10^{-4} \text{ lb/ft-sec} @ 70^\circ\text{F} \quad (8.19 \times 10^{-4} \text{ N-sec/m at } 294^\circ\text{K})$$

therefore,

$$Re_f = 13,418$$

III Application of Results (cont.)

the vapor pressures are:

$$P_{\text{vox}} = 15 @ 70^{\circ}\text{F} \quad (1.03 \times 10^5 \text{ N/m}^2 \text{ at } 294^{\circ}\text{K})$$

$$P_{\text{vf}} = 0.75 @ 70^{\circ}\text{F} \quad (5.2 \times 10^3 \text{ N/m}^2 \text{ at } 294^{\circ}\text{K})$$

therefore,

$$P_{\text{vox}} P_{\text{vf}} / P_c = 0.0268$$

$$\text{and } R = 13,418 \times 0.0268 = 360$$

From Table I,

$$a = 125$$

$$b = -0.05$$

therefore,

$$P_{\text{sep}} = 125 (360)^{-0.05} = 93 \text{ psia } (6.41 \text{ N/m}^2)$$

RSS would be expected to occur at chamber pressures above this.

IV RECOMMENDATIONS

The following recommendations are made;

1. The injector design criteria developed here should be incorporated into the JANNAF DER performance program to indicate when an injector design is operating in an RSS mode.

2. New correlation data should be generated for unlike doublet elements using orifice geometries representative of flight injectors.

IV Recommendations (cont.)

3. The influence of fuel type should be verified using N_2H_4 , A-50, and UDMH.
4. Large orifice diameter (.060 inch, .152 cm) unlike doublets should be tested to determine mechanisms controlling pops.
5. Experimental correlations between RSS severity and performance loss should be generated using the experimental techniques developed here and on the OMS Subscale Program (Ref. 26).
6. Additional refinement of the existing RSS correlations should be attempted in order to more completely characterize propellant stream/spray fan influences on RSS.

V TECHNICAL DISCUSSION

A. RSS Model

During Task III it was concluded that RSS was a surface reaction controlled phenomena and that it could be correlated with the Weber number. A Weber number model was constructed to explain the observed pressure and velocity influence on RSS. The Weber number is the ratio of aerodynamic surface forces to the surface tension forces acting on a liquid droplet or stream as shown in Figure 3. It is used to characterize droplet breakup due to aerodynamic force. A critical Weber number exists beyond which droplets are shed or stripped from the surface. It was postulated that RSS would occur beyond the critical Weber number due to increased interfacial surface area and hence increased reactivity as a result of jet shedding ahead of impingement.

Preliminary data plots showed good correlation with the Weber number. However, Task IV testing showed the need to include a fuel temperature effect over and above the surface tension

V Technical Discussion (cont.)

influence. This was accomplished by modeling the fuel and oxidizer vapor phase concentrations just ahead of impact as shown in Figure 4. It was assumed that the vapor phase is in equilibrium with the liquid phase such that the vapor phase concentration is described by the vapor pressure as determined by the bulk propellant temperature. With this model increasing fuel temperature decreases the vapor phase mixture ratio such that the vapor phase becomes more reactive. The vapor phase mixture ratio is described by the equation;

$$MR_{vp} = P_{ox} \frac{MW_{ox}}{P_f} \frac{T_f}{MW_f T_{ox}}$$

where:

- P_{ox} = oxidizer vapor pressure @ T_{ox} , psia
- P_f = fuel vapor pressure @ T_f , psia
- MW_{ox} = oxidizer molecular wt.
- MW_f = fuel molecular wt.
- T_{ox} = oxidizer injection temperature, °R
- T_f = fuel injection temperature, °R

As shown in Figure 5, the Weber number and propellant vapor pressure parameters correlate the unlike doublet RSS data well. However, correlations with other injector data were not as good and it was necessary to re-examine the model to derive new correlation parameters.

The dependence of RSS on the Weber number is due to the strong chamber pressure and injection velocity influence. It was found during the Task IV cold flow experiments that the onset of self-atomization and jet shedding depends on the Reynolds number as well as the Weber number. For example, it was found that jet shedding begins at a Weber number of 3 for turbulent jets produced

V Technical Discussion (cont.)

by sharp edged orifices as compared to 12 for jets produced by contoured inlet orifices. This led to the realization that the orifice geometry and the Reynolds number influence the interfacial surface area.

The chamber pressure and fuel temperature dependence suggests a gas phase reaction mechanism as described by the classical reaction rate equation;

$$\frac{d[C]}{dt} = -k C_1^m C_2^n$$

where:

- C_1, C_2 = concentration of reactants, (moles/cc)
- k = rate constant
- m, n = concentration exponent

For the classical gas phase reaction the reactant concentrations are related to the partial pressure through the ideal gas law;

$$C_1 = n_1/V = p_1/RT \text{ moles/cc}$$

where:

- n_1 = no. of moles of species 1
- V = reaction volume, cc
- p_1 = partial pressure of species 1, atm
- R = universal gas constant, atm/°K-mole/cc
- T = gas temperature, °K

The concentration is related to the total pressure through the mole fraction;

$$C_1 = x_1 P/RT$$

V Technical Discussion (cont.)

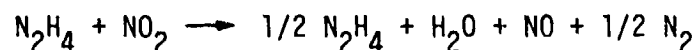
where:

$$x_1 = \text{mole fraction of species 1}$$

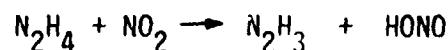
Therefore, increasing chamber pressure increases the reaction rate and heat release rate in the gas phase through increased reactant concentrations.

The fuel temperature also influences the gas phase reactions through its effect on reactant concentrations. The partial pressure of the reactants in the impingement zone are controlled primarily by the propellant vapor pressures. If it can be assumed that the propellant vapor is in equilibrium with the liquid phase then the vapor phase concentrations become direct functions of the propellant temperatures.

The functional relationship between the reaction rate and the concentrations depend upon the specific reaction mechanism. For example, the hydrazine/nitrogen dioxide gas phase reaction was found to exhibit two reaction steps depending on the gas temperature (Ref. 13). In the lower temperature regimes ($T < 900^\circ\text{K}$) the step I reaction stoichiometry is;



with the rate being controlled by the following hydrogen abstraction reaction;



The complete reaction mechanism is given in Ref. 13. The rate equation is;

$$d[\text{N}_2\text{H}_4]/dt = -k_1 [\text{N}_2\text{H}_4] [\text{NO}_2]$$

V Technical Discussion (cont.)

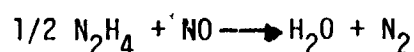
where:

$[N_2H_4]$ = Hydrazine concentration

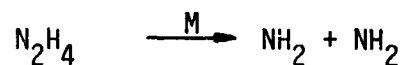
$[NO_2]$ = nitrogen dioxide concentration

$k_1 = 10^{15.83} e^{-26,700/RT}$

In the higher temperature regime ($T > 1000^\circ K$) the step II reaction stoichiometry is;



with the rate controlled by hydrazine decomposition;



The step II reaction rate is;

$$d [N_2H_4]/dt = - k_2 [N_2H_4]$$

where;

$$k_2 = 10^{10.17} e^{-39,600/RT}$$

The Step I reaction is more likely to occur under the relatively low temperature conditions of reactive stream impingement. The functional relationship between the propellant concentration and the reaction rate for the Step I reaction is:

$$\begin{aligned} d [N_2H_4]/dt &= k_1 [p_{N_2H_4}/RT] [p_{NO_2}/RT] \\ &= -k_1/(RT)^2 [p_{N_2H_4}] [p_{NO_2}] \end{aligned}$$

Therefore, it is expected that the propellant temperature influence should be correlated with a parameter containing the product of the

V Technical Discussion (cont.)

fuel and oxidizer partial pressure (i.e., vapor pressure).

The partial pressures were normalized by the chamber pressure for correlation purposes. It was found that the best correlation is obtained by plotting the chamber pressure versus the product of the fuel Reynolds number and the square root of the normalized partial pressure product.

B. Task I - Model and Data Review

The results of the literature review and model critique were the: (1) description of the problem, (2) tabulation and evaluation of results of all prior RSS work, (3) critical evaluation of two RSS models, (4) development of RSS data correlation for hydrazine fuel, and (5) development of new RSS model concepts. These results are discussed in detail.

1. Problem Description

Popping and RSS are important design considerations because they have been found to adversely affect performance and stability. RSS has been observed to significantly reduce the performance of the following types of impinging injectors:

- Unlike Doublet
- Triplet
- Like Doublet

Experimental test results from the various OMS Engine Technology programs have provided quantitative data of the effect of RSS on performance caused by increasing the fuel temperature from ambient to a level simulating operation with a fuel regeneratively-cooled chamber ($\sim 200^{\circ}\text{F}$, 367°K). In many cases, the

V Technical Discussion (cont.)

injector performance decreased almost linearly with increasing fuel temperature. For example, as shown in Figure 6 the energy release efficiency decreased from 2 to 5% as the fuel temperature was increased from 50 to 240°F with two ALRC subscale OMS unlike doublet injectors. Similar results were also obtained with the full scale OMS unlike doublet injector.

Performance reductions due to RSS have been found to be influenced by:

- Propellant Temperature
- Element Size
- Element Spacing
- Chamber Length

The effect of RSS was lessened with increasing chamber length and decreasing orifice size as shown in Figure 6. RSS was noted with both $N_2O_4/A-50$ and N_2O_4/MMH propellants.

Random high amplitude short duration (less than a millisecond) pressure and accelerometer disturbances were observed with the N_2O_4 /Amine fueled Apollo Spacecraft engines during their development phase and more recently on some OMS development engines. These disturbances, called pops, are undesirable because they may trigger damaging combustion instability. In all of the Apollo engines the pops were eliminated or reduced to acceptable levels through trial and error testing since there were no design criteria by which logical element or pattern changes could be made. The pops were generally attributed to random accumulation and mono-propellant explosion of fuel pockets or zones caused by such things as poor element or pattern design, orifice flow instabilities (hydraulic flip), plugged oxidizer orifices, or fuel leakage through weld cracks.

V Technical Discussion (cont.)

Although these causes are certainly possible recent investigations demonstrate that the observed pops are related to combustion phenomena associated with unlike hypergolic propellant stream impingement. For instance, it has been shown (Ref. 1-5) that small local explosions of fuel and oxidizer, rather than mono-propellant fuel explosions, are the source of pop triggers. The pops and/or pop triggers can upset engine stability by driving the feed system dynamics and/or chamber acoustics.

This program was undertaken because the mechanisms controlling pops and RSS were not thoroughly understood such that meaningful injector design criteria could be developed. The first task was to critically review the prior related work to develop a rational starting point. The results of this task are discussed below.

2. Literature Review

A summary of the literature review is contained in Appendix D. The key finding of each investigator is summarized under the comments heading. A summary of the range of parameters covered in these studies is shown in Table II.

Past investigations had (1) identified four operating regimes for hypergolic stream impingement, (2) demonstrated techniques of identifying and measuring RSS and pops, (3) identified many of the design and operating parameters influencing RSS and pops, (4) provided a multitude of RSS and pop data, and (5) postulated two theoretical models for the prediction of one or more of the reactive impingement regimes.

Experimental studies had identified four operating regimes for reactive stream impingement as illustrated in Figure 7. The occurrence of these regimes was found to be dependent on many

V Technical Discussion (cont.)

engine operating and design parameters. The conditions found to be conducive to the occurrence of each operating regime are listed in Figure 7, although a quantitative description of these operating regimes was found to be lacking.

The following sensing techniques had been used to identify the various impingement regimes in past investigations:

- Photographic
- Performance Measurement
- Pressure Measurement
- Accelerometers
- Mass Spectrometer
- Thermocouples

Photographic techniques provide a qualitative measure of the reactive stream flow characteristics and were used in the first reported evidence of stream separation (Ref. 6). High speed movies were also instrumental in the identification of cyclic blowpart of single doublet elements (Refs. 2, 3, 5 and 7). The performance measurement technique with conventional chambers provided a pseudo-quantitative measurement of stream separation by comparing the performance of a system while varying the parameter which influences stream separation (Ref. 8). For example, the decrease in performance with increasing fuel temperature resulted presumably from increasing reactive stream separation. Performance measurements with a baffle chamber provided more conclusive quantitative measurement and was capable of identifying the mixed, separated, and penetrated flow regimes (Ref. 9). The P_c and accelerometer measurements have been used to identify the "popping" regime (Refs. 1, 3, 5, and 9). Finally, the mass spectrometer and thermocouple measurement techniques were used to measure the mixture ratio distribution downstream of the reactive stream impingement providing a quantitative measurement of the reactive stream mixing process (Ref. 11).

V Technical Discussion (cont.)

It was concluded that photographic techniques could best be used to define RSS mechanisms because the flow field can be magnified for close examination.

3. Model Review and Critique

The JPL model and ALRC model were the only models reported in the literature for predicting the occurrence of RSS and pops. Each of these models were reviewed and checked for accuracy using the accumulated RSS data.

a. JPL Model

The first model is that developed by Kushida and Houseman (Ref. 12) of the JPL. It postulates two regimes of RSS, (1) a low pressure separation due to liquid phase reactions, and (2) a high pressure separation due to gas phase reactions. The model defines the regimes of mix and separation in terms of the chamber pressure, propellant temperature, and impingement contact time (D/V) as shown in Figure 8.

The low-pressure-separation condition was postulated to occur when the liquid phase reactions heat the propellants to the bubble point. This condition exists when:

$$T_{P_i} \geq T_B$$

where:

T_{P_i} = propellant temperature at the impingement point

T_B = propellant boiling temperature

V Technical Discussion (cont.)

Although Kushida does not develop this portion of the model in detail in the literature, it can be done as follows. T_{p_i} is determined by the liquid phase heat release rate and the contact time within the impingement zone.

$$T_{p_i} = T_{p_o} + \Delta T_p$$

where:

T_{p_o}	=	propellant injection temperature
ΔT_p	=	temperature rise due to liquid phase reactions
ΔT_p	=	$\dot{Q} \text{ tr } M_R/M_H C_p$
\dot{Q}	=	liquid phase heat release, K cal/sec-mole oxidizer
M_R	=	mass of propellant reacted, moles
M_H	=	mass of propellant heated, moles
C_p	=	specific heat of propellant, cal/mole-°K
tr	=	contact time, sec.

using:

C_p	=	23 cal/mole-°K for hydrazine
\dot{Q}	=	83×10^3 kcal/sec-mole of oxidizer*
ΔT_p	=	3.6×10^6 °K/sec x tr x M_R/M_H

The average contact time is:

$$\text{tr} = D/V$$

where:

D	=	jet diameter, feet
V	=	jet velocity, ft/sec

*Kushida's reported value of 83 kcal/sec-mole of oxidizer is in error.

V Technical Discussion (cont.)

To calculate ΔT_p requires that the ratio M_R/M_H be assumed. Although Kushida does not report the value used it was calculated to be about 0.27 as follows: Figure 8 shows separation is predicted to occur below pressures of about 38 psia ($2.6 \times 10^5 \text{ N/m}^2$) for a contact time of 40 μ sec and 40°F (278°K) propellants. Using this contact time and the N_2O_4 saturation temperature at 38 psia ($2.6 \times 10^5 \text{ N/m}^2$), the assumed ratio of M_R/M_H is about 0.27.

The second mode of stream separation is postulated to occur at high pressure due to gas phase reactions at the propellant stream interface. The interface is modeled as a stable gas film separating the liquid streams. The rate of fuel and oxidizer transport into the film is presumed to be limited by vaporization. The reaction film thickness is determined by a momentum balance on the reaction volume. Using a hydrazine/ N_2O_4 reaction mechanism and rate reported by Sawyer and Glassman (Ref. 13) Kushida is able to develop a relationship between the contact time (D/V) and the pressure through the gas density.

Separation is postulated to occur if:

$$D/V \sim 35 (100/P)^{-1.5}$$

where:

P = chamber pressure, psia

The weakness of this model is that it assumes the existence of a stable gas film with laminar flow at the contact point. Recent cold flow and reactive impingement experiments (Refs. 3, 4, and 5) show the impingement process to be highly unstable and cyclic in nature.

The RSS/pop data summarized in Appendix D were plotted on the pressure/contact time coordinates (Figure 9)

V Technical Discussion (cont.)

to check the Kushida model. It is evident that the model does not adequately correlate the mix and separation regimes.

b. ALRC Model

The second model, developed by Lawver (Ref. 1) of ALRC, describes regimes of mix, pop, and separation for reactive stream impingement as well as describing inter-element spacing requirements for preventing coupling of single-element pops with adjacent element sprays. Reactive stream impingement is postulated to be controlled by liquid-phase reaction kinetics and mixing. The onset of stream separation is postulated to occur when the stream residence time exceeds the ignition delay time:

$$\tau_{RES} \geq \tau_{ign}$$

where:

$$\tau_{RES} = D/V$$

and:

$$\tau_{ign} = \frac{1}{Y} e^{E/RT}$$

Y = reaction rate constant

E = activation energy

R = Universal gas constant

T = Propellant temperature

The pops are postulated to be due to the ignition of well-mixed fuel and oxidizer within the ligament formation zone. Pops occur when:

$$\tau_{lig} < \tau_{ign}$$

V Technical Discussion (cont.)

where:

$$\tau_{\text{lig}} = (200/V) (D/V)$$

For a complete derivation see Reference 1.

The regimes of mix, pop, and separation are defined by plotting $(1/T)$ versus (D/V) as illustrated in Figure 10.

Using this model of reactive stream impingement, a stream separation parameter, I , was defined such that:

$$I = (\tau_{\text{ign}})_{\text{SEP}} / (\tau_{\text{ign}})$$

where:

$$(\tau_{\text{ign}})_{\text{SEP}} = \text{ignition delay at separation limit}$$

$$\tau_{\text{ign}} = \text{ignition delay as determined by fuel temperature}$$

A large set of Apollo engine pop data, in the 100-150 psia pressure range, were correlated with this parameter (Ref. 1) to define regimes of engine popping. The data correlations showed that most engines operate within the regime of element popping, therefore, it became necessary to prevent inter-element coupling to eliminate the engine pops. On the basis of high speed movies (Ref. 2) which showed the element pops to be highly localized explosions it was theorized that the pops behave as spherical blastwaves in which the pressure decays rapidly with distance. With this theory it was possible to develop the following element spacing parameter.

$$D = R/S$$

V Technical Discussion (cont.)

where:

$$\begin{aligned} R &= 49.2 (D_f/P_c)^{1/3} \text{ in.} \\ D_f &= \text{fuel orifice diameter} \\ P_c &= \text{chamber pressure} \\ S &= \text{element spacing, in.} \end{aligned}$$

See Ref. 1 for a complete development. Coupling occurs when:

$$D \geq 1.2$$

Application of this spacing criteria to the Apollo SPS IOS injector (Ref. 1) eliminated the popping.

The RSS data tabulated in Appendix A were plotted on the $(1/T)$ versus (D/V) coordinates to check the Lawver theory. The results showed poor correlation especially at higher pressures. The atmospheric pressure data are shown in Figure 11. It is evident that this model does not adequately define the mix, pop, and separate regimes.

The model's failure to adequately correlate the various regime does not invalidate its ability to accurately predict pop coupling. The pop blastwave coupling model merely states that given a regime of element popping, coupling will or will not occur given certain conditions of chamber pressure, element size and spacing. The successful application on the Apollo IOS and the ALRC OME engine studies is ample verification.

Since neither the Kushida or Lawver models adequately correlate the operating regimes it was desirable to develop a new model of reactive stream impingement.

V Technical Discussion (cont.)

4. Development of Data Correlations

The data generated by the following investigators were tabulated for computerized data analysis:

- Lee and Houseman (Ref. 5)
- Zung and White (Ref. 27)
- Nurick and Cordill (Ref. 4)

A computer program was written to summarize the data and to calculate and plot correlation parameters. The computer listing and data summaries are included in Appendix D. Of all the correlations plotted the factors providing the best correlation are:

- Chamber Pressure
- Orifice Diameter
- Propellant Temperature

These factors were found to exhibit the controlling influence over the occurrence of the mix, pop, and separation regimes.

The data were used to construct correlation plots to define the first order controlling parameters. This was done by plotting the data versus several parameters including:

- $1/T_f$ vs. $(D/V)_f$
- P_c vs. D_f
- P/D vs. T_f
- P/D vs. $(D/V)_f$

V Technical Discussion (cont.)

- P/D^2 vs. T
- P_c vs. $1/V_f$
- P_c vs. T

The limits of occurrence of RSS and popping were observed to be primarily dependent on the chamber pressure and orifice diameter. The orifice diameter was found to exhibit an overwhelming influence on the pop limit as shown in Figures 12 and 13. These plots also indicate the existence of two distinct modes of RSS for hydrazine; a high-pressure mode and a low-pressure mode as previously reported by Zung (Ref. 3). The low pressure mode is associated with N_2O_4 boiling which prevents stream penetration and mixing. The second mode of RSS occurs at pressures above 300 psia ($2.1 \times 10^6 \text{ N/m}^2$). Modeling of this RSS mode is discussed in the subsequent section.

In summary the data correlations showed four distinct modes of reactive stream impingement for hydrazine fuel;

- Mixing
- Popping
- Low Pressure RSS ($P_c < P_{\text{sat ox}}$)
- High Pressure RSS ($P_c > 300 \text{ psia}, (2.1 \times 10^6 \text{ N/m}^2)$)

5. New Model Concepts

A new model of reactive stream impingement was formulated through the examination of the new RSS data correlations and re-examination of the physical and chemical processes involved. The new model presumes four regimes of reactive impingement as identified by the data correlations: (1) mixing or penetration, (2) popping (cyclic blowpart), (3) low pressure separation, and (4) high pressure separation. The existence of these regimes is believed

V Technical Discussion (cont.)

to be a consequence of the complex reaction mechanisms of the N_2O_4 /Amine propellants. Hypergolic reaction has been observed to occur in four basic modes:

- High enthalpy gas-phase oxidation reaction
- Low enthalpy surface reactions
- Monopropellant decomposition
- Liquid phase reactions

However, it is believed that liquid phase reactions are not a factor in RSS since true liquid-phase reactions have been observed only with dilute mixtures of N_2O_4 and N_2H_4 (Ref. 14). They are not likely to occur under reactive stream impingement conditions due to the immiscibility of the propellants and the apparent low degree of liquid/liquid mixing observed with cold flow liquid stream impingement. Also, it has been subsequently determined, based on the Task III work, that the postulated monopropellant decomposition model of RSS is incorrect (See Section V.C.2).

During the proposal phase, high speed color movies were taken of cold flow stream impingement using streams of water and dyed water in an effort to more fully understand the impingement process. It was found that impingement is cyclic in nature due to the normal streak breakup processes. The formation of surface waves is evident in the film sequence. The stream breakup process is controlled by the stream hydraulic and interfacial forces.

Examination of the movie sequence showed the existence of distinct globs of water and dyed water indicating a lack of liquid-phase mixing. Therefore, nonreactive impingement of miscible fluids appears to be dominated by interfacial forces. Likewise, reactive stream impingement of immiscible fluids is also expected to be

V Technical Discussion (cont.)

dominated by interfacial forces. Extensive liquid-phase mixing is not expected. This conclusion was also reached by Breen et. al., (Ref. 17), on the basis of reactive tests. Also reaction kinetic data derived from N_2O_4/N_2H_4 stream impingement data (Ref. 18) were found to agree well with the gas kinetic data of Sawyer and Glassman (Ref. 13). This observation along with that of Weiss, et al., (Ref. 19) showing N_2O_4/N_2H_4 to be immiscible led Breen to the conclusion that the reaction between liquid surfaces may be controlled by vapor phase kinetics.

Evidence in support of this theory is provided by the results of the hypergolic impingement experiments of Rodriguez (Ref. 20) meant to measure liquid phase reaction rate. The measured heat release rates reported by Rodriguez would suggest that MMH is more reactive than N_2H_4 which seemingly conflicts with other impingement test results which had indicated N_2H_4 to be more reactive. However, his results are reasonable if in fact a gas-phase reaction is measured. A rather clear correlation is provided by comparing the vapor pressure to the measured heat release rate as shown in Table III. These data show that the measured heat release rate increases with the fuel vapor pressure and show the MMH to be more reactive than the N_2H_4 .

Hypergolic surface reactions were first described by Lawver during a study of N_2H_4 droplet combustion (Ref. 21). Droplet ignition was observed to occur through a surface reaction which produces a white milky substance on the droplet surface which has since (Ref. 21) been identified as ammonium nitrate. Ignition proceeds by surface reactions which heat the liquid fuel to the ignition point as shown in Figure 14. At the ignition point the reaction changes from the surface to that of a droplet diffusion flame.

In a later study Zung (Ref. 22) characterized N_2O_4/N_2H_4 ignition. Photographs taken during this study clearly show

V Technical Discussion (cont.)

surface reaction phenomenon. Ignition was found to be a transition from low enthalpy surface reaction to high enthalpy vapor phase reaction as illustrated in Figure 14. Zung found the ignition point to depend on the N_2O_4 concentration and the N_2H_4 temperature, as shown in Figure 15.

At N_2H_4 temperatures below 107°F, ignition (i.e., transition from surface reaction to diffusion flame) is controlled by diffusion of N_2O_4 vapor to the liquid N_2H_4 surface. At temperatures higher than this, ignition is controlled by vapor phase kinetics. Above 187°F ignition is influenced by N_2H_4 decomposition kinetics.

It was concluded that RSS and popping are dominated by surface and gas phase reactions and that liquid phase reactions are not important. Controlling reaction mechanisms were postulated for each of the four impingement regimes as illustrated in Figure 16. The pop was postulated to be a consequence of the transition from a low enthalpy surface reaction mode to a high enthalpy gas-phase reaction mode with attendant ignition of explosive intermediates formed during the pre-ignition phase. The transition is a consequence of self-heating by the surface reactions. The heat-up time and the amount of heating both depend on the orifice size. Small orifices produce smaller ligaments which can heat to the ignition point before appreciable amounts of intermediates are formed. They also restrict the contact time (i.e., heat-up time). Larger jets produce correspondingly larger ligaments which allows sufficient self-heating and intermediate accumulation to produce pops. The self-heating by surface reaction is controlled by the volume to surface ratio and is therefore a function of the orifice diameter.

Stream mixing was postulated to occur when the orifice diameter is sufficiently small to prevent pops or when the pressure is sufficiently high to suppress transition from surface reaction to vapor-phase reaction until the ligaments are shed into droplets.

V Technical Discussion (cont.)

Low-pressure RSS occurs when the N_2O_4 temperature exceeds its saturation temperatures. The resultant two-phase stream enhances surface and/or gas phase reactions which prevent interdispersion of propellant droplets.

High-pressure RSS was postulated to be a consequence of monopropellant decomposition of the fuel vapor due to recirculation gas heating of the fuel stream ahead of impact. The resultant decomposition would produce gas-phase reactions upon contact with the oxidizer stream and the separation would be controlled by the fuel vapor pressure. However, subsequent analyses and data from the Task III effort indicates the recirculation gas heating to be of negligible consequence over the range of injection parameters surveyed.

POSTULATED POP MECHANISMS

Previous studies had identified the following pop characteristics:

- Cyclic nature
- Explosions originate close to impingement point
- Explosions emit high velocity blastwaves
- Explosions have been classified as to their strength
- Frequency of occurrence and strength depends on diameter, D/V, and fuel properties

Nurick and Cordill (Ref. 4) suggested the possibility that the explosions may be due to either ignition by impact of explosive intermediates or ignition of mixed propellants. The possibility of ignition of shock-sensitive intermediates by impact was shown to be unlikely since the calculated impact forces for typical stream velocities are orders of magnitude lower than that required for detonation. For instance the maximum impact energy for a 0.040 in. diameter

V Technical Discussion (cont.)

jet at 50 ft/sec velocity is 43×10^{-8} ft-lb. A minimum of 4-ft-lb is required (Ref. 16) to detonate the intermediates. Also it was observed that the popping frequency and strength decrease with increasing velocity which is the inverse of that expected for an impact ignition mechanism.

It was postulated during the proposal phase that the explosions may also be triggered by compression of the gas bubbles formed as a result of the surface reactions. Adiabatic compression of the gas bubbles by liquid stream impact could produce sufficiently high temperatures to trigger the explosive intermediates. This process would be expected to be highly random in nature. This mechanism was not experimentally investigated because it is believed that the explosions are most likely due to ignition by hypergolic self-heating through interfacial surface reactions as described above. This mechanism was not investigated.

POSTULATED HIGH PRESSURE RSS MECHANISM

The Task I data correlations showed that high pressure RSS occurs above 300 psia for N_2O_4/N_2H_4 . It was also noted that the N_2H_4 saturation temperature (450°F) at this pressure coincides with the N_2H_4 vapor decomposition temperature (Ref. 15). This observation suggested a monopropellant decomposition mechanism for high-pressure RSS. It was postulated that RSS could occur as illustrated in Figure 17 by recirculation gas heating of a significant portion of the N_2H_4 to its saturation temperature prior to impact. When the saturation temperature exceeds 450°F (506°K) then monopropellant decomposition occurs causing RSS, as illustrated in Figure 17. At 300 (2.1×10^6 N/m²) psia this would result in monopropellant decomposition of the N_2H_4 thus preventing stream mixing. Extending the theory to other fuels would suggest that separation with MMH and A-50 would occur at higher chamber pressure due to their higher vapor pressures as shown in Table IV.

V Technical Discussion (cont.)

It was decided to analytically verify this theory during the contract Task I effort. The results of the modeling presented in the following section indicates inadequate heating of the free stream to heat a significant portion of the fuel stream to the saturation temperature. The analytical results were experimentally verified in Task III by testing with elements having different free stream lengths (i.e., different impingement distance).

The recirculation gas heating of the free-stream was modeled as illustrated in Figure 18. Combustion recirculation gas heats the surface film of fuel to the boiling point in the free-stream ahead of impact. The resultant surface film monopropellant decomposition prevents inter-propellant stream mixing.

The fuel stream liquid temperature and fuel vapor temperature were calculated using the heat and mass transfer equations developed by Priem in NASA-TR67 for droplet vaporization. A cross-section of the fuel jet illustrated in Figure 18 shows the vaporization mechanism. Also shown are the typical temperature profiles for fuel vaporization with and without monopropellant decomposition of the fuel vapor. The influence of the decomposition is to increase the heat flux to the droplet by steepening of the temperature gradient. The equations describing the heat and mass transfer are also shown in Figure 18. These equations were computer coded for a finite difference calculation. The classic Ranz-Marshall correlation was used for fuel vaporization without decomposition and a predicted Nusselt number which accounts for decomposition was used for the case of monopropellant decomposition. A computer listing is included in Appendix D.

The computer results indicate that for a hydrazine free-jet length of 4 L/D's injected at 50 ft/sec (15.2 m/sec) the liquid temperature rise would be on the order of only 10°F (5.6%) for jets from .027 to .060 inches (.069 to .152 cm) diameter at 300 psia (2.1×10^6 N/m²) chamber pressure. Less than .1 percent of the liquid

V Technical Discussion (cont.)

mass would be expected to vaporize prior to impingement. These calculations indicate that insufficient pre-impingement heating occurs to heat the fuel stream to the saturation temperature. Thus it was concluded that free-stream heating by recirculation gas does not control high pressure RSS. Subsequent experiments support this conclusion. However, recirculation heating may increase the free jet boundary layer temperature and result in more droplet shedding due to increased Weber number due to lower jet boundary surface tension and add to gas generation.

The conclusions drawn from the Task I results were:

(1) Previous investigations had identified four operating regimes of reactive stream impingement; (1) Mixing - impingement results in uniform spray field mixture ratio distribution, (2) Separation - impingement results in non-uniform spray field mixture ratio distribution with fuel and oxidizer striations, (3) Penetration - impingement results in non-uniform spray field mixture ratio distribution with fuel and oxidizer "shoot through", and (4) Popping - Impingement results in random or cyclic explosion of spray field producing blowpart and blastwaves.

(2) Two regimes of RSS exist, a high pressure mode ($P_x > 300$ psia (2.1×10^6 N/m²) for hydrazine) due to reactive stream blowpart and a low pressure mode ($P_c < P_{sat_{ox}}$) due to oxidizer boiling.

(3) The operating and design factors exhibiting the greatest influence on RSS and popping with hydrazine fuel are: (1) chamber pressure, (2) orifice diameter, and (3) propellant temperature.

(4) Neither the Kushida (JPL) or the Lawler RSS models (ALRC) adequately correlate RSS and popping and a new RSS model is required.

V Technical Discussion (cont.)

(5) RSS appeared to be related to four possible mechanisms of hypergolic reaction; (1) high enthalpy gas-phase reaction, (2) low enthalpy surface reactions, (3) monopropellant decomposition, and (4) liquid phase reactions. High pressure RSS was concluded to be a result of monopropellant reaction, however, Task III and IV testing show this not to be case. It was also concluded that liquid phase reactions are not important due to the predominance of surface reactions and lack of liquid phase mixing.

C. Task III - Definition of Governing Mechanisms

During the Task III effort two unlike doublet injector elements were designed, fabricated, and tested in a photographic test chamber with four GN_2 cooled quartz windows. The injection elements were identical except for impingement length which was varied to test the monopropellant decomposition RSS model formulated in Task I.

High speed movies of reactive stream impingement were taken over the pressure range of 100-1000 psia ($6.89 - 68.9 \text{ N/m}^2$). It was observed that; (1) photographic clarity decreased with increasing chamber pressure, (2) RSS was not sensitive to impingement length for the range tested (0.160 - 0.060 inches .406 - .152 cm), (3) RSS occurs gradually as pressure is increased from 100-1000 psia ($6.89 - 68.9 \text{ N/m}^2$) and that RSS severity increases with chamber pressure, fuel temperature, and injection velocity, and (4) RSS severity depends on fuel composition - MMH separates at a lower chamber pressure and at lower propellant temperature than does A-50.

A temperature probe technique was used to measure impingement point reaction temperature rise. A discontinuity in the impingement point temperature rise as chamber pressure is increased was observed. These results are discussed in detail.

V Technical Discussion (cont.)

1. Task III Test Objectives and Conditions

The two unlike doublet injector elements were tested over the range of parameters listed in Table V in an effort to determine the mechanisms controlling RSS. The injectors, test chamber, and test setup are described in detail in Section V,E.

The Task III test objectives and results are included in Table V. A detailed test condition log was maintained and is included in Appendix B. The test data were stored in a computer data file for easy manipulation and correlation. A listing of the reduced data is included in Appendix E.

The objectives of the first series of tests (#101-106) were to verify proper test stand operation and to check the photographic equipment. The tests showed that the back-lighting intensity was too bright and that the test stand functioned as required. Examination of the movie pictures showed the unlike doublet to be separated at all of the conditions tested. Separation is defined as the appearance of unmixed oxidizer in the spray field evidenced by clouds of dark brown NO_2 . Although density gradients between the cold window purge gas and the hot combustion gas obscured detail in the impingement zone at the higher pressures, the spray field operating mode was readily discernible.

The backlight was modified prior to the next test series to improve photography. A sheet of polarized filter paper and a sheet of ground glass were placed between the Fresnel lense and the test chamber to reduce the backlighting intensity and to eliminate parallel light from the quartz lamp.

The next series of tests (#107-111) were run at lower pressures to determine the pressure limit of RSS. The onset of RSS was found to occur between 100 and 150 psia (6.89 and $10.3 \times 10^5 \text{ N/m}^2$) with ambient temperature MMH. The recirculation gas model

V Technical Discussion (cont.)

developed in Task I had predicted separation at about 400 psia ($2.75 \times 10^6 \text{ N/m}^2$) with MMH. RSS was found to gradually worsen with increasing pressure. The density gradients produced by the GN_2 purge were still visible but their intensity was found to decrease with pressure. Good, clear pictures were obtained at the lower pressures (100-200 psi, $6.89\text{-}13.78 \times 10^5 \text{ N/m}^2$).

The next set of tests (#112-123) were run with A-50 to determine the influence of fuel vapor pressure on RSS. The hot gas recirculation model had predicted that A-50 would separate at 500 psia ($3.44 \times 10^6 \text{ N/m}^2$) as compared to 400 psia ($2.75 \times 10^6 \text{ N/m}^2$) for MMH. The data showed ambient temperature A-50 to separate at about 200 psia ($13.78 \times 10^5 \text{ N/m}^2$) as compared to about 150 psia ($10.3 \times 10^5 \text{ N/m}^2$) for the MMH. The increase in separation severity with pressure is readily apparent in Figure 19 which shows a series of single movie frames from successive tests at increasing chamber pressures. Test No. 117 was run at a lower injection velocity to determine its influence on RSS. The movie film shows notably less separation at the lower velocity (88 ft/sec, 26.8 m/sec) than at the nominal velocity (125 ft/sec, 38.1 m/sec).

The next series of tests (No. 124-132) were run with the short impingement length doublet to determine the influence of recirculation gas heating on RSS. The Task I model had indicated that RSS should depend on impingement length. The movie data do not show any discernible difference in separation characteristics between the long impingement (0.160 in., .406 cm) and the short impingement (0.060 in., .152 cm) doublet elements.

Test Numbers 134-138 were run with heated fuel over the pressure range of 100-250 psia ($6.89 - 17.2 \times 10^5 \text{ N/m}^2$). The movies show a pronounced worsening of separation with increased fuel temperature. This influence is demonstrated by the movie frames shown in Figure 20. Also shown in Figure 20, is the thermocouple used to probe the impingement point in the subsequent set of tests.

V Technical Discussion (cont.)

The onset of separation was found to occur at 100 psia ($6.89 \times 10^5 \text{ N/m}^2$) with 200°F (367°K) MMH.

The final Task III test set (No. 139-152) was run with ambient temperature propellants and the thermocouple probe mounted as shown in Figure 20. Initially, a 0.010 in. (.0254 cm) dia. thermocouple was used, however, it lacked sufficient mechanical strength to remain in a fixed position. It was discarded for a more rigid 0.020 in. (.0508 cm) dia. thermocouple. The movies indicate some disruption of the impingement by the thermocouple particularly at the lower velocity conditions, however, the temperature data are reasonably consistent and orderly. The temperature probing technique appeared to offer significant quantitative data on impingement point heating and therefore was improved and used during the Task IV testing. These temperature data were used to correlate propellant stream temperature rise, ΔT , to fuel Weber number and is thought to contribute to RSS.

2. Task III Interpretation of Test Results

The impingement process was observed to be cyclic in nature in both the mixed and separated modes. Although detailed frequency measurements were not made, the characteristic frequencies appeared to correspond to the frequencies observed in cold flow suggesting that the cyclic nature is characteristic of the ligament shedding process. Energetic cyclic blowpart (i.e., popping) was not observed on any of the test. The Task I data correlations would indicate that none should have occurred with the small orifice diameter used.

The RSS data for both MMH and A-50 fuels were plotted on the pressure versus temperature scales as shown in Figure 2 for comparison with the Task I data correlations previously displayed in Figures 12 and 13. It is noted that the A-50 separates at a higher pressure than the MMH and that the pressure at which MMH

V Technical Discussion (cont.)

separates decreases with increasing fuel temperature. The N_2H_4 data of Zung (Ref. 27) did not reflect an influence of inlet temperature on high pressure RSS.

Listed in Table IV are the predicted and measured RSS pressure limits for ambient temperature MMH and A-50. Also included is Zung's N_2H_4 data for reference. The pressure limits were predicted on the basis of the recirculation gas heating model developed in Task I. The model states that separation should occur at the pressure corresponding to the fuel vapor pressure at 450°F, the vapor phase decomposition temperature. It is seen that the pressure levels do not agree and that the trend in fuel type is correct for MMH and A-50 but not for N_2H_4 . In view of this and the fact that the analytical calculations had indicated insufficient heating, the hot gas recirculation model does not appear valid.

The experimentally measured (Ref. 20) heat release rates and the onset of RSS as shown on Table VI would seem to indicate a correlation between RSS and propellant reactivities, as measured by the heat release rates. This dependence on propellant reactivity was experimentally observed by noting that the impingement point temperature was found to depend on the chamber pressure and the propellant velocity. These influences appeared to be accounted for by adding the propellant stream dynamic head to the chamber pressure as shown in Figure 22. Also included in Figure 22 are the saturation temperature lines for N_2O_4 and MMH. There appears to be two modes of RSS as evidenced by a step change in temperature. The temperature discontinuity is believed to be indicative of a change in reaction mechanism as suggested by the popping regime mechanism described in Section IV.B.3. It is believed that low enthalpy surface reactions heat the propellants to their saturation temperatures and that when the fuel saturation temperature is exceeded, the reaction switches to a high enthalpy gas phase reaction.

V Technical Discussion (cont.)

It was also thought that the onset of visual separation might occur when the impingement point temperature exceeds the N_2O_4 saturation temperature. However, as indicated in Figure 22 this was not found to be true as there are mixed conditions above the N_2O_4 saturation temperature. It was concluded that the onset of RSS could not be correlated with impingement point temperature rise alone.

The low enthalpy surface controlled reaction was expected to be primarily a function of the propellant interfacial surface area at the point of impact which is expected to be related to propellant stream turbulence level or Reynolds number. A dependence of impingement point ΔT_i on fuel Reynolds number (R_{ef}) up to the point of transition from low to high enthalpy reaction was found. Beyond the transition there was no dependence of impingement point ΔT_i on R_e . Examination of the movie film brought the realization that a free jet may also experience self-atomization prior to impingement, thus influencing the effective surface area at impact. The self-atomization of a free stream is characterized by the ratio of aerodynamic to surface tension forces as described by the Weber number. The impingement point temperature rise was plotted versus the fuel and the oxidizer Weber number as shown in Figure 23. There appeared to be some dependence of ΔT_i on Weber number. It is noted that the temperature transition occurs beyond the critical Weber No. which signifies the onset of self atomization and increased surface area. It was noted that the occurrence of visual RSS was also correlated by the critical Weber number.

The following conclusions were drawn from the Task III results:

- (1) The GN_2 window purge produces visible gas density gradients which obscure the spray field at higher pressures. Consequently Task IV tests were conducted both with helium purge gas and without purge gas which verified this conclusion.

V Technical Discussion (cont.)

(2) The proposed monopropellant decomposition model of RSS is incorrect.

(3) RSS depends on surface reactions.

(4) RSS occurs in two modes depending on chamber pressure: (1) low enthalpy reaction (fuel saturation temperature less than impingement point temperature), (2) high enthalpy reaction (fuel saturation temperature greater than the impingement point temperature).

(5) The Weber number is a useful RSS correlation parameter and Task IV testing should explore the surface controlled reaction mechanism further.

D. Task IV - Verification of Governing Mechanisms

The work conducted during this task resulted in; (1) the improvement of the photographic and temperature probe technique which significantly improved the spray field clarity and understanding of reaction mechanisms, (2) the completion of 130 hot fire tests and 10 cold flow tests covering a wide range of operating conditions using an unlike doublet element, two triplet elements, and two platelet elements, (3) the development of RSS correlations for injector design purposes.

1. Task IV Experimental Results

The Task IV test objectives, conditions, and results are summarized in Table VII. The significant results of each group of tests are described below.

The first series of tests were run to improve both the temperature probe technique and the photographic definition of the reactive spray field. It had been found in the Task III

V Technical Discussion (cont.)

testing that gas density gradients produced by the cold GN_2 purge gas were visible under the high speed photographic conditions. The density gradients obscured the field of view at the higher pressures such that detailed description of the operating mode could not be made. It was reasoned that the density difference between the cold GN_2 and the hot combustion gas could be reduced by using helium in place of nitrogen. Testing with helium purge verified that to be true.

The photographs obtained using Helium purge were significantly clearer than those obtained with the nitrogen purge. However, the cold window permitted condensation of small quantities of fuel. The cost of using helium for the entire test program would have been prohibitive within the cost constraints of the program. Therefore, with the concurrence of the NASA technical monitor, it was decided to test with no purge to evaluate the ability of the hardware to run uncooled. The tests were highly successful in that no hardware damage was incurred and the pictures obtained were of the highest quality ever produced. Therefore, all subsequent tests were run without purge. The improved visibility allowed the mode of operation to be more finely classified. For example, it was possible to identify four distinct modes; mixing, mixing with penetration, mixing with incipient separation and separation as illustrated in Figure 7.

The thermocouple probe was reduced in size from a 0.020 inch (.0508 cm) diameter to a 0.010 inch (.0254 cm) diameter probe as shown in Figure 24. to lessen its influence on the impingement process. Cold flow tests were run with the smaller probe to determine its influence on the spray character. It was found that no visible disruption of the spray was discernible at pressure drops above 10 psi ($6.89 \times 10^4 \text{ N/m}^2$). The 0.010 inch (.0254 cm) diameter probe was supported by a 0.020 inch (.0508 cm) diameter sheath for rigidity as shown in Figure 24.

V Technical Discussion (cont.)

The next three sets of tests were run to determine the influence of chamber pressure, propellant temperature, injection velocity, and fuel type on the impingement point temperature rise (ΔT_i) and RSS. It was initially concluded that these influences on ΔT_i and RSS could be correlated with the Weber number parameter which is discussed in detail in Section IV.E. However, closer examination of the data led to the conclusion that factors accounting for the fuel reactivity must be included.

It was found that the Weber number at which RSS occurs depends on the propellant temperature as shown in Figure 25. The influence of fuel temperature on RSS was believed to be due to the vapor pressure dependence. Increasing propellant temperature increases the vapor pressure and hence the propellant partial pressures and propellant concentrations, making a more reactive vapor mixture. A model of propellant reactivity in terms of a vapor phase mixture ratio is described in Section IV.E. This vapor phase mixture ratio parameter was used to correlate RSS for MMH as previously shown in Figure 5. The A-50 data were found to also be correlated with the Weber number and MR_{vp} .

Visual examination of the movies showed the character of the fuel and oxidizer stream to change with velocity as evidenced by the reflective quality of the stream surfaces. The fuel stream surface was observed to be smooth and highly reflective at fuel velocities below about 67 ft/sec (20 m/sec). The fuel stream appearance transitions to a rough and non-reflective surface at velocities greater than this. It was initially thought that this was evidence of the onset of self atomization due to the Weber number influence. However, cold flow tests, reported in Section V.D.2 will show that the observed behavior corresponds to a transition from laminar to turbulent flow. As discussed in Section V.D.2 the contoured inlet suppresses the transition from laminar to turbulent flow such that transition occurs at a Reynolds number dependent on the orifice

V Technical Discussion (cont.)

L/D. Transition was predicted to occur at a Reynolds number of 12,500 for the unlike doublet fuel orifice which is in excellent agreement with the experiment. The oxidizer stream exhibited a similar transition in stream surface quality indicating a similar flow transition.

The impingement point temperature rise, ΔT_i , was plotted versus the fuel orifice Reynolds number as shown in Figure 26. These data appear to indicate that there is both a Reynolds number and propellant temperature influence on RSS.

The next series of tests were run using water and freon as propellant simulants to determine the influence of reaction on the observed cyclic nature of stream impingement. It was found that the observed jet shedding and atomization frequencies were the same as observed in the hot fire tests. The fuel orifice transition Reynolds number was also verified.

The final series of tests were run to map the RSS regimes for the X-doublet and splash plate platelet elements and two different sized EDM triplet elements. These elements produced similar trends of RSS with chamber pressure and the dimensionless reactivity parameter, R . The similarity of the trends is reflected by the small differences between the correlation coefficients and exponents shown on Table I. These data indicate parallel trends of minimum chamber pressure for RSS as a function of R . (Constant correlation exponent). Also the intercepts of the correlation equations are within $\pm 16\%$ of each other for the three injector types. The data contained in Appendix B show the correlation equations for these injectors to adequately characterize RSS.

V Technical Discussion (cont.)

2. Cold Flow Results

The unlike doublet elements were cold flow tested during Task III to determine their hydraulic resistance. The results are plotted in Figure 27. The injector resistances determined during Task IV for the two triplet and two platelet injectors are shown in Figures 28 and 29.

The unlike doublet injector was visually observed to flow in an attached mode over the entire flow range as verified by the smooth data plot. It was visually observed that the flow experienced a transition from laminar to turbulent as the pressure drop was increased. The transition occurs at a pressure drop of about 40 psi as shown in Figure 30. The transition from laminar to turbulent flow was not apparent from the cold flow plots of Figure 27, since most of the pressure losses were turbulent entrance and exit losses rather than boundary layer shear losses. The flow model described in Reference 23 predicts that the contoured inlet should suppress transition from laminar to turbulent flow until a Reynolds number based on orifice length of about 300,000 is reached. This corresponds to a Reynolds number based on diameter of about 12,500. The experimental data is in excellent agreement. It was also observed that the onset of self-atomization occurs at a Weber Number of about 12 which is in agreement with that reported in Ref. 24 for drop-let breakup.

The fuel orifices in both the small and large triplet elements were also observed to flow attached over the flow range. However, the oxidizer orifices were observed to experience hydraulic flip. The larger oxidizer orifice ($D_o = 0.049$ inch, .124 cm) detached at a pressure drop of about 25 psi whereas the smaller orifice ($D_o = 0.033$, .084 cm) detached at about 45 psi ($3.1 \times 10^5 \text{ N/m}^2$). The transition from attached to detached flow is apparent in the strobe lighted photographs shown in Figure 31.

V Technical Discussion (cont.)

Hydraulic flip is a consequence of flowing to ambient backpressure which permits cavitation in the vena contracta. The orifices were observed to flow in the attached mode at the hot test conditions. The only consequence of flowing detached in cold flow is that the discharge coefficients may be significantly less (.6 compared to .8) than for the attached hot fire test conditions. This could lead to errors in predicted hot fire test injector admittance, if not recognized.

The important observation from an RSS standpoint is that the triplets' sharp edged short L/D orifices all exhibit turbulent flow. Another important observation is, that the onset of self-atomization was found to occur at a Weber number of 3 as compared to 12 for the contoured long L/D unlike doublet orifice. The significance of this observation is that it identifies orifice geometry dependence on the onset of self-atomization and/or critical Weber number. It was concluded that turbulent propellant streams will produce greater interfacial surface area on impact than coherent laminar streams such that RSS is expected to be aggravated.

The conclusions drawn from Task IV are:

1. The use of cold window purge gases should be avoided in photographic test chambers.
2. RSS is a gas phase/surface reaction related phenomena which is influenced by injector hydraulics, chamber pressure, fuel type and temperature, and injection velocity.
3. The chamber pressure and injector hydraulics exert the strongest influence on RSS, with RSS increasing with increasing chamber pressure, propellant temperature and propellant stream surface area.

V Technical Discussion (cont.)

4. The chamber pressure above which severe RSS occurs is predicted by a dimensionless propellant reactivity parameter, R , which accounts for injector hydraulics (through the fuel Reynolds number and the vapor phase propellant through a normalized propellant partial pressure product.

$$R = Re_f \sqrt{P_{v0} \times P_{vf}} / P_0$$

5. Cyclic blowpart (popping) was not observed on any of the tests.

E. Experimental Hardware and Test Setup

1. Test Apparatus

The test apparatus consists of a test chamber equipped with transparent viewing ports and removable injectors and nozzles as shown in Figure 32. The test chamber and an unlike doublet injector were designed and fabricated during the Task III effort. Two triplet and two platelet injectors were designed and fabricated during the Task IV effort.

a. Test Chamber

The test chamber was machined from a 4-inch square x 6-inch long block of 304 CRES. The combustion chamber section is 4 inches (10.16 cm) long, to which a 2 in. (5.08 cm) L^* spacer is bolted to increase the combustion zone length to 6 inches (15.2 cm). The block was bored to provide a 2.75 inch (6.99 cm) diameter combustion chamber. Four circular quartz windows were provided to facilitate photography and to allow flexibility in photographic lighting of the combustion process. The windows are 1/2 inch (1.27 cm) thick to provide a safety margin for 1000 psia ($6.89 \times 10^5 \text{ N/m}^2$) operation. The flat quartz windows are sandwiched

V Technical Discussion (cont.)

between durabula gaskets for cushioning against ignition shocks and uneven loading. A silicon "O" ring provides sealing on the window periphery. Quartz windows are used to provide good propellant compatibility and well defined optical properties.

The chamber was designed to provide an inert gas (GN_2) film purge to prevent obscuring the view by propellant spray impingement on the windows. The gas purge flow is injected through four inlets into an annular manifold. The gas is directed from the manifold through an annular gap and made to flow around the periphery of the chamber wall. The gas passages were sized such that the GN_2 is injected into the chamber at 50 ft/sec (15.2 m/sec) at 300 psia ($2.07 \times 10^6 \text{ N/m}^2$) chamber pressure to minimize mixing with the propellant spray and combustion gas. Subsequent testing showed that the cold GN_2 purge gas causes poor spray field visibility due to the density gradient created between it and the hot combustion gas. A significant improvement in visibility was made by matching the purge gas density to the combustion gas density by using a cold helium purge. However, the best photographs were obtained with no purge at all.

Provision was made for mounting both high and low frequency response pressure transducers and thermocouples. The nozzles consist of removable copper inserts drilled to provide the desired operating pressures.

b. Injectors

(1) Task III Injectors

The injector body was made in a cylindrical "piston" shape as shown in Figure 33 to fit into the chamber purge ring located at the forward end of the chamber. The injector is held in the purge ring by allen head screws. A silicon rubber O-ring seals the injector to the purge ring.

V Technical Discussion (cont.)

The injector consists of a main body with brazed-on inlet tubes. The injectors were made of 304 CRES to permit braze assembly and provide dimensionally stable orifices. Two injector patterns were incorporated in one body as shown in Figure 33 to reduce fabrication costs. The unlike doublet element design parameters are included in Figure 33. The orifice L/D's were made long (24/1) with rounded inlets to provide controlled hydraulics.

The injector orifices were flow tested prior to the Task III hot fire testing to measure Kw's and to verify impingement accuracy. The flow data are discussed in Section V.D.2. Subsequent cold flow tests were run during Task IV to characterize the rounded inlet orifice hydraulics. Results of these tests are also discussed in Section V.D.2.

A high frequency response Kistler pressure transducer mounting port was provided in the long impingement doublet as shown in Figure 33 to measure impingement point disturbances. This port was also used to install a high response thermocouple for measuring the impingement point temperature rise.

(2) Task IV Injectors

The triplet element and the self-atomizing platelet element were selected for test evaluation during Task IV. Two different triplet injectors were designed to evaluate the influence of orifice diameter on RSS. The larger triplet element uses 0.030 inch (.076 cm) diameter fuel orifices and the smaller one uses 0.020 (.05 cm) inch dia. fuel orifices. The design details are shown in the schematic of Figure 34. Both elements were FDM machined into the same injector body to reduce fabrication costs. Separate propellant inlets were provided.

V Technical Discussion (cont.)

The orifices were designed with short L/D's and sharp edged inlets to simulate typical rocket injector orifice hydraulics. The results of a cold flow evaluation of the triplet injectors are discussed in Section V.D.2.

The X-doublet and the splash plate self atomizing platelet elements shown in Figure 35 were selected for test evaluation. Both elements were photoetched into the same platelet stack for minimum cost. The platelet stack was bolted to an injector body provided with a single set of propellant inlets. Manifolding of the desired element was accomplished by rotation of the platelet stack. "O" rings were used to seal the platelet stack propellant passages to the injector body.

The X-doublet element consists of a parallel self-atomized fuel and a parallel self-atomized oxidizer stream placed in close proximity to one another such that mixing occurs by parallel stream momentum exchange. Self-atomization is accomplished by self-impingement within the platelet stack as shown in Figure 35. The resultant atomized stream is ejected perpendicularly from the platelet stack. The splash plate element consists of one self-atomized fuel stream impinging on one self-atomized oxidizer stream as shown in Figure 35. The self-atomization is produced by impinging the propellant stream against a splash plate as shown in Figure 35.

2. Hot Fire Test Facility Setup

The test apparatus was setup in the Research Physics Laboratory Test Bay 2 shown in Figure 36.

A schematic of the propellant system used is shown in Figure 37. Propellant (MMH/A-50/NT0) is stored in 50-gallon, 3000-psi run vessels. Gaseous nitrogen pressurization of these systems was used to provide controlled run conditions over a wide range of injector and chamber pressures.

V Technical Discussion (cont.)

Propellant conditioning was provided by installing in-line heat exchangers immediately upstream of the thrust chamber valves. A hot water circulation type temperature conditioning system was used to provide independent conditioning of the ox and fuel to temperatures from ambient to 300°F.

A separately regulated GN₂ supply was used to provide test chamber back pressure as well as provide window purge for the chamber viewports during the Task I testing. The window purge was eliminated early in the Task IV testing to improve photographic quality as discussed in Section V.D.1.

3. Cold Flow Test Setup

The cold flow tests were also conducted in the Research Physics Laboratory. Filtered, de-ionized water was used as the test fluid on most tests. Pressure measurements were made using Heise gages and flow rate was calculated using a time/volume technique, with run times of from 60 to 200 seconds. On some tests strobe light photographs were obtained in order to better evaluate propellant stream properties. Selected tests also were made with dyed water and freon as the test fluids in order to evaluate stream impingement characteristics.

4. Hot Fire Instrumentation

High speed color photographs of the spray field were taken with the photographic equipment shown in Figure 36. Pictures were taken over a wide range of exposures, from 8000 pictures per second (1.25 μ sec exposure) down to 400 PPS (25 μ sec exposure) with a Hycam Model 41-0004 high speed camera. Four hundred foot rolls of Ektachrome EF No. 7242 film were used which allows approximately 0.6 sec. of constant speed frame rate at 8000

V Technical Discussion (cont.)

PPS and approximately 30 sec. constant speed at 400 PPS.

Lighting of the spray field was accomplished with the use of three 1000-watt quartz iodine lamps focused with Fresnel lenses. One lamp was used to backlight the spray area with the second and third lamps used as top and front lighting to provide spray detail and definition.

The lighting was improved during the Task IV testing by replacing the front and top lights with smaller 750 watt lamps and adding another lamp to light the bottom port. The smaller lamps were placed within one inch of the top and bottom windows to maximize the illumination. The net effect was to improve the balance between the front lighting and the back lighting.

The high frequency and low frequency instrumentation listed in Tables VIII and IX were used in the locations shown in the schematic of Figure 38.

Low frequency response test parameters were recorded on a Consolidated Electrodynamic Corporation's direct writing oscillograph. The high frequency response data were recorded on a Sangamo Model 3564 analog tape recorder.

The operating point data indicated in Table IX were digitized and stored in the on-line HP 2100 A Computer/Real Time process controller for "quick look" test review.

REFERENCES

1. Lawver, B. R., "Rocket Engine Popping Phenomena", Aerojet-General Report No. TCER 9642:0095, March 1969
2. Mills, T. R., Tkachenko, E. A., Lawver, B. R., and Breen, B. P., "Transients Influencing Rocket Engine Ignition and Popping", Dynamic Science Report No. NAS 7-467, January 1969
3. Zung, L. B., "Hypergolic Impingement Mechanisms and Criteria for Jet Mixing or Separation", presented at the 6th ICRPG Liquid Propellant Combustion Instability Conference, 9-11 September 1969
4. Nurick, W. H. and Cosdill, J. D., "Reactive Stream Separation Photography", Final Report R-8490, Contract NAS 7-720, Rocketdyne, 1971
5. Lee, A., and J. Houseman, "Popping Phenomena with N_2O_4/N_2H_4 Injectors", presented at the Western States Section meeting of the Combustion Institute on Stable Combustion of Liquid Propellants, JPL, October 26-27 1970.
6. Elverum, G. W., Jr., and Staudhammer, P., "The Effect of Rapid Liquid-Phase Reactions on Injector Design and Combustion Rocket Motors", Progress Report 30-4. Jet Propulsion Laboratory, Pasadena, California, 25 August 1959
7. Burrows, M. C., "Mixing Reaction of Hydrazine and Nitrogen Tetroxide at Elevated Pressure", AIAA Journal, Volume 5, No. 9, pp. 1700-1701, September 1967
8. Weiss, R. R. and Klopotev, R. D., "Experimental Evaluation of the Titan III Transtage Engine Combustion Stability Characteristics", TR WO AFRPL TR-66-51, March 1966
9. Evans, D. D., Stanford, H. B., and Riebling, R. W., "The Effect of Injector Element Scale on the Mixing and Combustion of Nitrogen Tetroxide-Hydrazine Propellant", Technical Report 32-1178, Jet Propulsion Laboratory, Pasadena, California, 1 November 1967
10. Clayton, R., "Experimental Observations Relating the Inception of Liquid Rocket Engine Popping and Resonant Combustion to the Stagnation Dynamics of Injection Impingement", TR 32-1479. JPL, 15 December 1970
11. Houseman, J., "Jet Separation and Optimum Mixing for an Unlike Doublet", presented at the 6th ICRPG Liquid Propellant Combustion Instability Conference, 9-11 September 1969

References (cont.)

12. Kushida, R., Houseman, J., "Criteria for Separating of Impinging Streams of Hypergolic Propellants", Technical Memorandum 33-395, Jet Propulsion Laboratory, Pasadena, California, July 1968
13. Sawyer, R. F., Glassman, I., "Gas-Phase Reactions of Hydrazine with Nitrogen Dioxide, Nitric Oxide, and Oxygen", Eleventh International Symposium on Combustion, The Combustion Institute, 1967, p. 861
14. Nadiz, E. W., and Breen, B. P., "Liquid Phase Kinetics for the Hypergolic Nitrogen Tetroxide Hydrazine System", presented at ICRPG 3rd Combustion Conference, October 1966
15. Vander Wall, E. M., Suder, J. K., Beegle, R. L., Jr., and Cabeal, J. A., "Propellant/Material Compatibility Study", Technical Report AFRPL-TR-71-41, December 1971
16. Perlee, H., et al., "Hypergolic Ignition and Combustion Phenomena in the Propellant System AeroZINE-50 N_2O_4 ", Final Report No. 4019, Bureau of Mines, Pittsburgh, Pennsylvania, 1 April 1965 to 31 March 1967
17. Breen, B. P., Zung, L. B., Lawver, B. R., "Injection and Combustion of Hypergolic Propellants", AFRPL-TR-69-48, Dynamic Science, a Division of Marshall Industries, Monrovia, California, April 1969
18. Lawver, B. R. and Breen, B. P., "Hypergolic Stream Impingement Phenomena-Nitrogen Tetroxide/Hydrazine", NAS-CR-72444, Dynamic Science Division, Marshall Industries, Monrovia, California, October 1968
19. Weiss, H. G., and Johnson, B., "Modification of the Hydrazine-Nitrogen Tetroxide Ignition Delay", AIAA J. Volume 2, No. 12, December 1964
20. Rodriguez, S. E., and Axworthy, A. E., "Liquid Phase Reactions of Hypergolic Propellants", Rocketdyne Final Report R-8374, Contract NAS 7-739, 1 December 1970
21. Lawver, B. R., "Some Observations on the Combustion of N_2H_4 Droplets", AIAA Journal, Volume 4, No. 4, p. 659, April 1966
22. Zung, L. B., Breen, B. P., and Kushida, R., "A Basic Study of Ignition of Hypergolic Liquid Propellants", Paper No. 68-43 presented at Western States Section of the Combustion Institute, October 1968

References (cont.)

23. Ito, J. I., "A General Model Describing Hydraulic Flip in Sharp Edge Orifices", CPIA 204, Volume I, Feb. 1971
24. Wallis, B. W., "One Dimensional Two-Phase Flow", McGraw-Hill, 1969
25. Hines, W. S., "High Performance N_2O_4 /Amine Elements", Contract NAS 9-14126, Task III Data Dump, 10 March 1975
26. Ito, J. I., "OMS Subscale Stability Study", ALRC ADR to be published
27. Zung, L. B., and White, J. R., "Combustion Process of Impinging Hypergolic Propellants," NASA CR 1704, Marshall Industries, Irvine, California, May 1971

TABLE I. RSS CORRELATION CONSTANTS

FUEL	INJECTOR DESCRIPTION	a	b	FIGURE NO. SHOWING CORRELATION
MMH	Unlike Doublet - Rounded Inlet	3300	-0.447	1, B-1, B-2, B-3
A-50	Unlike Doublet - Rounded Inlet	3300	-0.447	B-4
N ₂ H ₄	Unlike Doublet - 100/1 L/D	387	-0.05	B-5, B-6, B-7, B-8
MMH	Triplet	255	-0.05	B-9, B-10
MMH	X-Doublet	206	-0.05	B-11, B-12, B-13
MMH	Splash Plate	193	-0.05	B-14
MMH	V-Doublet	157	-0.05	B-15, B-16, B-17
MMH	Like-on-Like	125	-0.05	B-18, B-19, B-20, B-21

Correlation Equation:

$$P_{CRSS} = a R^b$$

where:

$$P_{CRSS} =$$

$$R = Re_f \sqrt{\frac{P_{vo} \times P_{vf}}{P_c}}$$

$$Re_f = \text{Fuel orifice Reynolds Number}$$

$$P_c = \text{Chamber Pressure, psia}$$

$$P_{vo}, P_{vf} = \text{Oxidizer and fuel vapor pressure}$$

TABLE II

RSS STUDY PARAMETER SUMMARY

PARAMETER	RANGE INCLUDED IN STUDIES
Propellant Combination	N_2O_4/N_2H_4 , N_2O_4/MMH , $N_2O_4/A-50$, $N_2O_4/UDMH$, $N_2O_4/M-50$, $N_2O_4/Furfuryl\ Alcohol$, IRFNA/UDMH, ClF_5/N_2H_4
Element Type	Unlike Doublet, Like Doublet, Quadlet, F-O-F and O-F-O Triplets
Element Size	.020 to .236 in. Dia. (.051 - .599 cm)
Element Mixing Efficiency	0.1 to 1.0
Element Spacing	Not Specified
Injection Velocity	5 to 145 ft/sec (1.5 - 44.2 m/sec)
Chamber Pressure	Atmospheric to 500 psia ($3.44 \times 10^6\ N/m^2$)
Fuel Temperature	40 to 250°F (278 - 394°K)
Oxidizer Temperature	-10 to 140°F (250 - 333°K)
Mixture Ratio	0.5 to 3.0
Stream Dynamic Pressure Ratio	0.3 to 8.0 Fuel/Oxidizer

TABLE III
COMPARISON OF HEAT RELEASE RATES AND VAPOR PRESSURE

Fuel	P_V at 100°F, (311°K) psia (N/m ²)	Heat Release Rate Kcal/sec-mole of NTO
N ₂ H ₄	0.65 (4.47 x 10 ³)	4 x 10 ⁴
UDMH	5.9 (40.7 x 10 ³)	14 x 10 ⁴
MMH	1.9 (13.1 x 10 ³)	20 x 10 ⁴

TABLE IV
PREDICTED AND MEASURED RSS PRESSURE LIMITS

Fuel	P_V at 100°F (311°K) psia (N/m ²)	Predicted Limit psia (N/m ²)	Measured* Limit psia (N/m ²)	Orifice Type
N ₂ H ₄ ***	.65 (4.47 x 10 ³)	300 (2.07 x 10 ⁶)	300 (2.02 x 10 ⁶)	Unlike Doublet L/D = 100
MMH	1.9 (40.7 x 10 ³)	400 (2.76 x 10 ⁶)	150 (1.03 x 10 ⁶)	Unlike Doublet L/D = 24**
A-50	4.5 (13.1 x 10 ³)	500 (3.45 x 10 ⁶)	200 (1.38 x 10 ⁶)	Unlike Doublet L/D = 24**

*Ambient temperature propellants

**Contoured Inlets

***Zung, Ref. 27

TABLE V
TASK III TEST OBJECTIVES AND RESULTS

TEST OBJECTIVE	FUEL	INJECTOR	T _O (°F)	T _F (°F)	P _C (PSIA)	NO. TESTS	RESULTS
CHECKOUT TESTS	MMH	LONG IMPING.	AMB	AMB	300-1000	6	ALL SHOWED SEP.
DETERMINE PRESSURE LIMIT FOR RSS	MMH	LONG IMPING.	AMB	AMB	100-300	5	SEP ABOVE 150 PSI
DETERMINE VAPOR PRESSURE EFFECT ON RSS	A-50	LONG IMPING.	AMB	AMB	100-1000	12	SEP ABOVE 200 PSI
DETERMINE EFFORT OF IMPING. LENGTH ON RSS	MMH	SHORT IMPING.	AMB	AMB	100-1000	9	NO DISCERNIBLE DIFFERENCE
DETERMINE EFFECT OF TEMPERATURE ON RSS	MMH	LONG IMPING.	AMB	200	100-250	5	SEP AT 100 PSI
MEASURE IMPINGEMENT POINT TEMPERATURE	MMH	LONG IMPING.	AMB	AMB	100-1000	14	IMPING. PT. TEMPERATURE IS VELOCITY & PRESSURE DEPENDANT

TABLE VI
COMPARISON OF HEAT RELEASE RATES AND RSS LIMIT

<u>Fuel</u>	<u>Heat Release Rate Kcal/sec-mole of NTO</u>	<u>Measured RSS Limit psia, N/m²</u>
N ₂ H ₄	4×10^4	300 (2.07×10^6)
A-50	-	200 (1.38×10^6)
UDMH	14×10^4	-
MMH	20×10^4	150 (1.03×10^6)

TABLE VII

TASK IV - TEST OBJECTIVES AND RESULTS

Test Objective	Fuel	Injector	D_f (in)	T_o (°F)	T_F (°F)	P_c (psia)	ΔP_f (psi)	No. of Tests	Results
Improve Photography & T/C Probe Technique	MMH	Unlike Doublet	0.020	70	70	100-300	30-100	10	1. Best pictures are obtained without GN ₂ Purge 2. Helium purge is better than GN ₂ purge. 3. 0.010 in. dia. T/C Probe provides reliable T_i with minimum disruption.
Propellant Temp Effect on T_i	MMH	Unlike Doublet	0.020	70-124	70-200	100-200	30	8	Temperature appears to influence ΔT_i through propellant property changes.
Velocity Effect on T_i	MMH	Unlike Doublet	0.020	Amb	Amb	100-300	10-100	27	Velocity and chamber pressure exhibit controlling influence on ΔT_i & RSS
Fuel Effect on T_i	A-50	Unlike Doublet	0.020	Amb	Amb	100-500	20-150	20	Fuel effect can be correlated with vapor phase mixture ratio
Jet Shedding Freq.	Water	Unlike Doublet	0.020	Amb	Amb	14.7	10-100	10	Jet Shedding frequencies same as hot fire
Map RSS	MMH	XDT1-Platelet	0.021	70-150	70-290	80-150	20-125	16	
Map RSS	MMH	Splash Plate Platelet	0.021	70-150	70-290	80-150	20-125	15	
Map RSS	MMH	Triplet	0.029	70-150	70-290	80-150	20-125	11	
Map RSS	MMH	Triplet	0.020	70-150	70-290	80-150	20-125	12	
*Map RSS	MMH	Unlike Doublet	0.020	70-150	70-290	80-150	20-125	11	RSS correlates with Weber Number and vapor phase MR
									140

* Bonus Tests

TABLE VIII
HIGH FREQUENCY RESPONSE INSTRUMENTATION

<u>Test Parameter</u>	<u>Symbol</u>	<u>Instrument Make</u>	<u>Model</u>	<u>Range</u>	<u>Accuracy</u>
Oxidizer Manifold Pressure	POJHF	Kistler	601	0-3000 psi (P-P)	$\pm 0.5\%$
Fuel Manifold Pressure	PFJHF	Kistler	601	0-3000 psi (P-P)	$\pm 0.5\%$
Chamber Pressure	PCHF	Kistler	601	0-3000 psi (P-P)	$\pm 0.5\%$
Injector Acceleration	ACC	-	-	0-500 g's	$\pm 0.5\%$
Injector Probe Temperature	TP1	C/A		0-500 °F	$\pm 1\%$

TABLE IX
LOW FREQUENCY RESPONSE INSTRUMENTATION

TEST PARAMETER	SYMBOL	RANGE	UNITS	RECORDER			
				"O"	GRAPH	TAPE	DIGITAL
Oxid. Tank Pressure	POT	0-1500	PSIA	X			
Fuel Tank Pressure	PFT	0-1500	PSIA	X			
Oxid. Injector Pressure	POJ	0-1500	PSIA	X			X
Fuel Injector Pressure	PFJ	0-1500	PSIA	X			X
Chamber Pressure	PC	0-1000	PSIA	X			X
Window Purge Pressure	PNZ	0-2000	PSIA	X			X
Oxid. Flowrate	WO	0-0.1	LB/SEC	X			X
Fuel Flowrate	WF	0-0.1	LB/SEC	X			X
Oxid. Flowmeter Temp.	TOFM	0-500	°F	X			X
Fuel Flowmeter Temp.	TFFM	0-500	°F	X			X
Oxid. Injector Temp.	TOJ	0-500	°F	X			
Fuel Injector Temp.	TFJ	0-500	°F	X			
Oxid. Valve Voltage	VOV			X			
Fuel Valve Voltage	VFW			X			
Wind Purge Valve Voltage	VWPV			X			
Camera Voltage	VCAM			X		X	
Injector Purge Valve Voltage	VIPV			X			

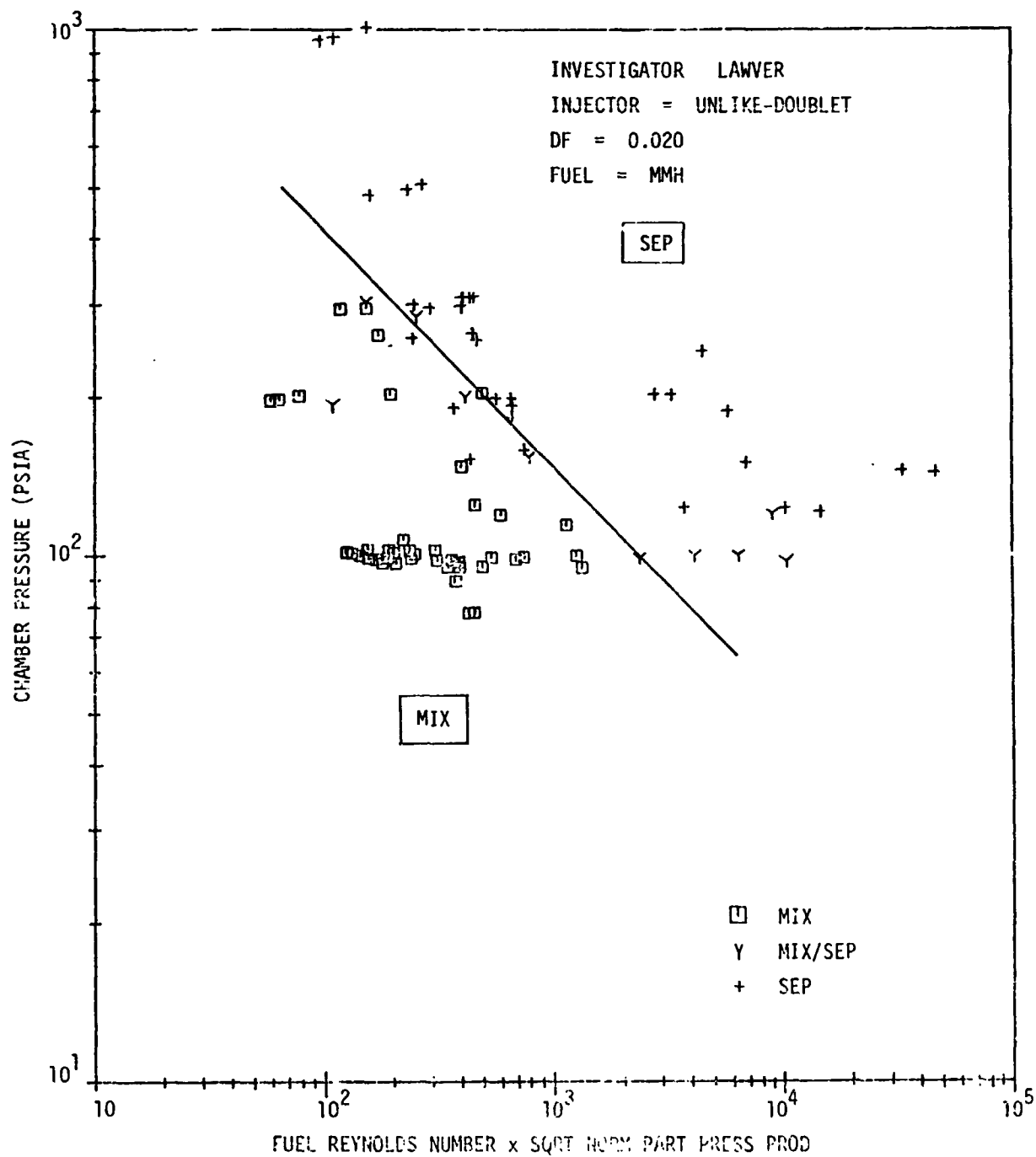


Figure 1. Effect of Chamber Pressure and Reactivity on RSS

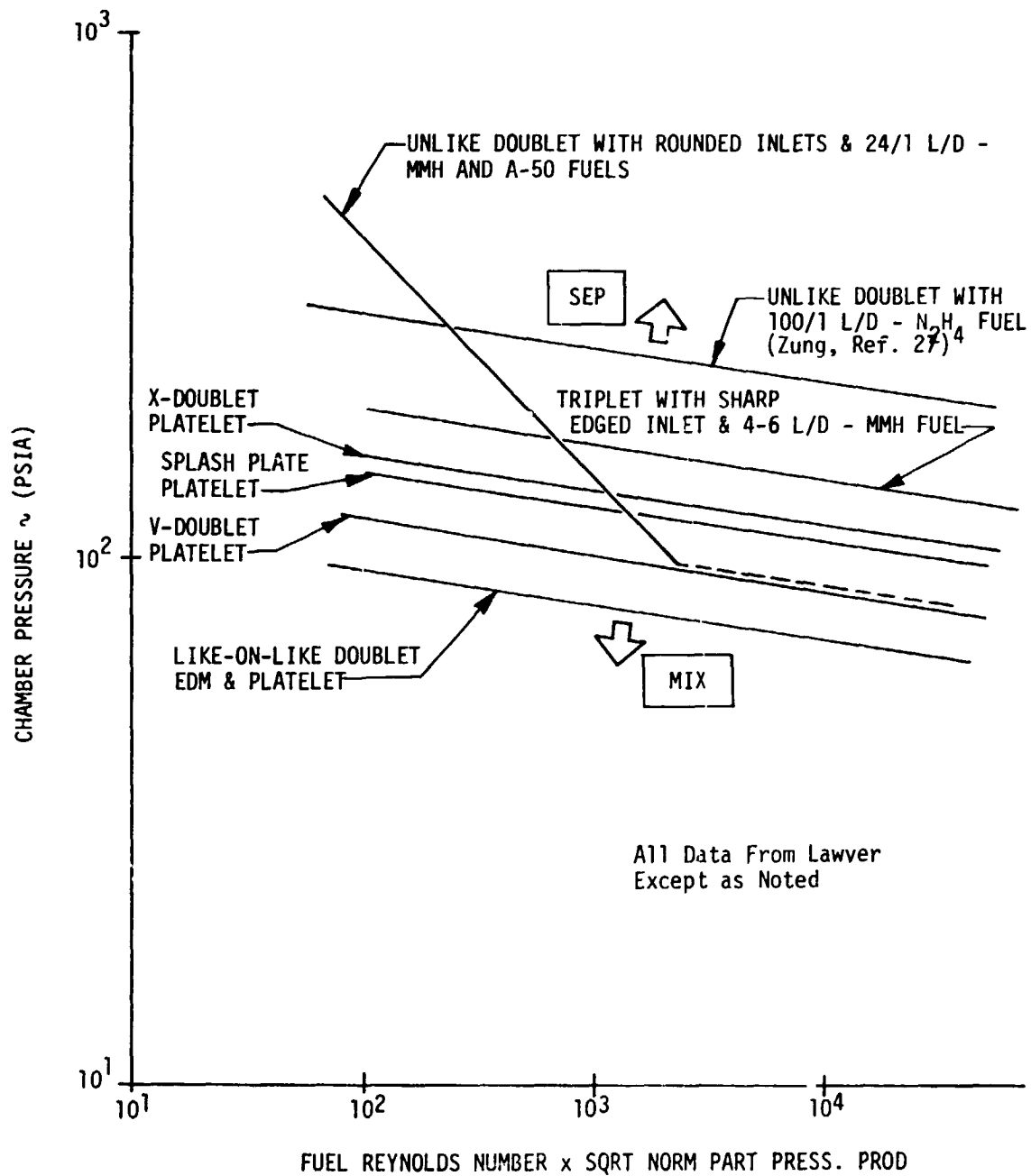
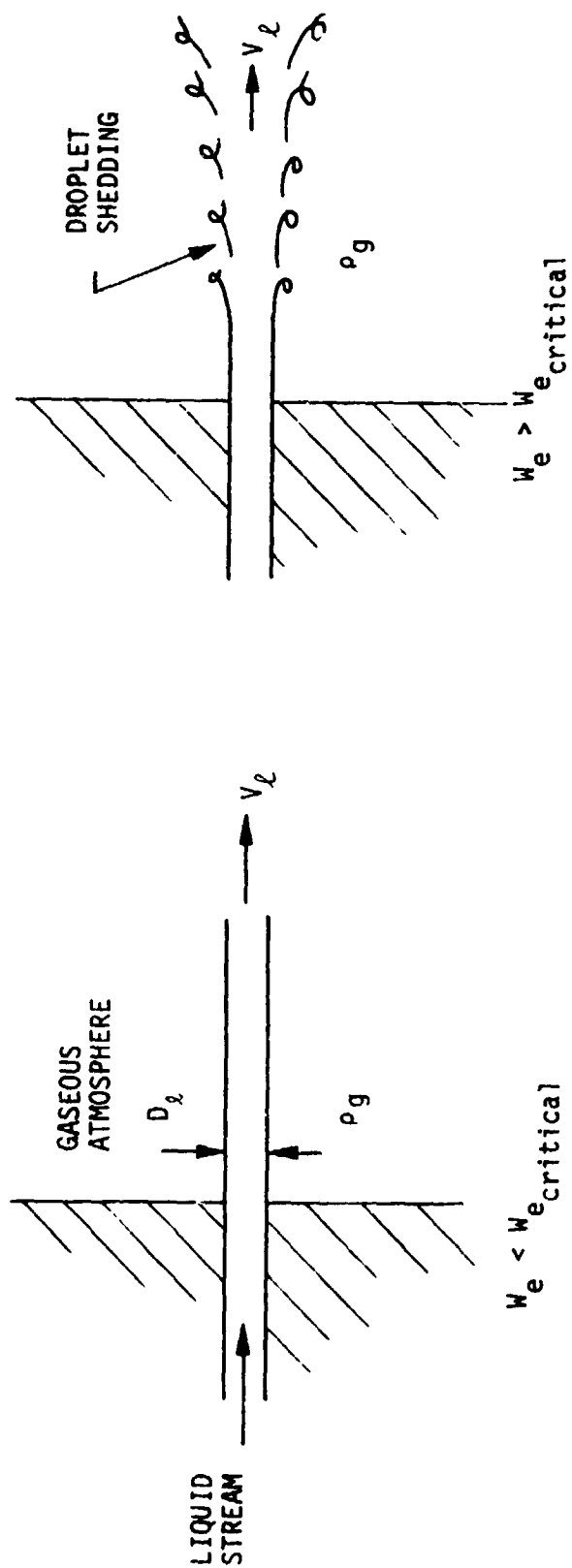


Figure 2. RSS Correlation Plot



$$W_e < W_{e \text{ critical}}$$

$$W_e = \text{AERODYNAMIC SURFACE FORCE / SURFACE TENSION}$$

$$W_e = \frac{\rho_g \Delta V_l^2 D_l}{\sigma_l g_c}$$

$$W_{e \text{ crit}} = 1.0 + \left(\frac{\mu_l^2}{\rho_l D_l \sigma_l g} \right)^{0.36} \quad \text{[(Ref. 24)]}$$

$$\rho_g = \frac{P_c}{R T_c}$$

Figure 3. Weber Number Model

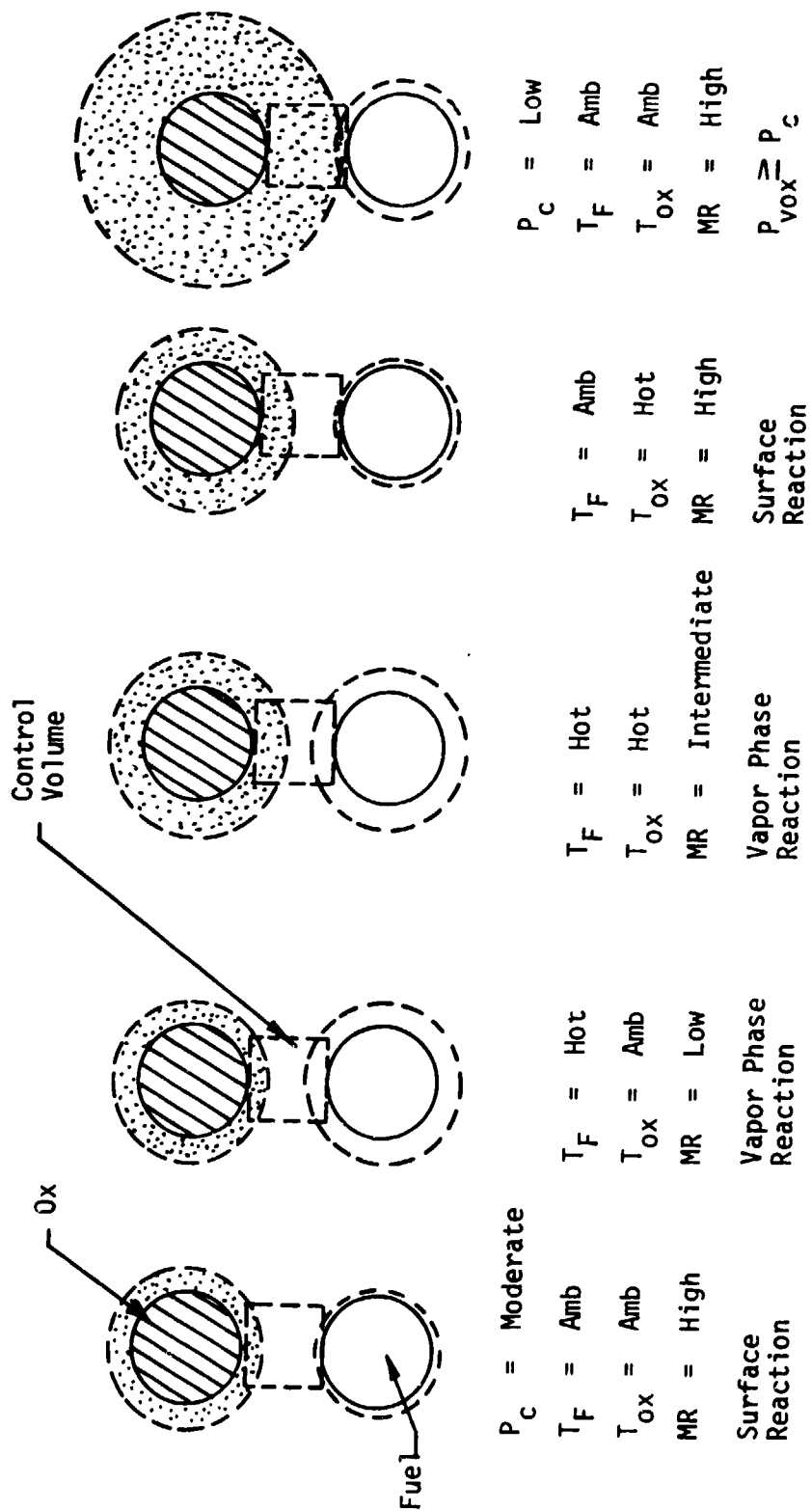


Figure 4. Vapor Phase Reaction Model

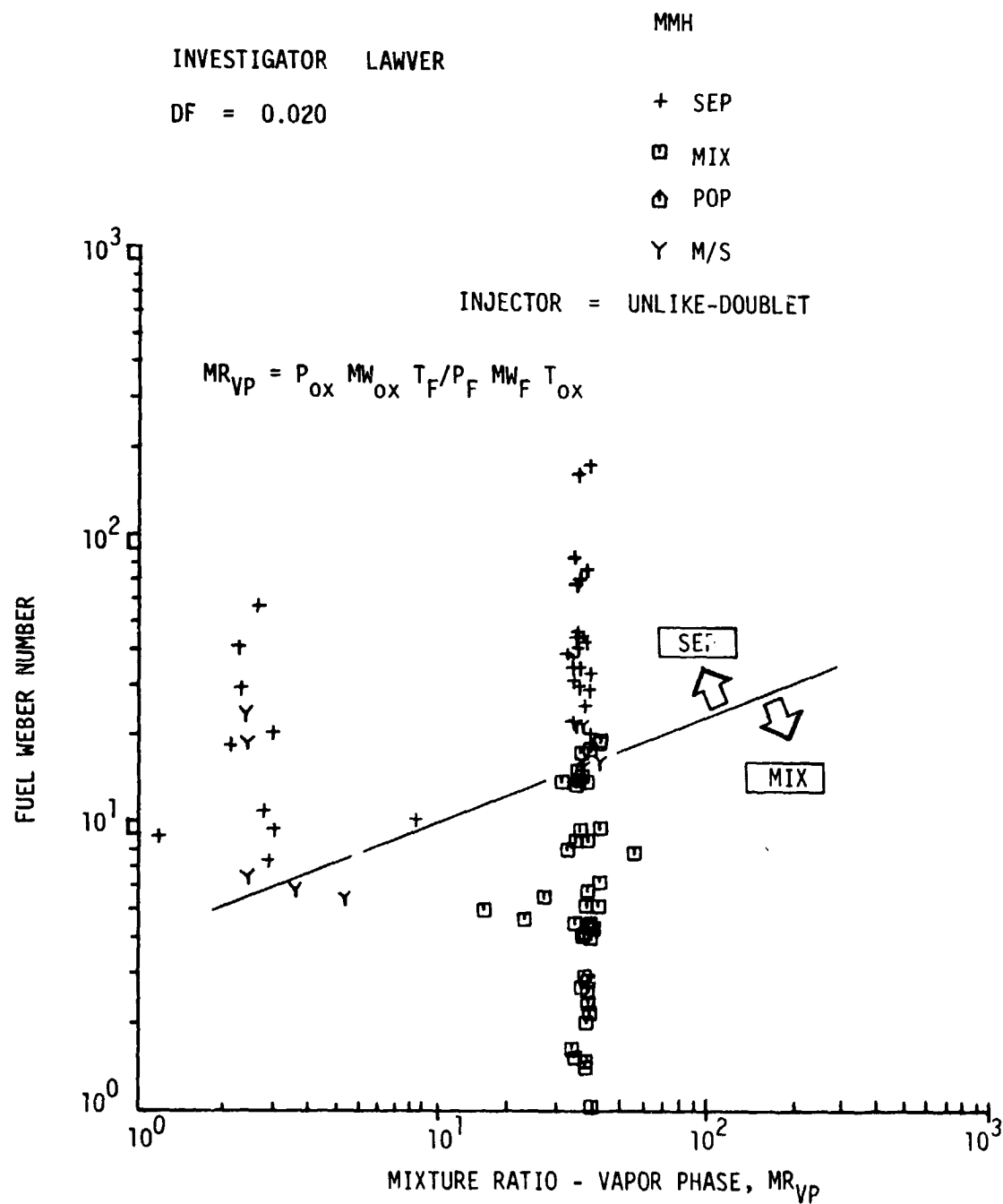


Figure 5. Effect of Fuel Weber No. and Vapor Phase Mixture Ratio on RSS

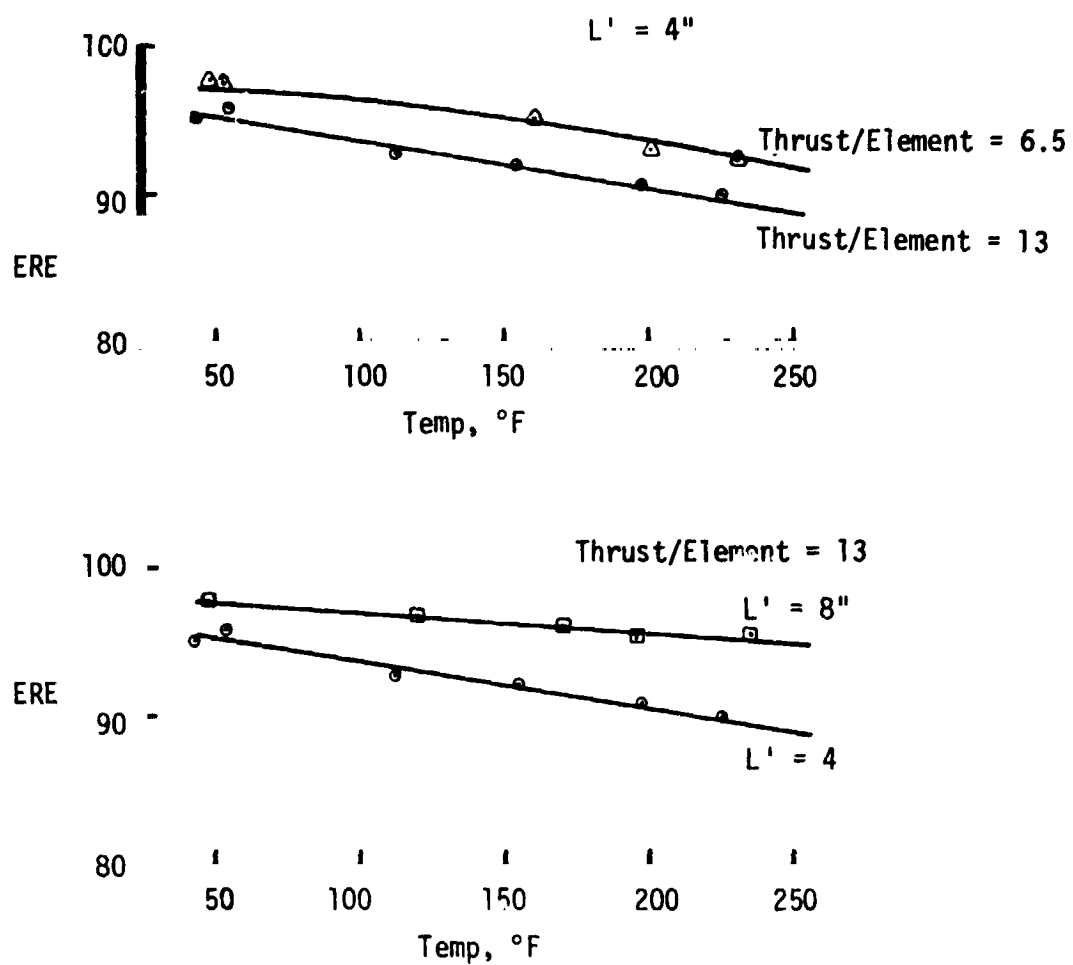
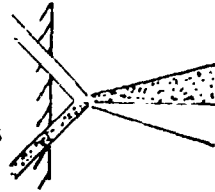


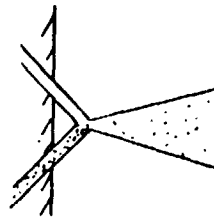
Figure 6. Effect of Fuel Temperature on ERE

- **PENETRATED** - Injection process results in the penetration of the fuel and oxidizer with propellants unmixed on the opposite side of the element. Same effect on combustion efficiency as separated flow regime.



Low Reactivity Propellants
Small Orifice Size
Low Injection Velocity

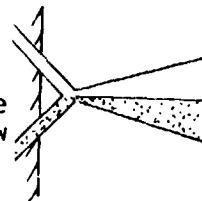
- **MIXED** - Injection process results in uniform mixture of fuel and oxidizer and high combustion efficiency.



Lower Propellant Temperatures
Lower Reactivity Propellants
Smaller Orifice Size
Unequal Stream Dynamic Pressures

- **PENETRATED/MIXED** - Injection process results in some propellant penetration, but with a mixed region also.

- **SEPARATED** - Injection process results in the separation of fuel and oxidizer before mixing with the propellants remaining in their respective side of the element. Results in a low combustion efficiency which will improve to some degree with increasing chamber length.



High Fuel Temperature
High Oxidizer Temperature (oxidizer vaporization)
Highly Reactive Propellants
Larger Orifice Size

- **POPPING** - Cyclic blowapart of a single element or the spray detonation of a large group of elements. Random occurrence which could lower the overall time averaged combustion efficiency and effect combustion stability.



Lower Propellant Temperatures
Lower Chamber Pressure
Larger Orifice Size
Higher Contact Time (D/V)
Reactive Propellants
Equal Stream Dynamic Pressures

Figure 7. Reactive Stream Impingement Regime

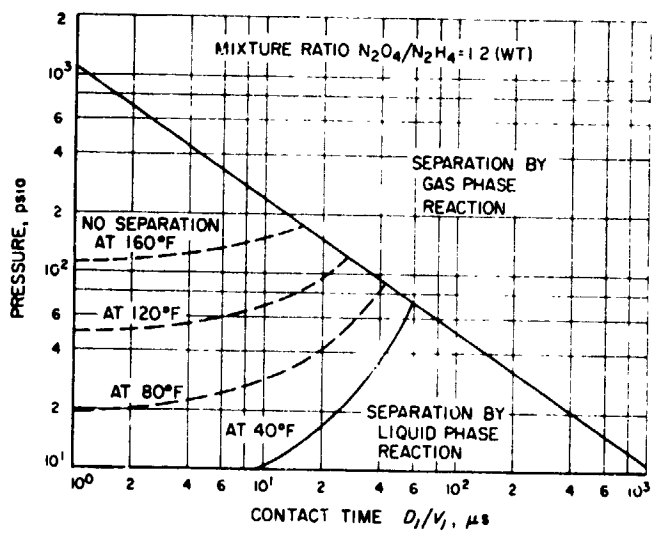


Figure 8. Predicted RSS Limits - JPL Model

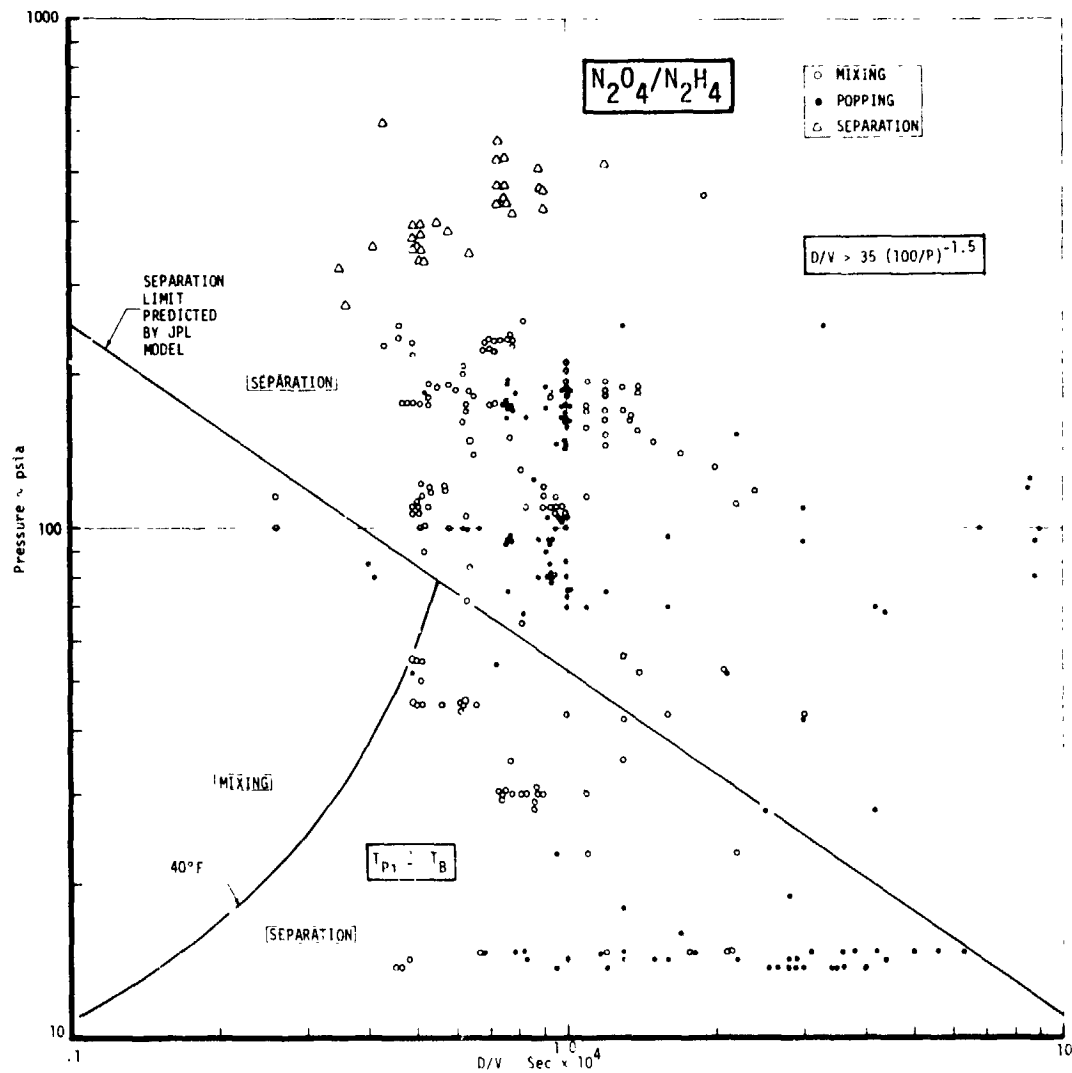


Figure 9. RSS Data Correlation - JPL Model

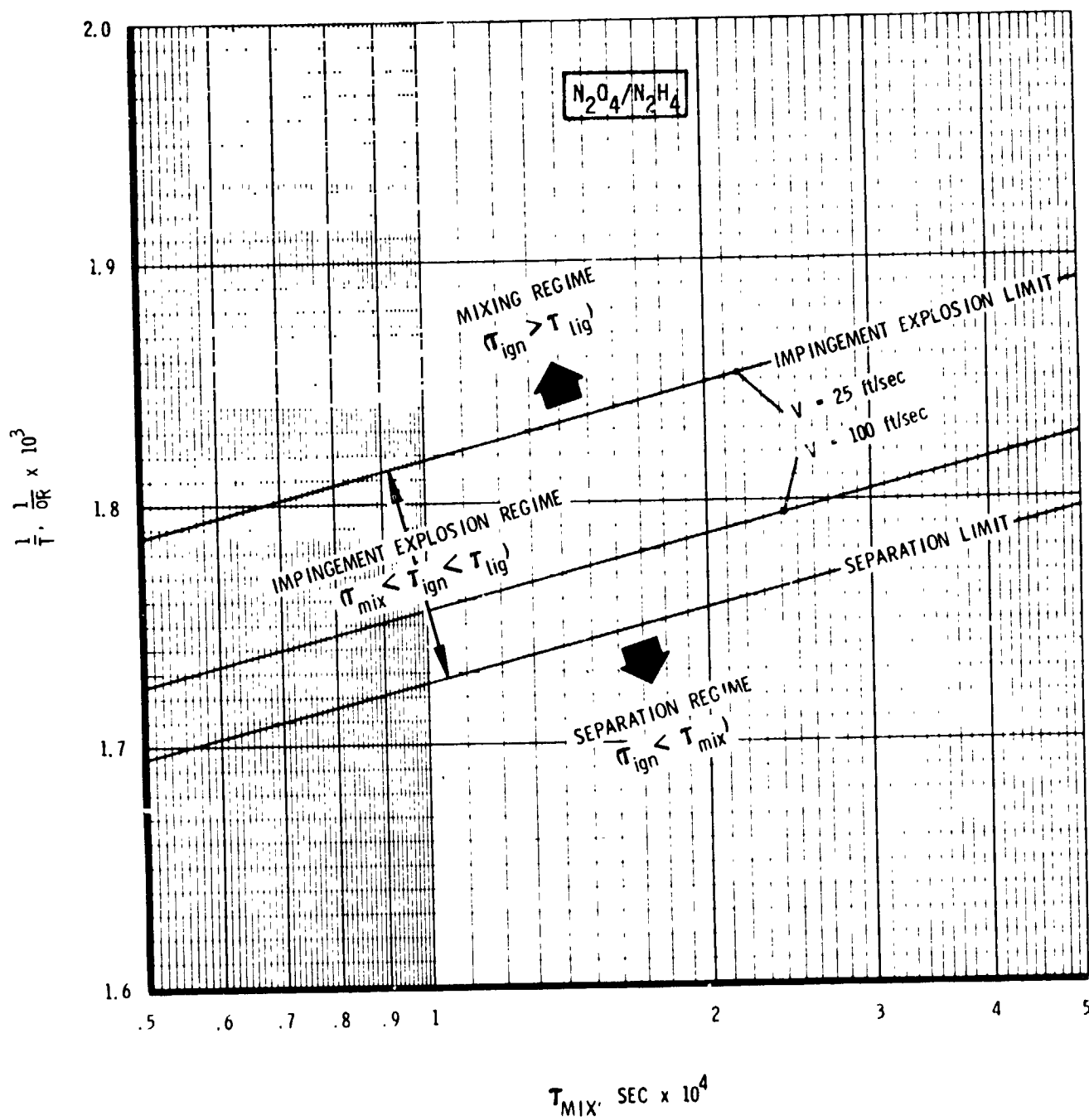


Figure 10. Predicted RSS and Pop Limits - ALRC Model

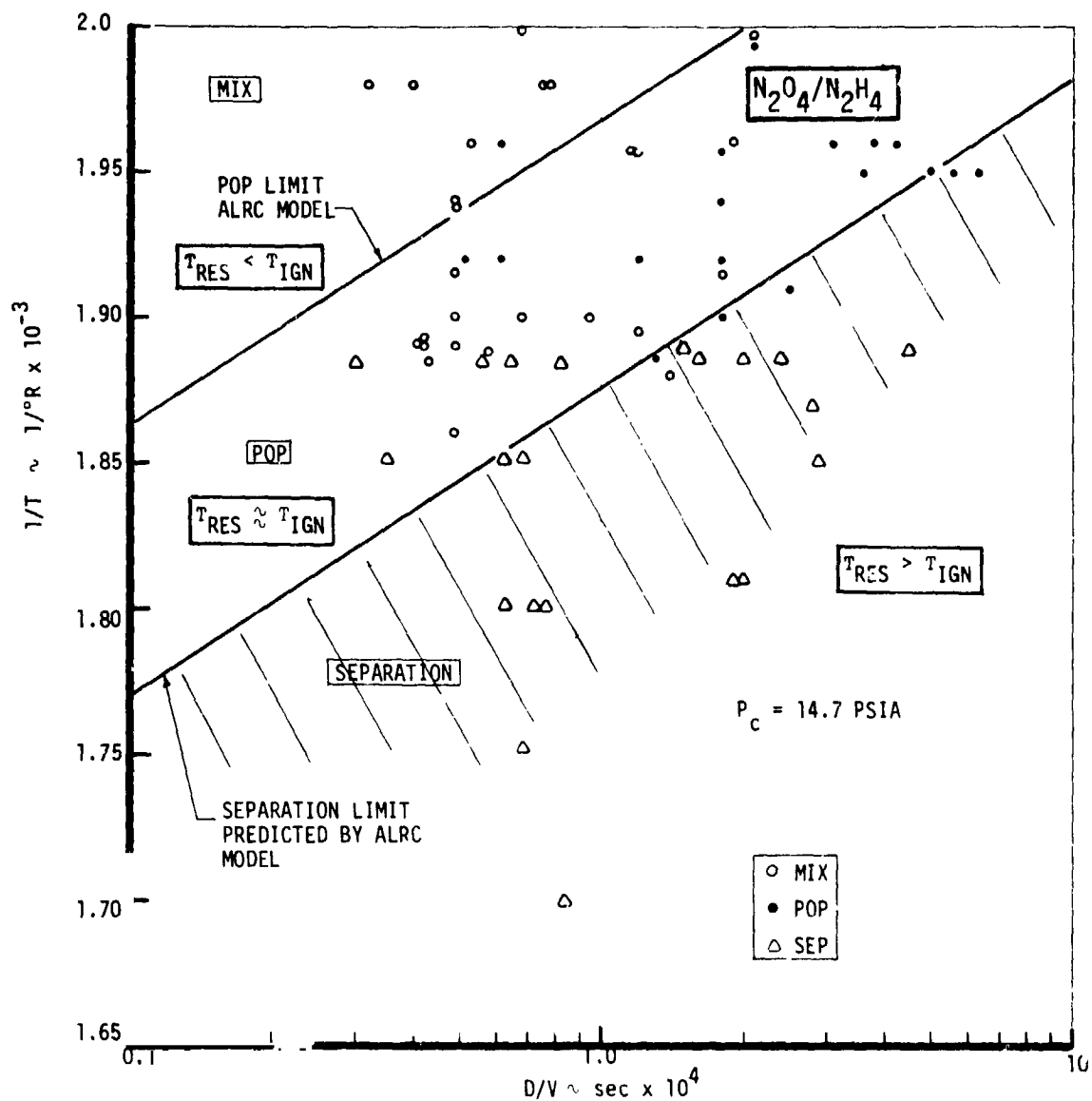


Figure 11. RSS and Pop Data Correlations - ALRC Model

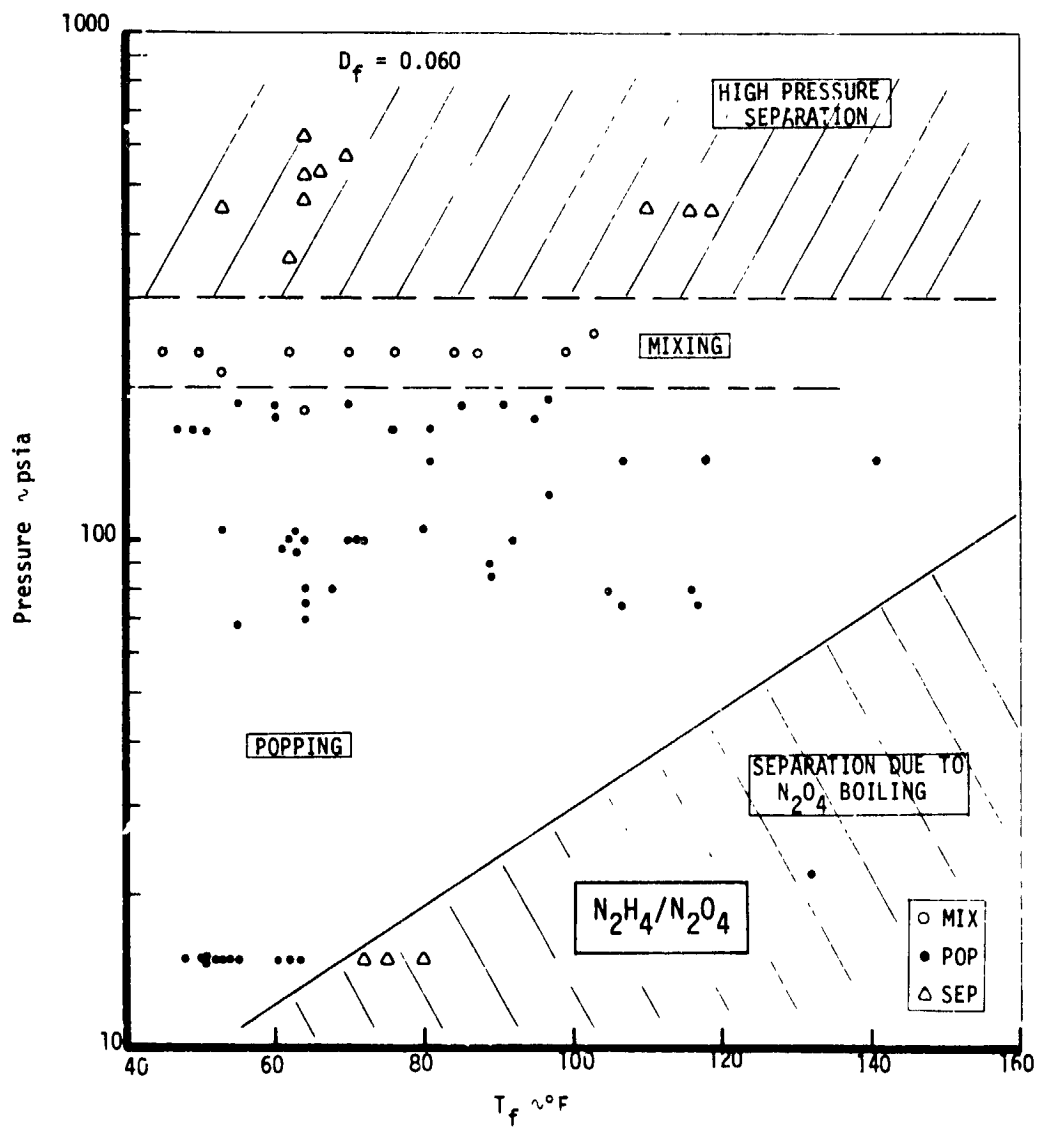


Figure 12. Pressure vs Temperature - $D_f = 0.060$

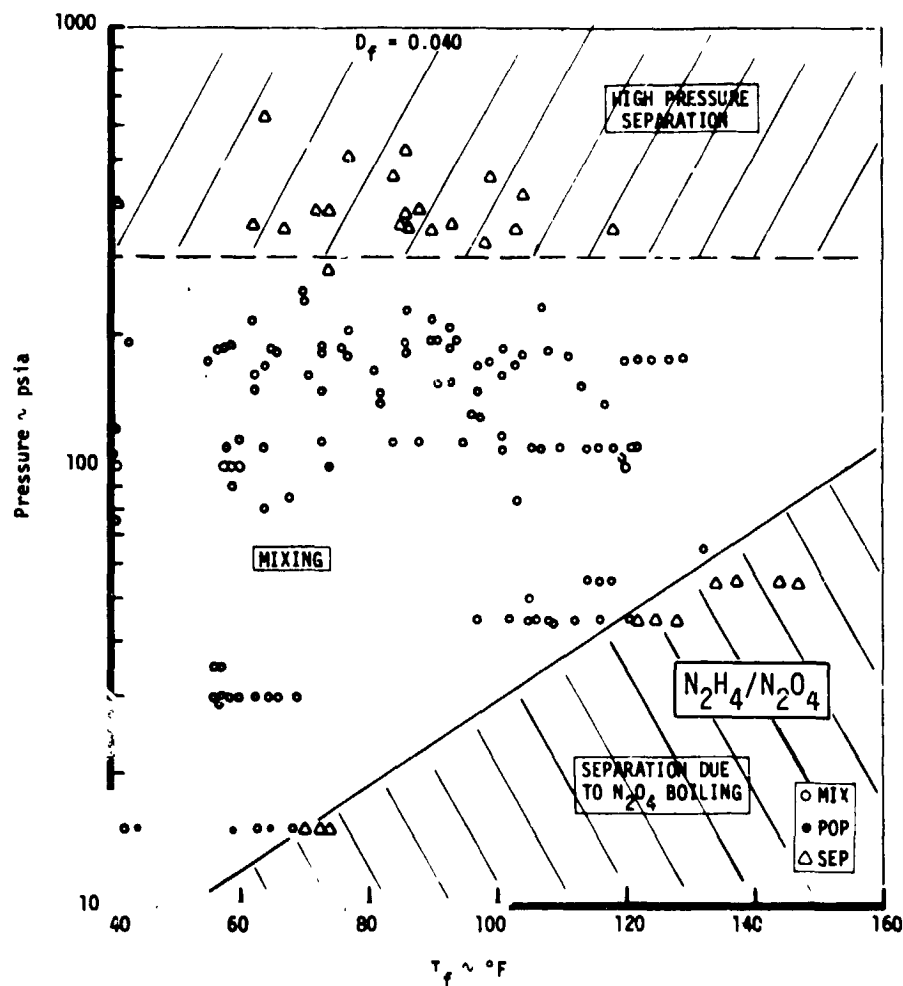


Figure 13. Pressure vs Temperature - $D_f = 0.040$

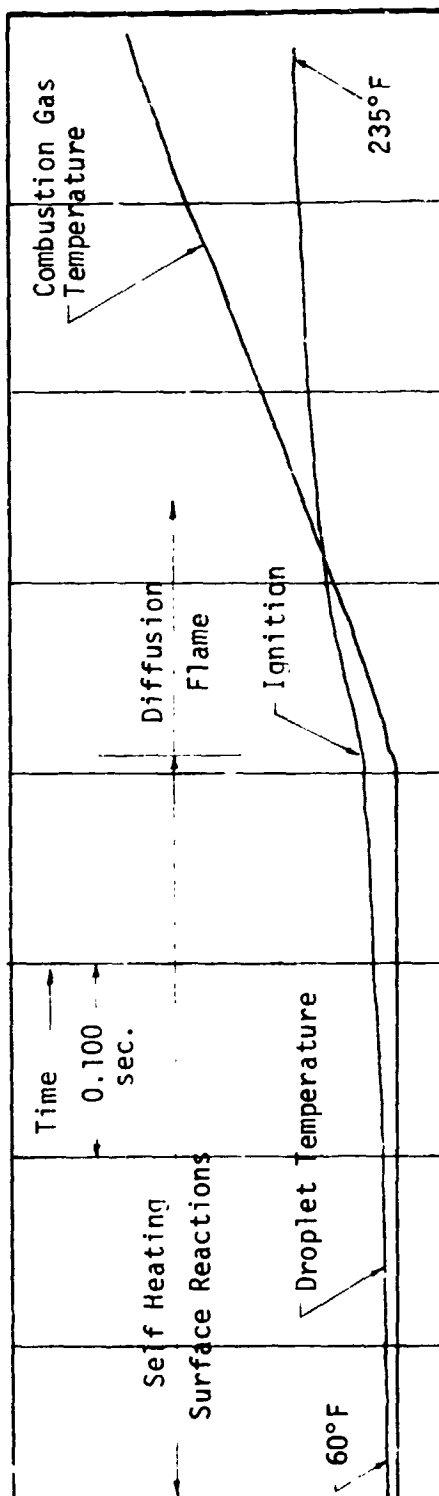
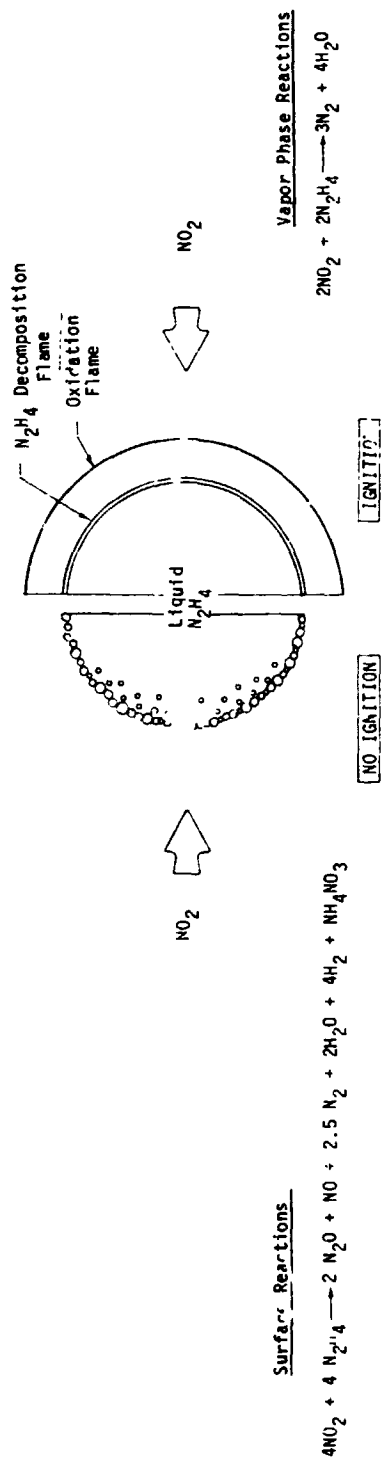


Figure 14. Hypergolic Ignition Process

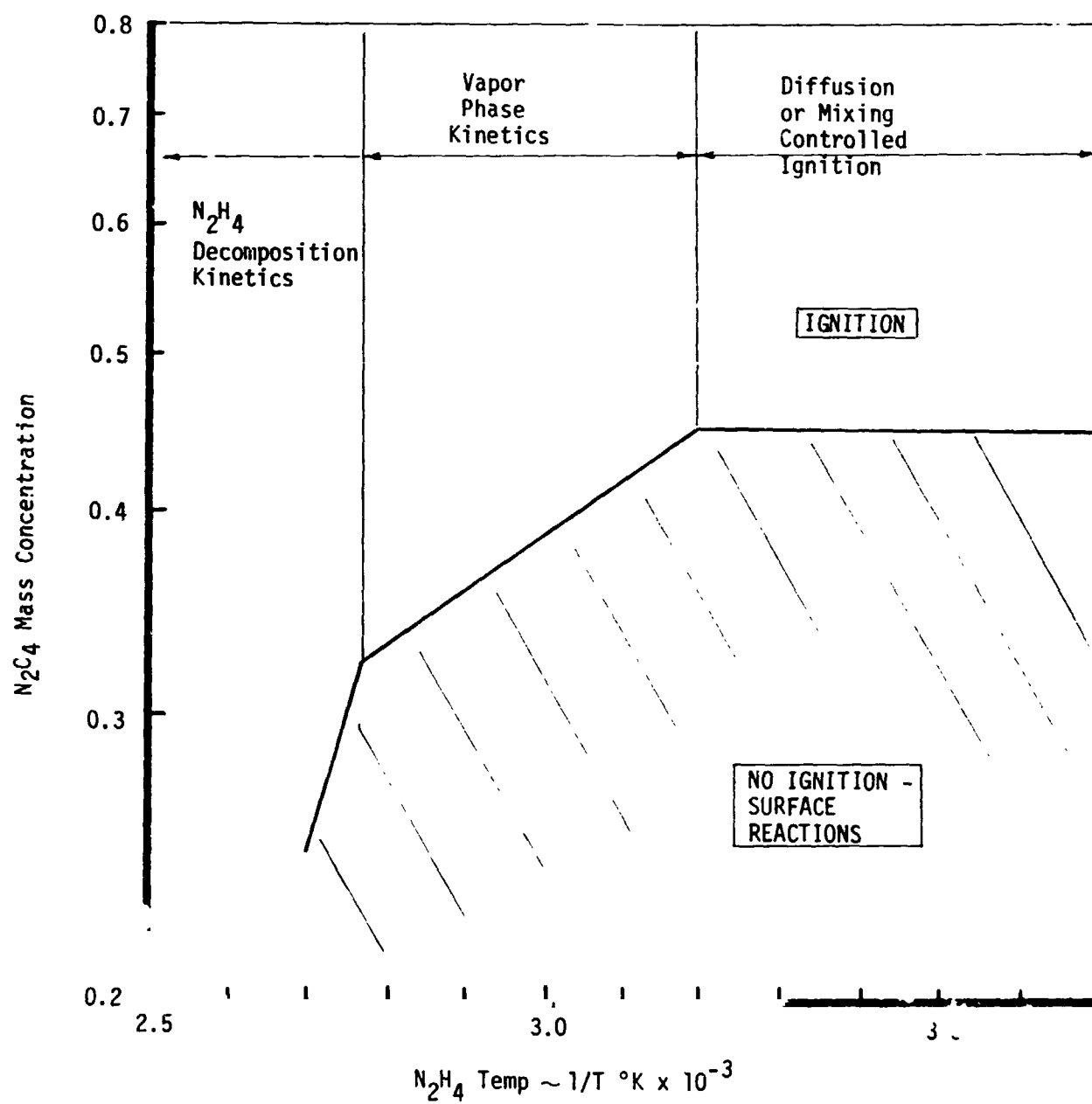
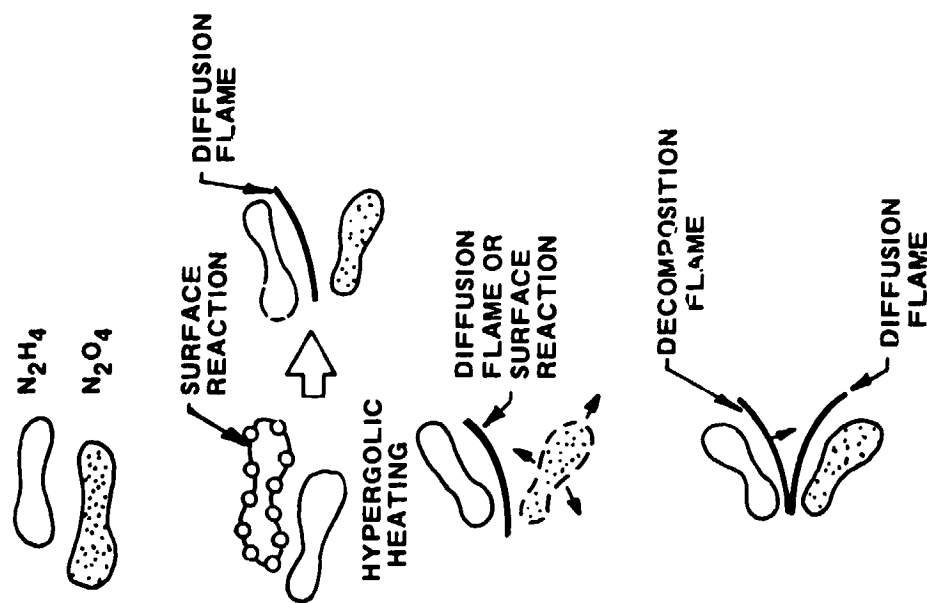


Figure 15. $\text{N}_2\text{H}_4/\text{N}_2\text{O}_4$ Ignition Limits



MIXING REGIME

VAPOR PHASE REACTION CONTROLLED

- HIGHER PRESSURE
- SMALLER ORIFICE DIAMETER
- LOWER PROPELLANT TEMP.

POPPING REGIME

SURFACE REACTION CONTROLLED

- LOWER PRESSURE
- LARGER ORIFICE DIAMETER
- LOWER TEMP.

LOW PRESSURE SEPARATION REGIME

N_2O_4 VAPOR PRESSURE CONTROLLED

- LOWER PRESSURE
- HIGHER TEMP.

HIGH PRESSURE SEPARATION REGIME

MONOPROPELLANT DECOMPOSITION CONTROLLED

- HIGHER PRESSURE

Figure 16. Postulated RSS and Pop Controlling Mechanisms

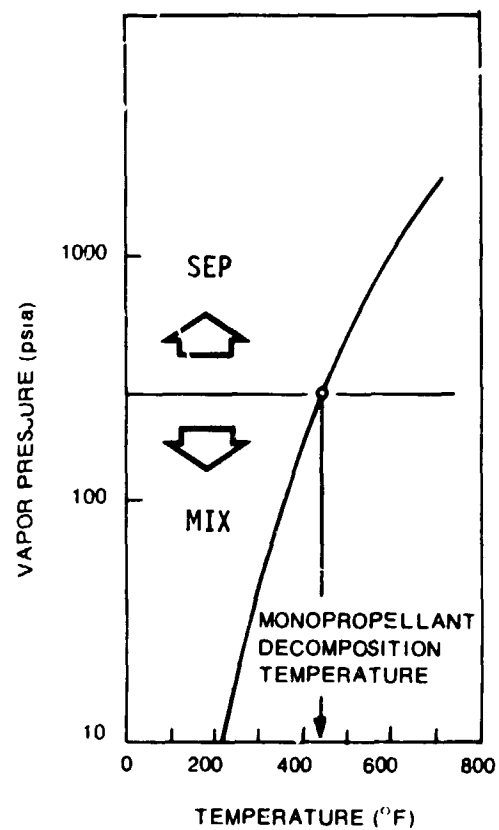
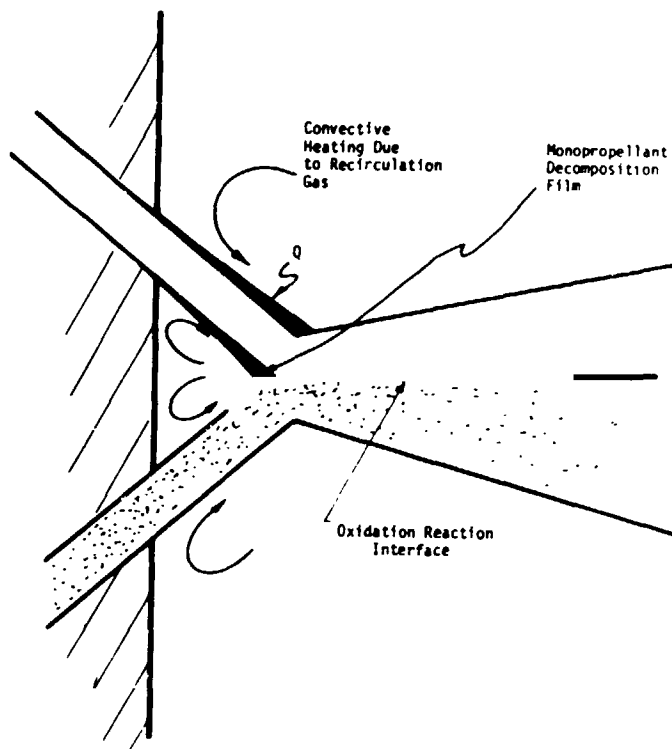


Figure 17. High Pressure RSS Model

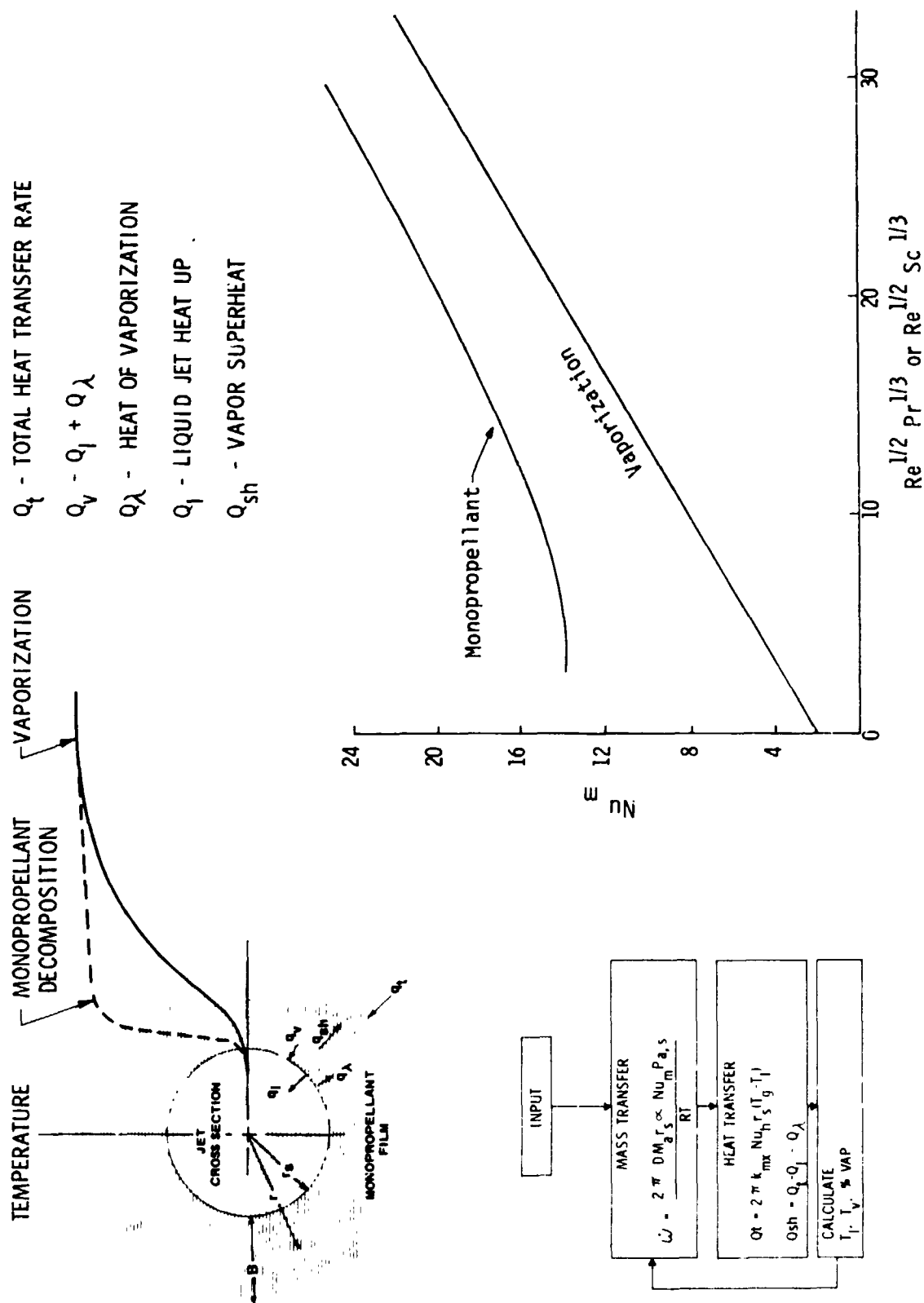


Figure 18. High Pressure RSS Vaporization/Decomposition Model

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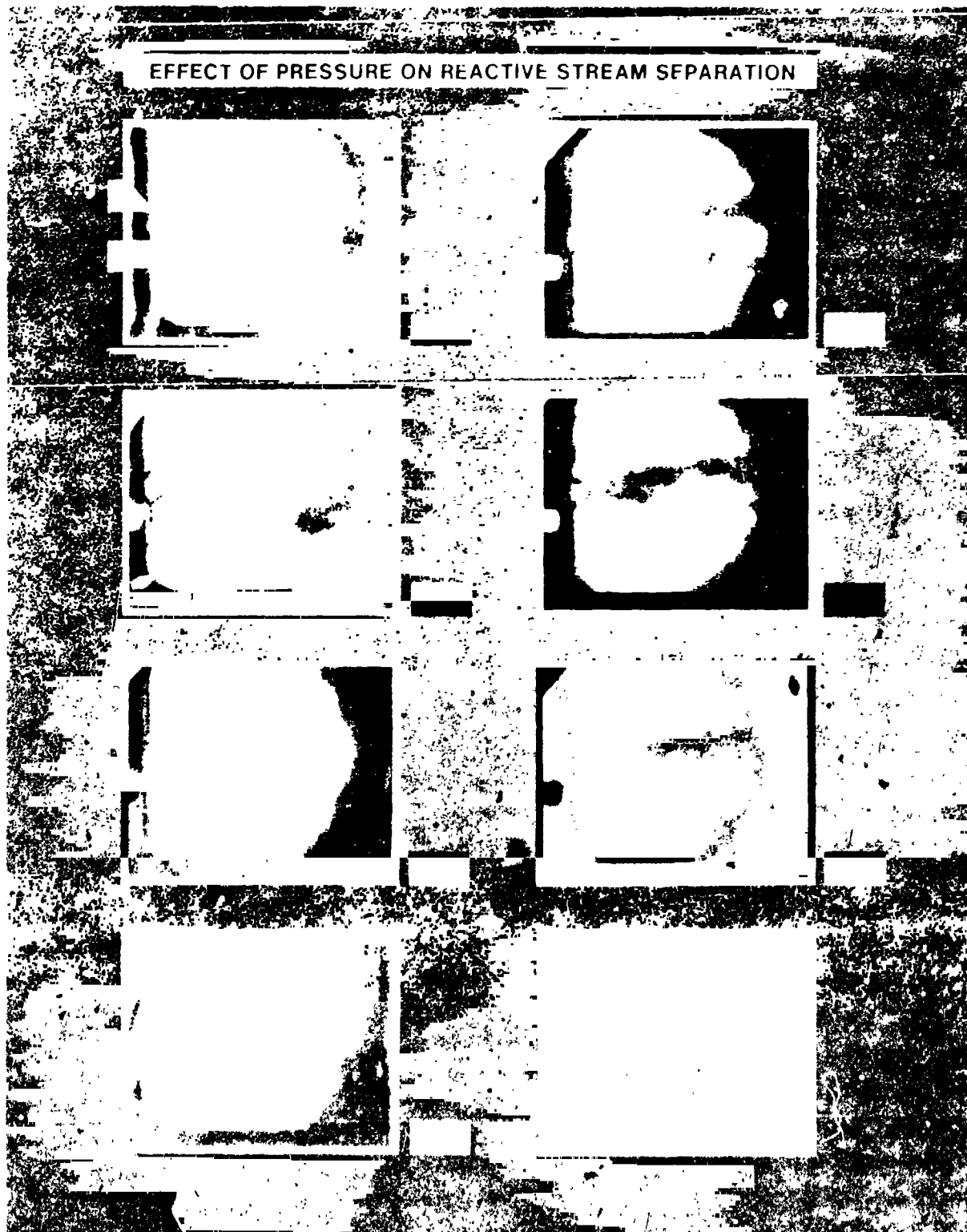


Figure 19. Effect of Pressure on Reactive Stream Separation

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EFFECT OF TEMPERATURE ON REACTIVE STREAM SEPARATION

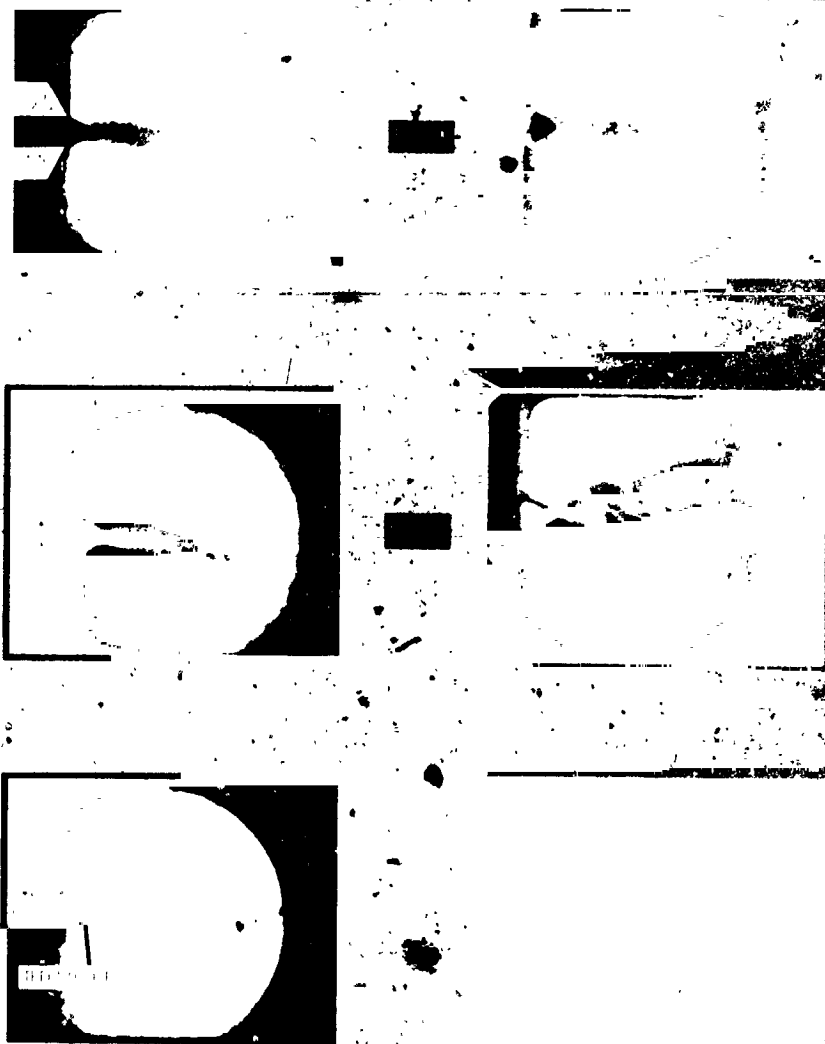


Figure 20. Effect of Temperature on Reactive Stream Separation

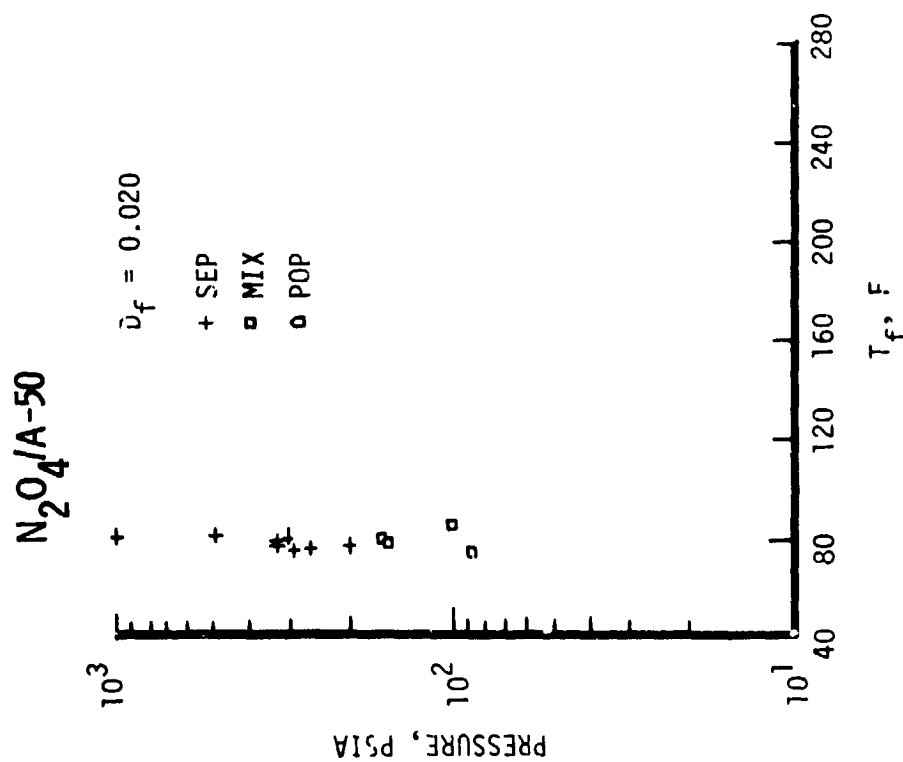
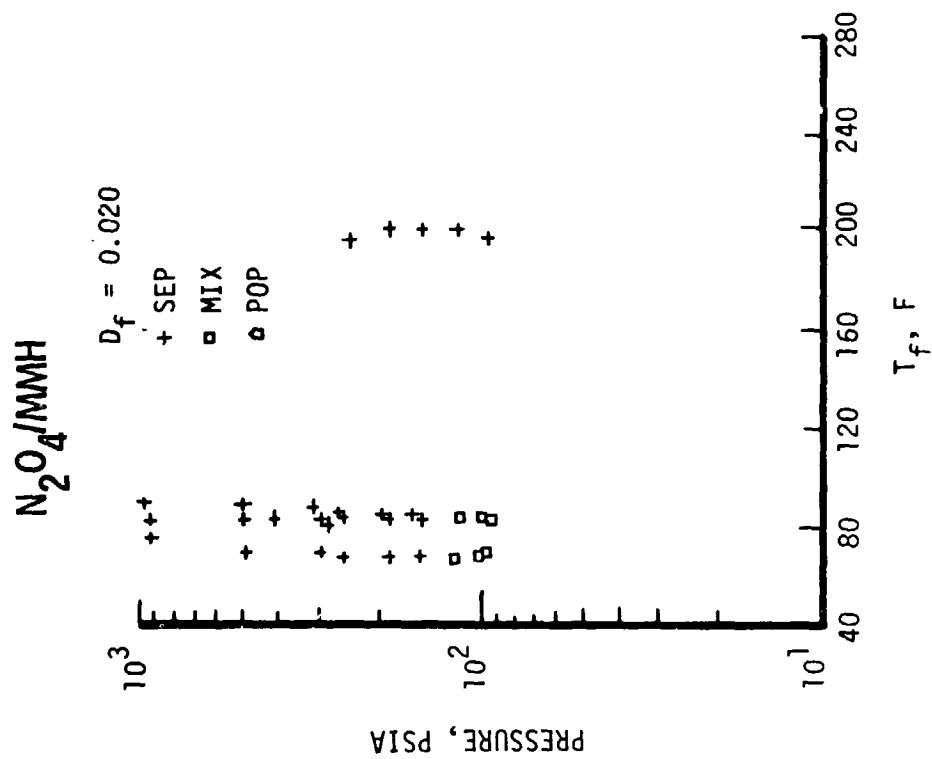


Figure 21. Effect of Pressure and Temperature on RSS

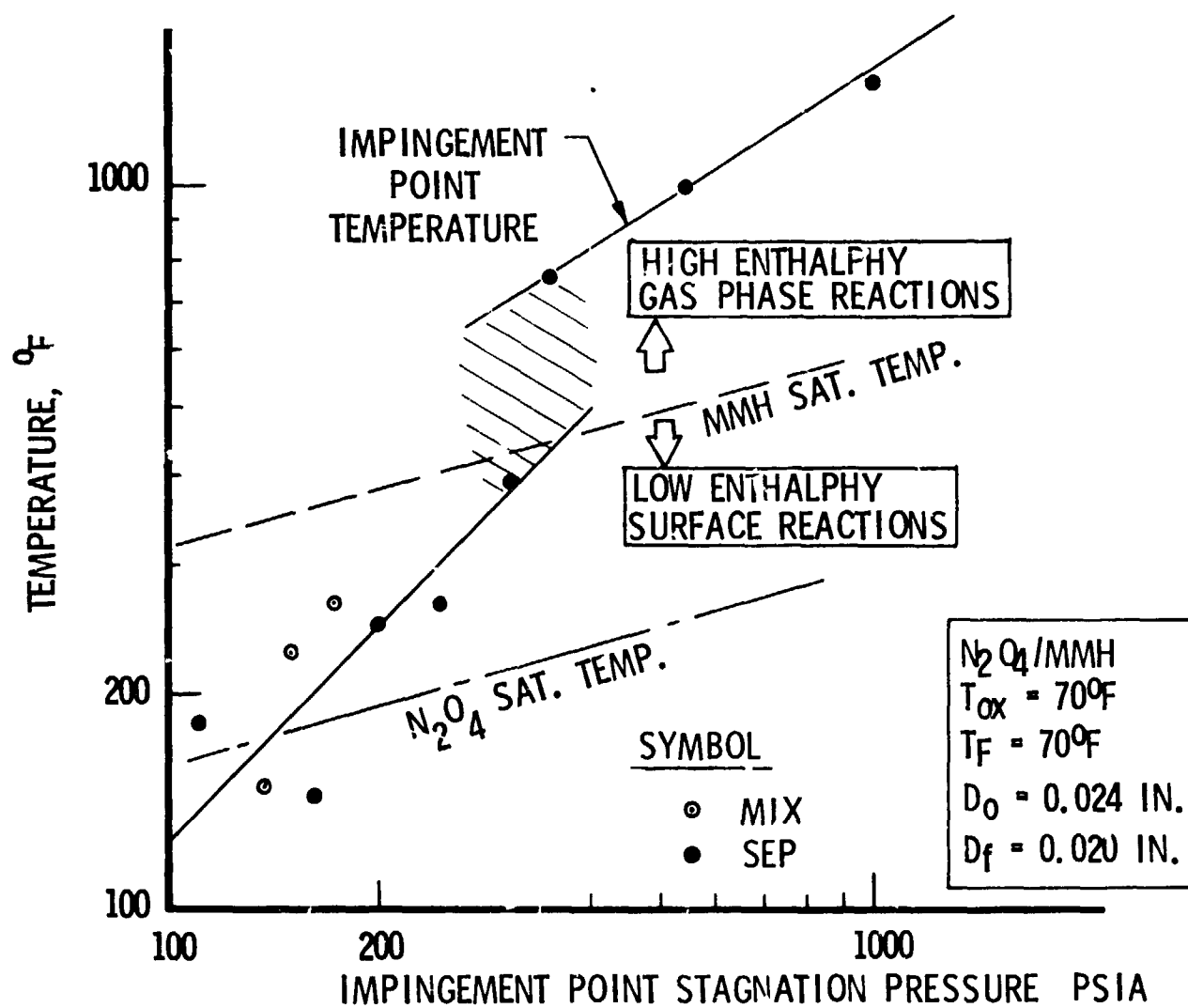


Figure 22. Effect of Pressure on Impingement Point Temperature

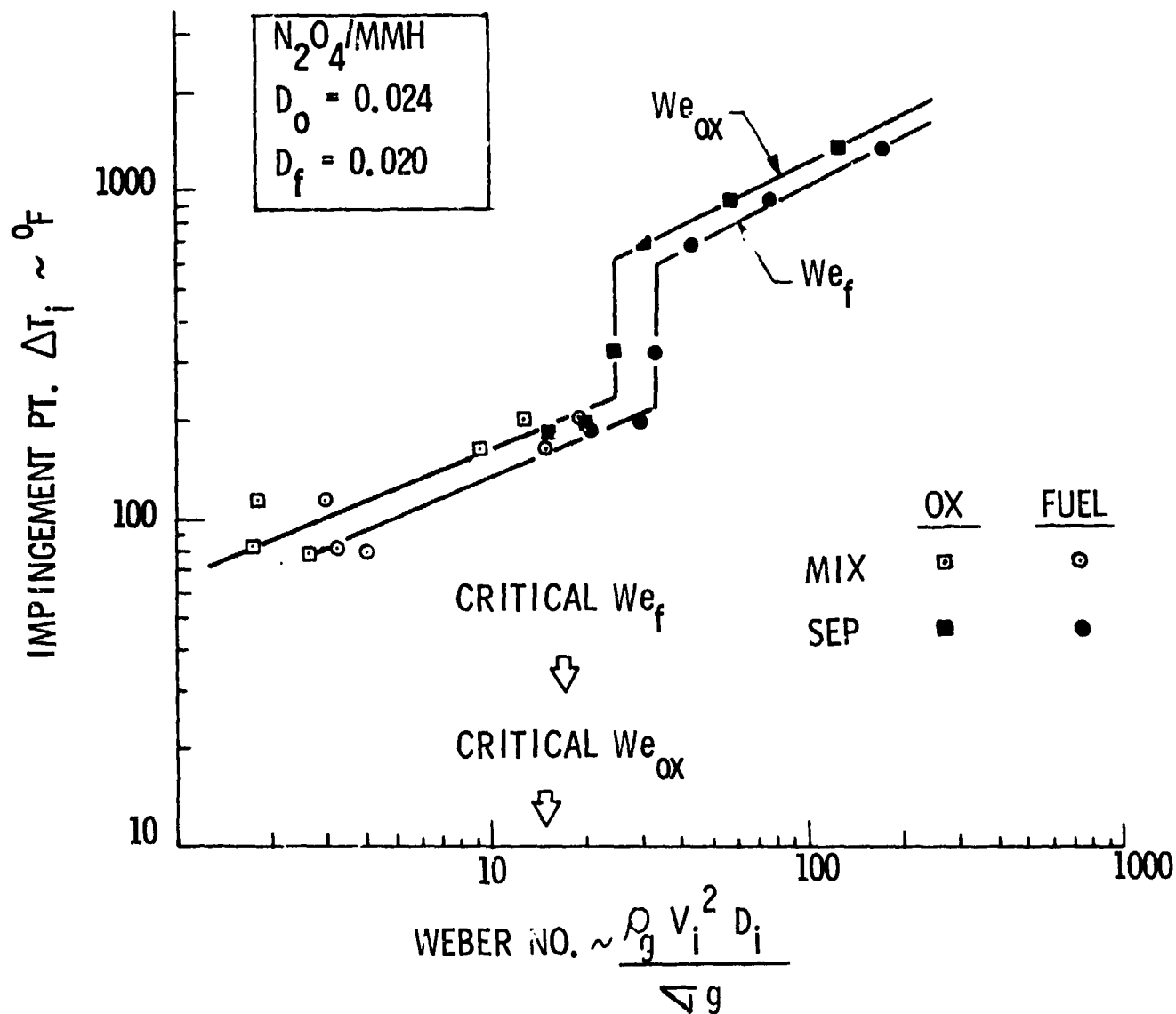


Figure 23. Effect of Weber No. on Impingement Pt. Temperature Rise

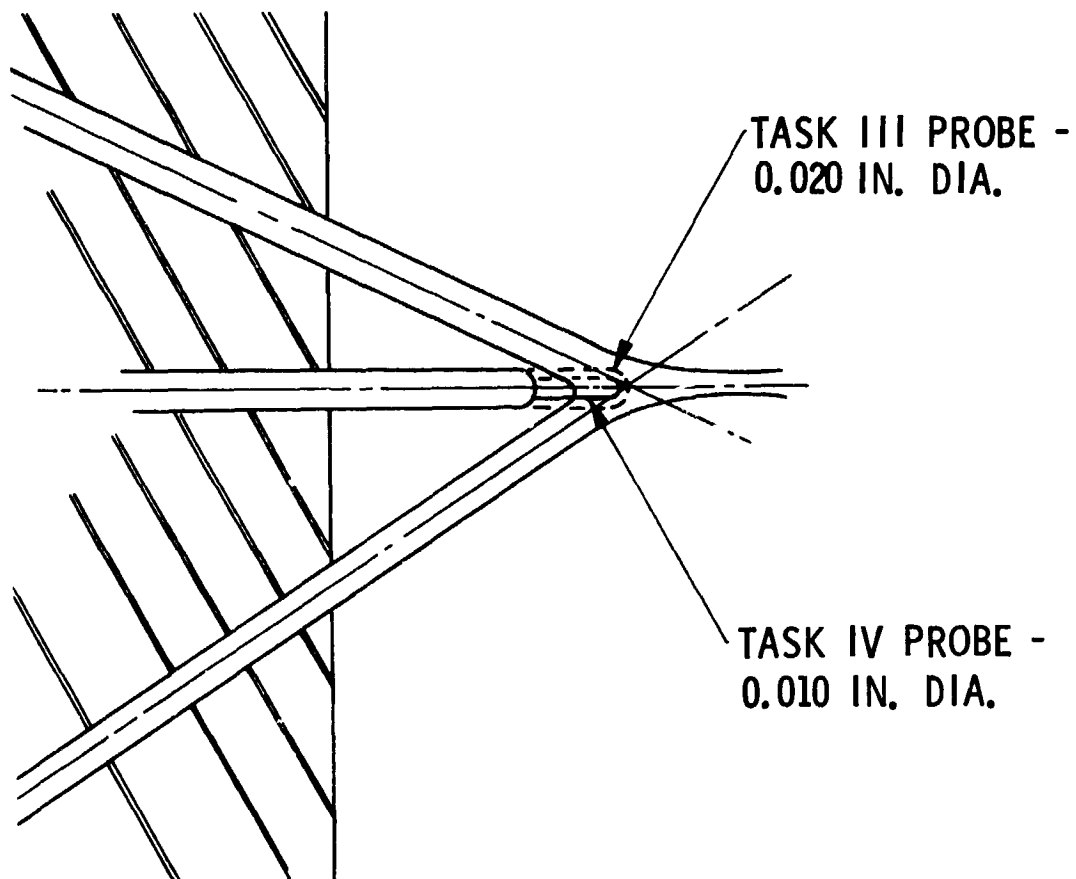


Figure 24. Impingement Point T/C Probe Designs

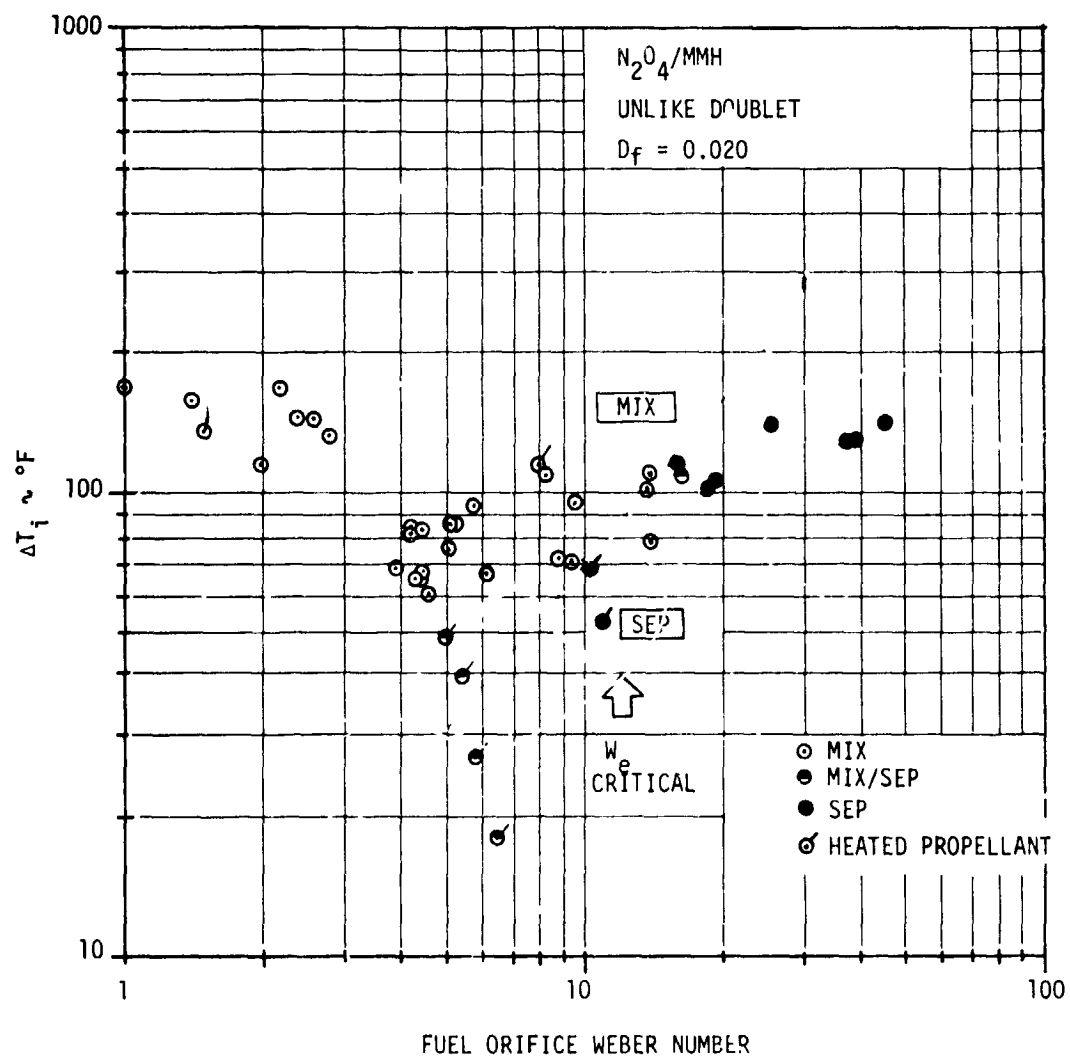


Figure 25. Effect of Fuel Orifice Weber Number on Impingement Point Temperature Rise

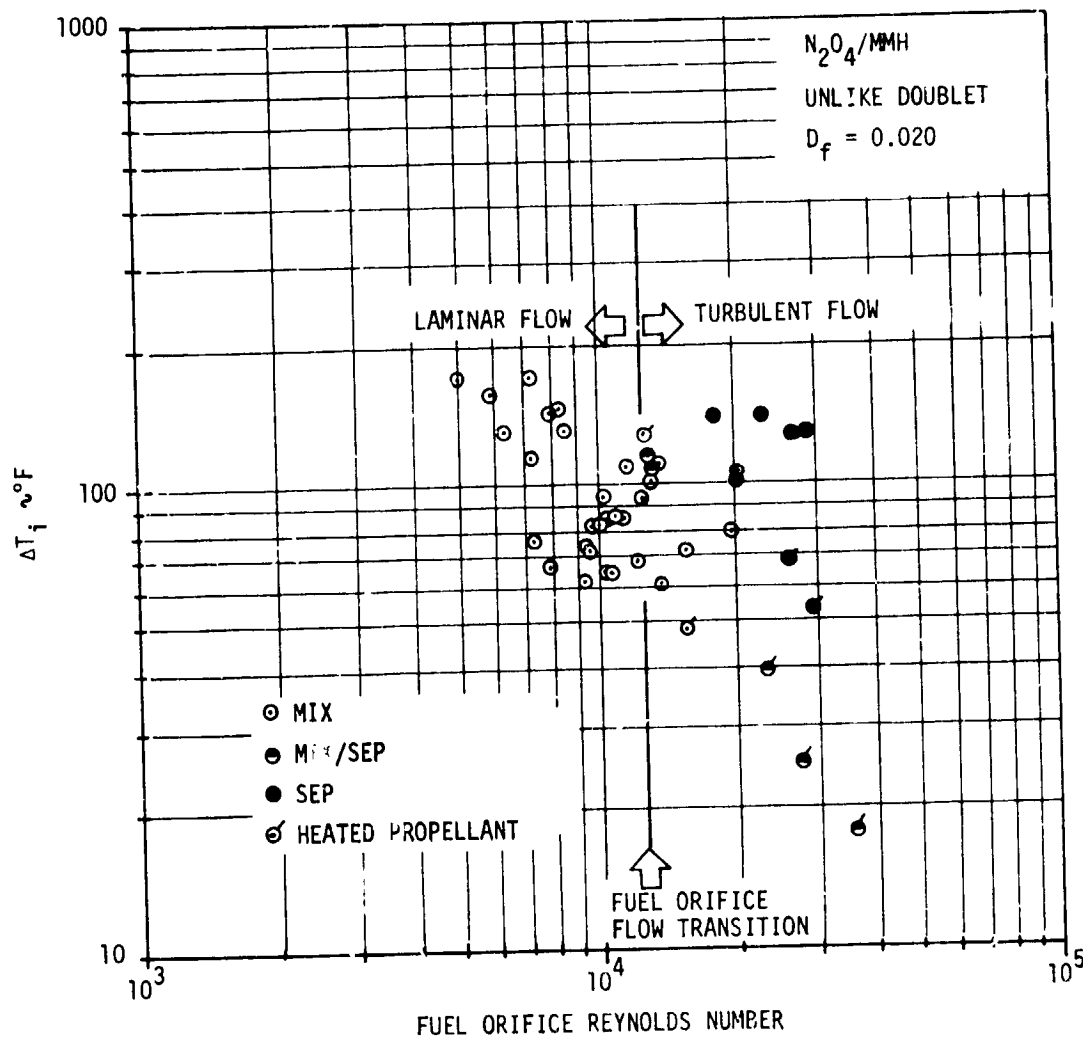


Figure 26. Effect of Fuel Orifice Reynolds Number on Impingement Point Temperature Rise

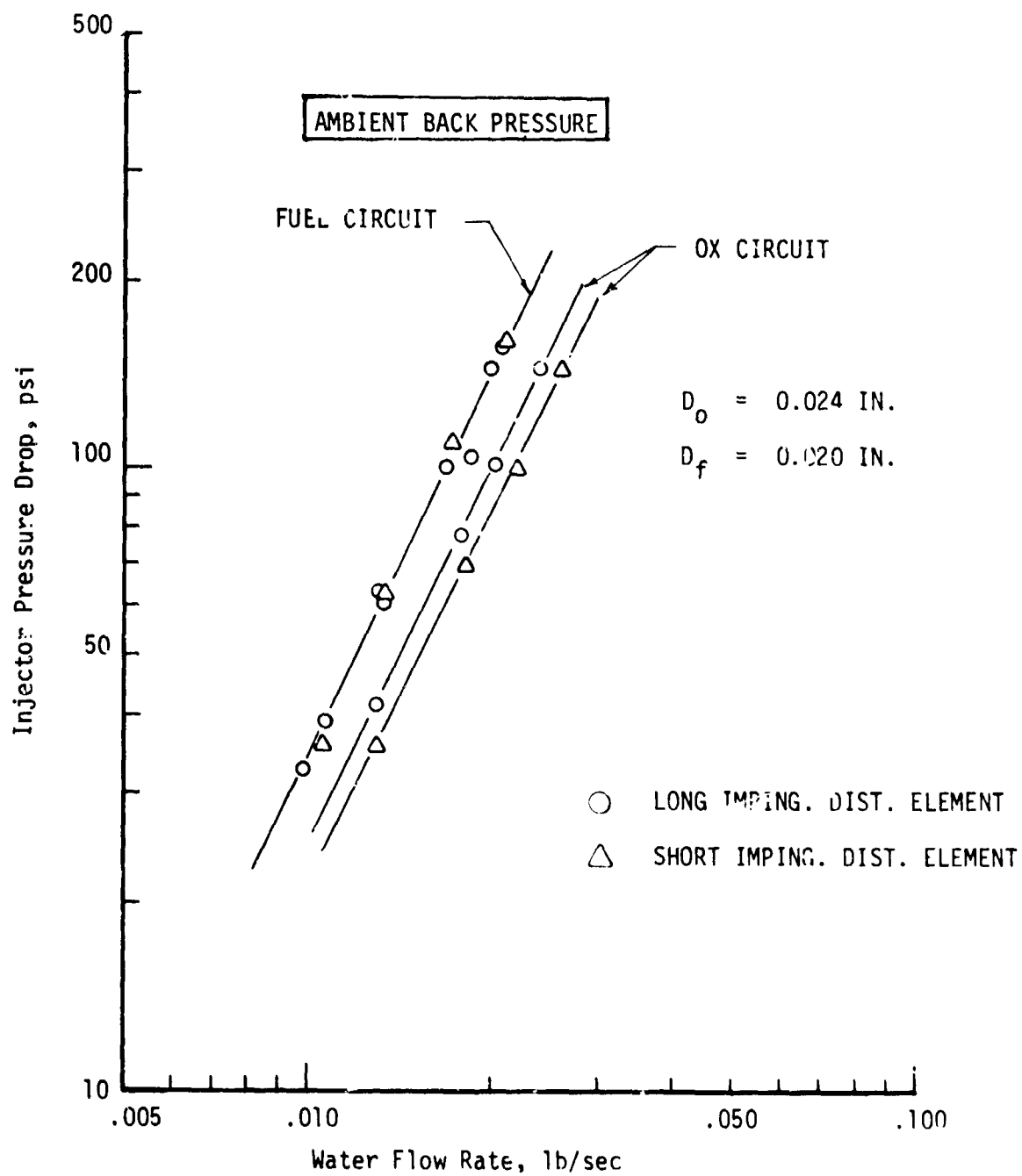


FIGURE 27. PRESSURE DROP CHARACTERISTICS OF UNLIKE DOUBLET INJECTOR ELEMENTS

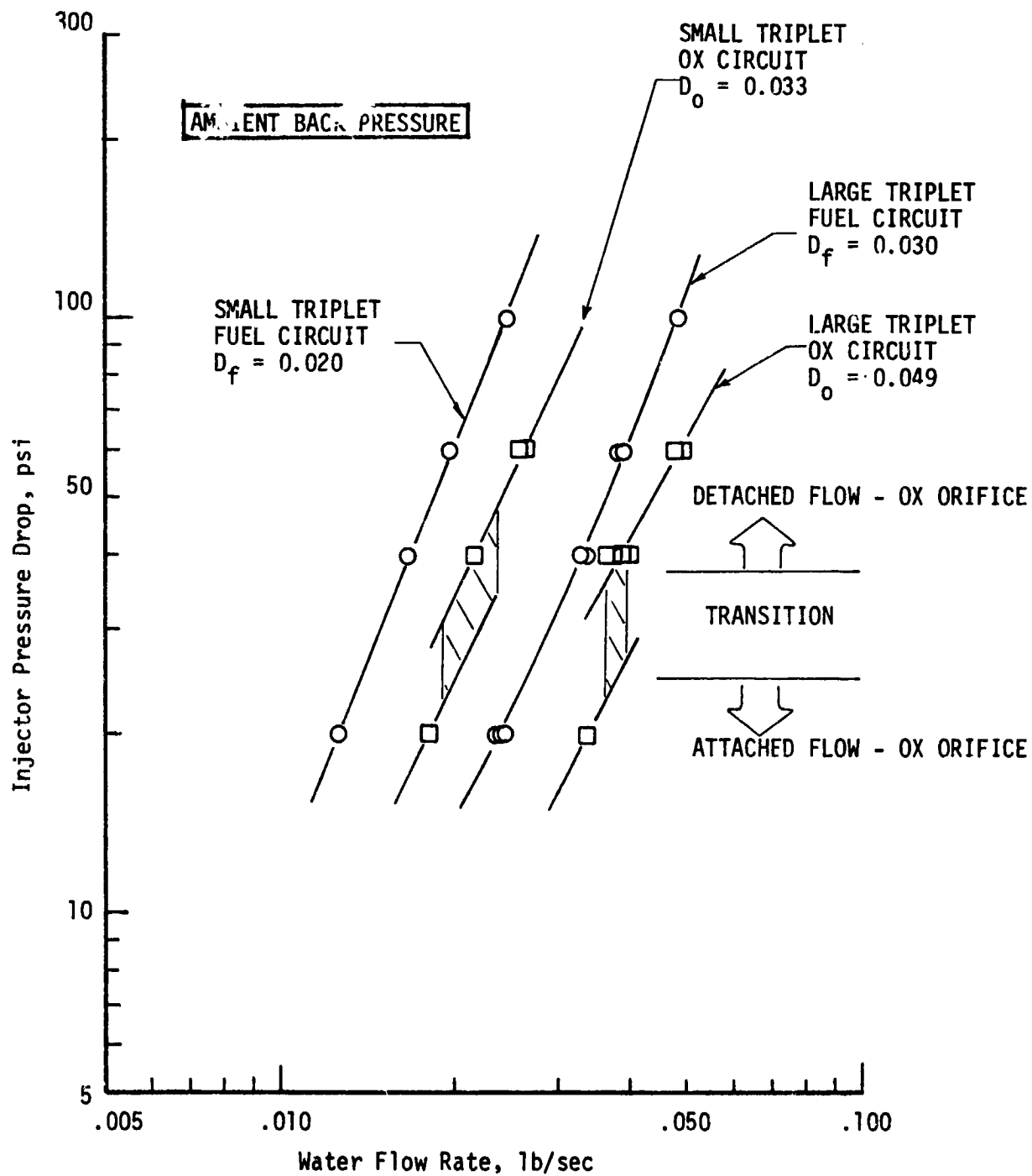


FIGURE 28. PRESSURE DROP CHARACTERISTICS OF TRIPLET INJECTOR ELEMENTS

0-2

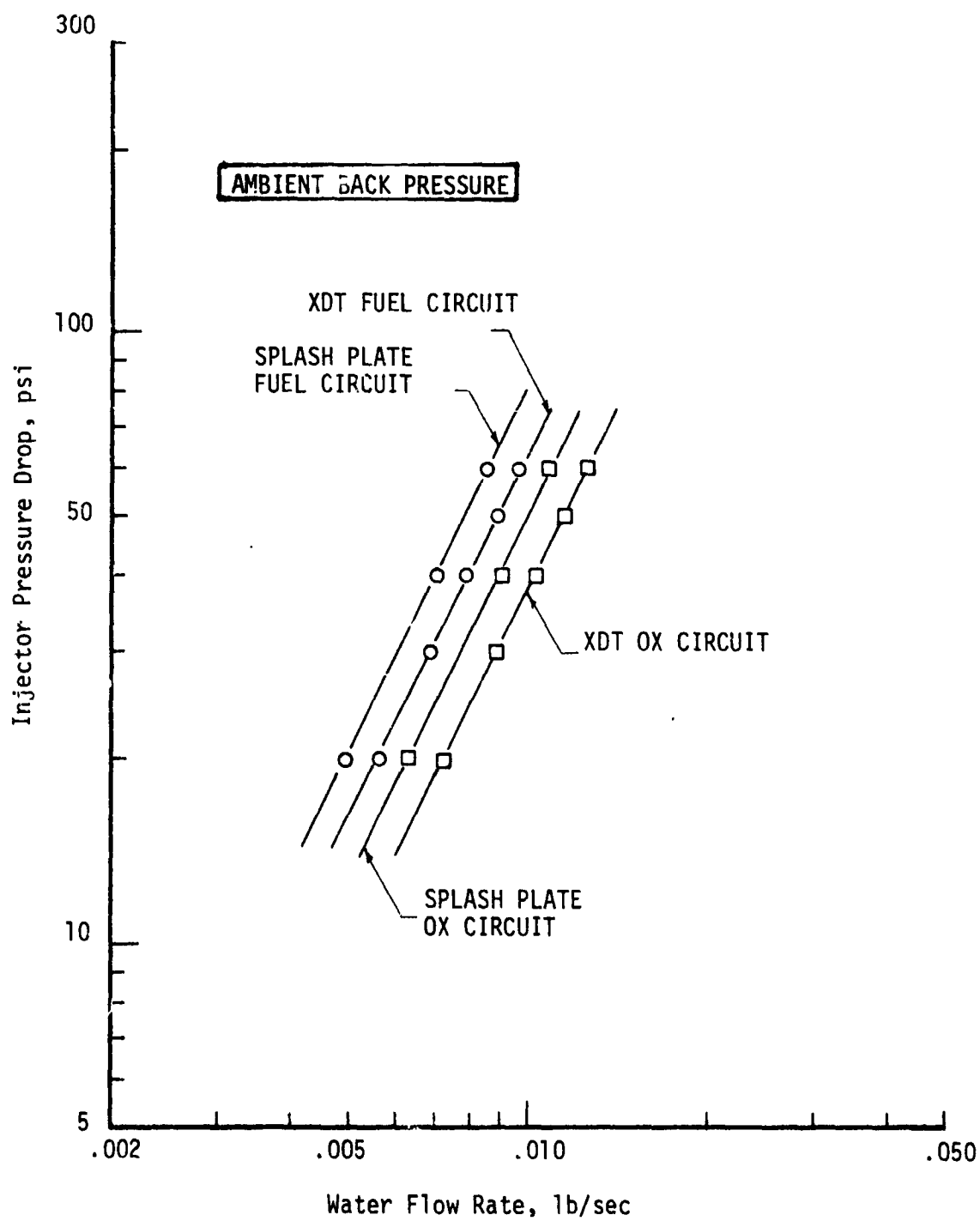


FIGURE 29. PRESSURE DROP CHARACTERISTICS OF PLATELET ELEMENT

UNLIKE DOUBLET HYDRAULIC CHARACTERISTICS

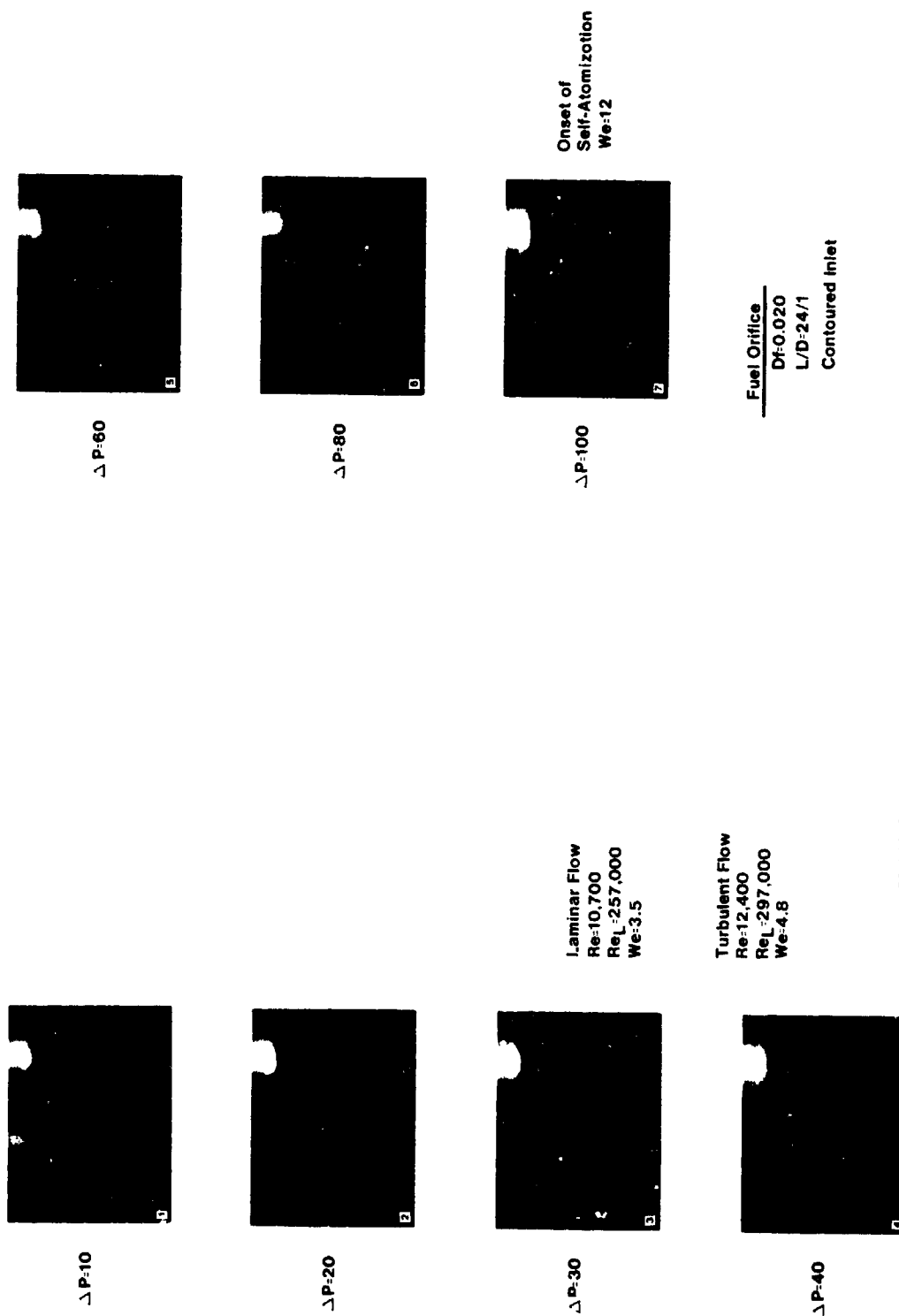


FIGURE 20. UNLIKE DOUBLET HYDRAULIC CHARACTERISTICS

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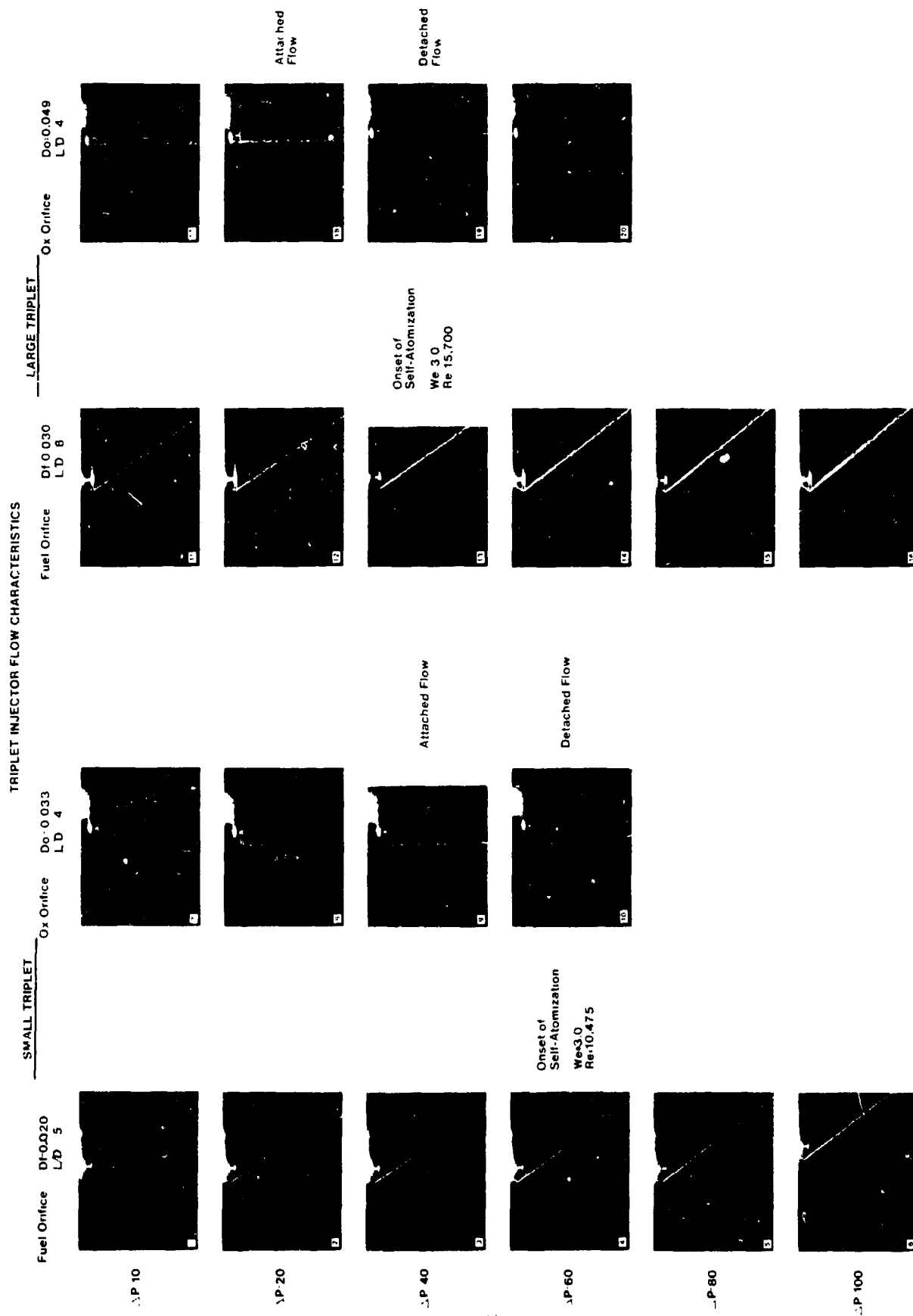


FIGURE 11 TRIPLET INJECTOR FLOW CHARACTERISTICS

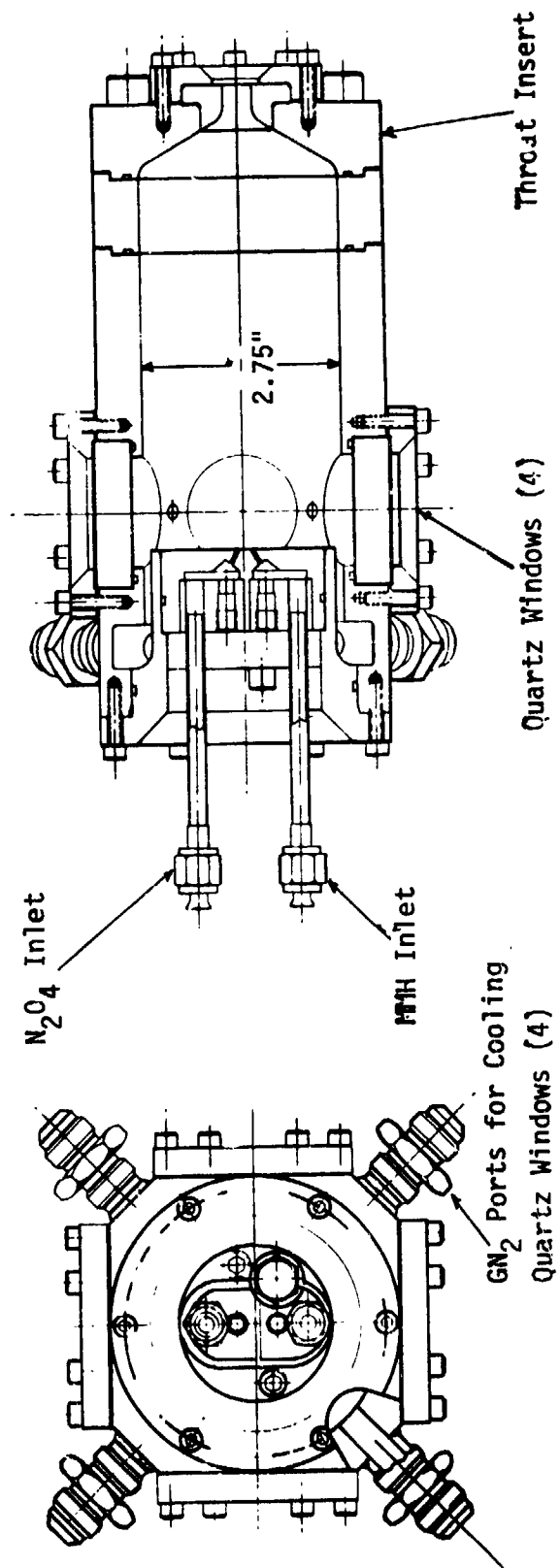


Figure 32. Windowed Photographic Test Chamber

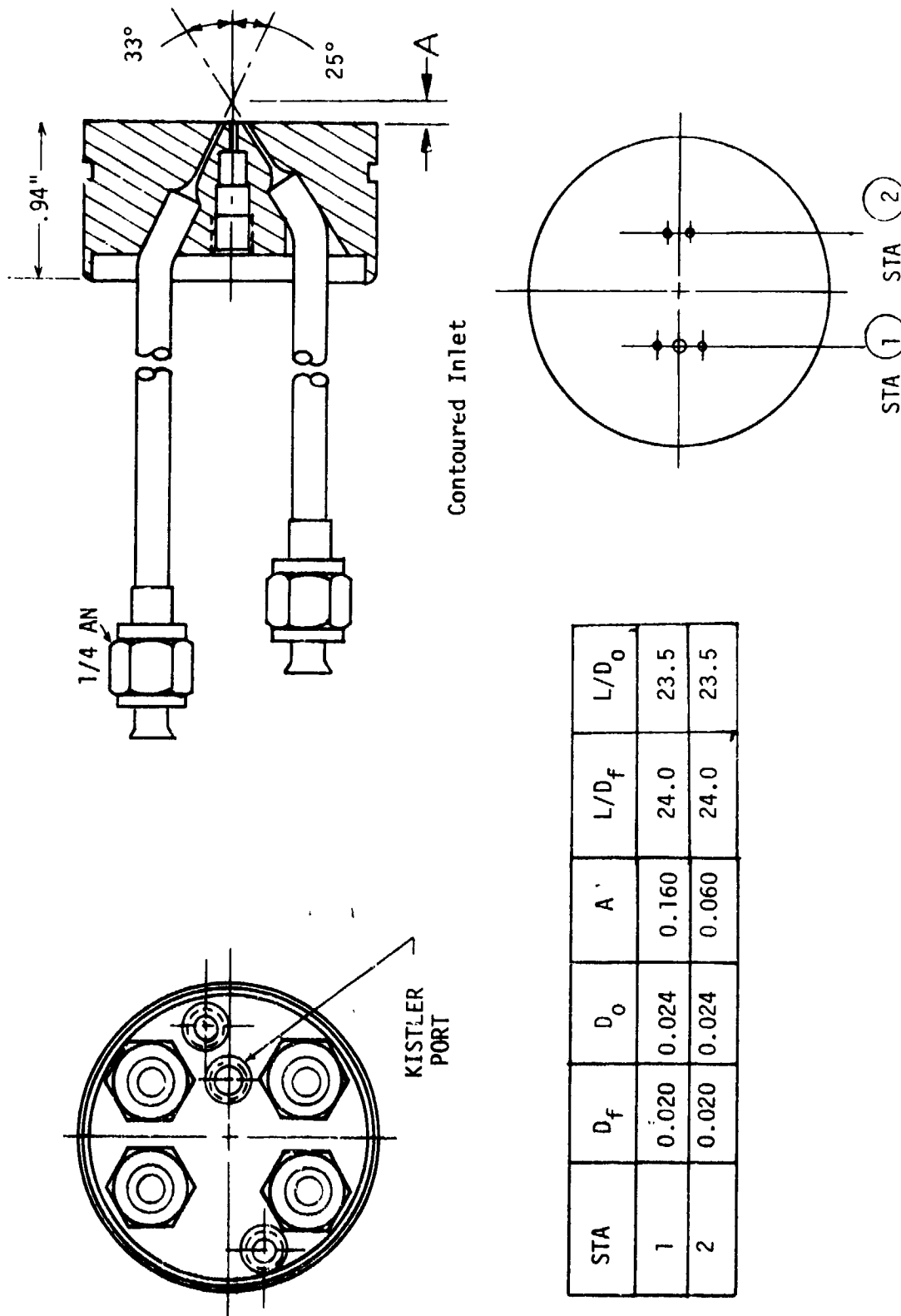


Figure 33. Unlike Doublet Injector

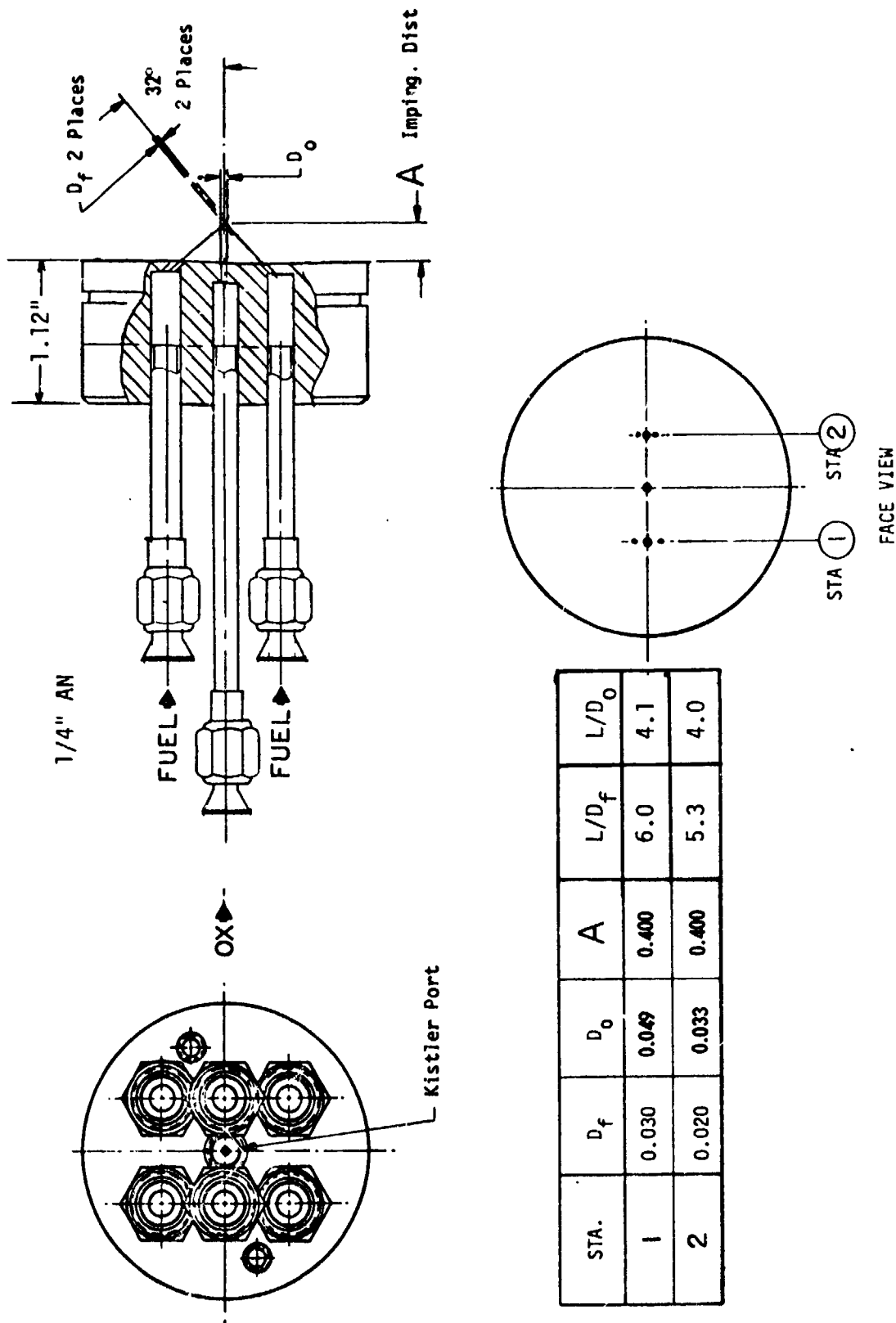


Figure 34. Triplet Injector

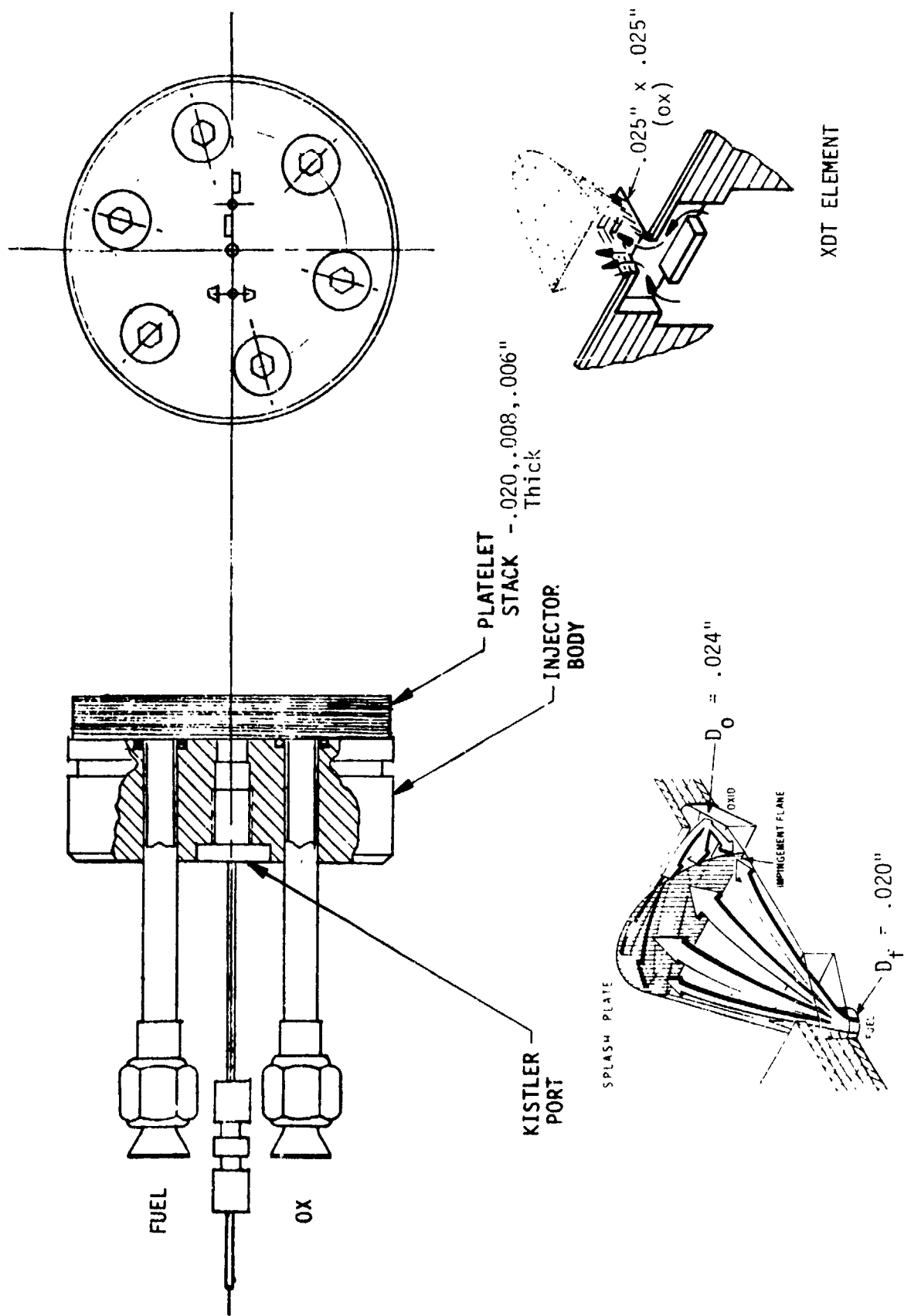


Figure 35. Self-Atomizing Platelet Injector Elements

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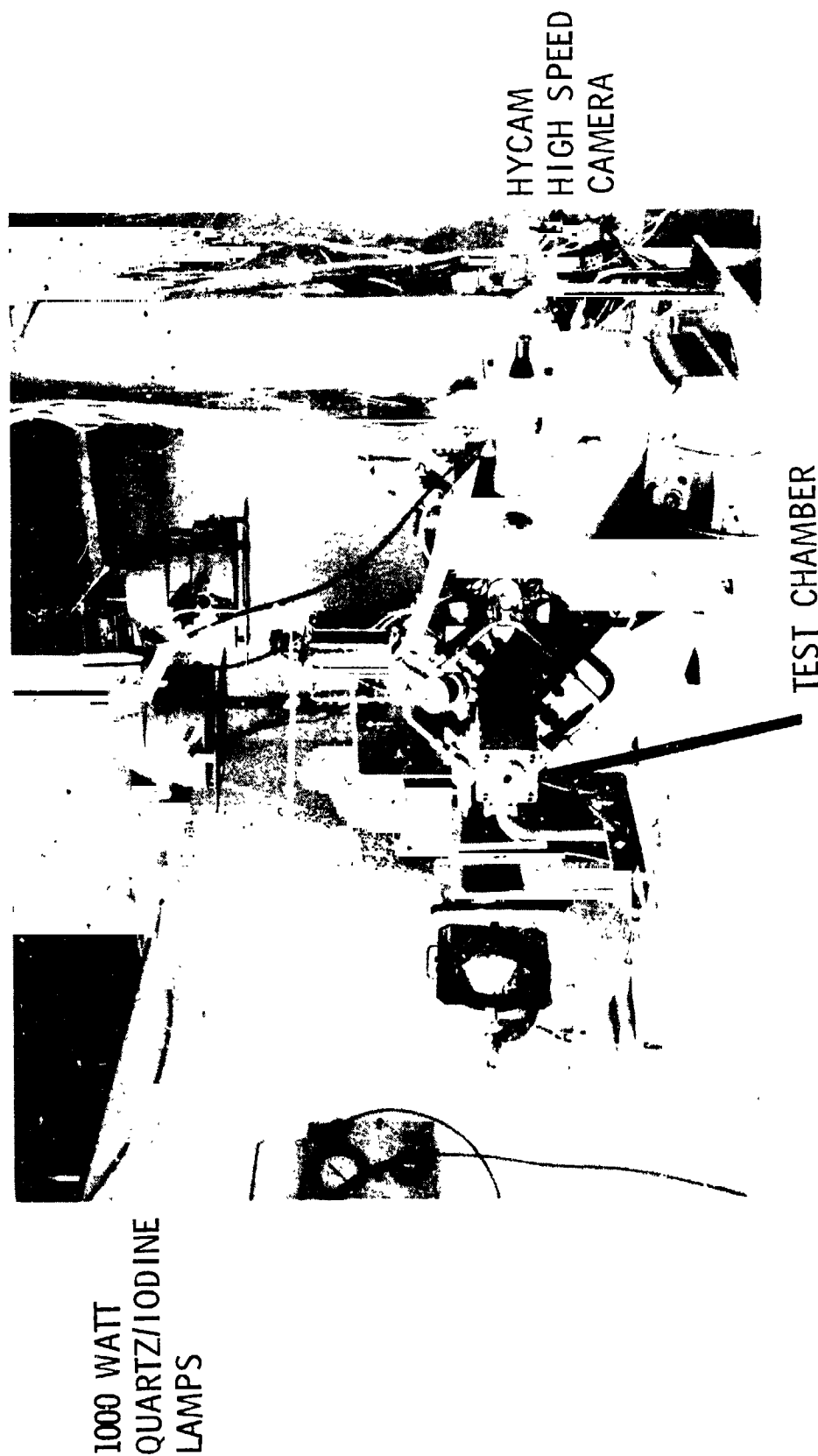


Figure 36. Photographic Test Setup

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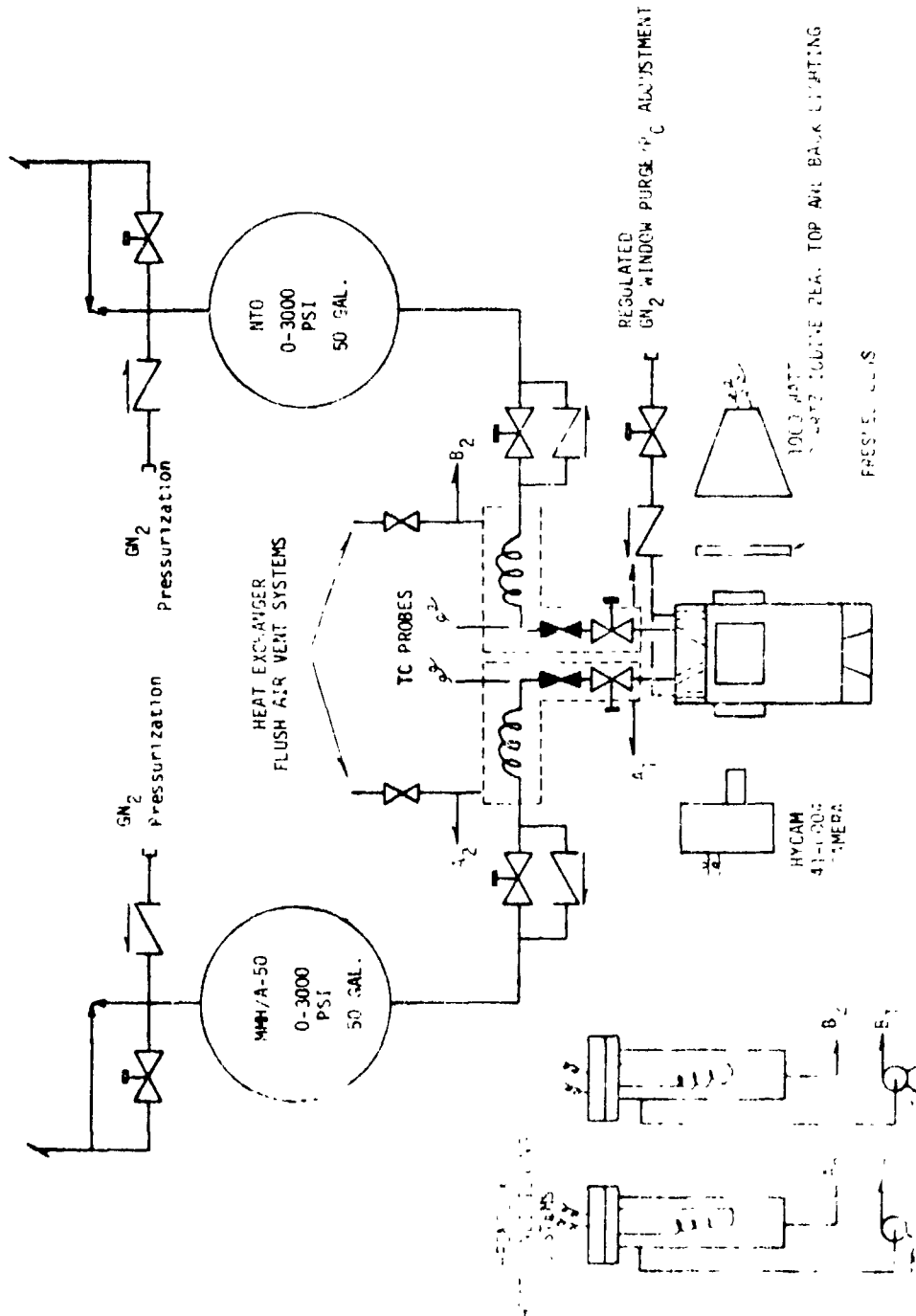


Figure 37. Propellant Flow System Schematic

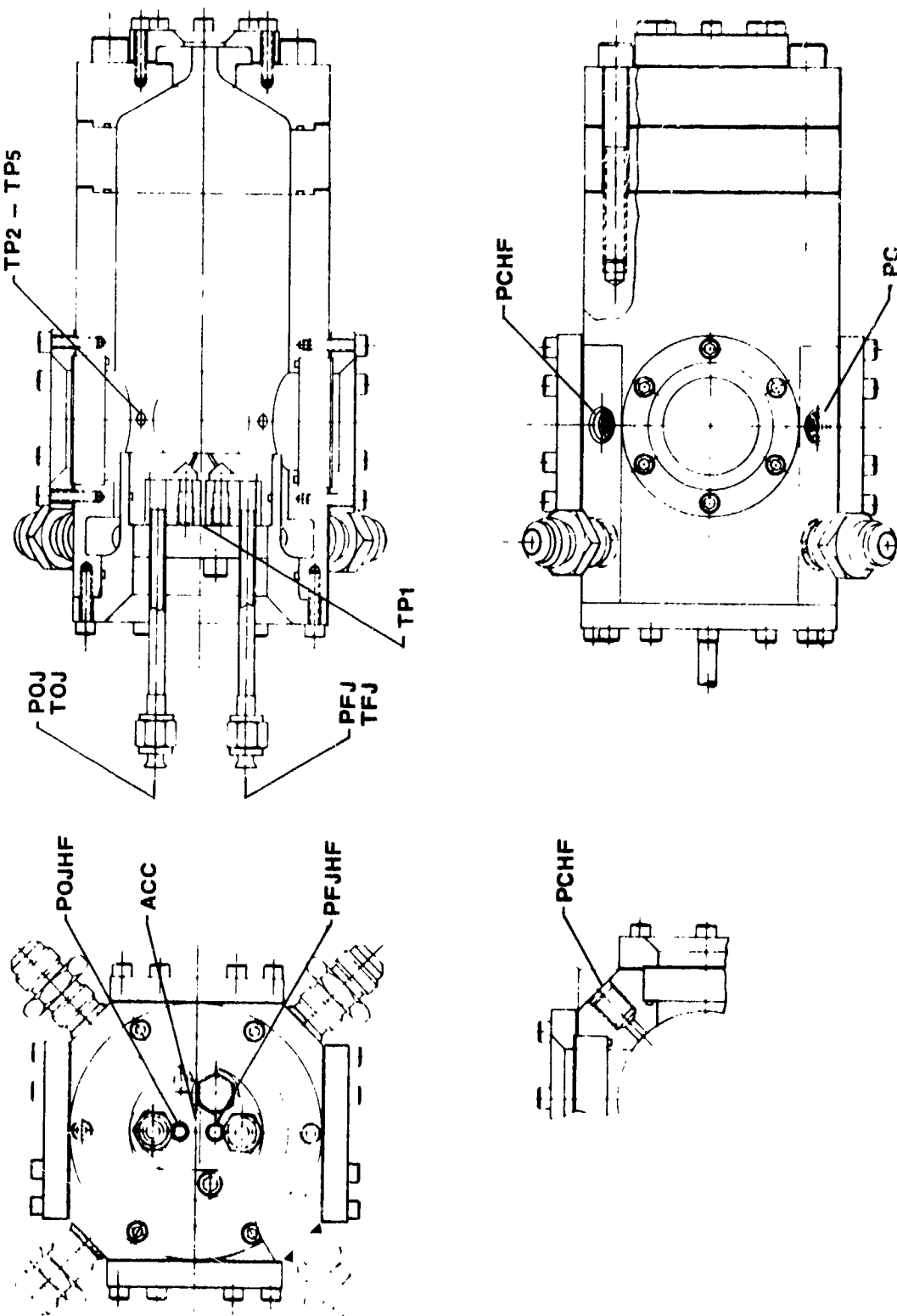


Figure 38. Instrumentation Schematic

APPENDIX A

NOMENCLATURE

NOMENCLATURE

a	Coefficient in power form separation pressure correlation
b	Exponent in power form separation pressure correlation
C	Reactant concentration
C _p	Heat capacity
d	Orifice diameter
D	Orifice spacing parameter
DELTI	Impingement point temperature rise
EM	Rupe mixing parameter
g	Gravitational constant
IS	$\ln (D/V \sin 1/2 \text{ imp. angle}) + 46.8 - 21800 (1/T)$
K	Rate constant
L	Orifice length
M	Mass
m	Concentration exponent, propellant 1
MR	Mixture ratio oxidizer/fuel
MW	Molecular weight
n	Concentration exponent, propellant 2
P	Pressure
PD	Pc/D
ΔP	Pressure difference
PPF	Fuel partial pressure
PPO	Oxidizer partial pressure
Q	Liquid phase heat release
R	Reactivity, gas constant, or element spacing coefficient
RE	Orifice Reynolds number based on diameter
REL	Orifice Reynolds number based on length
RESID	Propellant stream contact time
S	Element spacing

Nomenclature (cont.)

SPR	Fuel to oxidizer momentum flux ratio
T	Temperature
ΔT	Temperature difference
t	Time
t_r	D/V, propellant stream contact time
V	Velocity or volume
W_e	Weber Number
\dot{W}	Weight flow rate
X	Mole fraction
XP	Square root of fuel and oxidizer mole fraction product

Greek

μ	Viscosity
ρ	Density
τ	Time
σ	Surface tension

Subscripts

a	Ambient conditions
Avg	Average conditions
c	Chamber conditions
f	fuel
g	Gas phase
i	Impingement point condition
ign	Ignition
J	Manifold value
L	Liquid
LIG	Ligament

Nomenclature (cont.)

ox	Oxidizer
p	Partial pressure
RES	Stream residence (contact)
RSS	Separated conditions
V	Vapor
VP	Vapor phase

APPENDIX B

FINAL DATA CORRELATIONS

C

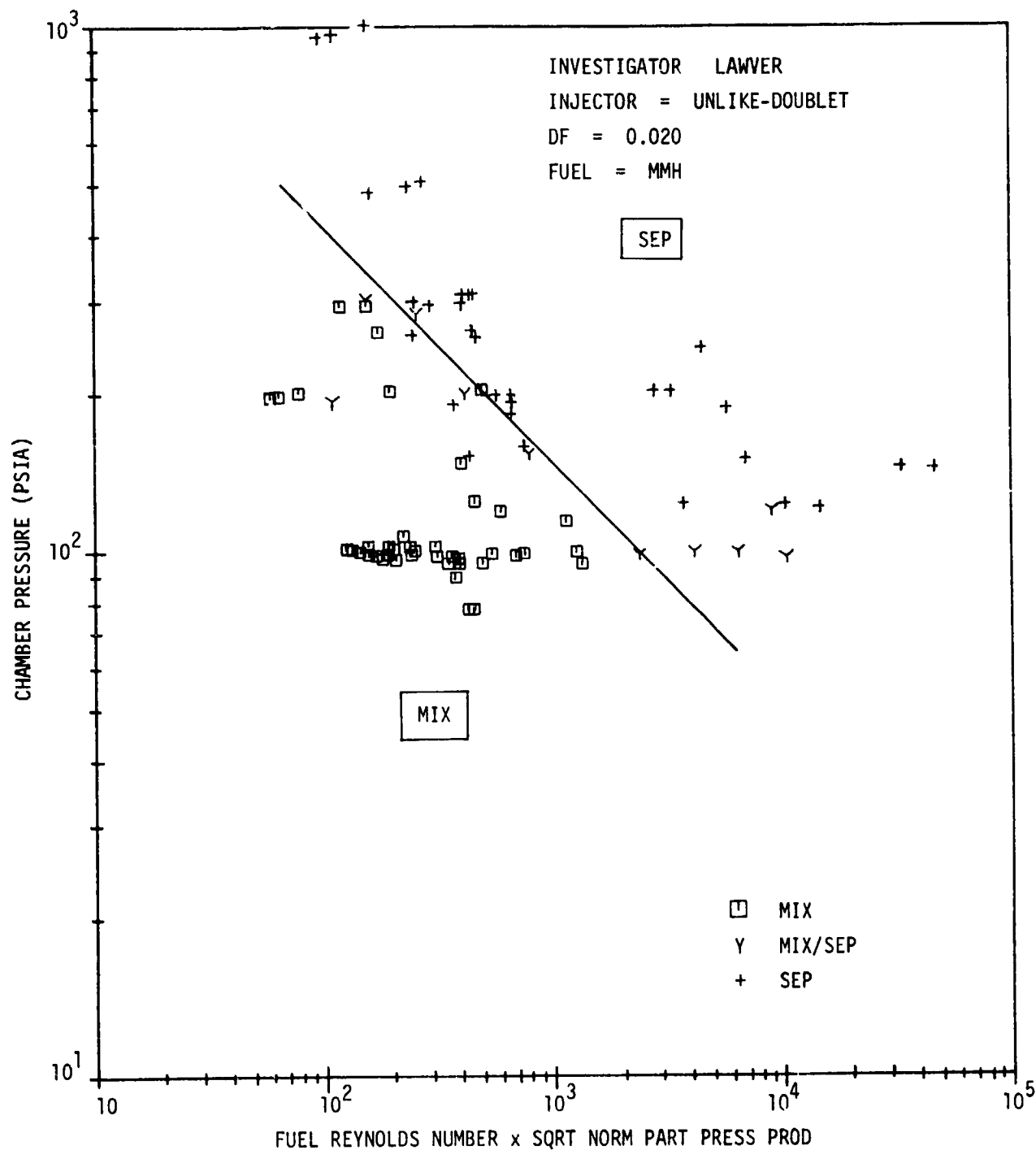


Figure B-1. Effect of Chamber Pressure and Reactivity on RSS

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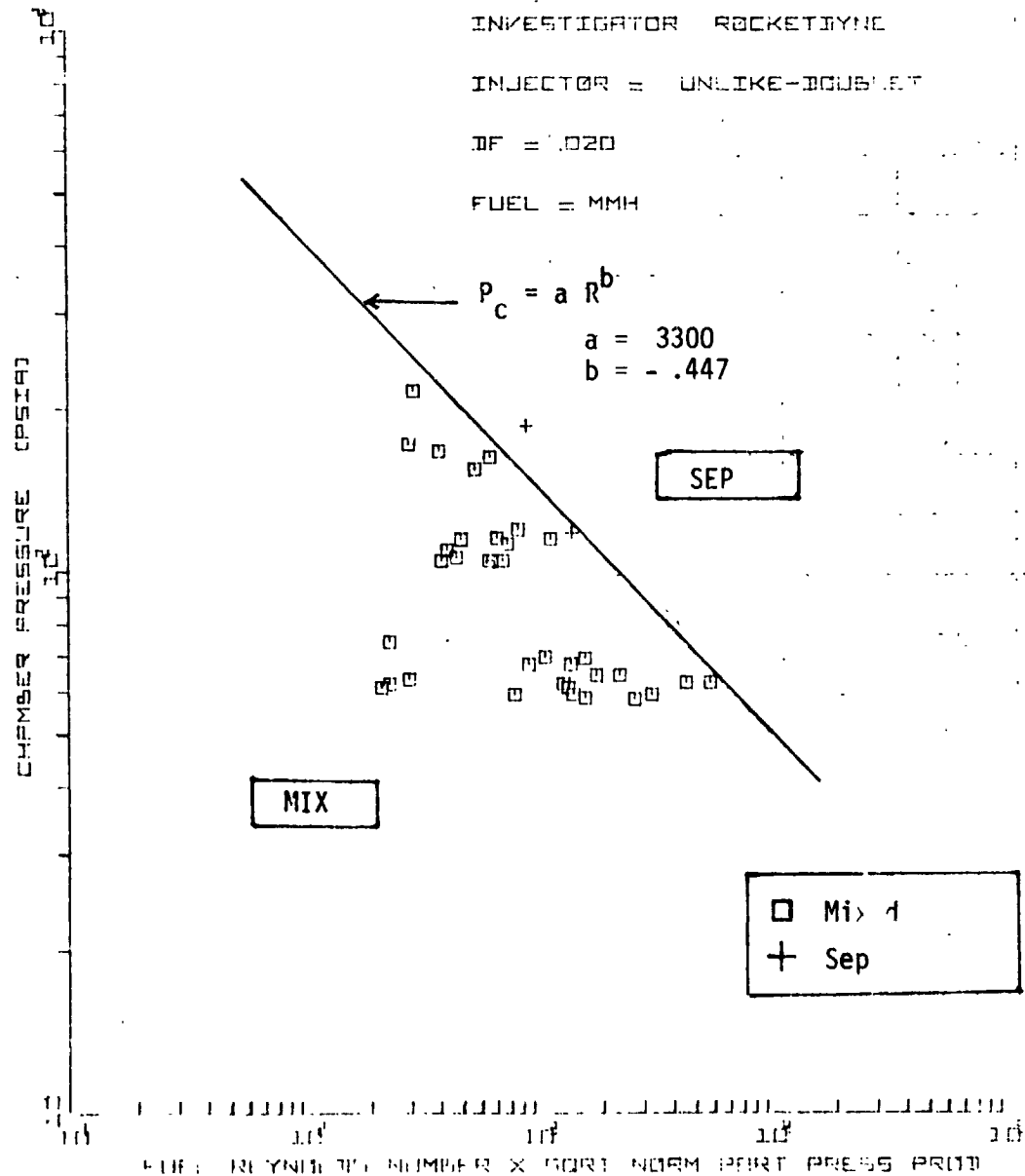


Figure B-2. Rocketdyne Unlike-Doublet, $D_f = .020$

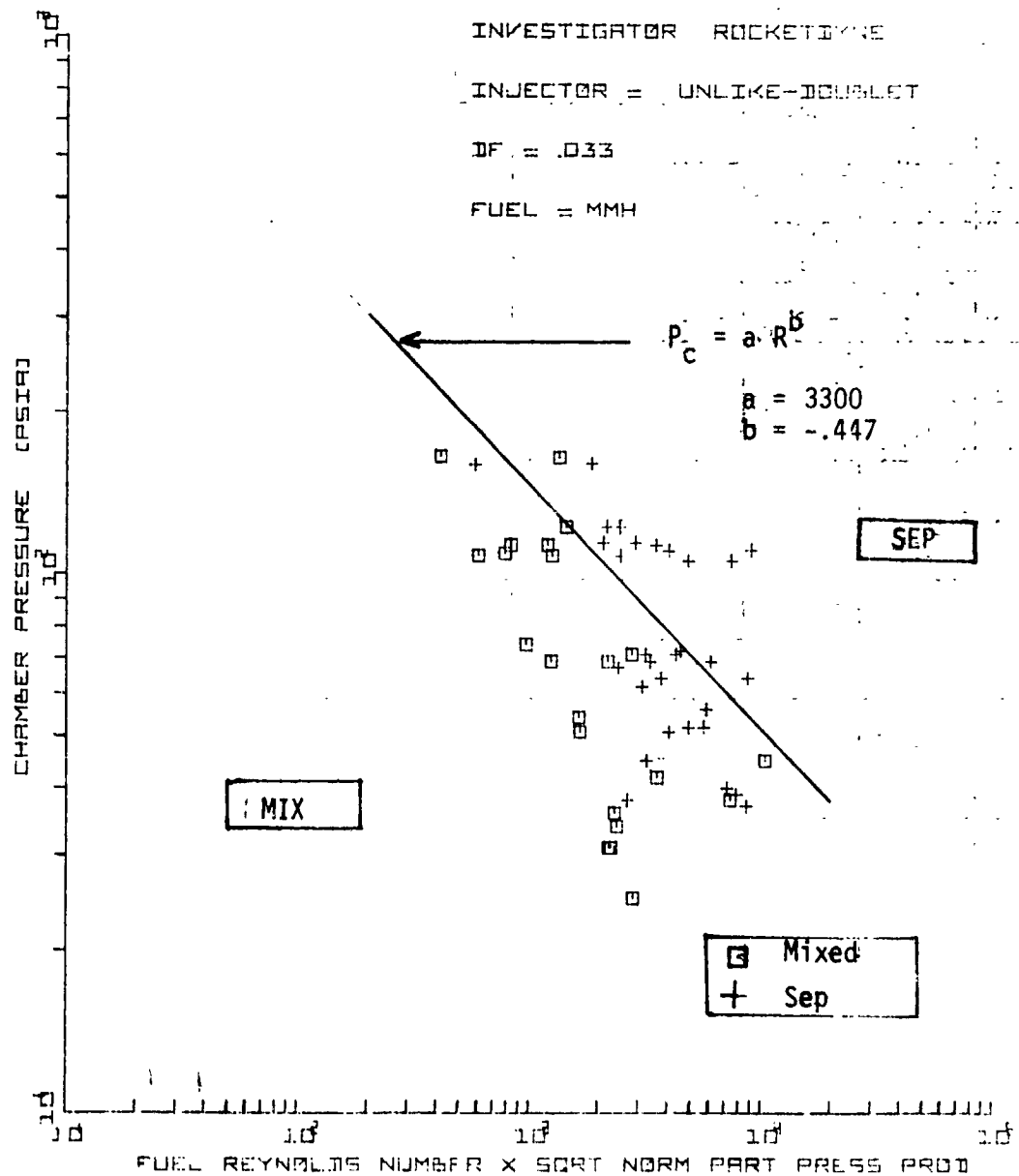


Figure B-3. Rocketdyne Unlike-Douplet, $D_f = .033$

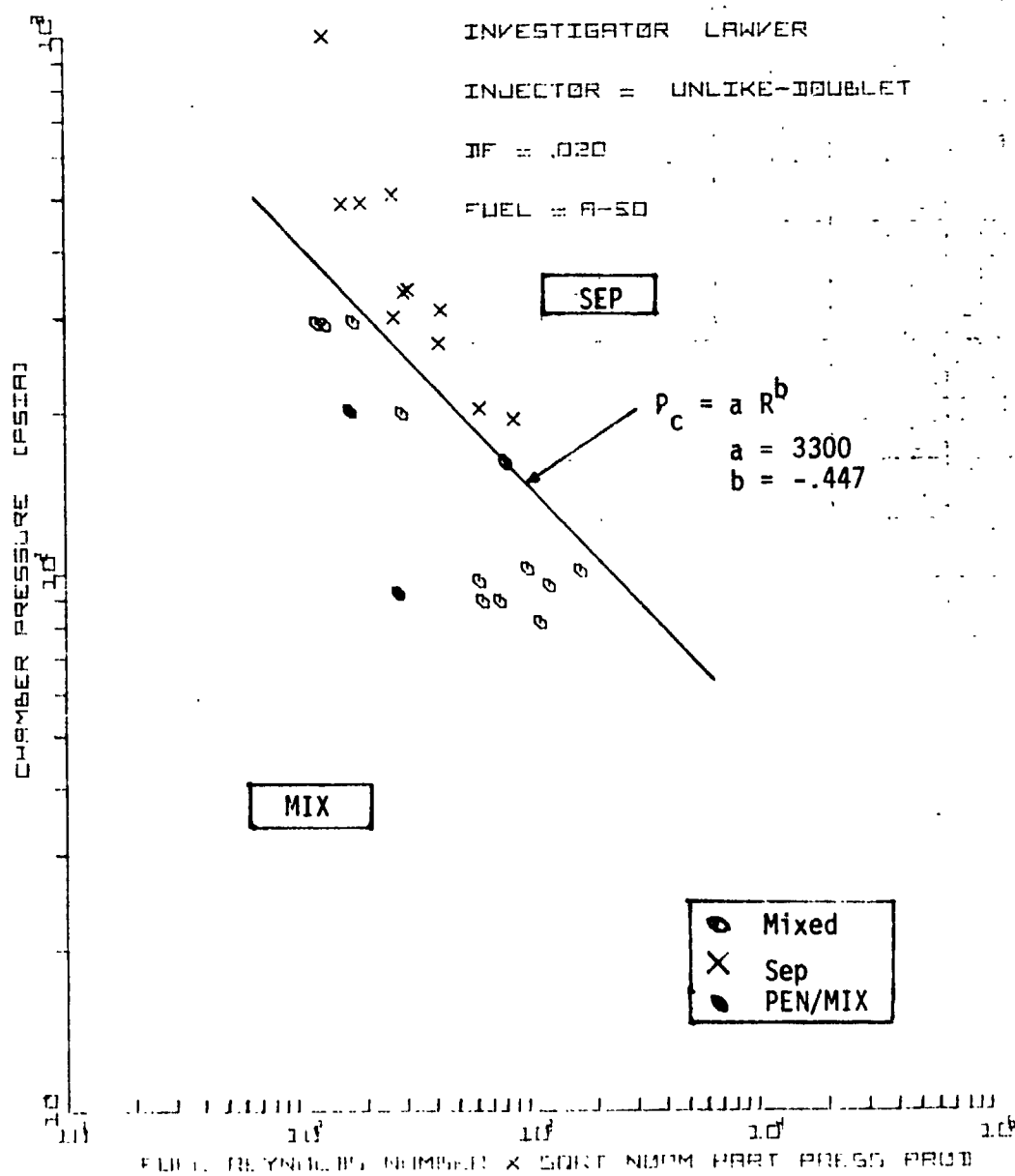


Figure B-4. Lawver Unlike Doublet, $D_f = .020$

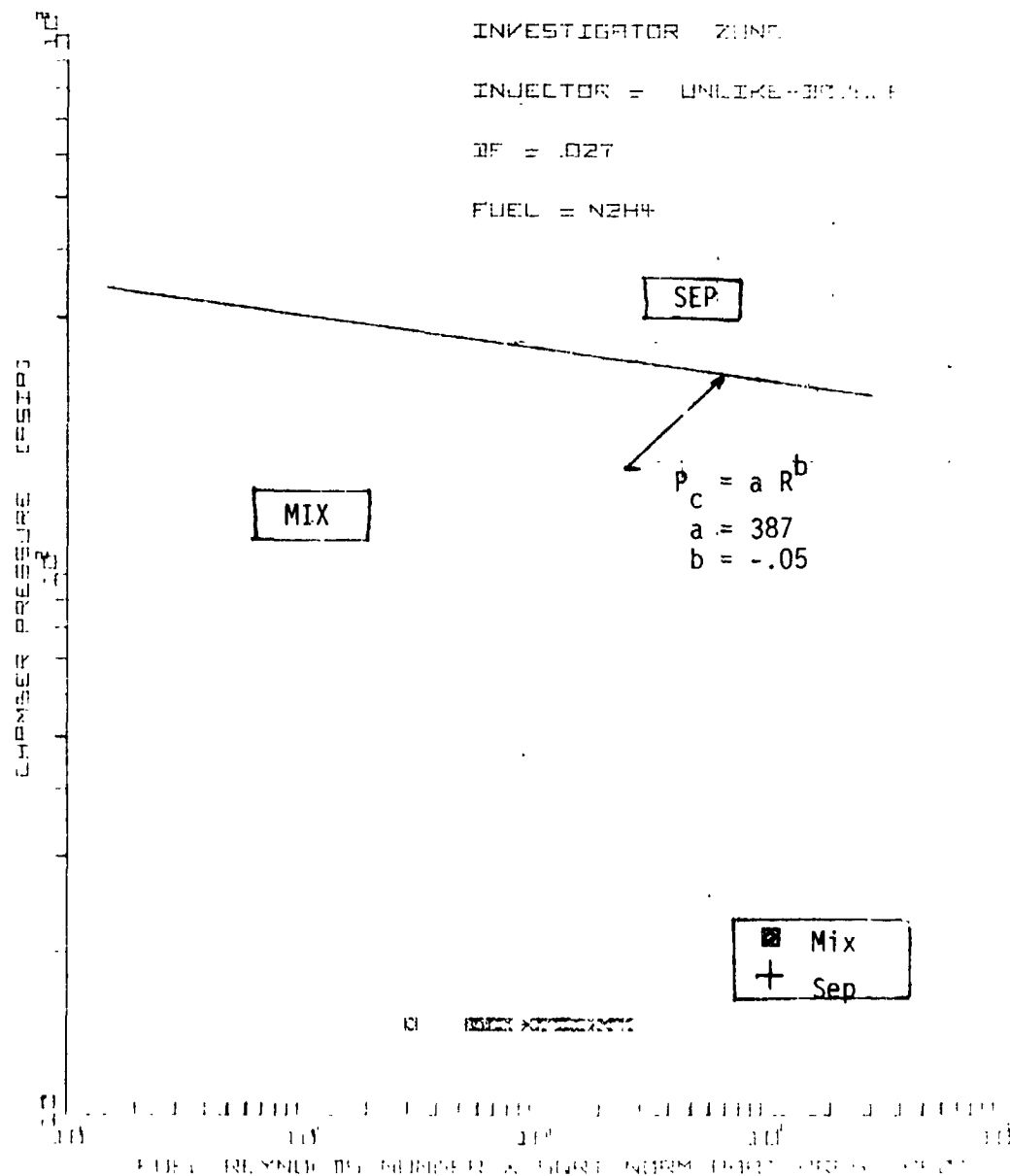


Figure B-5. Zung Unlike-Doublet, $D_f = .027$

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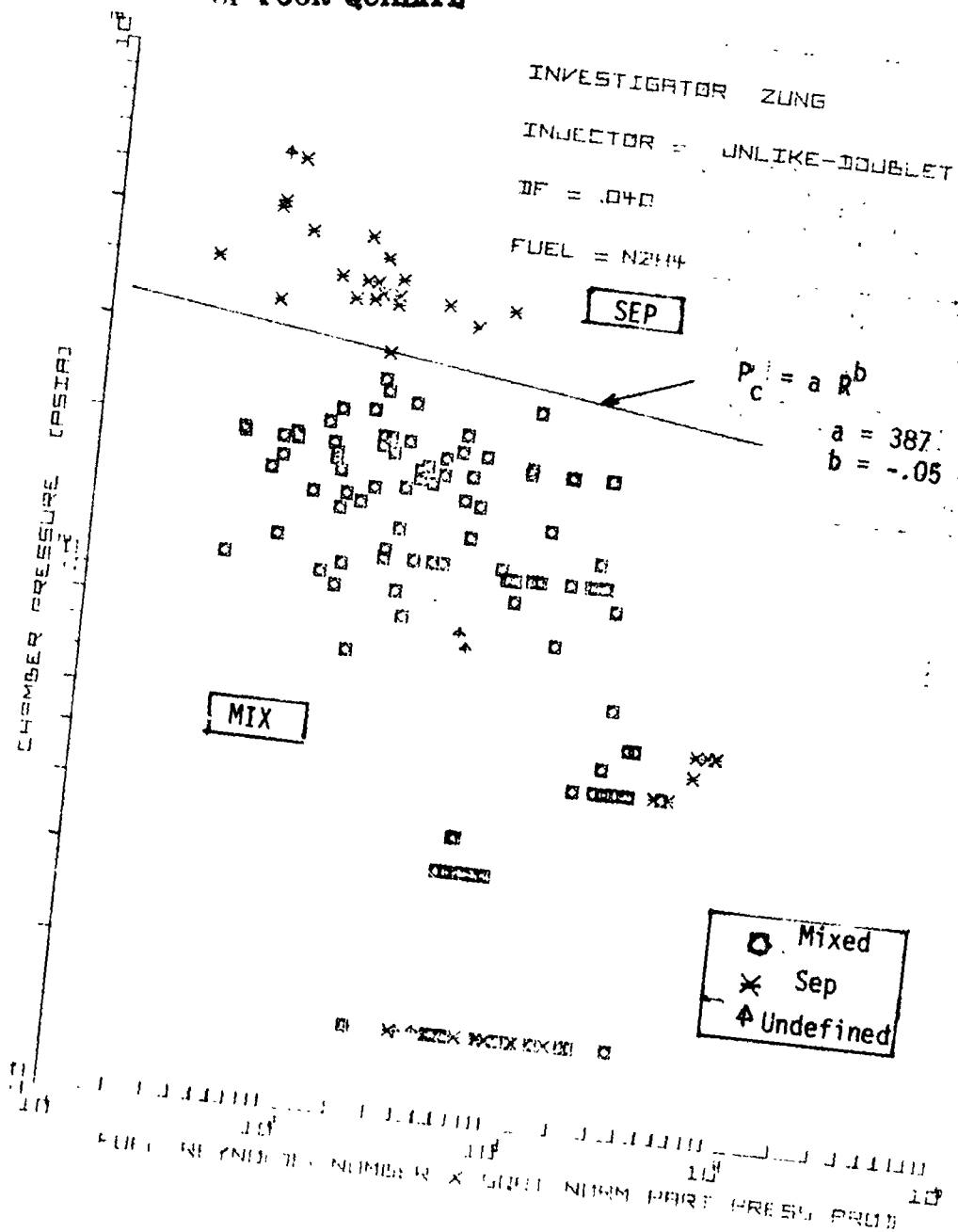


Figure B-6. Zung Unlike-Doublet, $D_f = .040$

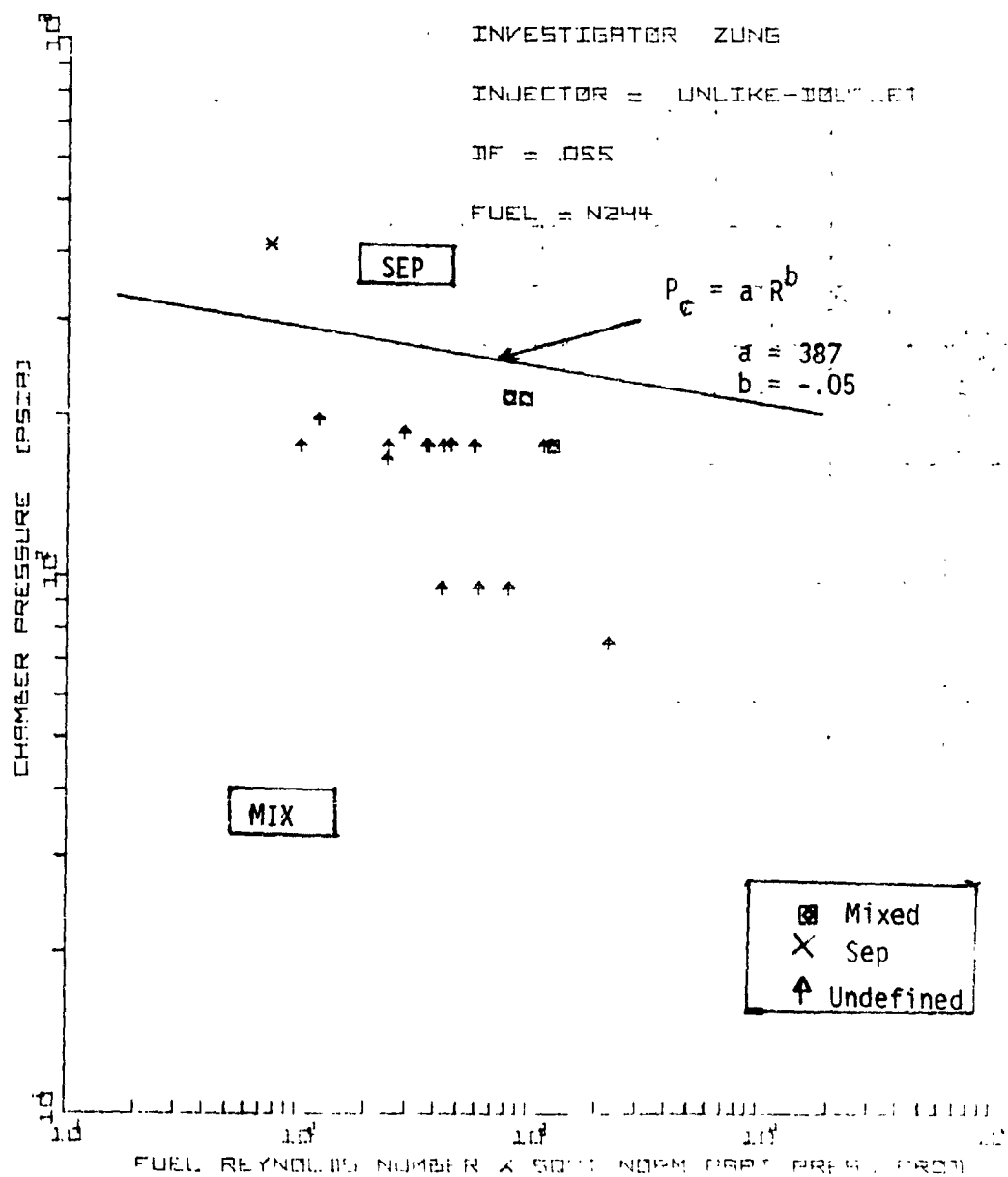


Figure B-7. Zung Unlike-Doublet, $D_f = .055$

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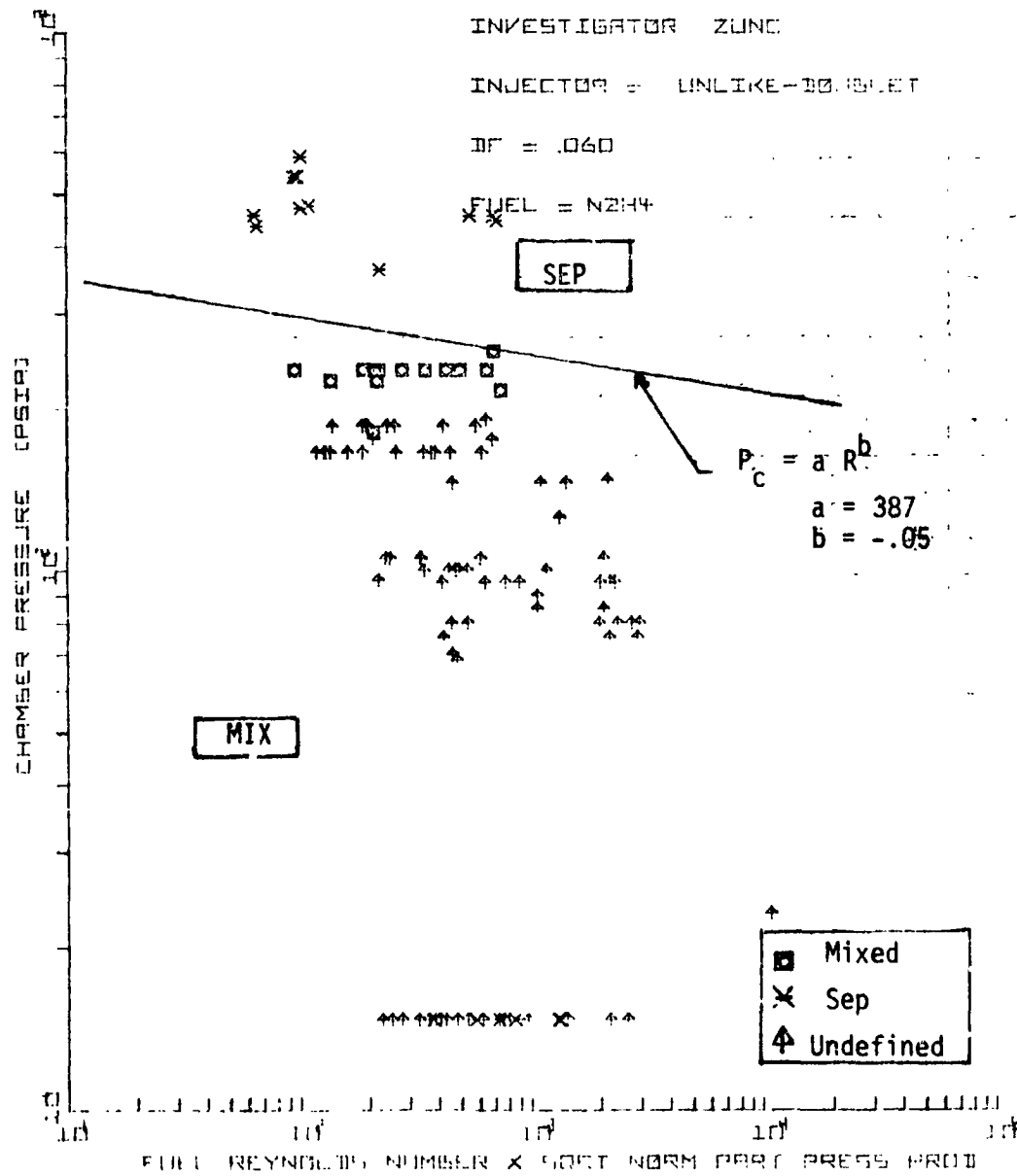


Figure B-8. Zung Unlike-Doublet, $D_f = .060$

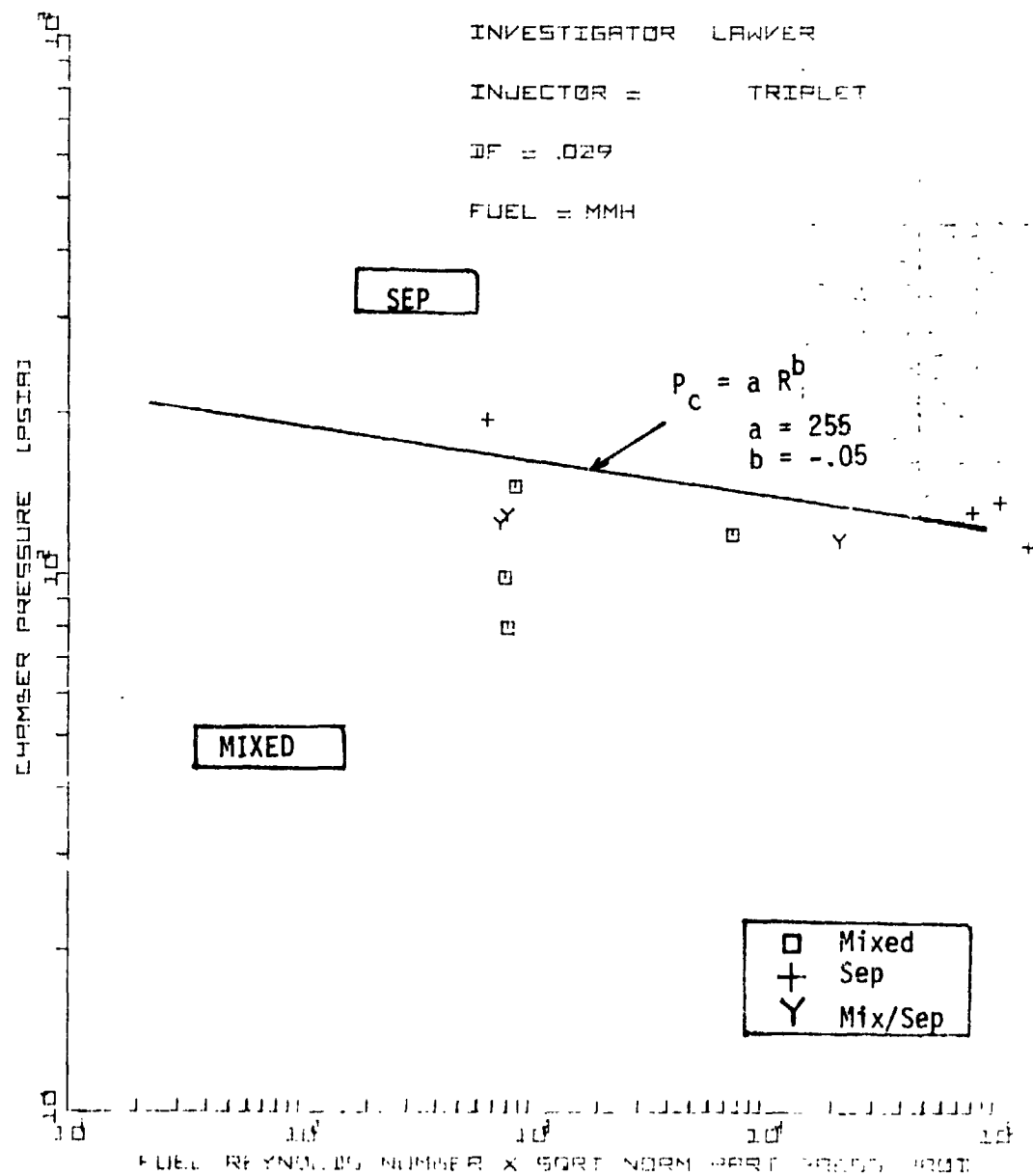


Figure B-9. Lawver Triplet, $D_f = .029$

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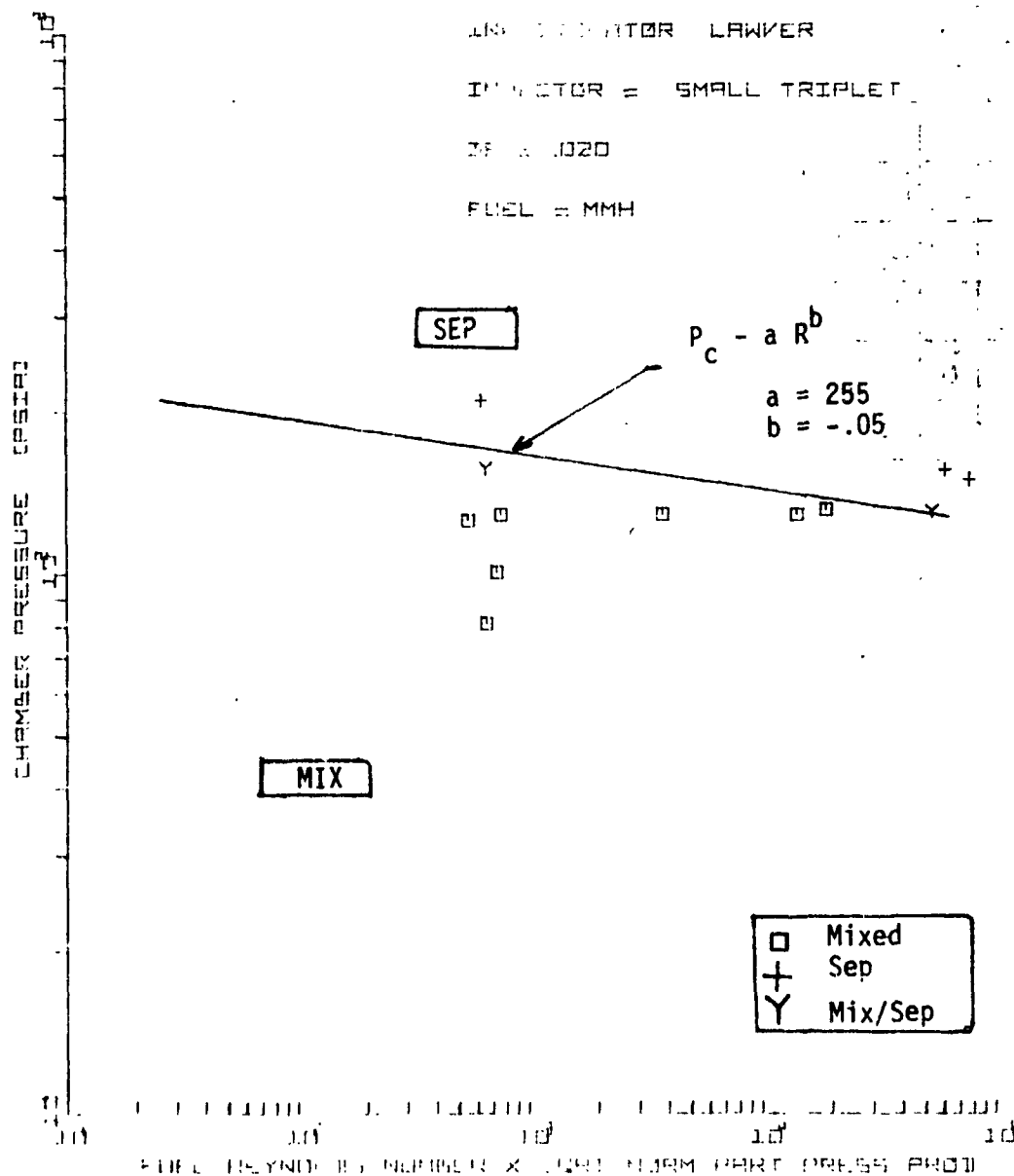


Figure 10. Lawver Small Triplet, $D_f = .020$

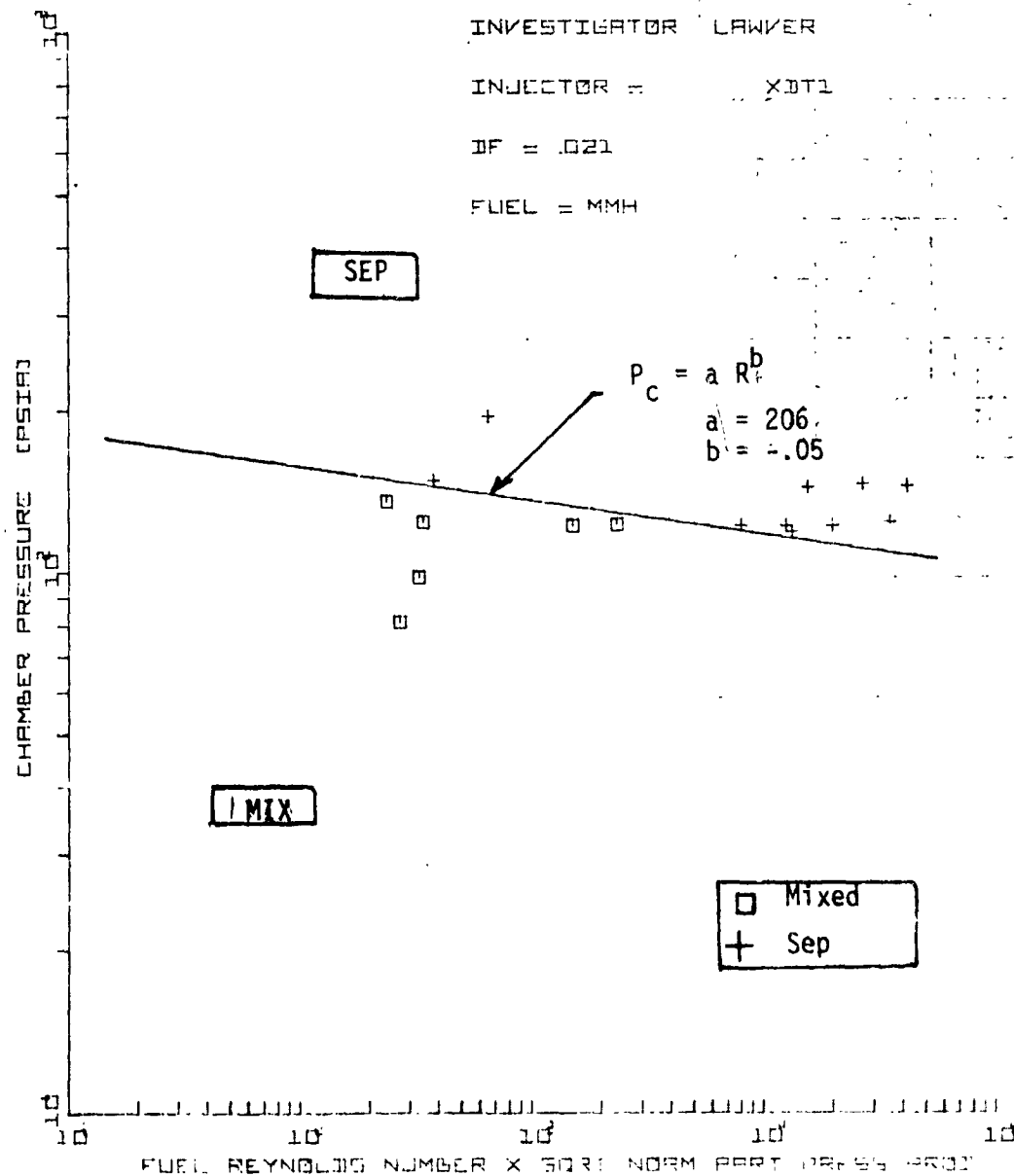


Figure B-11. Lawver XDT-1, $D_f = .021$

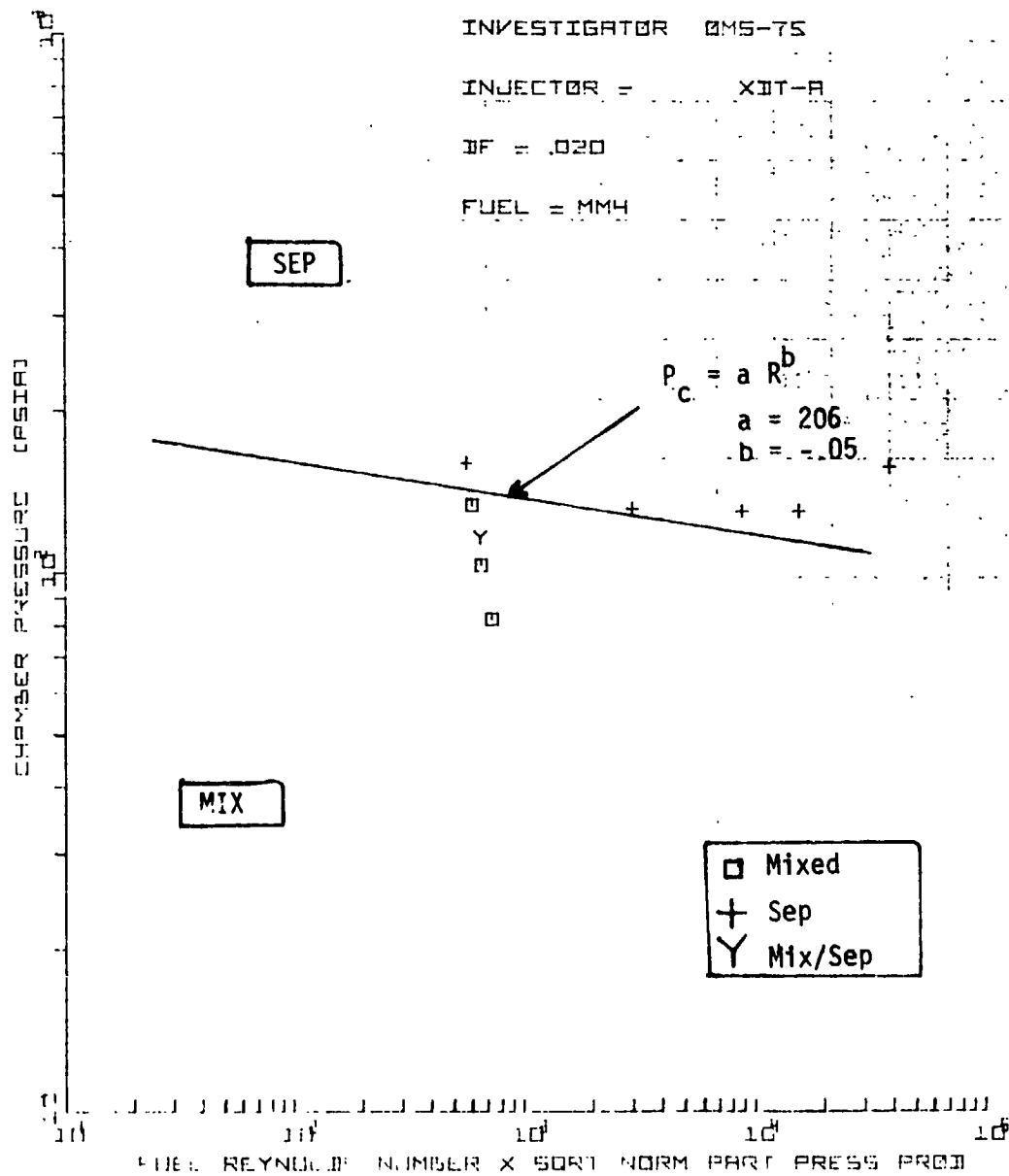


Figure B-12. OMS-75, XDT-A, $D_f = .020$

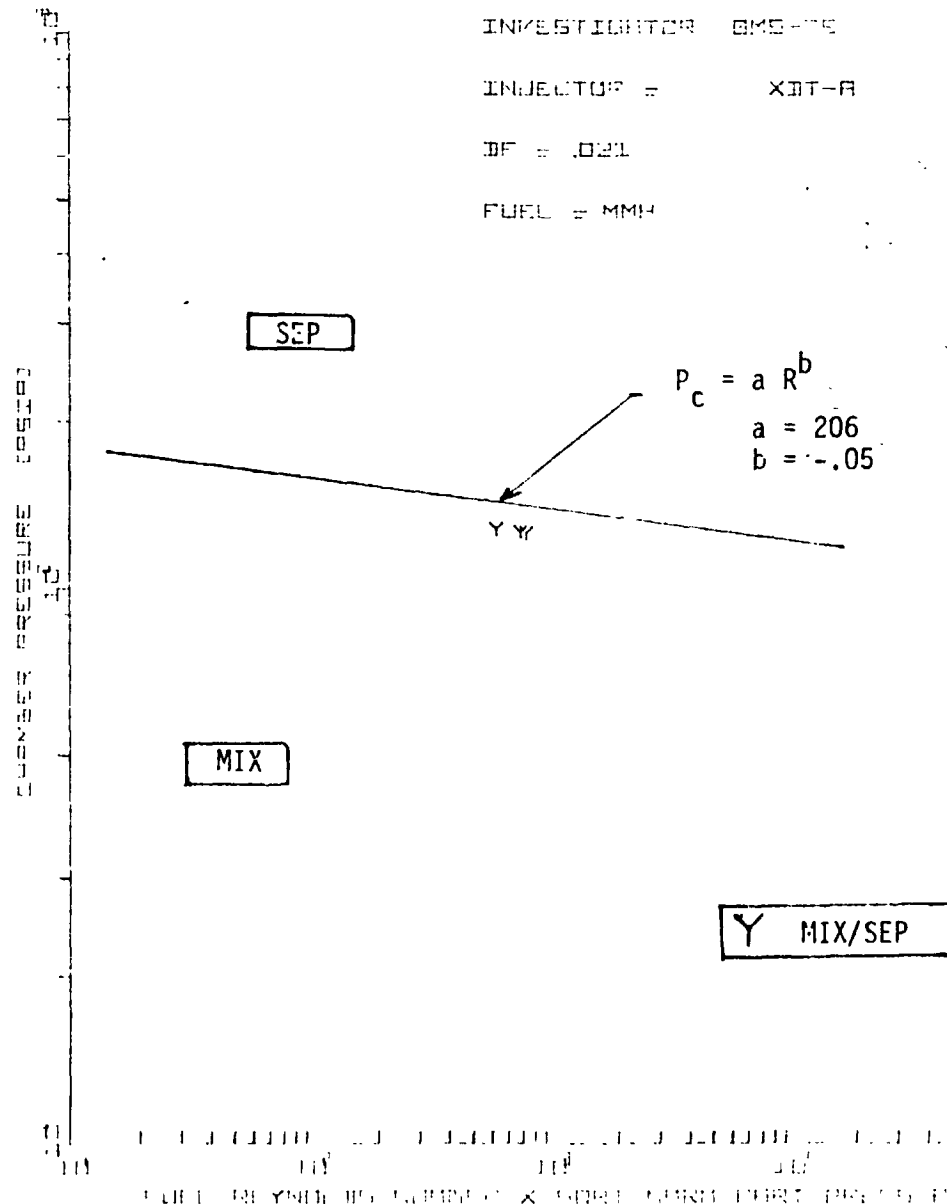


Figure B-13. OMS 75 XDT-A, $D_f = .021$

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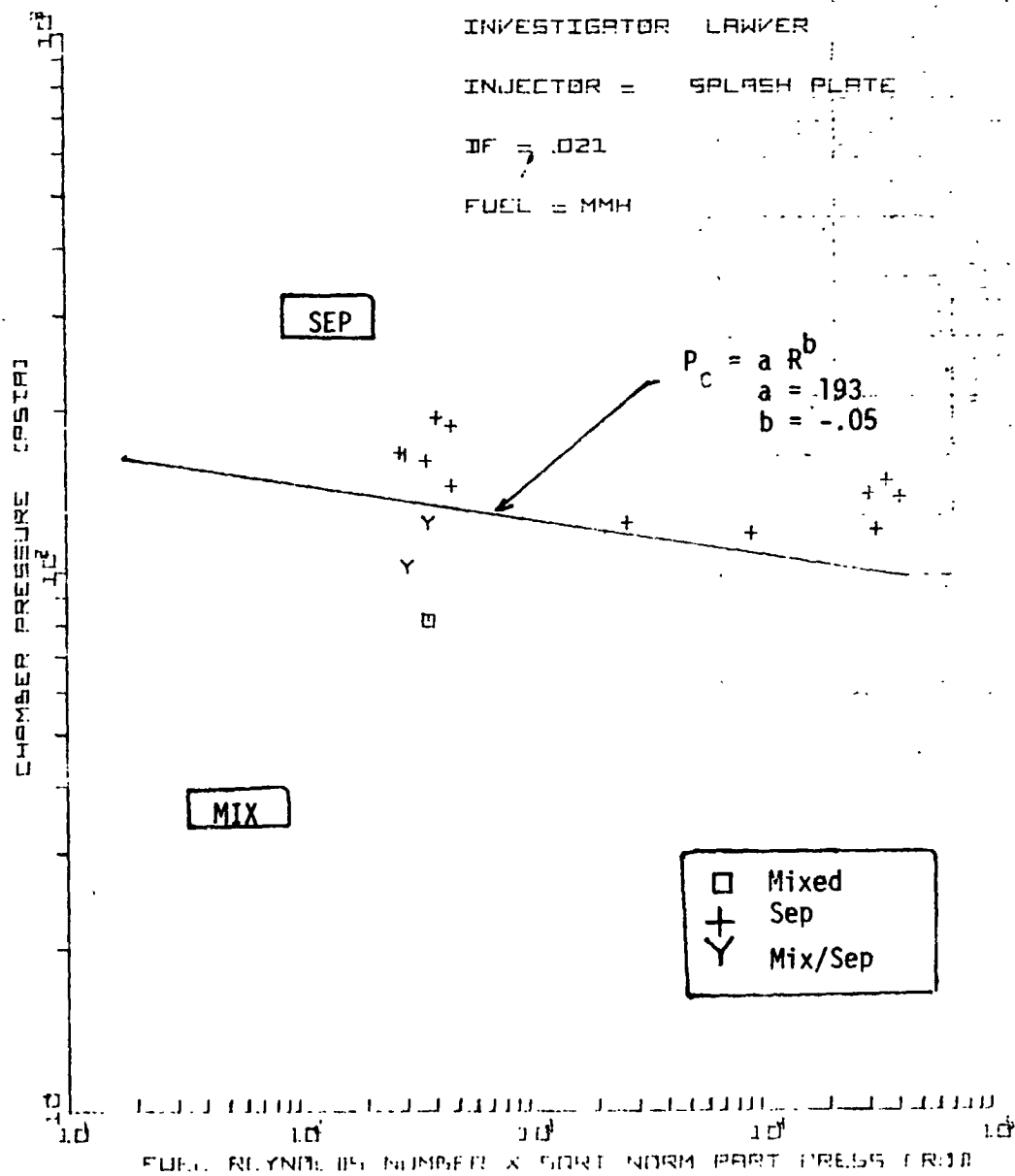


Figure B-14. Lawver Splash Plate, $D_f = .021$

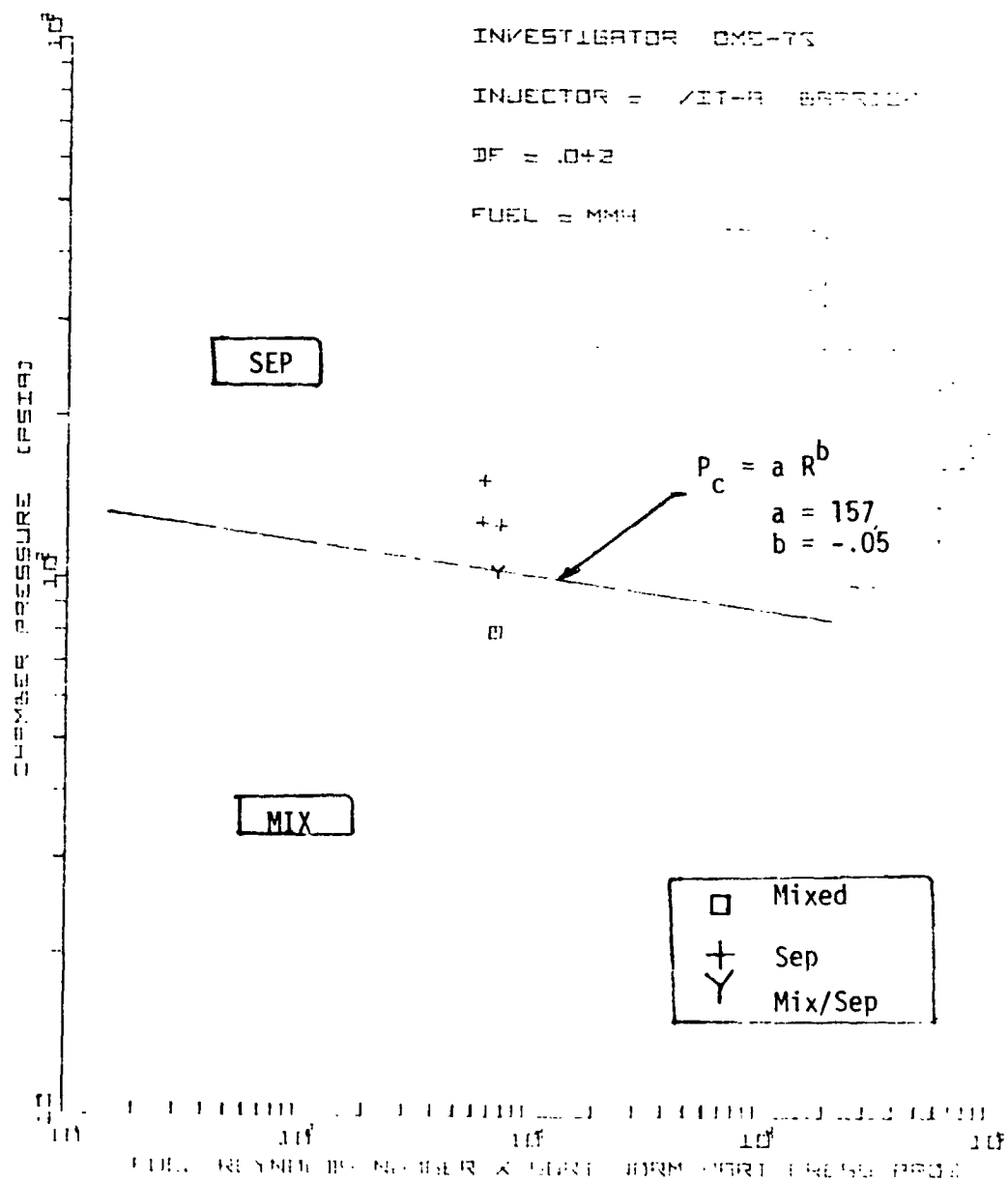


Figure B-15. OMS-75, VDT-A Barrier, $D_f = .042$

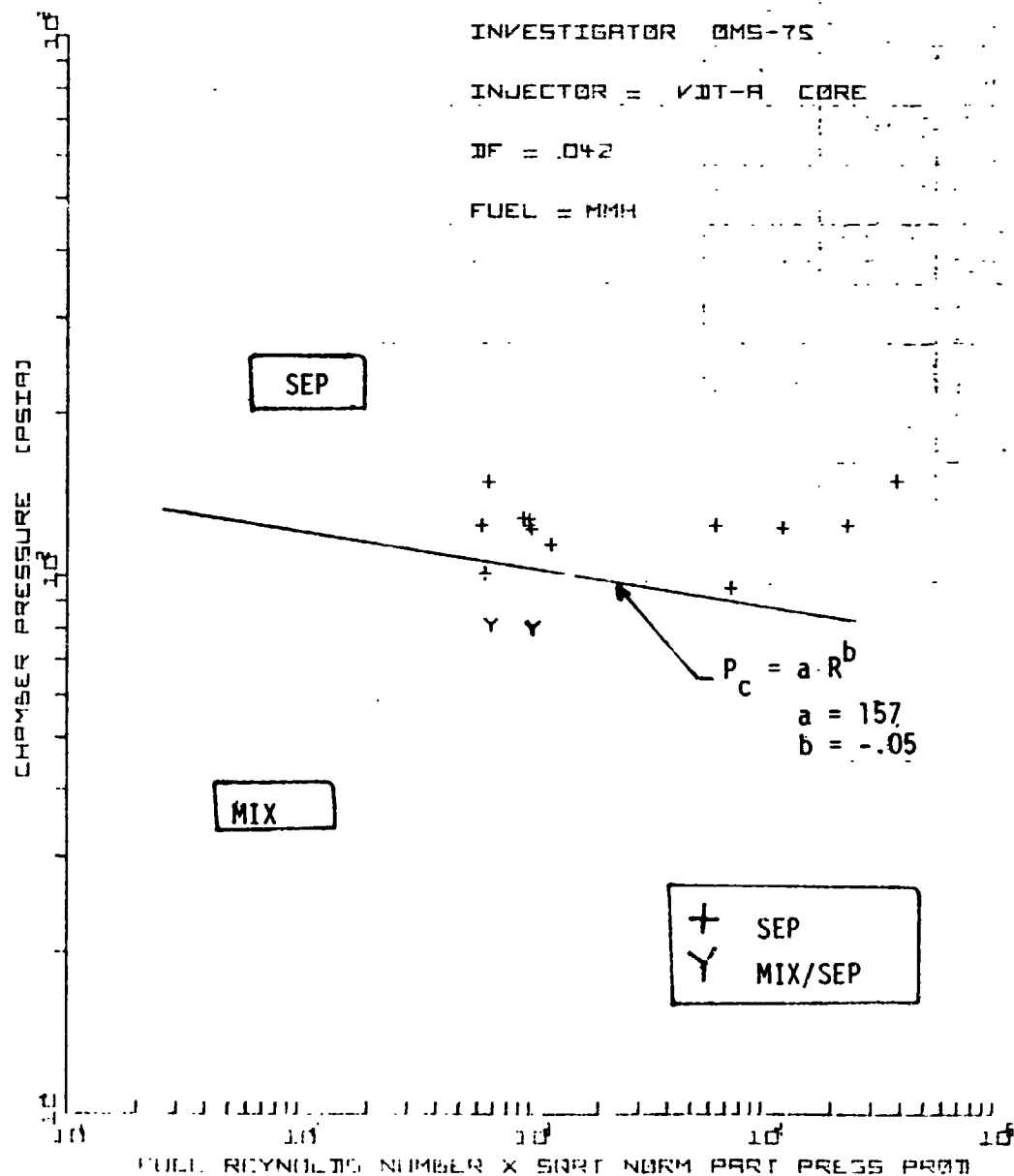


Figure B-16. OMS-75, VDT-A Core, $D_f = .042$

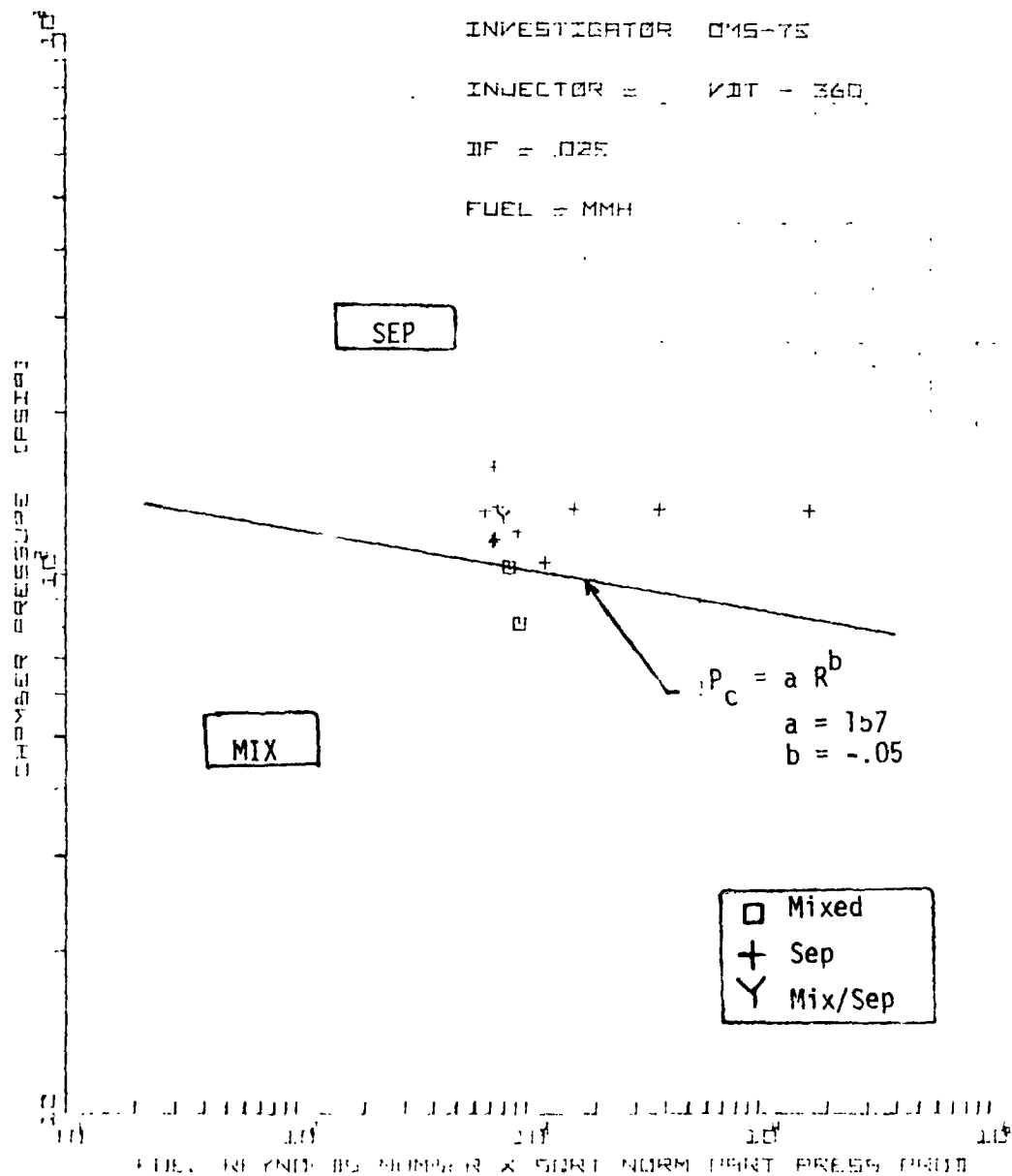


Figure B-17. OMS-75 VDT-360, $D_f = .025$

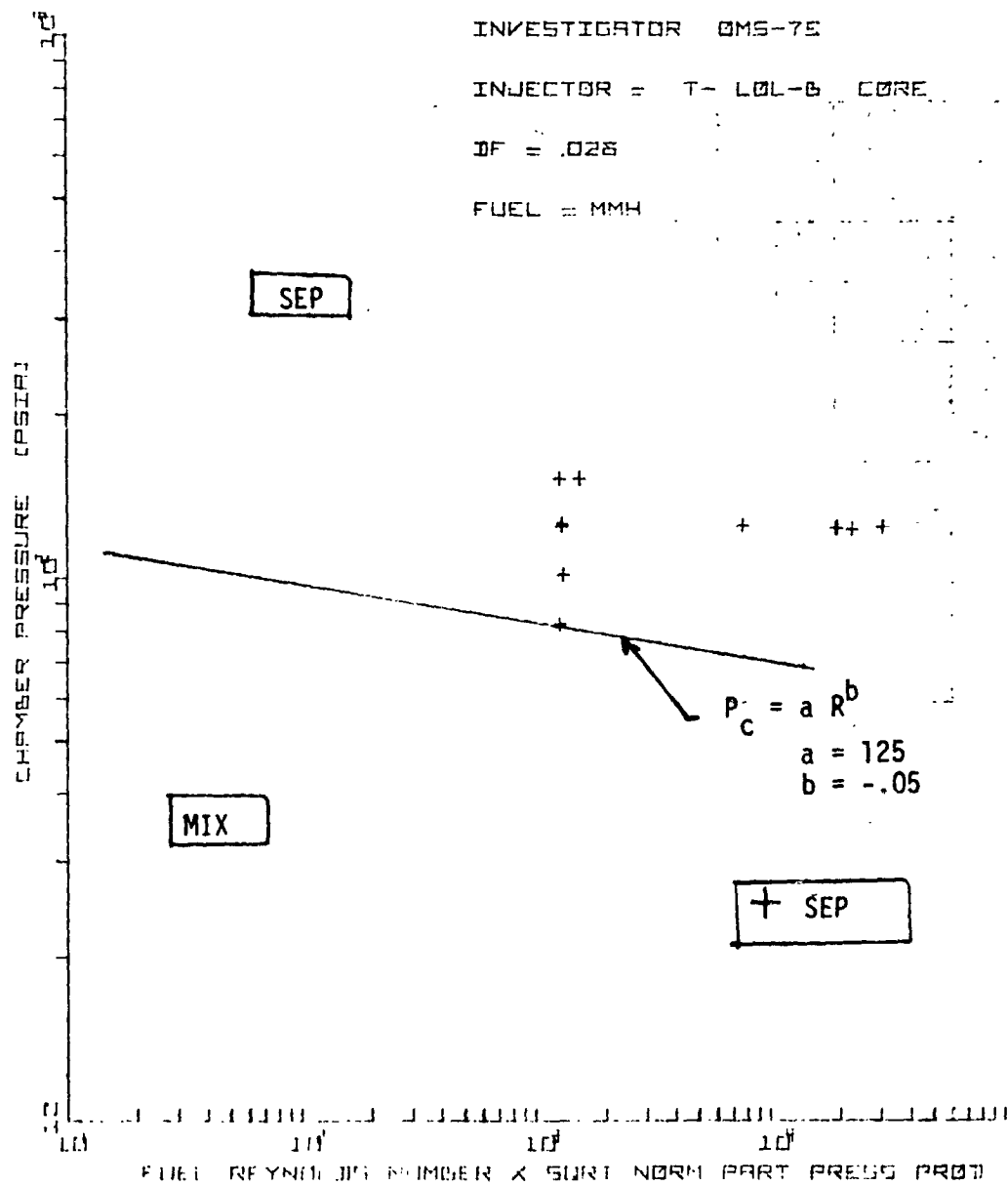


Figure B-18. OMS-75 T-L0L-B Core, $D_f = .028$

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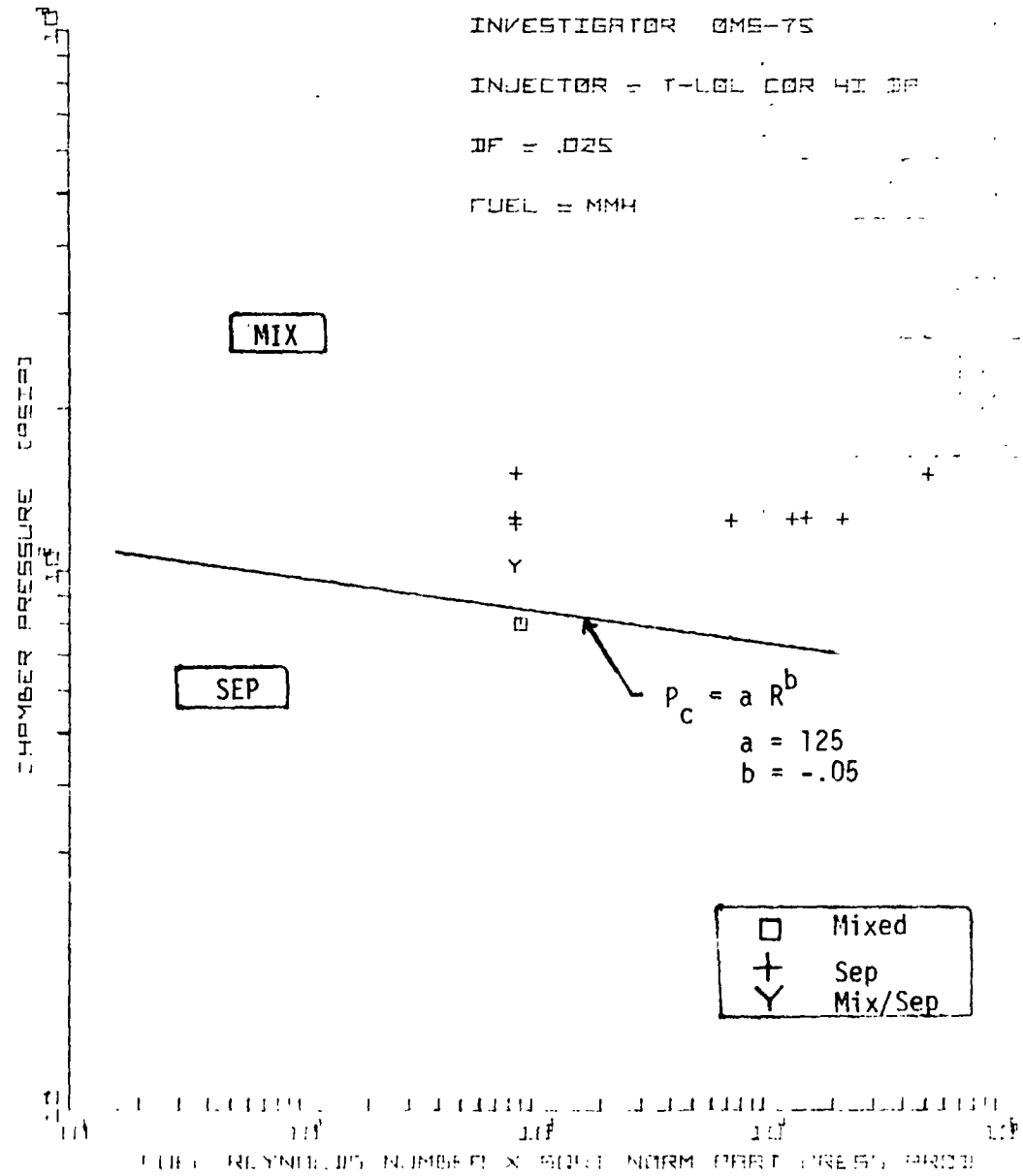


Figure B-19. OMS-75 T-LOL COR HI DP, $D_f = .025$

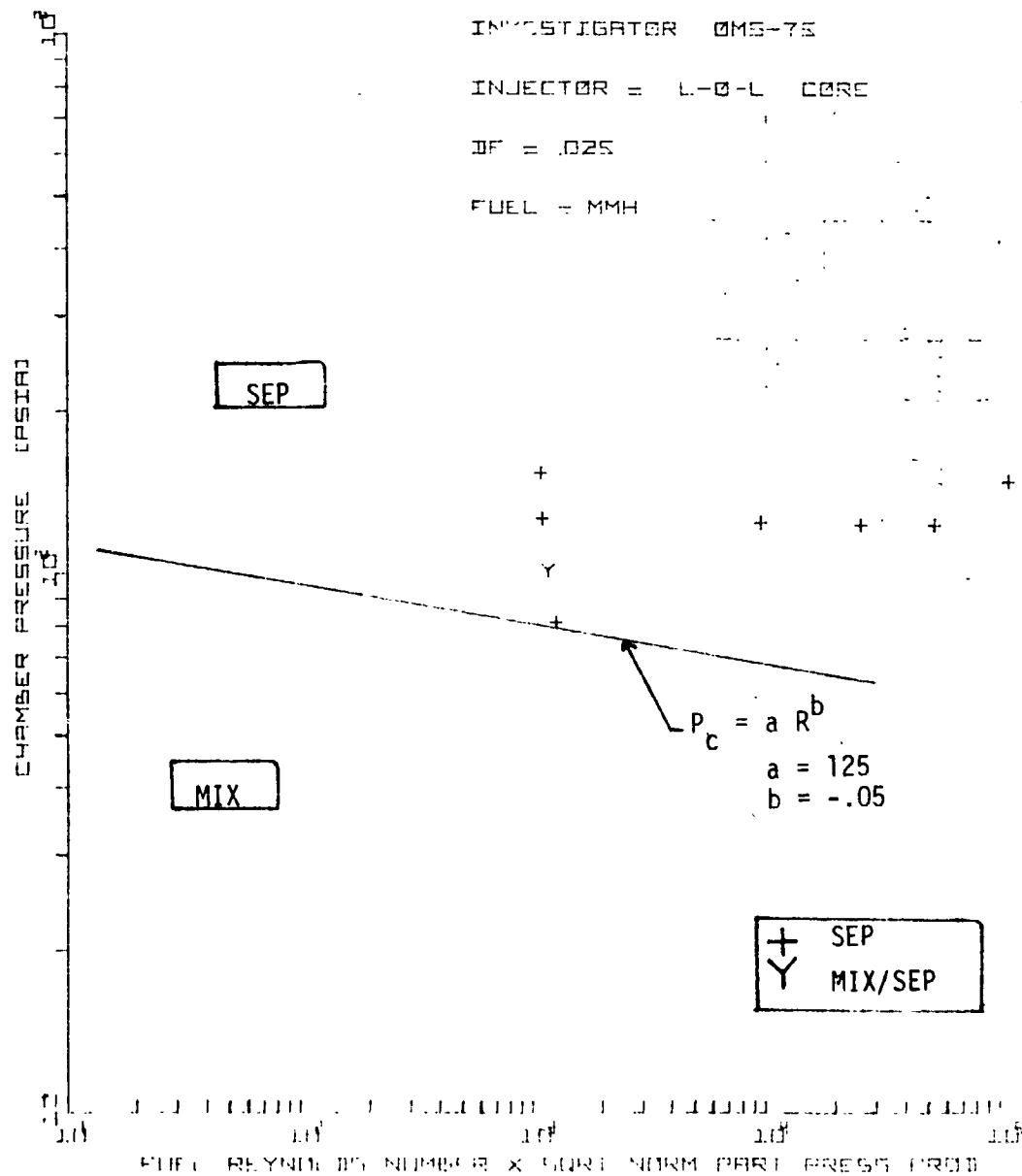


Figure B-20. OMS-75 LOL-CORE, $D_f = .025$

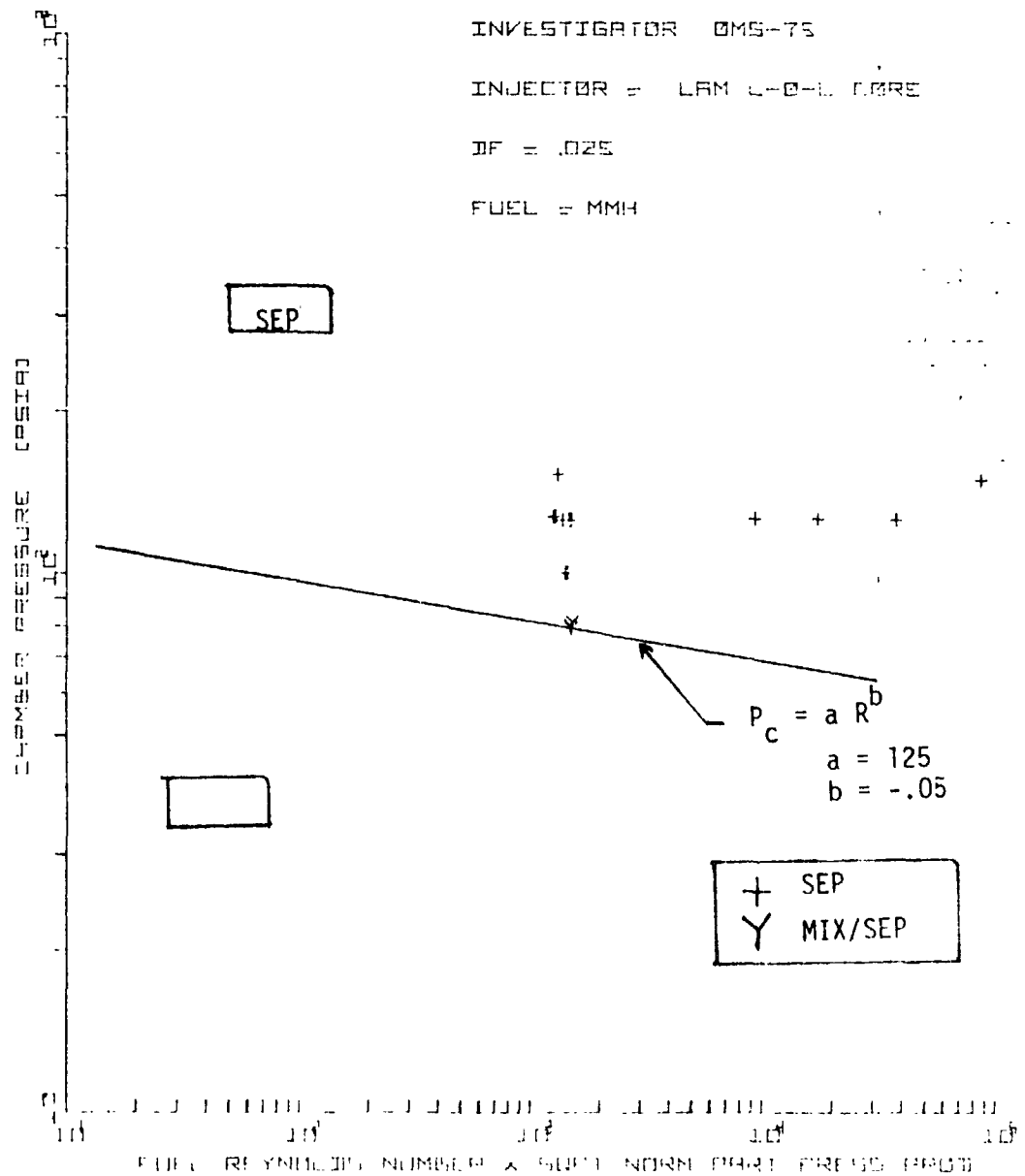


Figure B-21. OMS-75 LAR LOL CORE, $D_f = .025$

APPENDIX C

LIST OF CORRELATIONS EVALUATED

LIST OF CORRELATIONS EVALUATED

CORRELATION

$$P_c \text{ vs. } D_F$$

$$P_c/D \text{ vs. } T_f$$

$$P_c/D \text{ vs. } (D/V)_f$$

$$P_c/D^2 \text{ vs. } T_f$$

$$P_c \text{ vs. } L/V_f$$

$$P_c \text{ vs. } T_F$$

$$P_c \text{ vs. } \sqrt{XP}$$

$$P_c \text{ vs. } W_{ef} \sqrt{R_{ef}} \sqrt{P_{vo} P_{vf}}$$

$$P_c \text{ vs. } W_{ef} \sqrt{P_{vo} P_{vf}} \frac{(D/V)_f}{P_c}$$

$$*P_c \text{ vs. } R_{ef} \sqrt{P_{vo} P_{vf}/P_c}$$

$$1/T_f \text{ vs. } (D/V)_f$$

$$W_{ef} \sqrt{R_{ef}} \text{ vs. } \frac{P_{vo} P_{vf}}{P_c}$$

$$W_{ef} \sqrt{R_{ef}} \text{ vs. } \sqrt{P_{vo} P_{vf}}$$

$$W_{ef} \sqrt{R_{ef}} \text{ vs. } \sqrt{P_{vo} P_{vf}}$$

*Best Correlation

APPENDIX D

TASK I - COMPUTER LISTINGS AND TASK I DATA SUMMARIES

List of Appendix D Symbols

Reactive Stream Separation and
Popping Chronology (Table D-I)

Hypergolic Stream Impingement Data
Compilation (Table D-II)

List of Appendix D Data Sources

Propellant Stream Heating Model

RSS Data Storage and Reduction Program

APPENDIX D LIST OF SYMBOLS

D_F	Fuel orifice diameter, in
D_O	Oxidizer orifice diameter, in.
DV	Stream contact time, sec.
EM	Rupe Mixing efficiency
IS	$\ln (D/V \sin 1/2 \text{ Imp. Angle}) + 46.8 - 21800 1/T$
Imp. Angle	Stream impingement included angle, °
L/D	Orifice length/diameter ratio
MF/MO	Fuel to oxidizer momentum ratio
MR	Mixture ratio, oxidizer/fuel
P_C	Chamber pressure, psia
R	Element spacing correlation coefficient, $R = 49.2 \text{ DF}/[PC]^{1/3}$
SPR	Fuel to oxidizer momentum flux ratio $(\rho V^2)_F/(\rho V^2)_{ox}$
TF	Fuel temperature, °F
TO	Oxidizer temperature, °F
VF	Fuel injection velocity, ft/sec
VO	Oxidizer injection velocity, ft/sec

TABLE D-1

REACTIVE STREAM SEPARATION AND POPPING CHRONOLOGY

DATE	INVESTIGATOR	ORGAN.	CONTRACT NO.	REFERENCE	PROPELLANTS	SENSING TECHNIQUES	ELEMENT TYPES	ELEMENT SIZE	PC RANGE	TEMP. RANGE	COMMENTS
1959	Elverum & Staudhammer	JPL	NAS 7-100	Prog. Rpt. 30-4	H_2O_4/H_2H_4	Photographic	Single Element Unlike Doublet	100 Given	Atmos.	Ambient	First report of stream separation with hypergolic propellants.
1965	Johnson	JPL	NAS 7-100	TR No. 32-609	N_2O_4/H_2H_4 , MMH, UDMH, Furfuryl Alch. Corporal	Performance with Baffled Chamber	Single Element Unlike Doublet	.0236 Dia. 2000 lbf/ele.	150 PSIA	Ambient	Quantitative performance measurement of stream separation. N_2O_4/H_2H_4 , N_2O_4/MMH , and $N_2O_4/UDMH$ showed some degree of blowpart. Corporal propellant showed blowpart to a lesser degree and $N_2O_4/Furfural$ alcohol indicated no blowpart.
1966	Heiss & Popotter	AFRL	Proj. 624A	AFRPL-R-56-51	$H_2O_4/A-50$, H_2O_4/H_2H_4 , H_2O_4/MMH , $H_2O_4/H-50$	Pc Measurement	Transtage Quadlet	24 lbf/ele.	100 PSI	37-86°F	Provides popping data for Transtage injectors. Includes variations in propellant combination, injector type, manufacturing variations, mixture ratio and film cooling.
1967	Evans, Stanford & Retlin	JPL	NAS 7-100	TR No. 32-117	H_2O_4/H_2H_4	Performance with Baffled Chamber	Unlike Doublet & Imping. Sheet	10, 100, 2000 lbf/ele. .022, .064, .236 Dia.	150 PSIA	Ambient	Stream separation found to be element size dependent; increasing with increasing element size. Impinging sheets exhibited less blowpart than impinging jets.
1967	Larrows	NASA Lewis	In-house	NASA TR 52-244	H_2O_4/H_2H_4	Photographic, Thermocouple and emission spectra	Quadlet	.068 in. Dia.	19 Atmos.	Ambient	Stream separation observed with a unlike impingement quadlet element. Combustion product gases eventually mixed approximately 18 in. downstream of the injection point. Mixing obtained with like-on-like impingement quadlet element.
1967	Johnson & Rousman	JPL	NAS 7-100	NASA TR 33-395	H_2O_4/H_2H_4	None	Unlike Doublet	10-2000 lbf/ele.	10-2000 PSIA	40-160°F	First theoretical model of stream separation for unlike doublet elements. Two separation regimes postulated; liquid phase reactions controlling which applies at

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Table D-1 (cont.)

DATE	INVESTIGATORS	ORGAN.	CONTRACT NO.	REFERENCE	PROPELLANTS	SENSING TECHNIQUES	ELEMENT TYPES	ELEMENT SIZE	PC RANGE	TEMP. RANGE	COMMENTS
1968	Burrows	NASA/Lewis	In-house	NASA TMX-52483	N_2O_4/N_2H_4	Photographic	O-F-O & F-O-F Triplet	.025-.035-.025 Dia.	150-250 PSIA	Ambient	Lower pressures where separation is dependent on contact time (U/V); and propellant temperature, and if gas phase reaction controlling which applies at higher pressures where separation is dependent on (U/V) and independent on propellant temperature. Model correlates previous JPL separation data.
1968	Lawler & Breen	Dynamic Science	NASA-467	NASA-CR-72444	N_2O_4/N_2H_4	Photographic	Unlike Doublet	.025, .050, .100 in.	Atmos.	40-90°F	Developed a semi-empirical model in which separation is controlled by liquid phase mixing and kinetics, (U/V) and propellant temperature. A combustion "popping" regime was noted at lower temperatures and higher U/V values. Popping appeared to be caused by ignition within the entire impingement region of mixed liquid ligaments.
1969	Breen, Zung, Lawler, Kosvic & Coats	Dynamic Science	F04611-68-C-0040	AFRPL-TR-69-48	H_2O_4/N_2H_4 N_2O_4/MMH $N_2O_4/A-50$ ClF_5/N_2H_4	Photographic	Unlike Doublet Triplet	.025, .050, .150 Dia.	15-500 PSI	40-120 °F	Separation limits were defined for H_2O_4/N_2H_4 experimentally which were correlated with the liquid phase reaction chemical kinetics. Stream separation was observed to be independent of chamber pressure. Separation limits were observed at higher temperatures with $H_2O_4/A-50$ and H_2O_4/MMH as compared with the H_2O_4/N_2H_4 system. ClF_5/N_2H_4 always exhibited stream separation "popping" was postulated to be a function of element mixing level as well as U/V and propellant temperature.

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TABLE D-I (cont.)

DATE	INVESTIGATORS	ORGAN.	CONTRACT NO.	REFERENCE	PROPELLANTS	SENSING TECHNIQUES	ELEMENT TYPES	ELEMENT SIZE	PC RANGE	TEMP RANGE	COMMENTS
1969	Rupe, Dipprey, Kushida, & Clayton		NAS7-100	CPIA Pub. No. 192	$N_2O_4/A-50$	PC Measurement	Unlike Doublet	.142 & .073 Dia.	120 PSIA	30-105 F	Postulated that unlike doublet streams with equal stagnation pressure result in unstable operation or "pop". Correlation was demonstrated when a single element in an injector containing 72 elements was designed to operate at the unity pressure ratio. Pop free operation was attained without the element; 8 pops/second were obtained with the element.
1969	Housen	JPL	NAS7-100	CPIA Pub. No. 192	N_2O_4/N_2H_4	Mass Spectrometer & C* performance	Unlike Doublet	.020, .029, .073 Dia.	75-185 PSIA	40 F	Shown experimentally that jet mixing of hypergolic fluids can result in either penetrative, mixed, or separated regimes while operating at optimum cold flow mixing conditions
1969	Lawer	ALRC	NAS9-8285	ALRC TCER 9642:0106	$N_2O_4/A-50$	Photographic & PC and Accelerometer measurements	Unlike Doublet (Apollo 10S)	21.5 lbf/ele	100 PSIA	Ambient	Developed semi-empirical model in which "popping" is the result of spray detonation which is triggered by a blastwave generated by small explosions associated with hypergolic stream impingement. High speed movie of a single doublet element impinging into and A-50 verified the cyclic separation or "popping" which could provide the trigger source. Model showed above conditions must be met for the occurrence of negative popping; (1) stream impingement process must produce triggers, (2) element separation must provide for coupling of the trigger explosion, and (3) element impingement must be sufficient to produce pop. Model data from Apollo 10S, and 10S.

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TABLE D-I (cont.)

DATE	INVESTIGATORS	ORGAN.	CONTRACT NO.	REFERENCE	PROPELLANTS	SENSING TECHNIQUES	ELEMENT TYPES	ELEMENT SIZE	PC RANGE	TEMP. RANGE	COMMENTS
1970	Rodriguez and Axworthy	Rockwell Dyn.	NASA-7-139	NASA-CR-115863	H_2O_4/H_2H_4 H_2O_4/HHH $H_2O_4/UDMH$	Calorimetric & PVT measurements	Unlike Doublet	.028 Dia.	Atmos.	50°F	Experimentally measured the heat and gas release rates from hypergolic propellants reacting in the liquid phase. The reactivity with H_2O_4 was found to increase from hydrazine to UDMH to HHH.
1970	Claxton	JPL	NASA-7-100	JPL TR-32-1479	$H_2O_4/A-50$ $H_2O_4/Furfuryl alcohol$	Pc Measurement	Unlike Doublet	.042 & .073 Dia.	120-300 PSIA	40-100°F	Provides popping data from 130 tests with a 18 in. dia engine which was fired in a cylindrical and annular configuration. Separate flow control permitted changes in the relative boundary, outer core, and inner core, elements flow conditions. Concludes that impingement stream stagnation dynamics are significant in engine "popping". Equal dynamic pressure maximizes the tendency to produce pops. Popping occurrence and frequency was temperature sensitive. No pops were obtained at Pc = 300 psia and reduced popping was noted with furfuryl alcohol.
1972	Lee & Householder	JPL	NASA-7-100	Paper presented at West. Status Conf. Inst. Meeting Oct. 1970	H_2O_4/H_2H_4 H_2O_4/HHH $H_2O_4/UDMH$	Pc Meas. & Photographic	Unlike Doublet	.073, .100, & .173 Dia.	Atmos. to 450 PSIA	40-140°F	Presented experimental data and correlations which show that Pc, D ₁ , and V are controlling parameters for "popping" with high pressure or low contact time (D/V) eliminating pops. H_2H_4 and HHH exhibit similar popping trends, UDMH has relatively little tendency to pop. Popping rate noted to decrease with temperature and stagnation pressure ratio had only a small effect on popping rate.

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TABLE D-I (cont.)

DATE	INVESTIGATORS	ORGAN.	CONTRACT NO.	REFERENCE	PROPELLANTS	SENSING TECHNIQUES	ELEMENT TYPES	ELEMENT SIZE	PC RANGE	TEMP. RANGE	COMMENTS
1971	Zung & White	Dynamic Science	NAS3-12031	NAS CR-1704	N_2O_4/N_2H_4	Photographic & Chamber Pressure Measurement	Unlike Doublet	.027, .040, .055, & .060 Dia.	15-500 PSIA	40-140 °F	Presents results from nearly 500 tests with variation in temp, Pc, and D _j . Empirically determined operating regimes for "popping", stream mixing, and stream separation. The occurrence of popping was found to be chamber pressure and orifice size dependent; occurring at low Pc and Large D _j and absent at high Pc (> 185 psia) and/or small D _j . Identified two regimes of separated flow; one at low pressures and N ₂ O ₄ temperatures above the boiling point resulting in significant N ₂ O ₄ vaporization prior to impingement and one at high pressures with conventional liquid-liquid impingement. Mixed flow was observed at low pressures (< 230 psi) and temperatures below the -204 boiling point.
1971	Murick & Cordill	Rocketdyne	NAS7-720	NASA-CR-119246	N_2O_4/N_2H_4 $N_2O_4/A-50$ IRFNA/UDMH ClF ₅ /N ₂ H ₄	Photographic	Unlike Doublet	.030, .072, & .173 Dia.	Atmos. -10 to & 200 PSIA	80°F	Observed cyclic blowpart ("popping") with N ₂ O ₄ /N ₂ H ₄ (most violent), N ₂ O ₄ /A-50 and IRFNA/UDMH (least violent). Operating conditions reduced both the strength and frequency of "popping", decreasing with decreasing D _j and increasing V _j . Chamber pressure (to 200 PSIA) had little effect of "popping". Dynamic pressure ratio (0.9-1.5) produced some variation in the popping frequency. No continuous stream separation noted with N ₂ O ₄ , but continuous stream separation was observed with ClF ₅ /N ₂ H ₄ at all conditions except small D _j and high dynamic pressure ratio.

TABLE D-I (cont.)

DATE	INVESTIGATORS	ORGAN.	CONTRACT NO.	REFERENCE	PROPELLANTS	SENSING TECHNIQUES	ELEMENT TYPES	ELEMENT SIZE	°C RANGE	TEMP. RANGE	COMMENTS
1973	OMS Progra.	ALRC	In-House	-	N ₂ O ₄ /MMH N ₂ O ₄ /A-50	Performance Measurement (Isp & C*)	Unlike Doublet	D ₀ =.033, D _F =.028; D ₀ =.031, D _F =.028; D ₀ =.024, D _F =.020	125-150 PSIA	40-250°F Fuel Ambient Oxidizer	Subscale (1000 lbf) and fullscale (6000 lbf) OMS injectors demonstrated decreasing performance with increasing fuel temperature. Performance data showed a continuously decreasing performance efficiency with increasing fuel temperature. Performance decrease is lessened with increasing chamber length (4-12 in.). A-50 data showed slight increase in efficiency to 100°F then continuously decreasing efficiency.
1973	OMS Program	Rocket-dyne	NAS9-12802	ASR73-27	N ₂ O ₄ /MMH	Performance Measurement (C*)	Like-on-Like Doublet	D ₀ =.026, D _F =.024	100-140 PSIA	80-250°F Fuel Ambient Oxidizer	Single element testing with a 3.9 inch chamber showed continuous C* efficiency decrease with increasing fuel temperature for all elements tested. No performance change noted with full scale OMS injector tested with long chambers (12-20 in.)
1973	OMS Program	Rel Aero-Space Co.	NAS9-12803	Data Dump	N ₂ O ₄ /MMH	Performance Measurement (C*)	F-O-F Triplet	D ₀ =.05, D _F =.03	125 PSIA	40-200°F Fuel Ambient Oxidizer	C* efficiency decrease of 1/2-1% as fuel temperature increased from ambient to 200°F.
1973	OMS Program	ALRC	In-House	-	N ₂ O ₄ /MMH	Performance & Pc Measurement	Like-On-Like Doublet	D ₀ =.027, D _F =.025	110-140 PSIA	50-210°F Fuel Ambient Oxidizer	Full scale OMS injector showed slight decrease (0.1%) in performance efficiency from ambient to hot fuel with long chamber (16 in.). Axial Pc measurements indicate lower combustion near injector with hot fuel which is nearly damped at the end of the combustion chamber.

TABLE NO. D-II
HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION**

INVESTIGATOR		A L R C MODEL CORRELATION PARAMETERS																		
NIRICK		14**																		
FUEL TYPE	TEST NO.	JO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	COMMENTS	PD	IS (IN)	R (IN)	EM	SPR	NV (SEC)	
N2H4	9	.173	.173 10J.		45	13.7	33.0	43.0	55.0	60.0	.000	.860	POP	*	79.2	.115	3.560	.987	1.173	34-03
N2H4	10	.173	.173 10J.		45	13.7	33.0	43.0	55.0	62.0	.000	.860	POP	*	79.2	.135	3.560	.988	1.172	34-03
N2H4	11	.173	.173 10J.		45	13.7	33.0	41.0	55.0	58.0	.000	.910	POP	*	79.2	.103	3.560	.998	1.067	35-03
N2H4	12	.173	.173 10J.		45	13.7	33.0	41.0	52.0	67.0	.000	.930	POP	*	79.2	.210	3.560	.998	1.060	35-03
N2H4	13	.173	.173 10J.		45	13.7	30.0	36.0	55.0	55.0	.000	1.000	POP	*	79.2	.091	3.560	1.000	.997	40-03
N2H4	14	.173	.173 10J.		45	13.7	30.0	36.0	55.0	55.0	.000	1.000	POP	*	79.2	.091	3.560	1.000	.997	40-03
N2H4	15	.173	.173 10J.		45	13.7	30.0	40.0	55.0	55.0	.000	.850	POP	*	79.2	.082	3.560	.979	1.231	36-03
N2H4	16	.173	.173 10J.		45	13.7	28.0	41.0	55.0	65.0	.000	.690	POP	*	79.2	.180	3.560	.927	1.477	35-03
N2H4	17	.173	.173 10J.		45	13.7	28.0	43.0	55.0	65.0	.000	.630	POP	*	79.2	.171	3.560	.890	1.625	34-03
N2H4	18	.173	.173 10J.		45	13.7	30.0	43.0	55.0	65.0	.000	.830	POP	*	79.2	.171	3.560	.942	1.416	34-03
N2H4	19	.173	.173 10J.		45	13.7	32.0	43.0	55.0	65.0	.000	.880	POP	*	79.2	.171	3.560	.976	1.244	34-03
N2H4	20	.173	.173 10J.		45	13.7	33.0	43.0	55.0	65.0	.000	.910	POP	*	79.2	.171	3.560	.988	1.170	34-03
N2H4	21	.173	.173 10J.		45	13.7	33.0	41.0	55.0	65.0	.000	.910	POP	*	79.2	.171	3.560	.988	1.170	34-03
N2H4	22	.072	.072 10J.		60	13.7	38.0	50.0	45.0	55.0	.000	.850	POP	*	190.3	.021	1.482	.985	1.189	12-03
N2H4	23	.072	.072 10J.		60	13.7	51.0	63.0	45.0	55.0	.000	.940	POP	*	190.3	.017	1.482	.999	1.048	95-04
N2H4	24	.030	.030 10J.		60	13.7	47.0	56.0	45.0	55.0	.000	1.000	MIX	*	456.7	.008	.617	1.000	.975	45-04
N2H4	25	.030	.030 10J.		60	13.7	47.0	56.0	45.0	55.0	.000	1.000	MIX	*	456.7	.008	.617	.998	.946	46-04
N2H4	26	.030	.030 10J.		60	13.7	46.0	54.0	45.0	55.0	.000	1.030	POP	*	79.2	.013	3.560	.974	1.257	26-03
N2H4	27	.173	.173 10J.		60	13.7	41.6	56.2	40.0	40.0	.000	1.260	POP	*	79.2	.014	3.560	.991	1.142	28-03
N2H4	28	.173	.173 10J.		60	13.7	40.3	51.9	40.0	40.0	.000	1.150	POP	*	79.2	.014	3.560	.992	1.131	28-03
N2H4	29	.173	.173 10J.		60	13.7	40.5	51.9	40.0	40.0	.000	1.140	POP	*	79.2	.014	3.560	.995	.905	30-03
N2H4	30	.173	.173 10J.		60	13.7	42.3	48.5	40.0	40.0	.000	.900	POP	*	79.2	.014	3.560	1.000	1.014	29-03
N2H4	31	.173	.173 10J.		60	13.7	39.2	49.2	40.0	40.0	.000	1.020	POP	*	79.2	.014	3.560	.997	1.085	29-03
N2H4	32	.173	.173 10J.		60	13.7	36.0	52.0	40.0	50.0	.000	1.280	POP	*	79.2	.032	3.560	.990	1.131	28-03
A-5U	33	.173	.173 10J.		60	13.7	40.2	51.3	40.0	53.0	.000	1.040	POP	*	79.2	.032	3.560	1.000	1.001	28-03
A-5U	34	.173	.173 10J.		60	13.7	40.2	51.6	40.0	50.0	.000	1.050	POP	*	79.2	.032	3.560	1.000	1.012	28-03
A-5U	35	.173	.173 10J.		60	13.7	40.2	51.4	40.0	50.0	.000	1.040	POP	*	79.2	.032	3.560	1.000	1.005	28-03
A-5U	36	.173	.173 10J.		60	235.0	40.4	53.0	40.0	45.0	.000	1.500	POP	*	1358.4	.017	1.342	.892	1.617	22-03
A-5U	37	.173	.173 10J.		60	225.0	40.4	53.0	40.0	45.0	.000	1.100	POP	*	1300.6	.020	1.402	.987	1.176	26-03
A-5U	38	.173	.173 10J.		60	225.0	40.4	53.0	40.0	45.0	.000	.990	POP	*	1300.6	.021	1.402	.998	1.060	27-03
A-5U	39	.173	.173 10J.		60	220.0	40.4	52.0	40.0	45.0	.000	.960	POP	*	1271.7	.021	1.412	1.000	1.020	28-03

*See Appendix A Nomenclature for Symbol Definitions
**Ref. 4

HYPERGOLIC STREAM IMPINGEMENT DATA COMPIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR HOUSEMAN *

FUEL TYPE	TEST NO.	JO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	W/MO	COMMENTS	PD	IS	R (IN)	EM	SPR	DV (SEC)
N2H4	1	.073	.073	10J.	45.	14.2	38.2	38.2	70.0	70.0	1.200	1.000	POP	* 194.5	.120	1.484	.937	.696	.16-03
N2H4	2	.073	.073	100.	45.	16.0	36.1	36.1	70.0	70.0	1.200	1.000	POP	* 219.2	.127	1.427	.937	.596	.17-03
N2H4	3	.073	.073	100.	45.	42.0	36.8	36.8	70.0	70.0	1.200	1.000	MIX	* 575.3	.125	1.035	.937	.696	.17-03
N2H4	4	.073	.073	100.	45.	14.2	46.3	46.3	70.0	70.0	1.200	1.000	POP	* 194.5	.099	1.484	.937	.696	.13-03
N2H4	5	.073	.073	100.	45.	18.0	45.1	45.1	70.0	70.0	1.200	1.000	POP	* 246.6	.102	1.372	.937	.696	.13-03
N2H4	6	.073	.073	100.	45.	56.0	46.3	46.3	70.0	70.0	1.200	1.000	MIX	* 767.1	.099	.940	.937	.696	.13-03
N2H4	7	.073	.073	100.	45.	14.2	58.2	58.2	70.0	70.0	1.200	1.000	POP	* 194.5	.075	1.484	.937	.696	.10-03
N2H4	8	.073	.073	100.	45.	23.0	55.9	55.9	70.0	70.0	1.200	1.000	MIX	* 315.1	.082	1.264	.937	.696	.11-03
N2H4	9	.073	.073	100.	45.	73.0	60.8	60.8	70.0	70.0	1.200	1.000	POP	* 1000.0	.076	.861	.937	.696	.10-03
N2H4	10	.073	.073	100.	45.	14.2	73.0	73.0	70.0	70.0	1.200	1.000	POP	* 194.5	.063	1.484	.937	.696	.83-04
N2H4	11	.073	.073	100.	45.	29.0	70.8	70.8	70.0	70.0	1.200	1.000	MIX	* 397.3	.065	1.170	.937	.696	.86-04
N2H4	12	.073	.073	100.	45.	14.2	127.8	127.8	70.0	70.0	1.200	1.000	POP	* 194.5	.036	1.484	.937	.696	.48-04
N2H4	13	.073	.073	100.	45.	52.0	123.9	123.9	70.0	70.0	1.200	1.000	POP	* 712.3	.037	.944	.937	.696	.49-04
N2H4	14	.073	.073	100.	45.	120.0	115.2	115.2	70.0	70.0	1.200	1.000	MIX	* 1643.8	.040	.729	.937	.696	.53-04
N2H4	15	.100	.100	100.	45.	14.2	19.1	19.1	70.0	70.0	1.200	1.000	POP	* 142.0	.330	2.034	.937	.696	.44-03
N2H4	16	.100	.100	100.	45.	28.0	19.8	19.8	70.0	70.0	1.200	1.000	POP	* 280.0	.316	1.622	.937	.696	.42-03
N2H4	17	.100	.100	100.	45.	68.0	19.0	19.0	70.0	70.0	1.200	1.000	POP	* 680.0	.332	1.207	.937	.696	.44-03
N2H4	18	.100	.100	100.	45.	14.2	28.5	28.5	70.0	70.0	1.200	1.000	POP	* 142.0	.221	2.034	.937	.696	.29-03
N2H4	19	.100	.100	100.	45.	19.0	29.6	29.6	70.0	70.0	1.200	1.000	POP	* 190.0	.213	1.846	.937	.696	.28-03
N2H4	20	.100	.100	100.	45.	42.0	27.9	27.9	70.0	70.0	1.200	1.000	POP	* 420.0	.226	1.417	.937	.696	.30-03
N2H4	21	.100	.100	100.	45.	110.0	28.3	28.3	70.0	70.0	1.200	1.000	POP	* 1100.0	.223	1.028	.937	.696	.29-03
N2H4	22	.100	.100	100.	45.	14.2	37.3	37.3	70.0	70.0	1.200	1.000	POP	* 142.0	.169	2.034	.937	.696	.22-03
N2H4	23	.100	.100	100.	45.	52.0	39.9	39.9	70.0	70.0	1.200	1.000	MIX	* 230.0	.163	1.732	.937	.696	.22-03
N2H4	24	.100	.100	100.	45.	154.0	37.8	37.8	70.0	70.0	1.200	1.000	POP	* 1540.0	.167	.919	.937	.696	.21-03
N2H4	25	.100	.100	100.	45.	14.2	66.3	66.3	70.0	70.0	1.200	1.000	POP	* 142.0	.095	2.034	.937	.696	.22-03
N2H4	26	.100	.100	100.	45.	35.0	62.9	62.9	70.0	70.0	1.200	1.000	MIX	* 350.0	.100	1.506	.937	.696	.13-03
N2H4	27	.100	.100	100.	45.	250.0	63.2	63.2	70.0	70.0	1.200	1.000	POP	* 2500.0	.100	.782	.937	.696	.13-03
N2H4	28	.100	.100	100.	45.	54.0	117.4	117.4	70.0	70.0	1.200	1.000	POP	* 540.0	.054	1.303	.937	.696	.71-04
N2H4	29	.100	.100	100.	45.	118.0	145.0	145.0	70.0	70.0	1.200	1.000	MIX	* 1180.0	.043	1.005	.937	.696	.57-04
N2H4	30	.100	.100	100.	45.	14.2	40.0	40.0	70.0	70.0	1.200	1.000	POP	* 194.5	.115	1.484	.937	.696	.15-03
N2H4	31	.073	.073	100.	45.	450.0	32.3	32.3	70.0	70.0	1.200	1.000	POP	* 6164.4	.142	.470	.937	.696	.19-03
N2H4	32	.073	.073	100.	45.	14.2	21.6	21.6	70.0	70.0	1.200	1.000	POP	* 194.5	.213	1.484	.937	.696	.28-03
N2H4	33	.073	.073	100.	45.	94.0	20.5	20.5	70.0	70.0	1.200	1.000	POP	* 1287.7	.224	.791	.937	.696	.30-03
N2H4	34	.073	.073	100.	45.	250.0	18.7	18.7	70.0	70.0	1.200	1.000	POP	* 3424.7	.246	.571	.937	.696	.33-03
N2H4	35	.073	.073	100.	45.	28.0	71.0	71.0	70.0	70.0	1.200	1.000	MIX	* 383.6	.065	1.184	.937	.696	.86-04
N2H4	36	.073	.073	100.	45.	28.0	24.0	24.0	70.0	70.0	1.200	1.000	POP	* 383.6	.192	1.184	.937	.696	.25-03
N2H4	37	.073	.073	100.	45.	28.0	20.0	20.0	70.0	70.0	1.200	1.000	POP	* 280.0	.315	1.622	.937	.696	.42-03
N2H4	38	.100	.100	100.	45.	28.0	20.0	20.0	70.0	70.0	1.200	1.000	POP	* 161.8	2.019	2.806	.937	.696	.27-02
N2H4	39	.173	.173	100.	45.	28.0	5.4	5.4	70.0	70.0	1.200	1.000	POP	* 580.0	.077	1.026	.937	.696	.10-03
N2H4	40	.073	.073	100.	45.	43.0	60.0	60.0	70.0	70.0	1.200	1.000	MIX						

*Ref. 5

HYPERGOLIC STREAM IMPINGEMENT DATA COMPIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR HOUSEMAN

FUEL TYPE	TEST NO.	QO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	COMMENTS *	PD	IS	R (IN)	EM	SPR	OV (SEC)
N2H4	41	.073	.073	100.	45.	43.0	37.0	37.0	70.0	70.0	1.200	1.000	MIX *	589.0	.124	1.026	.937	.696	.16-03
N2H4	42	.100	.100	100.	45.	43.0	28.0	28.0	70.0	70.0	1.200	1.000	MIX *	430.0	.225	1.406	.937	.696	.30-03
N2H4	43	.100	.100	100.	45.	43.0	28.0	28.0	70.0	70.0	1.200	1.000	MIX *	430.0	.225	1.406	.937	.696	.30-03
N2H4	44	.173	.173	100.	45.	43.0	8.0	8.0	70.0	70.0	1.200	1.000	POP *	248.6	1.363	2.433	.937	.696	.18-02
N2H4	45	.073	.073	100.	45.	52.0	124.0	124.0	70.0	70.0	1.200	1.000	POP *	712.3	.037	.964	.937	.696	.49-04
N2H4	46	.073	.073	100.	45.	52.0	44.0	44.0	70.0	70.0	1.200	1.000	MIX *	712.3	.105	.964	.937	.696	.18-03
N2H4	47	.100	.100	100.	45.	52.0	40.0	40.0	70.0	70.0	1.200	1.000	MIX *	520.0	.158	1.320	.937	.696	.21-03
N2H4	48	.100	.100	100.	45.	52.0	40.0	40.0	70.0	70.0	1.200	1.000	POP *	520.0	.158	1.320	.937	.696	.21-03
N2H4	49	.173	.173	100.	45.	52.0	10.0	10.0	70.0	70.0	1.200	1.000	POP *	300.6	1.090	2.283	.937	.696	.14-02
N2H4	50	.073	.073	100.	45.	70.0	61.0	61.0	70.0	70.0	1.200	1.000	POP *	958.9	.075	.873	.937	.696	.10-03
N2H4	51	.100	.100	100.	45.	70.0	51.0	51.0	70.0	70.0	1.200	1.000	POP *	700.0	.124	1.195	.937	.696	.16-03
N2H4	52	.100	.100	100.	45.	70.0	20.0	20.0	70.0	70.0	1.200	1.000	POP *	700.0	.315	1.195	.937	.696	.42-03
N2H4	53	.073	.073	100.	45.	100.0	91.0	91.0	70.0	70.0	1.200	1.000	POP *	1369.9	.051	.775	.937	.696	.67-04
N2H4	54	.100	.100	100.	45.	100.0	80.0	80.0	70.0	70.0	1.200	1.000	POP *	1000.0	.079	1.062	.937	.696	.10-03
N2H4	55	.173	.173	100.	45.	100.0	21.0	21.0	70.0	70.0	1.200	1.000	POP *	578.0	.519	1.837	.937	.696	.69-03
N2H4	56	.173	.173	100.	45.	100.0	18.0	16.0	70.0	70.0	1.200	1.000	POP *	578.0	.682	1.837	.937	.696	.90-03
N2H4	57	.073	.073	100.	45.	120.0	115.0	115.0	70.0	70.0	1.200	1.000	MIX *	1643.8	.040	.729	.937	.696	.53-04
N2H4	58	.100	.100	100.	45.	120.0	145.0	145.0	70.0	70.0	1.200	1.000	MIX *	1200.0	.043	.999	.937	.696	.57-04
N2H4	59	.173	.173	100.	45.	120.0	17.0	17.0	70.0	70.0	1.200	1.000	POP *	693.6	.641	1.728	.937	.696	.85-03

HYPERGOLIC STREAM IMPINGMENT DATA COMPIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	COMMENTS	PO	IS	P (IN)	EM	SPR	DV (SEC)
N2H4	15	.027	.027	100.	60.	14.7	23.0	35.0	70.0	70.0	.940	1.620	SEP	.544.4	.037	.543	.893	1.612	.64-04
N2H4	28	.027	.027	100.	60.	14.7	44.0	75.0	70.0	70.0	.840	2.050	SEP	.544.4	.017	.543	.784	2.023	.30-04
N2H4	58	.027	.027	100.	60.	14.7	55.0	46.0	69.0	69.0	1.780	.470	MIX	.544.4	.026	.543	.776	.467	.49-04
N2H4	61	.027	.027	100.	60.	14.7	42.0	40.0	70.0	70.0	1.500	.640	SEP	.544.4	.033	.543	.901	.632	.56-04
N2H4	63	.027	.027	100.	60.	14.7	41.0	52.0	67.0	70.0	1.140	1.100	MIX	.544.4	.025	.543	.994	1.117	.43-04
N2H4	64	.027	.027	100.	60.	14.7	45.0	53.0	65.0	68.0	1.210	1.020	MIX	.544.4	.021	.543	.999	.963	.42-04
N2H4	68	.027	.027	100.	60.	14.7	71.0	78.0	70.0	70.0	1.290	.860	SEP	.544.4	.017	.543	.985	.840	.29-04
N2H4	79	.027	.027	100.	60.	14.7	20.0	25.0	70.0	70.0	1.110	1.160	SEP	.544.4	.052	.543	.996	1.088	.90-04
N2H4	80	.027	.027	100.	60.	14.7	41.0	54.0	68.0	68.0	1.090	1.220	MIX	.544.4	.021	.543	.982	1.207	.42-04
N2H4	81	.027	.027	100.	60.	14.7	34.0	53.0	70.0	70.0	.910	1.750	SEP	.544.4	.025	.543	.872	1.692	.42-04
N2H4	82	.027	.027	100.	60.	14.7	40.0	52.0	70.0	70.0	.020	1.700	SEP	.544.4	.025	.543	.987	1.177	.43-04
N2H4	83	.027	.027	100.	60.	14.7	38.0	48.0	70.0	70.0	1.080	1.200	SEP	.544.4	.027	.543	.994	1.111	.47-04
N2H4	85	.027	.027	100.	60.	14.7	19.0	24.0	67.0	68.0	1.100	1.180	MIX	.544.4	.046	.543	.995	1.109	.94-04
N2H4	89	.027	.027	100.	60.	14.7	53.0	94.0	67.0	67.0	.820	2.180	MIX	.544.4	.011	.543	.742	2.188	.24-04
N2H4	90	.027	.027	100.	60.	14.7	53.0	94.0	70.0	70.0	.820	2.180	SEP	.544.4	.014	.543	.741	2.190	.24-04
N2H4	91	.027	.027	100.	60.	14.7	65.0	92.0	70.0	70.0	1.020	1.380	SEP	.544.4	.014	.543	.946	1.395	.24-04
N2H4	93	.027	.027	100.	60.	14.7	43.0	60.0	60.0	60.0	1.040	1.310	MIX	.544.4	.010	.543	.956	1.351	.37-04
N2H4	94	.027	.027	100.	60.	14.7	39.0	58.0	45.0	45.0	1.310	.630	MIX	.544.4	.003	.543	.940	1.423	.40-04
N2H4	95	.027	.027	100.	60.	14.7	61.0	67.0	45.0	45.0	1.080	1.230	MIX	.544.4	.006	.543	.984	1.199	.34-04
N2H4	96	.027	.027	100.	60.	14.7	23.0	29.0	45.0	45.0	1.110	1.170	MIX	.544.4	.006	.543	.996	1.097	.78-04
N2H4	98	.027	.027	100.	60.	14.7	23.0	29.0	45.0	45.0	1.110	1.170	SEP	.544.4	.044	.543	.846	1.789	.35-04
N2H4	99	.027	.027	100.	60.	14.7	40.0	64.0	80.0	80.0	.890	1.800	SEP	.544.4	.085	.543	.818	1.854	.68-04
N2H4	100	.027	.027	100.	60.	14.7	20.0	33.0	75.0	80.0	.840	2.200	SEP	.544.4	.097	.543	.724	2.259	.62-04
N2H4	101	.027	.027	100.	60.	14.7	20.0	36.0	79.0	83.0	.800	2.280	SEP	.544.4	.135	.543	.977	1.239	.94-04
N2H4	102	.027	.027	100.	60.	14.7	18.0	24.0	78.0	82.0	1.100	1.200	UNDEF	.544.4	.231	.543	.671	.403	.62-04
N2H4	108	.027	.027	100.	60.	14.7	47.0	36.0	70.0	95.0	1.890	.410	SEP	.544.4	.277	.543	.591	.351	.75-04
N2H4	109	.027	.027	100.	60.	14.7	42.0	30.0	70.0	95.0	2.010	.410	SEP	.544.4	.037	.543	.998	1.057	.73-04
N2H4	110	.027	.027	100.	60.	14.7	25.0	31.0	70.0	95.0	1.180	1.060	SEP	.544.4	.027	.543	.962	.757	.94-04
N2H4	111	.027	.027	100.	60.	14.7	23.0	24.0	65.0	65.0	1.380	.760	MIX	.544.4	.078	.543	.999	1.038	.68-04
N2H4	112	.027	.027	100.	60.	14.7	27.0	33.0	65.0	65.0	1.160	1.070	MIX	.544.4	.061	.543	.968	1.290	.69-04
N2H4	113	.027	.027	100.	60.	14.7	24.0	33.0	70.0	110.0	1.040	1.330	SEP	.544.4	.617	.543	.912	1.539	.83-04
N2H4	114	.027	.027	100.	60.	14.7	18.0	27.0	70.0	105.0	.960	1.590	SEP	.544.4	.757	.543	1.000	1.022	.10-03
N2H4	116	.027	.027	100.	60.	14.7	16.0	22.0	70.0	105.0	.960	1.580	UNDEF	.544.4	.112	.543	.872	1.691	.90-04
N2H4	117	.027	.027	100.	60.	14.7	52.0	70.0	70.0	80.0	.960	1.580	UNDEF	.544.4	.089	1.206	.975	1.255	.71-04
N2H4	118	.060	.060	100.	60.	14.7	17.0	17.0	70.0	80.0	1.070	1.280	UNDEF	.544.4	.365	1.206	1.000	1.022	.29-04
N2H4	119	.060	.060	100.	60.	14.7	17.0	22.0	70.0	80.0	1.230	.950	SEP	.544.4	.282	1.206	.989	1.160	.23-03
N2H4	120	.060	.060	100.	60.	14.7	17.0	22.0	70.0	80.0	1.120	1.150	UNDEF	.544.4	.129	1.206	.979	1.232	.10-03
N2H4	121	.060	.060	100.	60.	14.7	36.0	48.0	70.0	80.0	1.090	1.210	UNDEF	.544.4	.067	1.206	.967	1.296	.54-04
N2H4	122	.060	.060	100.	60.	14.7	68.0	93.0	70.0	80.0	1.060	1.280	UNDEF	.544.4	.067	1.206	.967	1.296	.54-04

*Ref. 27

HYPERGOLIC STREAM IMPINGEMENT DATA COMPIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	JO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	WR MF/MO	COMMENTS *	PD	IS (IN)	R (IN)	EM	SPR	DV (SEC)
N2H4	123	.060	.060	100.	60.	14.7	13.0	18.0	75.0	75.0	1.160	1.060	245.0	.236	1.206	.959	1.337	.28-03
N2H4	124	.060	.060	100.	60.	14.7	8.3	13.0	71.0	72.0	.950	1.060	245.0	.260	1.206	.868	1.708	.38-03
N2H4	125	.060	.060	100.	60.	14.7	6.6	11.0	71.0	72.0	.890	1.810	245.0	.307	1.206	.808	1.934	.45-03
N2H4	126	.060	.060	100.	60.	14.7	5.4	8.6	72.0	72.0	.900	1.770	245.0	.393	1.206	.852	1.767	.58-03
N2H4	127	.060	.060	100.	60.	14.7	4.3	5.6	72.0	72.0	1.100	1.170	245.0	.603	1.206	.986	1.182	.69-03
N2H4	128	.060	.060	100.	60.	14.7	13.0	16.0	40.0	50.0	1.110	1.150	245.0	.036	1.206	.999	1.038	.31-03
N2H4	129	.060	.060	100.	60.	14.7	9.1	14.0	42.0	52.0	.960	1.560	245.0	.049	1.206	.890	1.623	.50-03
N2H4	130	.060	.060	100.	60.	14.7	8.3	10.0	42.0	52.0	1.180	1.010	245.0	.068	1.206	1.000	.996	.56-03
N2H4	131	.060	.060	100.	60.	14.7	8.0	9.0	42.0	52.0	1.270	.880	245.0	.076	1.206	.990	.868	.63-03
N2H4	132	.060	.060	100.	60.	14.7	10.0	7.9	45.0	52.0	1.830	.430	245.0	.086	1.206	.706	.429	.52-03
N2H4	133	.060	.060	100.	60.	14.7	9.9	12.0	45.0	50.0	1.180	1.000	245.0	.043	1.206	1.000	1.011	.38-03
N2H4	134	.060	.060	100.	60.	14.7	9.3	13.0	47.0	50.0	1.000	1.420	245.0	.044	1.206	.957	1.347	.58-03
N2H4	135	.060	.060	100.	60.	14.7	4.3	8.6	63.0	62.0	.710	2.810	245.0	.179	1.206	.606	2.779	.36-03
N2H4	136	.060	.060	100.	60.	14.7	12.0	14.0	54.0	54.0	1.170	1.050	245.0	.057	1.206	.998	.942	.42-03
N2H4	137	.060	.060	100.	60.	14.7	9.6	12.0	55.0	55.0	1.040	1.540	245.0	.073	1.206	.926	1.483	.25-03
N2H4	138	.060	.060	100.	60.	14.7	8.2	12.0	61.0	61.0	1.130	1.110	245.0	.119	1.206	.997	1.084	.13-03
N2H4	139	.060	.060	100.	60.	14.7	14.0	20.0	63.0	63.0	.980	1.470	245.0	.083	1.206	.941	1.417	.20-03
N2H4	140	.060	.060	100.	60.	14.7	22.0	26.0	66.0	70.0	1.200	.990	367.5	.074	.804	1.000	.969	.16-03
N2H4	141	.040	.040	100.	60.	14.7	18.0	21.0	70.0	70.0	1.220	.950	367.5	.092	.804	.999	.948	.24-03
N2H4	142	.040	.040	100.	60.	14.7	13.0	17.0	70.0	70.0	1.070	1.250	367.5	.114	.804	.985	1.191	.20-03
N2H4	143	.040	.040	100.	60.	14.7	12.0	14.0	70.0	70.0	1.180	1.030	367.5	.138	.804	.999	.948	.20-03
N2H4	144	.040	.040	100.	60.	14.7	21.0	17.0	72.0	92.0	1.170	.490	367.5	.059	.804	.736	.452	.12-03
N2H4	145	.040	.040	100.	60.	14.7	24.0	28.0	68.0	68.0	1.210	.970	367.5	.029	.804	.999	.947	.12-03
N2H4	146	.040	.040	100.	60.	14.7	24.0	28.0	67.0	59.0	1.200	.980	367.5	.029	.804	.999	.950	.18-03
N2H4	147	.040	.040	100.	60.	14.7	16.0	19.0	62.0	65.0	1.220	1.290	367.5	.069	.804	1.000	.978	.15-03
N2H4	148	.040	.040	100.	60.	14.7	14.0	19.0	63.0	63.0	1.010	1.400	367.5	.059	.804	.970	1.279	.15-03
N2H4	149	.040	.040	100.	60.	14.7	16.0	22.0	69.0	73.0	1.070	1.250	3725.0	.111	.372	.964	1.513	.14-03
N2H4	150	.040	.040	100.	60.	14.7	23.0	23.0	67.0	73.0	1.440	.690	4725.0	.106	.344	.936	.693	.14-03
N2H4	151	.040	.040	100.	60.	187.0	24.0	23.0	66.0	73.0	1.470	.650	4675.0	.106	.345	.904	.636	.14-03
N2H4	152	.040	.040	100.	60.	189.0	23.0	25.0	47.0	59.0	1.350	.790	4725.0	.032	.344	.978	.811	.13-03
N2H4	153	.040	.040	100.	60.	209.0	24.0	27.0	40.0	52.0	1.310	.840	5225.0	.017	.332	.990	.867	.22-03
N2H4	154	.040	.040	100.	60.	111.0	15.0	15.0	40.0	58.0	1.450	.680	2775.0	.050	.410	.930	.683	.12-03
N2H4	155	.040	.040	100.	60.	191.0	23.0	27.0	43.0	58.0	1.250	.920	4775.0	.028	.342	.998	.943	.12-03
N2H4	156	.040	.040	100.	60.	172.0	22.0	27.0	54.0	64.0	1.200	.990	4300.0	.045	.354	.999	1.038	.12-03
N2H4	157	.040	.040	100.	60.	187.0	22.0	27.0	57.0	65.0	1.200	.990	4675.0	.048	.345	.999	1.040	.14-03
N2H4	158	.040	.040	100.	60.	163.0	21.0	23.0	56.0	63.0	1.500	.840	4075.0	.048	.361	.982	.828	.14-03
N2H4	159	.040	.040	100.	60.	163.0	19.0	24.0	76.0	81.0	1.550	1.070	4125.0	.186	.359	.995	1.111	.17-03
N2H4	160	.040	.040	100.	60.	140.0	18.0	20.0	77.0	82.0	1.230	.940	3500.0	.240	.380	.989	.860	.16-03
N2H4	161	.040	.040	100.	60.	259.0	20.0	19.0	81.0	87.0	1.470	.660	6475.0	.365	.309	.899	.629	.11-03
N2H4	162	.040	.040	100.	60.	157.0	19.0	29.0	85.0	93.0	.940	1.610	3925.0	.368	.365	.890	1.625	.11-03

HYPERGOLIC STREAM IMPINGEMENT DATA COMPIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/WO	COMMENTS	Pn	IS	P (IN)	EM	SPR	DV (SEC)
N2H4	184	.040	.040	100.	60.	119.0	17.0	14.0	95.0	101.0	1.830	.430	MIX	* 2975.0	1.33A	.401	.763	.475	.24-.03
N2H4	185	.040	.040	100.	60.	131.0	17.0	17.0	107.0	96.0	1.410	.720	MIX	* 3275.0	.777	.348	.943	.710	.20-.03
N2H4	186	.040	.040	100.	60.	155.0	22.0	24.0	77.0	91.0	1.310	.840	MIX	* 3875.0	.386	.367	.982	.825	.14-.03
N2H4	187	.040	.040	100.	60.	194.0	27.0	27.0	91.0	91.0	1.410	.720	MIX	* 4850.0	.343	.341	.940	.702	.12-.03
N2H4	188	.040	.040	100.	60.	194.0	26.0	29.0	86.0	90.0	1.240	.920	MIX	* 4850.0	.297	.341	.990	.870	.11-.03
N2H4	191	.040	.040	100.	60.	170.0	25.0	25.0	96.0	103.0	1.420	.710	MIX	* 4250.0	.860	.346	.939	.701	.13-.03
N2H4	192	.040	.040	100.	60.	152.0	24.0	27.0	104.0	113.0	1.270	.890	MIX	* 3800.0	1.566	.369	.993	.889	.12-.03
N2H4	193	.040	.040	100.	60.	174.0	25.0	31.0	94.0	99.0	1.140	1.100	MIX	* 4350.0	.526	.353	.997	1.078	.11-.03
N2H4	194	.040	.040	100.	60.	169.0	24.0	30.0	100.0	104	1.130	1.120	MIX	* 4225.0	.769	.347	.996	1.098	.11-.03
N2H4	196	.040	.040	100.	60.	194.0	24.0	32.0	76.0	94.0	1.090	1.210	MIX	* 4850.0	.358	.341	.979	1.229	.10-.03
N2H4	197	.040	.040	100.	60.	187.0	16.0	32.0	106.0	76.0	.720	2.790	MIX	* 4675.0	.096	.345	.588	2.866	.10-.03
N2H4	198	.040	.040	100.	60.	164.0	24.0	27.0	91.0	101.0	1.250	.920	MIX	* 4100.0	.694	.360	.992	.884	.10-.03
N2H4	199	.040	.040	100.	60.	181.0	23.0	32.0	93.0	108.0	1.030	1.360	MIX	* 4525.0	.945	.349	.956	1.340	.12-.03
N2H4	200	.040	.040	100.	60.	203.0	23.0	32.0	64.0	77.0	1.040	1.320	MIX	* 5075.0	.103	.345	.959	1.336	.10-.03
N2H4	201	.040	.040	100.	60.	178.0	20.0	32.0	62.0	77.0	.920	1.680	MIX	* 4450.0	.103	.340	.853	1.764	.10-.03
N2H4	204	.040	.040	100.	60.	110.0	31.0	36.0	52.0	64.0	1.230	.950	MIX	* 2750.0	.033	.411	.997	.927	.03-.04
N2H4	205	.040	.040	100.	60.	115.0	33.0	35.0	76.0	84.0	1.360	1.360	MIX	* 2875.0	.159	.405	.970	.782	.95-.04
N2H4	206	.040	.040	100.	60.	115.0	33.0	30.0	91.0	95.0	1.590	1.590	MIX	* 2875.0	.410	.405	.863	.579	.11-.03
N2H4	207	.040	.040	100.	60.	110.0	29.0	34.0	110.0	107.0	1.230	1.230	MIX	* 2750.0	.831	.411	1.000	.973	.08-.04
N2H4	208	.040	.040	100.	60.	110.0	28.0	40.0	102.0	108.0	.990	1.440	MIX	* 2750.0	.756	.411	.937	1.434	.83-.04
N2H4	209	.040	.040	100.	60.	110.0	29.0	35.0	105.0	110.0	1.190	.990	MIX	* 2750.0	.989	.411	1.000	1.025	.95-.04
N2H4	210	.040	.040	100.	60.	110.0	29.0	37.0	121.0	110.0	1.120	1.130	MIX	* 2750.0	.935	.411	.989	1.162	.90-.04
N2H4	211	.040	.040	100.	60.	110.0	30.0	35.0	121.0	116.0	1.240	.930	MIX	* 2750.0	1.472	.411	.999	.969	.95-.04
N2H4	212	.040	.040	100.	60.	110.0	30.0	35.0	121.0	118.0	1.240	.930	MIX	* 2750.0	1.678	.411	.999	.969	.95-.04
N2H4	215	.040	.040	100.	60.	110.0	56.0	68.0	108.0	118.0	1.170	.850	MIX	* 2750.0	.864	.411	.999	1.036	.49-.04
N2H4	217	.040	.040	100.	60.	110.0	56.0	68.0	93.0	101.0	1.170	.850	MIX	* 2750.0	.275	.411	1.000	1.031	.49-.04
N2H4	219	.040	.040	100.	60.	110.0	56.0	63.0	123.0	121.0	1.270	.790	MIX	* 2750.0	1.133	.411	.994	.900	.53-.04
N2H4	220	.040	.040	100.	60.	110.0	54.0	68.0	117.0	122.0	1.130	.880	MIX	* 2750.0	1.119	.411	.993	1.121	.49-.04
N2H4	221	.040	.040	100.	60.	110.0	59.0	68.0	103.0	110.0	1.240	.810	MIX	* 2750.0	.509	.411	.998	.933	.49-.04
N2H4	222	.040	.040	100.	60.	110.0	55.0	68.0	119.0	114.0	1.160	.860	MIX	* 2750.0	.664	.411	.997	1.087	.49-.04
N2H4	223	.040	.040	100.	60.	110.0	59.0	68.0	119.0	114.0	1.520	.800	UNDEF	* 2750.0	.664	.411	.998	.945	.49-.04
N2H4	231	.040	.040	100.	60.	350.0	51.0	66.0	79.0	67.0	1.100	1.180	SEP	* 8750.0	.023	.240	.987	1.177	.51-.04
N2H4	232	.040	.040	100.	60.	380.0	55.0	66.0	85.0	72.0	1.180	1.000	SEP	* 9500.0	.034	.272	1.000	1.015	.51-.04
N2H4	234	.040	.040	100.	60.	353.0	50.0	68.0	118.0	118.0	1.120	.880	SEP	* 8825.0	.864	.279	.943	.709	.49-.04
N2H4	235	.040	.040	100.	60.	353.0	50.0	68.0	84.0	103.0	1.050	.950	SEP	* 8825.0	.316	.279	.970	1.282	.49-.04
N2H4	236	.040	.040	100.	60.	346.0	48.0	63.0	71.0	90.0	1.080	.920	SEP	* 8650.0	.137	.281	.985	1.189	.53-.04
N2H4	237	.040	.040	100.	60.	362.0	56.0	68.0	63.0	93.0	1.180	.840	SEP	* 9050.0	.157	.277	1.000	1.009	.49-.04
N2H4	238	.040	.040	100.	60.	361.0	50.0	68.0	66.0	93.0	1.050	.950	SEP	* 9025.0	.088	.277	.971	1.274	.49-.04
N2H4	239	.040	.040	100.	60.	145.0	22.0	27.0	72.0	82.0	1.190	1.000	MIX	* 3625.0	.178	.375	.999	1.044	.12-.03
N2H4	240	.040	.040	100.	60.	238.0	65.0	68.0	63.0	67.0	1.370	.760	UNDEF	* 5950.0	.022	.318	.963	.759	.49-.04

HYPERGOLIC STREAM IMPINGMENT DATA COMPILIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	COMMENTS *	PN	IS	P (IN)	EM	SPR	DV (SEC)
N2H4	241	.040	.040	100.	60.	230.0	74.0	68.0	115.0	117.0	1.560	.590	MIX	* 5750.0	.809	.322	.877	.597	.49-04
N2H4	242	.040	.040	100.	60.	176.0	55.0	69.0	146.0	127.0	1.140	1.110	MIX	* 4400.0	1.518	.352	.991	1.140	.48-04
N2H4	243	.040	.040	100.	60.	176.0	55.0	69.0	117.0	122.0	1.140	1.110	MIX	* 4400.0	1.103	.352	.994	1.113	.48-04
N2H4	244	.040	.040	100.	60.	338.0	39.0	65.0	44.0	54.0	.870	1.890	SEP	* 8450.0	.008	.283	.815	1.507	.51-04
N2H4	246	.040	.040	100.	60.	400.0	57.0	61.0	39.0	41.0	1.310	.820	SEP	* 10000.0	.003	.288	.972	.788	.55-04
N2H4	247	.040	.040	100.	60.	215.0	58.0	66.0	51.0	62.0	1.250	.920	MIX	* 5375.0	.016	.329	.993	.891	.51-04
N2H4	248	.040	.040	100.	60.	115.0	61.0	65.0	51.0	60.0	1.340	.790	MIX	* 2875.0	.013	.405	.970	.782	.51-04
N2H4	250	.040	.040	100.	60.	218.0	61.0	68.0	58.0	70.0	1.290	.890	MIX	* 5450.0	.028	.328	.988	.856	.49-04
N2H4	251	.040	.040	100.	60.	384.0	56.0	65.0	62.0	74.0	1.230	.940	SEP	* 9600.0	.040	.271	.997	.930	.51-04
N2H4	252	.040	.040	100.	60.	307.0	45.0	50.0	65.0	76.0	1.280	.870	UNDEF	* 7675.0	.061	.292	.988	.853	.67-04
N2H4	253	.040	.040	100.	60.	507.0	35.0	38.0	68.0	77.0	1.290	.850	SEP	* 12675.0	.087	.247	.980	.816	.88-04
N2H4	254	.040	.040	100.	60.	520.0	36.0	28.0	77.0	86.0	1.330	.590	SEP	* 13000.0	.230	.245	.695	.420	.12-04
N2H4	255	.040	.040	100.	60.	84.0	49.0	52.0	106.0	103.0	1.320	.820	MIX	* 2100.0	.414	.450	.974	.796	.64-04
N2H4	256	.040	.040	100.	60.	100.0	56.0	65.0	126.0	120.0	1.260	.920	MIX	* 2500.0	1.029	.425	.999	.961	.51-04
N2H4	257	.040	.040	100.	60.	122.0	56.0	66.0	124.0	120.0	1.200	.980	MIX	* 3050.0	1.014	.367	1.000	.964	.51-04
N2H4	258	.040	.040	100.	60.	423.0	32.0	37.0	93.0	104.0	1.220	.950	SEP	* 10575.0	.623	.263	.998	.934	.90-04
N2H4	259	.040	.040	100.	60.	461.0	32.0	37.0	94.0	99.0	1.220	.840	SEP	* 11525.0	.441	.255	.998	.937	.90-04
N2H4	260	.040	.040	100.	60.	461.0	32.0	37.0	74.0	84.0	1.240	.920	SEP	* 11525.0	.150	.255	.997	.928	.90-04
N2H4	261	.040	.040	100.	60.	115.0	31.0	37.0	79.0	88.0	1.180	1.010	MIX	* 2875.0	.201	.405	1.000	.891	.90-04
N2H4	262	.040	.040	100.	60.	130.0	34.0	41.0	92.0	97.0	1.180	1.040	MIX	* 3250.0	.346	.389	1.000	1.018	.81-04
N2H4	263	.040	.040	100.	60.	149.0	40.0	52.0	90.0	97.0	1.100	1.180	MIX	* 3725.0	.273	.372	.986	1.181	.64-04
N2H4	264	.040	.040	100.	60.	169.0	46.0	53.0	90.0	97.0	1.240	.930	MIX	* 4225.0	.267	.357	.997	.928	.63-04
N2H4	265	.040	.040	100.	60.	188.0	51.0	61.0	84.0	93.0	1.170	1.030	MIX	* 4700.0	.175	.344	1.000	.937	.55-04
N2H4	266	.040	.040	100.	60.	203.0	56.0	68.0	83.0	93.0	1.160	.860	MIX	* 5075.0	.155	.335	.998	1.057	.48-04
N2H4	267	.040	.040	100.	60.	388.0	58.0	68.0	81.0	88.0	1.220	.950	SEP	* 9700.0	.110	.270	.999	.958	.49-04
N2H4	268	.040	.040	100.	60.	381.0	58.0	57.0	78.0	86.0	1.440	.680	SEP	* 9525.0	.113	.272	.924	.672	.58-04
N2H4	269	.040	.040	100.	60.	353.0	55.0	52.0	77.0	86.0	1.490	.570	SEP	* 8825.0	.124	.279	.894	.621	.64-04
N2H4	270	.040	.040	100.	60.	176.0	49.0	71.0	138.0	129.0	.970	1.520	MIX	* 4400.0	1.673	.352	.920	1.508	.47-04
N2H4	272	.040	.040	100.	60.	179.0	56.0	63.0	106.0	111.0	1.280	.870	MIX	* 4475.0	.587	.350	.993	.891	.53-04
N2H4	273	.040	.040	100.	60.	176.0	58.0	63.0	120.0	124.0	1.310	.820	MIX	* 4400.0	1.374	.352	.944	.836	.53-04
N2H4	274	.040	.040	100.	60.	176.0	55.0	46.0	117.0	120.0	1.700	.490	MIX	* 4407.0	1.454	.352	.785	.495	.72-04
N2H4	275	.040	.040	100.	60.	138.0	42.0	51.0	114.0	117.0	1.170	1.030	MIX	* 3450.0	1.079	.351	.999	1.042	.65-04
N2H4	276	.040	.040	100.	60.	279.0	75.0	73.0	70.0	70.0	1.450	.680	UNDEF	* 6975.0	.026	.302	.918	.660	.46-04
N2H4	278	.040	.040	100.	60.	237.0	61.0	73.0	70.0	70.0	1.190	.980	MIX	* 5925.0	.026	.319	1.000	.937	.46-04
N2H4	279	.040	.040	100.	60.	250.0	66.0	73.0	70.0	70.0	1.290	1.160	MIX	* 6250.0	.025	.313	.987	.852	.46-04
N2H4	281	.040	.040	100.	60.	100.0	101.0	127.0	69.0	74.0	1.130	1.120	MIX	* 2500.0	.021	.425	.996	1.008	.26-04
N2H4	282	.040	.040	100.	60.	115.0	117.0	127.0	69.0	73.0	1.540	.870	MIX	* 2875.0	.019	.405	.980	.918	.26-04
N2H4	283	.040	.040	100.	60.	326.0	84.0	98.0	84.0	98.0	1.220	.940	SEP	* 8150.0	.155	.246	.998	.946	.34-04
N2H4	284	.040	.040	100.	60.	282.0	76.0	93.0	54.0	74.0	1.160	1.060	SEP	* 7050.0	.023	.301	1.000	1.026	.36-04
N2H4	285	.040	.040	100.	60.	228.0	63.0	78.0	54.0	86.0	1.160	1.060	MIX	* 5700.0	.083	.323	.999	1.044	.43-04

HYPERGOLIC STREAM IMPINGEMENT DATA COMPIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	UO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	COMMENTS	PO	IS	R (IN)	EM	SPR	DV (SEC)
N2H4	286	.040	.040	100.	60.	192.0	51.0	63.0	54.0	86.0	1.140	1.060	MIX	.102	.142	.999	1.040	.53-04
N2H4	287	.040	.040	100.	60.	182.0	44.0	51.0	55.0	86.0	1.250	.920	MIX	.126	.348	.996	.916	.65-04
N2H4	288	.040	.040	100.	60.	186.0	46.0	52.0	62.0	57.0	1.890	.600	MIX	.013	.345	.993	.885	.64-04
N2H4	289	.040	.040	100.	60.	174.0	49.0	53.0	62.0	55.0	1.320	.820	MIX	.011	.353	.979	.815	.63-04
N2H4	290	.040	.040	100.	60.	186.0	54.0	56.0	96.0	101.0	1.030	.780	MIX	.335	.345	.961	.754	.60-04
N2H4	291	.040	.040	100.	60.	162.0	44.0	54.0	106.0	71.0	1.130	1.090	MIX	.039	.362	.997	1.082	.62-04
N2H4	293	.060	.060	100.	60.	362.0	92.0	121.0	60.0	62.0	1.090	1.200	SEP	.013	.415	.984	1.149	.41-04
N2H4	294	.060	.060	100.	60.	180.0	46.0	54.0	60.0	64.0	1.240	.940	MIX	.033	.524	.999	.954	.93-04
N2H4	295	.060	.060	100.	60.	14.7	64.0	73.0	59.0	53.0	1.440	.780	POP	.010	1.206	.995	.905	.68-04
N2H4	296	.060	.060	100.	60.	68.0	49.0	61.0	57.0	55.0	1.230	1.010	POP	.014	.724	.997	1.076	.82-04
N2H4	297	.060	.060	100.	60.	76.0	46.0	53.0	52.0	41.0	1.200	.960	MIX	.003	.445	.997	1.075	.82-04
N2H4	298	.040	.040	100.	60.	106.0	36.0	53.0	53.0	40.0	.930	1.590	MIX	.003	.416	.919	1.509	.63-04
N2H4	299	.040	.040	100.	60.	122.0	46.0	37.0	53.0	41.0	1.110	1.050	MIX	.005	.397	.733	.450	.90-04
N2H4	300	.040	.040	100.	60.	130.0	47.0	57.0	53.0	41.0	1.160	1.050	MIX	.005	.425	1.000	1.024	.58-04
N2H4	301	.040	.040	100.	60.	190.0	36.0	57.0	55.0	43.0	1.070	1.740	MIX	.004	.342	.858	1.746	.59-04
N2H4	302	.040	.040	100.	60.	192.0	36.0	57.0	55.0	43.0	1.040	1.410	POP	.003	.704	.935	1.443	.79-04
N2H4	304	.040	.040	100.	60.	14.7	29.0	42.0	37.0	37.0	1.000	1.270	MIX	.012	.814	.999	1.045	.21-03
N2H4	305	.040	.040	100.	60.	14.7	13.0	16.0	43.0	42.0	1.260	.800	POP	.014	.814	.932	.687	.21-03
N2H4	306	.040	.040	100.	60.	14.7	16.0	16.0	39.0	44.0	1.580	.590	MIX	.002	.04	.886	.610	.61-04
N2H4	307	.040	.040	100.	60.	14.7	52.0	49.0	32.0	32.0	1.580	.800	POP	.015	.304	.994	.897	.51-04
N2H4	308	.040	.040	100.	60.	14.7	49.0	55.0	58.0	59.0	1.300	.860	POP	.012	.804	.994	.897	.51-04
N2H4	309	.040	.040	100.	60.	14.7	58.0	66.0	58.0	59.0	1.250	.860	POP	.052	.804	.999	.953	.49-04
N2H4	310	.040	.040	100.	60.	14.7	58.0	68.0	70.0	78.0	1.250	1.170	MIX	.021	.804	.999	.951	.49-04
N2H4	311	.040	.040	100.	60.	14.7	58.0	68.0	60.0	66.0	1.250	1.170	MIX	.021	.804	.999	.951	.49-04
N2H4	312	.040	.040	100.	60.	14.7	64.0	68.0	66.0	61.0	1.250	.800	MIX	.014	.904	.972	.787	.49-04
N2H4	313	.040	.040	100.	60.	14.7	64.0	68.0	53.0	55.0	1.250	.800	MIX	.009	.804	.970	.780	.49-04
N2H4	314	.040	.040	100.	60.	14.7	63.0	68.0	53.0	55.0	1.290	.800	MIX	.009	.804	.977	.805	.49-04
N2H4	316	.040	.040	100.	60.	14.7	63.0	63.0	48.0	51.0	1.440	.700	MIX	.007	.804	.934	.630	.53-04
N2H4	317	.040	.040	100.	60.	14.7	54.0	55.0	45.0	51.0	1.450	.710	POP	.008	.804	.945	.714	.61-04
N2H4	318	.040	.040	100.	60.	14.7	23.0	28.0	48.0	51.0	1.150	1.030	SEP	.015	.804	1.000	1.022	.12-03
N2H4	319	.040	.040	100.	60.	14.7	27.0	28.0	45.0	51.0	1.360	.770	SEP	.015	.804	.956	.740	.12-03
N2H4	320	.040	.040	100.	60.	14.7	18.0	19.0	55.0	55.0	1.400	.760	POP	.031	.804	.967	.772	.18-03
N2H4	321	.040	.040	100.	60.	14.7	16.0	19.0	57.0	59.0	1.400	.760	POP	.042	.804	.967	.771	.18-03
N2H4	322	.040	.040	100.	60.	14.7	18.0	19.0	58.0	61.0	1.290	.820	MIX	.050	.804	.934	.691	.18-03
N2H4	323	.040	.040	100.	60.	14.7	19.0	19.0	50.0	51.0	1.460	.700	POP	.023	.804	.934	.691	.19-03
N2H4	324	.040	.040	100.	60.	14.7	18.0	18.0	50.0	51.0	1.570	.620	SEP	.034	.804	.961	.733	.14-03
N2H4	325	.040	.040	100.	60.	14.7	23.0	24.0	63.0	72.0	1.370	.780	SEP	.026	1.206	.982	.828	.21-03
N2H4	326	.060	.060	100.	60.	14.7	22.0	24.0	58.0	51.0	1.300	.850	POP	.021	1.206	.983	.829	.22-03
N2H4	327	.060	.060	100.	60.	14.7	21.0	23.0	48.0	48.0	1.300	.840	POP	.036	1.206	.980	.819	.21-03
N2H4	328	.060	.060	100.	60.	14.7	22.0	24.0	47.0	55.0	1.310	.830	POP	.036	1.206	.980	.819	.21-03

HYPERGOLIC STREAM IMPINGMENT DATA COMPIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	COMMENTS	P1	IS	P (IN)	EM	SPR	OV (SEC)
N2H4	329	.060	.060	100.	60.	14.7	21.0	24.0	47.0	53.0	1.270	.900	POP	* 245.0	.031	1.206	.994	.899	.21-03
N2H4	330	.060	.060	100.	60.	14.7	21.0	24.0	48.0	53.0	1.270	.900	POP	* 245.0	.021	1.206	.994	.900	.21-03
N2H4	331	.060	.060	100.	60.	14.7	27.0	31.0	64.0	61.0	1.240	.920	POP	* 1250.0	.046	1.206	.996	.917	.16-03
N2H4	332	.060	.060	100.	60.	75.0	34.0	43.0	63.0	64.0	1.110	1.160	POP	* 1333.3	.036	.686	.997	1.110	.12-03
N2H4	333	.060	.060	100.	60.	80.0	40.0	50.0	63.0	64.0	1.130	1.120	POP	* 1333.3	.036	.686	.997	1.085	.10-03
N2H4	334	.060	.060	100.	60.	95.0	34.0	57.0	61.0	63.0	.820	2.080	POP	* 1583.3	.029	.648	.804	1.949	.16-03
N2H4	335	.060	.060	100.	60.	96.0	26.0	31.0	64.0	61.0	1.230	1.570	POP	* 1600.0	.046	.646	1.000	.989	.16-03
N2H4	336	.060	.060	100.	60.	70.0	34.0	44.0	63.0	64.0	1.110	1.150	POP	* 1166.7	.041	.717	.989	1.163	.11-03
N2H4	337	.060	.060	100.	60.	80.0	36.0	57.0	61.0	66.0	.890	1.780	POP	* 1333.3	.037	.646	.861	1.736	.16-03
N2H4	338	.060	.060	100.	60.	85.0	43.0	50.0	86.0	89.0	1.240	.940	POP	* 1416.7	.240	.672	.998	.946	.10-03
N2H4	339	.060	.060	100.	60.	125.0	45.0	58.0	108.0	97.0	1.110	1.160	POP	* 2083.3	.366	.591	.986	1.180	.16-04
N2H4	340	.060	.060	100.	60.	75.0	36.0	50.0	108.0	117.0	1.050	1.310	POP	* 1250.0	1.651	.701	.955	1.356	.10-03
N2H4	341	.060	.060	100.	60.	75.0	41.0	50.0	100.0	107.0	1.160	1.060	POP	* 1250.0	.848	.701	.999	1.044	.10-03
N2H4	342	.060	.060	100.	60.	145.0	42.0	50.0	99.0	107.0	1.210	.990	POP	* 2416.7	.848	.563	1.000	.994	.10-03
N2H4	343	.060	.060	100.	60.	145.0	41.0	50.0	103.0	118.0	1.160	1.060	POP	* 2416.7	1.762	.563	.999	1.041	.10-03
N2H4	344	.060	.060	100.	60.	145.0	43.0	50.0	80.0	81.0	1.240	.810	POP	* 2416.7	.134	.563	.998	.944	.10-03
N2H4	345	.060	.060	100.	60.	165.0	43.0	49.0	78.0	81.0	1.260	.900	POP	* 2750.0	.136	.519	.995	.905	.10-03
N2H4	346	.060	.060	100.	60.	165.0	45.0	50.0	76.0	76.0	1.280	.880	POP	* 2750.0	.092	.539	.989	.862	.10-03
N2H4	347	.060	.060	100.	60.	165.0	44.0	49.0	78.0	80.0	1.250	.920	POP	* 2750.0	.127	.539	.990	.865	.10-03
N2H4	348	.060	.060	100.	60.	165.0	42.0	50.0	83.0	93.0	.910	1.720	POP	* 2750.0	.155	.539	.868	1.708	.10-03
N2H4	349	.060	.060	100.	60.	165.0	33.0	51.0	90.0	91.0	.930	1.640	POP	* 2750.0	.272	.539	.877	1.675	.10-03
N2H4	350	.060	.060	100.	60.	185.0	40.0	51.0	94.0	91.0	1.120	1.130	POP	* 3083.3	.272	.519	.991	1.144	.10-03
N2H4	351	.060	.060	100.	60.	185.0	47.0	50.0	70.0	72.0	1.350	7.820	POP	* 3083.3	.064	.519	.972	.787	.10-03
N2H4	352	.060	.060	100.	60.	165.0	41.0	49.0	68.0	72.0	1.210	.980	UNDEF	* 2750.0	.069	.519	1.000	.992	.10-03
N2H4	353	.060	.060	100.	60.	165.0	42.0	49.0	66.0	70.0	1.230	.938	POP	* 2750.0	.059	.519	.998	.944	.10-03
N2H4	354	.060	.060	100.	60.	165.0	42.0	51.0	50.0	49.0	1.190	1.020	POP	* 2750.0	.010	.519	1.000	1.020	.10-03
N2H4	355	.060	.060	100.	60.	175.0	44.0	51.0	47.0	47.0	1.230	.930	UNDEF	* 2916.7	.009	.529	.997	.928	.10-03
N2H4	356	.060	.060	100.	60.	165.0	36.0	51.0	50.0	49.0	1.030	1.350	POP	* 2750.0	.010	.519	.948	1.388	.10-03
N2H4	357	.060	.060	100.	60.	165.0	41.0	49.0	50.0	51.0	1.230	.970	POP	* 2750.0	.013	.539	1.000	.987	.10-03
N2H4	358	.060	.060	100.	60.	165.0	42.0	51.0	50.0	51.0	1.190	1.020	POP	* 2750.0	.012	.519	1.000	1.019	.10-03
N2H4	359	.060	.060	100.	60.	165.0	42.0	51.0	48.0	47.0	1.190	1.020	POP	* 2750.0	.003	.519	1.000	1.019	.10-03
N2H4	360	.060	.060	100.	60.	165.0	33.0	60.0	57.0	53.0	.079	1.020	POP	* 2750.0	.012	.519	.715	2.295	.10-03
N2H4	361	.060	.060	100.	60.	165.0	42.0	51.0	55.0	55.0	1.020	1.190	POP	* 2750.0	.017	.519	1.000	1.021	.10-03
N2H4	362	.060	.060	100.	60.	105.0	40.0	52.0	55.0	53.0	1.120	1.140	POP	* 1750.0	.014	.627	.988	1.171	.10-03
N2H4	363	.060	.060	100.	60.	105.0	44.0	51.0	53.0	53.0	1.230	.930	POP	* 1750.0	.015	.627	.997	.930	.10-03
N2H4	364	.060	.060	100.	60.	105.0	44.0	52.0	60.0	63.0	1.230	.950	POP	* 1750.0	.032	.627	.999	.968	.10-03
N2H4	365	.060	.060	100.	60.	105.0	43.0	51.0	60.0	63.0	1.220	.960	POP	* 1750.0	.033	.627	1.000	.974	.10-03
N2H4	366	.060	.060	100.	60.	105.0	45.0	52.0	60.0	63.0	1.250	.920	POP	* 1750.0	.032	.627	.997	.925	.10-03
N2H4	367	.060	.060	100.	60.	105.0	45.0	52.0	76.0	80.0	1.250	.930	POP	* 1750.0	.110	.627	.997	.930	.10-03
N2H4	368	.060	.060	100.	60.	185.0	45.0	49.0	68.0	70.0	1.290	.850	POP	* 3083.3	.059	.519	.981	.824	.10-03

HYPERGOLIC STREAM IMPINGMENT DATA COMPIATION

INVESTIGATOR														ZUNG														A L R C MODEL CORRELATION PARAMETERS													
FUEL TYPE	TEST NO.	DO (IN)	DF (IN)	L/O	IM ² ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	COMMENTS	PN	IS	R (IN)	EM	SPR	DV (SEC)																						
N2H4	309	.060	.060	100.	60.	185.0	43.0	50.0	70.0	69.0	1.240	.930	POP	* 3083.3	.054	.519	.928	.942	.10-03																						
N2H4	370	.060	.060	100.	60.	185.0	41.0	49.0	65.0	64.0	1.210	.980	POP	* 3083.3	.037	.519	1.000	.993	.10-03																						
N2H4	371	.060	.060	100.	60.	185.0	37.0	49.0	63.0	63.0	1.060	1.230	POP	* 3083.3	.034	.519	.981	1.218	.10-03																						
N2H4	372	.060	.060	100.	60.	185.0	34.0	49.0	55.0	55.0	.980	1.450	POP	* 3083.3	.014	.519	.936	1.438	.10-03																						
N2H4	373	.055	.055	100.	60.	95.0	52.0	60.0	63.0	64.0	1.230	1.030	POP	* 1727.3	.028	.594	.997	.924	.76-04																						
N2H4	374	.055	.055	100.	60.	95.0	51.0	60.0	63.0	64.0	1.190	.990	POP	* 1727.3	.028	.594	.999	.961	.76-04																						
N2H4	375	.055	.055	100.	60.	95.0	54.0	60.0	63.0	64.0	1.270	.790	POP	* 1727.3	.028	.594	.988	.857	.76-04																						
N2H4	376	.055	.055	100.	60.	165.0	54.0	60.0	63.0	64.0	1.290	.850	POP	* 3000.0	.028	.494	.988	.857	.76-04																						
N2H4	377	.055	.055	100.	60.	175.0	54.0	60.0	65.0	82.0	1.270	.880	POP	* 3181.8	.110	.485	.987	.851	.76-04																						
N2H4	378	.055	.055	100.	60.	175.0	56.0	61.0	68.0	89.0	1.310	.830	POP	* 3181.8	.181	.485	.990	.817	.75-04																						
N2H4	379	.055	.055	100.	60.	175.0	55.0	61.0	104.0	82.0	1.290	.860	POP	* 3181.8	.109	.485	.991	.877	.75-04																						
N2H4	380	.055	.055	100.	60.	175.0	53.0	61.0	114.0	105.0	1.240	.930	POP	* 3181.8	.556	.485	.998	.942	.75-04																						
N2H4	381	.055	.055	100.	60.	175.0	54.0	61.0	114.0	107.0	1.270	.880	POP	* 3181.8	.637	.485	.995	.907	.75-04																						
N2H4	382	.055	.055	100.	60.	175.0	52.0	61.0	111.0	110.0	1.210	1.080	POP	* 3181.8	.780	.485	1.000	.974	.75-04																						
N2H4	383	.055	.055	100.	60.	175.0	56.0	60.0	65.0	66.0	1.320	.810	POP	* 3181.8	.032	.485	.975	.797	.76-04																						
N2H4	384	.055	.055	100.	60.	175.0	55.0	60.0	88.0	78.0	1.290	.990	POP	* 3181.8	.082	.485	.945	.838	.76-04																						
N2H4	385	.055	.055	100.	60.	175.0	53.0	60.0	76.0	78.0	1.260	.910	POP	* 3181.8	.082	.485	.994	.893	.76-04																						
N2H4	388	.055	.055	100.	60.	175.0	42.0	46.0	48.0	51.0	1.290	.848	POP	* 3181.8	.012	.485	.982	.827	.10-03																						
N2H4	389	.055	.055	100.	60.	175.0	37.0	46.0	48.0	51.0	1.130	1.100	POP	* 3181.8	.012	.485	.998	1.065	.10-03																						
N2H4	390	.055	.055	100.	60.	175.0	35.0	46.0	48.0	51.0	1.710	.484	POP	* 3181.8	.012	.485	.985	1.191	.10-03																						
N2H4	391	.055	.055	100.	60.	195.0	49.0	60.0	48.0	52.0	1.170	1.040	POP	* 3545.5	.010	.467	.999	1.034	.76-04																						
N2H4	392	.055	.055	100.	60.	195.0	49.0	60.0	48.0	52.0	1.160	1.060	POP	* 3545.5	.010	.467	.999	1.034	.76-04																						
N2H4	393	.055	.055	100.	60.	415.0	37.0	59.0	55.0	60.0	.881	1.840	SEP	* 7545.5	.020	.364	.855	1.756	.78-04																						
N2H4	394	.060	.060	100.	60.	235.0	61.0	71.0	56.0	62.0	1.210	.967	MIX	* 3916.7	.022	.479	.998	.936	.70-04																						
N2H4	395	.060	.060	100.	60.	225.0	51.0	71.0	45.0	53.0	1.020	1.370	MIX	* 3750.0	.010	.486	.960	1.332	.70-04																						
N2H4	396	.060	.060	100.	60.	235.0	60.0	71.0	38.0	45.0	1.200	.991	MIX	* 3916.7	.005	.479	.999	.961	.78-04																						
N2H4	397	.060	.060	100.	60.	225.0	59.0	71.0	63.0	62.0	1.190	1.520	MIX	* 3750.0	.022	.486	1.000	1.006	.78-04																						
N2H4	398	.060	.060	100.	60.	235.0	56.0	71.0	71.0	70.0	1.350	.931	MIX	* 3916.7	.041	.479	.994	1.120	.70-04																						
N2H4	399	.060	.060	100.	60.	235.0	54.0	71.0	79.0	76.0	1.090	1.190	MIX	* 3916.7	.065	.479	.982	1.210	.70-04																						
N2H4	400	.060	.060	100.	60.	235.0	53.0	71.0	78.0	84.0	1.060	1.240	MIX	* 3916.7	.118	.479	.976	1.250	.70-04																						
N2H4	401	.060	.060	100.	60.	235.0	51.0	71.0	83.0	87.0	1.020	1.360	MIX	* 3916.7	.146	.479	.955	1.353	.70-04																						
N2H4	402	.060	.060	100.	60.	235.0	51.0	69.0	83.0	99.0	1.600	1.320	MIX	* 3916.7	.354	.479	.972	1.270	.72-04																						
N2H4	403	.060	.060	100.	60.	255.0	50.0	61.0	98.0	103.0	1.160	1.050	MIX	* 4250.0	.529	.466	.999	1.045	.82-04																						
N2H4	404	.060	.060	100.	60.	215.0	49.0	50.0	97.0	108.0	1.160	.882	MIX	* 3583.3	.907	.494	.951	.728	.10-03																						
N2H4	405	.060	.060	100.	60.	235.0	48.0	69.0	99.0	48.0	.993	1.440	MIX	* 3916.7	.007	.479	.924	1.492	.72-04																						
N2H4	406	.060	.060	100.	60.	235.0	56.0	66.0	63.0	66.0	1.020	.980	MIX	* 3916.7	.032	.479	.999	.963	.76-04																						
N2H4	407	.060	.060	100.	60.	235.0	52.0	65.0	63.0	66.0	1.140	1.090	MIX	* 3916.7	.033	.479	.997	1.084	.77-04																						
N2H4	408	.060	.060	100.	60.	235.0	54.0	65.0	63.0	66.0	1.110	1.080	MIX	* 3916.7	.033	.479	1.000	1.005	.77-04																						
N2H4	409	.060	.060	100.	60.	235.0	50.0	65.0	63.0	66.0	1.090	1.210	MIX	* 3916.7	.033	.479	.988	1.172	.77-04																						
N2H4	410	.060	.060	100.	60.	185.0	36.0	54.0	76.0	85.0	.946	1.620	POP	* 3083.3	.166	.519	.906	1.563	.93-04																						

HYPERGOLIC STREAM IMPINGMENT DATA COMPILIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	L/D	PC ANGLE (DEG)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR MF/MO	COMMENTS	PD	IS (IN)	R (IN)	EM	SPR	RV (SEC)
N2H4	411	.060	60	175.0	34.0	55.0	89.0	95.0	POP	2916.7	.336	.529	.836	1.829	.91-04
N2H4	412	.060	60	190.0	36.0	55.0	87.0	97.0	POP	3166.7	.386	.514	.889	1.627	.91-04
N2H4	413	.060	60	90.0	40.0	55.0	83.0	49.0	POP	1500.0	.218	.660	.963	1.319	.91-04
N2H4	414	.060	60	80.0	39.0	54.0	94.0	105.0	POP	1333.3	.685	.686	.958	1.339	.93-04
N2H4	415	.060	60	85.0	44.0	54.0	98.0	108.0	POP	1416.7	.840	.672	.999	1.054	.93-04
N2H4	416	.060	60	80.0	43.0	54.0	101.0	110.0	POP	1333.3	.961	.686	.995	1.106	.93-04
N2H4	417	.060	60	80.0	41.0	55.0	109.0	116.0	POP	1333.3	1.405	.686	.972	1.267	.91-04
N2H4	418	.060	60	80.0	39.0	55.0	102.0	116.0	POP	1333.3	1.405	.686	.947	1.392	.91-04
N2H4	419	.060	60	80.0	37.2	54.3	109.0	116.0	POP	1750.0	1.424	.686	.924	1.491	.92-04
N2H4	423	.060	60	105.0	51.7	54.3	109.0	112.0	POP	1583.3	1.248	.627	.969	1.778	.92-04
N2H4	424	.060	60	95.0	50.3	54.3	103.0	114.0	POP	1583.3	.225	.648	.980	.817	.92-04
N2H4	425	.060	60	95.0	36.2	53.5	75.0	81.5	POP	1583.3	.130	.648	.920	1.505	.93-04
N2H4	426	.060	60	95.0	34.9	54.7	79.0	81.9	POP	1583.3	.131	.648	.866	1.714	.91-04
N2H4	427	.060	60	95.0	39.7	54.7	104.0	108.0	POP	1583.3	.829	.648	.959	1.336	.91-04
N2H4	428	.060	60	95.0	36.3	54.7	106.0	114.0	POP	1581.3	1.239	.648	.897	1.596	.91-04
N2H4	429	.060	60	185.0	48.7	58.2	68.0	75.0	POP	3363.6	.067	.476	1.000	.990	.79-04
N2H4	430	.055	60	175.0	46.7	58.5	81.7	93.0	POP	3181.8	.251	.465	.996	1.090	.78-04
N2H4	431	.055	60	215.0	65.4	74.0	92.5	101.0	MIX	3909.1	.348	.453	.994	.895	.62-04
N2H4	432	.055	60	215.0	64.4	73.5	97.5	106.0	MIX	3909.1	.494	.453	.996	.912	.62-04
N2H4	433	.055	60	175.0	51.0	64.5	101.0	114.0	MIX	3181.8	.963	.485	.994	1.119	.71-04
N2H4	434	.055	60	75.0	50.1	60.4	96.0	106.0	POP	1363.6	.601	.643	1.000	1.017	.76-04
N2H4	435	.055	60	95.0	45.2	59.8	72.5	86.5	POP	1727.3	.154	.594	.982	1.211	.77-04
N2H4	436	.055	60	95.0	45.0	59.8	68.0	77.5	POP	1727.3	.079	.594	.982	1.212	.77-04
N2H4	437	.055	60	45.0	54.1	50.8	99.0	105.0	MIX	1125.0	.486	.554	.892	.619	.66-04
N2H4	438	.040	60	45.0	49.5	53.4	94.0	112.0	MIX	1125.0	.741	.554	.978	.810	.62-04
N2H4	439	.040	60	45.0	57.4	59.7	97.5	102.0	MIX	1125.0	.336	.554	.963	.759	.56-04
N2H4	440	.040	60	45.0	48.8	53.4	91.0	97.0	MIX	1125.0	.265	.554	.984	.838	.62-04
N2H4	441	.040	60	45.0	56.5	53.4	107.0	116.0	MIX	1125.0	.335	.554	.898	.628	.62-04
N2H4	442	.040	60	45.0	54.1	53.4	114.0	125.0	SEP	1125.0	1.727	.554	.932	.686	.62-04
N2H4	443	.040	60	45.0	60.0	53.4	108.0	108.0	MIX	1125.0	.566	.554	.847	.580	.62-04
N2H4	444	.040	60	45.0	50.0	57.4	114.0	122.0	MIX	1125.0	1.326	.554	.997	.929	.58-04
N2H4	445	.040	60	45.0	49.3	57.4	116.0	128.0	SEP	1125.0	1.944	.554	.999	.755	.58-04
N2H4	446	.040	60	45.0	50.6	66.6	108.0	122.0	MIX	1125.0	1.143	.554	.981	1.215	.50-04
N2H4	447	.040	60	55.0	54.1	66.6	104.0	114.0	MIX	1375.0	.678	.518	.998	1.063	.50-04
N2H4	448	.040	60	55.0	57.2	66.7	106.0	116.0	MIX	1375.0	.773	.518	.999	.955	.50-04
N2H4	449	.040	60	55.0	54.2	66.6	104.0	118.0	MIX	1375.0	.882	.518	.998	1.037	.50-04
N2H4	450	.040	60	45.0	44.7	68.2	101.0	106.0	MIX	1125.0	.387	.554	.905	1.565	.49-04
N2H4	451	.040	60	45.0	45.4	65.1	97.5	108.0	MIX	1125.0	.465	.554	.921	1.504	.51-04
N2H4	452	.040	60	50.0	44.9	65.1	91.0	105.0	MIX	1250.0	.379	.535	.930	1.465	.51-04

HYPERGOLIC STREAM IMPINGMENT DATA COMPIATION

A L R C MODFL CORRELATION PARAMETERS

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	COMMENTS	PN	IS	R (IN)	EM	SPR	OV (SEC)
N2H4	454	.040	.040	100.	60.	65.0	49.8	40.9	114.0	132.0	1.000	1.430	MIX	* 1625.0	3.504	.490	.761	.473	.81-04
N2H4	455	.040	.040	100.	60.	50.0	67.7	68.2	122.0	134.0	1.410	.712	SEP	* 1250.0	2.379	.535	.946	.716	.49-04
N2H4	456	.040	.040	100.	60.	55.0	58.5	66.6	129.0	137.0	1.250	.913	SEP	* 1375.0	2.929	.518	.996	.919	.50-04
N2H4	457	.040	.040	100.	60.	55.0	60.5	66.3	135.0	144.0	1.290	.852	SEP	* 1375.0	4.493	.518	.988	.853	.50-04
N2H4	458	.040	.040	100.	60.	55.0	60.0	64.1	131.0	144.0	1.330	.801	SEP	* 1375.0	4.647	.518	.978	.808	.52-04
N2H4	459	.040	.040	100.	60.	55.0	58.5	62.3	137.0	147.0	1.340	.797	SEP	* 1375.0	5.715	.518	.977	.806	.54-04
N2H4	460	.060	.060	100.	60.	470.0	37.9	66.3	61.5	64.0	.814	2.150	SEP	* 7833.3	.027	.380	.728	2.122	.75-04
N2H4	461	.060	.060	100.	60.	535.0	65.1	66.9	62.5	66.0	1.340	.789	SEP	* 891.7	.032	.364	.945	.732	.75-04
N2H4	462	.060	.060	100.	60.	535.0	48.4	68.8	62.5	66.0	1.000	1.420	SEP	* 891.7	.031	.364	.945	1.401	.73-04
N2H4	463	.060	.060	100.	60.	535.0	46.6	68.8	67.0	64.0	.960	1.540	SEP	* 8916.7	.026	.364	.917	1.518	.73-04
N2H4	464	.060	.060	100.	60.	585.0	56.5	68.8	66.0	69.5	1.170	1.040	SEP	* 9750.0	.040	.354	1.000	1.029	.73-04
N2H4	465	.060	.060	100.	60.	475.0	61.3	68.8	67.0	64.0	1.260	.890	SEP	* 7916.7	.026	.379	.991	.877	.73-04
N2H4	466	.060	.060	100.	60.	435.0	54.9	65.7	46.5	52.5	1.190	1.000	SEP	* 7250.0	.011	.390	1.000	.985	.76-04
N2H4	467	.060	.060	100.	60.	455.0	55.9	65.7	48.0	52.5	1.210	.970	SEP	* 7583.3	.011	.385	.999	.912	.76-04
N2H4	468	.060	.060	100.	60.	455.0	61.9	65.7	108.0	110.0	1.350	.790	SEP	* 7583.3	.790	.385	.974	.795	.76-04
N2H4	469	.060	.060	100.	60.	455.0	60.4	68.8	114.0	116.0	1.250	.910	SEP	* 7583.3	1.124	.385	.996	.914	.73-04
N2H4	470	.060	.060	100.	60.	445.0	58.2	68.8	114.0	119.0	1.210	.980	SEP	* 7416.7	1.367	.387	1.000	.984	.73-04
N2H4	471	.060	.060	100.	60.	100.0	46.6	52.8	96.0	92.0	1.260	.900	POP	* 1666.7	.282	.637	.995	.904	.95-04
N2H4	472	.060	.060	100.	60.	100.0	42.6	52.6	64.5	72.0	1.150	1.070	POP	* 1666.7	.064	.637	.999	1.055	.95-04
N2H4	473	.060	.060	100.	60.	100.0	46.6	52.6	63.0	70.0	1.260	.900	POP	* 1666.7	.055	.637	.992	.882	.95-04
N2H4	474	.060	.060	100.	60.	100.0	42.8	52.6	59.5	62.0	1.160	1.060	POP	* 1666.7	.029	.637	.999	1.046	.95-04
N2H4	475	.040	.040	100.	60.	630.0	59.5	77.9	68.0	64.0	1.090	1.200	SEP	* 15750.0	.015	.230	.984	1.195	.43-04
N2H4	476	.040	.040	100.	60.	80.0	73.5	81.7	68.0	64.0	1.280	.870	MIX	* 2000.0	.015	.457	.989	.861	.41-04
N2H4	477	.040	.040	100.	60.	85.0	61.3	83.2	61.0	66.0	1.050	1.290	MIX	* 2125.0	.017	.448	.971	1.275	.40-04
N2H4	478	.040	.040	100.	60.	640.0	65.4	70.0	59.5	66.0	1.330	.807	MIX	* 16000.0	.020	.229	.973	.792	.48-04
N2H4	479	.060	.060	100.	60.	100.0	46.5	52.8	88.0	71.5	1.250	.905	POP	* 1666.7	.062	.637	.994	.896	.95-04
N2H4	480	.060	.060	100.	60.	100.0	48.4	53.7	98.0	64.0	1.280	.860	POP	* 1666.7	.034	.637	.991	.873	.93-04
N2H4	481	.060	.060	100.	60.	23.0	26.5	35.1	128.0	132.0	1.060	1.240	POP	* 383.3	6.125	1.039	1.246	1.246	.14-03
N2H4	482	.050	.060	100.	60.	147.0	25.4	34.9	135.0	141.0	1.030	1.330	POP	* 2450.0	0.692	.540	.957	1.344	.14-03
N2H4	483	.060	.060	100.	60.	175.0	43.1	65.1	51.5	60.5	.946	1.600	POP	* 2916.7	.021	.529	.904	1.571	.77-04
N2H4	484	.060	.060	100.	60.	185.0	54.1	64.1	51.5	60.5	1.200	.986	POP	* 3083.3	.021	.519	.999	.967	.78-04
N2H4	485	.040	.040	100.	60.	220.0	54.1	62.6	57.5	60.5	1.230	.938	POP	* 5500.0	.015	.327	.997	.927	.53-04
N2H4	486	.040	.040	100.	60.	90.0	54.1	64.1	54.5	58.5	2.510	.471	MIX	* 2250.0	.012	.440	.970	.970	.52-04
N2H4	487	.040	.040	100.	60.	100.0	54.1	63.1	56.0	58.5	1.220	.952	MIX	* 2500.0	.012	.425	.998	.941	.53-04
N2H4	488	.040	.040	100.	60.	100.0	54.1	63.8	56.0	58.5	1.210	.970	MIX	* 2500.0	.012	.425	.999	.962	.52-04
N2H4	489	.040	.040	100.	60.	100.0	54.1	63.8	56.0	58.5	1.210	.970	MIX	* 2500.0	.012	.425	.999	.962	.52-04
N2H4	490	.040	.040	100.	60.	15.0	39.8	43.4	62.5	62.5	1.310	.830	MIX	* 3750.0	.025	.371	.982	.826	.77-04
N2H4	491	.040	.040	100.	60.	30.0	38.0	43.4	51.5	56.5	1.250	.910	MIX	* 750.0	.015	.634	.994	.900	.77-04
N2H4	492	.040	.040	100.	60.	30.0	38.6	41.4	64.5	60.0	1.370	.750	MIX	* 750.0	.021	.634	.976	.801	.81-04
N2H4	493	.040	.040	100.	60.	30.0	33.2	40.8	56.0	60.0	1.160	1.060	MIX	* 750.0	.021	.634	.999	1.044	.82-04

HYPERGOLIC STREAM IMPINGEMENT DATA COMPIATION

A L R C MODEL CORRELATION PARAMETERS

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	COMMENTS	PH	IS	P (IN)	EM	SPR	OV (SEC)
N2H4	494	.040	.040	100.	60.	30.0	35.9	40.8	56.0	62.0	1.160	1.040	MIX	* 750.0	.025	.634	1.000	1.000	.82-04
N2H4	495	.040	.040	100.	60.	35.0	38.3	43.4	56.0	64.5	1.260	.900	MIX	* 875.0	.029	.602	.993	.886	.77-04
N2H4	496	.040	.040	100.	60.	35.0	36.6	43.4	56.0	66.0	1.210	.980	MIX	* 875.0	.033	.602	1.000	.969	.77-04
N2H4	497	.040	.040	100.	60.	30.0	39.8	44.8	56.0	66.0	1.270	.890	MIX	* 750.0	.032	.634	.991	.873	.74-04
N2H4	498	.040	.040	100.	60.	30.0	39.8	44.8	58.0	66.0	1.270	.890	MIX	* 750.0	.032	.634	.991	.875	.74-04
N2H4	499	.040	.040	100.	60.	30.0	39.6	43.4	56.0	64.5	1.310	.830	MIX	* 750.0	.029	.634	.982	.829	.77-04
N2H4	500	.040	.040	100.	60.	30.0	39.6	43.4	56.0	66.0	1.310	.830	MIX	* 750.0	.033	.634	.982	.829	.77-04
N2H4	501	.040	.040	100.	60.	30.0	30.6	38.3	62.5	74.0	1.140	1.090	MIX	* 750.0	.063	.634	.997	1.082	.87-04
N2H4	502	.040	.040	100.	60.	30.0	37.3	37.3	64.5	69.0	1.270	.880	MIX	* 750.0	.048	.634	.991	.875	.89-04
N2H4	503	.040	.040	100.	60.	30.0	33.2	38.3	66.0	74.0	1.230	.930	MIX	* 750.0	.063	.634	.997	.922	.87-04
N2H4	504	.040	.040	100.	60.	30.0	33.7	31.2	64.5	74.0	1.540	.598	MIX	* 750.0	.084	.634	.874	.593	.11-03
N2H4	505	.040	.040	100.	60.	14.7	20.7	17.2	72.0	92.0	1.710	.488	SEP	* 367.5	.578	.804	.764	.476	.19-03
N2H4	506	.040	.040	100.	60.	14.7	21.4	21.7	72.0	72.0	1.400	.721	SEP	* 367.5	.104	.804	.946	.716	.15-03
N2H4	507	.027	.027	100.	60.	14.7	41.4	27.4	71.0	71.0	2.200	.296	SEP	* 544.4	.051	.543	.513	.305	.82-04
N2H4	508	.027	.027	100.	60.	14.7	40.3	39.2	68.0	70.0	1.480	.650	MIX	* 544.4	.033	.543	.917	.658	.57-04
N2H4	509	.027	.027	100.	60.	14.7	47.1	53.2	68.0	68.0	1.270	.890	MIX	* 544.4	.021	.543	.993	.488	.42-04

LIST OF APPENDIX D
DATA SOURCES

1. Investigator - Nurich, W. H. and Cosdill, J. P., "Reactive Stream Separation Photography", Final Report R8490, Contract NAS 7-720, Rocketdyne, 1971
2. Investigator - Houseman
Lee, A., and Houseman, J., "Popping Phenomena with N_2O_4/N_2H_4 Injectors" presented at the Western States Section Meeting of the Combustion Institute on Stable Combustion of Liquid Propellant, JPL, Oct. 26-27, 1970
3. Investigator-Zung
Zung, L. B., and White, S. R., "Combustion Process of Impinging Hypergolic Propellants", NASA CR 1704, Marshall Industries, Irvine, California, May 1971

PROPELLANT STREAM HEATING
MODEL

QIR FOR,* MAIN,MAIN
DATE, TIME, LEVEL OF OUTPUT ELEMENT: 11 DEC 73 11:14(03)
FORTRAN V: ISD VERSION 2.9

MAIN PROGRAM

STORAGE USED (BLOCK, NAME, LENGTH)

*CODE	000423
*DATA	002565
*BLANK	000000
DATA1	000015
BLK2	000040

EXTERNAL REFERENCES (BLOCK, NAME)

0005	DATA
0006	NR00\$
0007	NR01\$
0010	NR02\$
0011	NR04\$
0012	SIN
0013	ALG
0014	EXP
0015	NEXP6\$
0016	NR00\$
0017	NSTOP\$

STORAGE ASSIGNMENT FOR VARIABLES (BLOCK, TYPE, RELATIVE LOCATION, NAME)

00000	002305	1000F	0000	002366	1001F	0001	000006	1176	0001	000027	133G	0001	000053	1426
00001	000014	2L	0000	002501	2001F	0001	00323	2066	0000	002525	1000F	0000	002530	5000F
00003	000001	LF	0003	000000	DO	0000	062345	DV	0000	002334	NVF	0000	002341	EM
00000	000030	A	0003	000001	ICOM	0000	002306	ICOMN	0003	000012	IFUEL	0000	0000	304
00000	002324	II	0003	000013	IMP	0000	002327	INDEX	0000	000034	INV	0000	0000	IFUELN
00000	002312	IIIVN	0000	002321	IPAGE	0000	002346	IPRNT	0000	002326	IPROC	0000	0000	INVEST
00000	000024	ITEST	0000	002316	IS	0000	002323	J	0000	002322	K	0000	0000	IS
00003	000007	4R	0000	002317	LN	0000	002317	NSETS	0000	002325	NTEST	0003	0000	PC
00000	002342	P3	0000	002336	R	0003	000014	RHF	0000	002333	RHOO	0000	002340	RN
00000	002337	R4IN	0004	000030	SGA50	0004	000010	SGN2H	0000	002332	SGN204	0000	002343	SPR
00000	002335	T	0004	000020	TASO	0003	000006	TF	0000	000000	TITLE	0004	000000	TN2H4
00000	000005	TO	0003	000004	VF	0003	000003	VO	0000	002320	YLO	0000	000000	TN2H4

```

10 DIMENSION TITLE(20), ITEST(200), IPIV(1000),
11 IFUELW(2), ICONF(4), INVR(3)

```

```
REAL IVP,MONT,VPR,I$  
COMMON/DATA1/DO,J,HF,PC,VO,VF,TQ,TF,VPR,WCV,ICOM,IFUEL,IMP,RHF  
CURVOT,BLKZ/J,MN2(8),SGM2(8),TASO(8),SGM50(6)  
DATA IVP/, .UJG., MURICK, .HOUSEM./  
DATA ICOWN/, VIX,,, SEP,,, POP,,, UNDEF./  
DATA IFUELIN2M4,,A-50,,/
```

ORIGINAL PAGE IS
OF POOR QUALITY

```

00112 C NAMELIST/INPUT/TITLE,ITEST,NSETS,INV,
00113 C READ(5,1000) TITLE
00114 C
00115 C READ(5,INPUT)
00116 C
00117 C HERE WE SET THE CASES TO BE PROCESSED
00118 C NSETS = NO. OF GROUPS OF SEQUENTIAL TESTS
00119 C ITEST = RANGE OF TESTS IN SEQUENCE ITEST(1)-ITEST(N)
00120 C INV = INVESTIGATORS FLAG 1 ZUNG, 2 HURICK
00121 C
00122 C XLC=100.
00123 C IPAGE=0
00124 C K=0
00125 C J=1
00126 C DO 1 I=1,NSETS
00127 C ITEST=ITEST(J)+1-ITEST(J)+1
00128 C IPROC=ITEST+ITEST(J)-1
00129 C INDEX=ITEST(J)
00130 C J=J+2
00131 C
00132 C HERE WE START PROCESSING EACH TEST OF ISETS
00133 C
00134 C DO 2 I=INDEX,IPROC
00135 C CHECK FOR INVESTIGATORS
00136 C
00137 C IPAGE=IPAGE+1
00138 C K=K+1
00139 C INVEST=INV(K)
00140 C CALL DATA(INVEST,I)
00141 C IF(V0 .LE. 0. ) IPAGE=IPAGE-1
00142 C IF(V0 .LE. 0. ) GO TO 2
00143 C
00144 C CALCULATE ALRC MODEL CORRELATION PARAMETERS
00145 C
00146 C SGR204=(12.5L-4*(11.0-T0)+1.515)+(11.2E-6*100.)
00147 C RH00=SGR204*0.240
00148 C DVF = DF/12./VF/SIN(0.5*IMP/57.3)
00149 C T= TF*460.
00150 C IS = EXP(LALOG(DVF)+...J-(21800./T)))
00151 C K = 49.2 * DF/ PC**0.333
00152 C RNN =(RH00* V0 * V0 * DO)/( RHF * VF * VF * DF)
00153 C RH = 1.0 / (1.0 + RNN )
00154 C EM = (1.0- (1.0-2.0*RN)**2)**2
00155 C PD=PC/DF
00156 C
00157 C CALCULATE RUPE STAGNATION PRESSURE RATIO
00158 C
00159 C SPR =(RHF * VF * VF)/(RH00 * V0 * V0)
00160 C
00161 C CALCULATE JPL MODEL PARAMETERS
00162 C
00163 C IF(INVEST .EQ. 3) PINE=AN
00164 C IF(INVEST .EQ. 3) NNE=
00165 C DV = JF/12.0/VF
00166 C IF(IPAGE .GT. 40) IPAGE=1
00167 C IF( K .GT. 1 .AND. (INV(K)-IPRINT) .NE. 0 ) ,PAGE=1
00168 C IF( K .GT. 1 .AND. IPAGE .EQ. 1 ) WRITE(6,5000)
00169 C IF(IPAGE .EQ. 1) ,RITE(6,3000) TITLE
00170 C IF(IPAGE .EQ. 1) ,RITE(6,1001) INV(INVEST),N
00171 C
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*NEW

WFOR, IS VAPOR
FORTRAN V: ISO VERSION 4.4S-09/03/74-09:05:38 (0)

MAIN PROGRAM

STORAGE USED: CODE(1) 000326; DATA(0) 000300; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 LIQ 000002
0004 VAP 000005
0005 GAS 000004
0006 MIX 000010

EXTERNAL REFERENCES (BLOCK, NAME)

0007 PROPMX
0010 MINTR\$
0011 NR'IL\$
0012 INRIL\$
0013 SIN
0014 SCRT
0015 CURT
0016 ALOG
0017 EXP
0020 NSTOPS

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0001	000045	1UL	0001	000236	15L	0001	000320	20L	0001	000324	30L	0001	000002	SL				
0000	K	000014	ALPHA	0005	000000	CPG	0003	R	000000	CPL	0006	R	000000	CPV				
0000	K	000004	UAX	0000	K	000004	DO	0000	R	000032	DT	0000	R	000034	DTV			
0000	K	000031	UX	0004	R	000004	HV	0000	I	000042	INPUT	0000	I	000007	NSTEP			
0000	R	000074	OUT1	0000	000134	OUT2	0000	000155	OUT3	0000	000207	OUT4	0000	R	000002	PC		
0000	R	000033	PCTV	0000	R	000016	PR	0006	R	000006	PV	0000	R	000024	OSH			
0000	K	000022	WIOT	0000	K	000021	QV	0000	R	000012	RED	0003	R	000001	RHOL			
0000	K	000005	T-ETA	0000	K	000011	SC	0000	R	000006	SPACE	0006	R	000007	TAVE			
0000	K	000027	WL	0000	K	000040	WLV	0000	R	000041	WL1	0000	R	000015	WV			
0000	R	000001	XKMX	0004	000001	XKV	0005	000002	XMUG	0000	R	000003	U	0000	R	000001	XKG	
0005	000003	XKMG	0006	K	000000	XKMX	0004	R	000003	XMWV	0006	R	000003	XMUMX	0004	000002	XNUM	
0000	K	000036	XSTEP	0000	K	000021	Z	0000	R	000020	ZZ	0000	R	000013	XNUM	0000	000001	XNUM

00100	1*	C	PROGRAM TO CALCULATE THE HEAT UP OF A CYLINDRICAL JET
00100	2*	C	PRIOR TO IMPINGEMENT
00100	3*	C	
00100	4*	C	
00101	5*		NAMLIST/INPUT/TL,TB,PC,U,DO,THETA,SPACE,NSTEP
00103	6*		NAMLIST/OUT1/XMXX,XKMX,CPMX,XKUMX,DMX,PV,TAVE,RHOMX,U,RM
00104	7*		NAMLIST/OUT2/SC,RED,XKUM,ALPHA,WV
00105	8*		NAMLIST/OUT3/PR,XKUM,ZZ,Z,OTOT,QV,OSH,GHV
00105	9*		NAMLIST/OUT4/DTL,RHOL,CPL,ML,TL,YI,PP,DX,DT,PCTV,DTV,TV

```

04/03/74 15:35 1K55 000427139 000427 S30 150 DATE 090374
J0107 10* COMON/LIG/CPL,RHOL
J0110 11* COMON/VAP/CPV,XKV,XMUV,XMUV,HV
J0111 12* COMON/GAS/CPG,XKG,XMUG,XMWG
J0112 13* COMON/MIX/XMXX,XKX,CPMX,XMUMX,DHX,RHOMX,PV,TAVE
J0113 14* 5 CONTINUE
J0114 15* READ (5,INPUT,END=30)
J0117 16* WRITE(6,INPUT)
J0118 17* RM=3./4.*DO
J0122 18* U=U*12.
J0123 19* XSTEP=NSTEP
J0124 20* THETA=THETA/57.3
J0125 21* XIMP=SPACE/2./SIN(THETA)
J0126 22* DX=XIMP/XSTEP
J0127 23* DT=DX/U
J0130 24* PCTV=0.
J0131 25* NS=0
J0132 26* NLV=0.0
J0133 27* 10 CONTINUE
J0134 28* NS=NS+1
J0135 29* CALL PROFMX(TL,TB,PC)
J0136 30*
J0136 31* CALCULATE MASS TRANSFER RATE
J0136 32*
J0137 33* SC=XMUMX/DHX/RHOMX
J0140 34* REJ=2.*RM*U*RHOMX/XMUMX
J0141 35* XNUM=2.*0.6*SQRT(RED)*CBRT(SC)
J0142 36* ALPHA=PC/PV*ALOG(PC/(PC-PV))
J0143 37* WFV=2.*3.14159*DMX*XMW*RM*ALPHA*XNUM*PV/18510./TAVE*1.25
J0143 38*
J0143 39* CALCULATE HEAT TRANSFER RATE
J0143 40*
J0144 41* PR=CPMX*XMUMX/XKMX
J0145 42* XNUH=2.*0.6*CBRT(PR)*SQRT(RED)
J0146 43* ZZ=WFV*CPV/2./3.14159/XKMX/RM/XNUH
J0147 44* IF(ZZ.GE.80) ZZ=80.
J0151 45* Z=ZZ/(EXP(ZZ)-1.)
J0152 46* OTOT=2.*3.14159*XKMX*XNUH*RM*(TB-TL)*1.25
J0153 47* QV=OTOT*Z
J0154 48* QSH=OTOT-QV
J0155 49* QNV=WFV*HV
J0156 50* QTV=QSH/CPV/WFV
J0157 51* TV=TL+QTV
J0160 52* IF(TV.LT.2500) GO TO 15
J0162 53* DTV=2500.-TV
J0163 54* TV=2500.
J0164 55* QSH=(2500.-TV)*CPV*WFV
J0165 56* QV=OTOT-QSH
J0165 57* 15 CONTINUE
J0167 58* NL1=3.14159/4.*(DO**2.*XIMP*RHOL
J0170 59* *L=4./3.*3.14159*(RM**3.)*RHOL
J0171 60* ALV=NLV*WFV*DT
J0172 61* PCTV=(QV-WFV*HV)/CPL/NL*DT
J0174 62* DTLE=(QV-WFV*HV)/CPL/NL*DT
J0174 63* TL=TL+CTL
J0174 64* WRITE(6,OUT1)
J0174 65* WRITE(6,OUT2)
J0174 66* WRITE(6,OUT3)
J0174 67*

```


09/03/74 09:05:35 1R55 000427139 000427 S30 15

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00205 67*
00211 68*
00213 69*
00214 70*
00215 71*
00216 72*
00217 73*
00220 74*

```

WRITE(6 OUT4)
IF (MS.GE.MSTEP) GO TO 20
GO TO 10
20 CONTINUE
THETA=THETA*57.3
GO TO 5
30 CONTINUE
END

END OF COMPILATION: NO DIAGNOSTICS.

DATE 090374
000305
000311
000316
000320
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000325
000325

PA

4

CEFOR, IS PROPMX
FORTRAN V: ISO VERSION 1.45-09/03/74-09:05:42 (.0)

SUBROUTINE PROPMX ENTRY POINT 000167

STORAGE USED: CODE(1) 000202; DATA(0) 000064; BLANK COMMON(2) 000000

COMMON BLOCKS:

0003 LIQ 000002
0004 VAP 000005
0005 GAS 000004
0006 MIX 000010

EXTERNAL REFERENCES (BLOCK, NAME)

0007 EXP
0010 XPMX
0011 HE H33

STORAGE ASSIGNMENT (BLOCK, TYPE, RELATIVE LOCATION, NAME)

0005 K 000000 CPG 0003 K 000000 CPL 0006 R 000002 CPMX 0004 R 000000 CPV 0000 R 000000 C1
0000 K 000001 C2 0006 K 000004 DMX 0004 R 000004 HV 0000 000042 INJPS
0003 K 000001 RHOL 0006 K 000005 RHOMX 0006 R 000001 XKG 0006 R 000001 XKMX
0004 K 000001 XKV 0005 K 000002 XMUG 0006 R 000003 XMMG
0006 K 000000 XMMX 0004 K 000003 XMMV 0005 R 000002 XMUV 0004 R 000002 XMUV

00101 1* SUBROUTINE PROPMX (TL, TB, PC)
00101 2* C
00101 3* C
00101 4* C
00101 5* C
00101 6* C
00101 7* C
00101 8* C
00101 9* C
00101 10* C
00101 11* C
00101 12* C
00101 13* C
00101 14* C
00101 15* C
00101 16* C
00101 17* C
00101 18* C
00101 19* C
00101 20* C
00101 21* C
00101 22* C

SUBROUTINE TO CALCULATE N2H4 LIQUID AND VAPOR PROPERTIES
AND N2H4/NTO GAS PROPERTIES
COMMON/LIQ/CPL, RHOL
COMMON/VAP/CPV, XKV, XMUV, XMMV, HV
COMMON/GAS/CPG, XKG, XMUG, XMMG
COMMON/MIX/XMMX, XKMX, CPMX, XMUNX, DMX, RHOMX, PV, TAVE
TAVE = (TL + TB) / 2.

CALC LIQUID PROPERTIES

CPV = 587125 + 2.0708E-4 * TL
RHOL = 5.062318E-24 + 0.28897E-5 * TL - 5.54321E-9 * (TL ** 2.)

CALC VAPOR PROPERTIES

CPV = 335 + 1.804E-4 * TAVE
XKV = 1.23753E-12 + 2.230358E-10 * TAVE
XMMV = 4.19981E-8 + 9.581164E-10 * TAVE
XMMV = 32.004

000003
000007
000007
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000007
000020
000023
000027
000033

09/03/74	05:35	1R55	000427139	000427	S30	150	DATE	090374
00116	23*						000035	
00117	24*						000047	
00117	25*						000047	
00117	26*						000047	
00120	27*						000056	
00121	28*						000060	
00122	29*						000062	
00123	30*						000064	
00123	31*						000064	
00123	32*						000064	
00123	33*						000064	
00124	34*						000066	
00125	35*						000073	
00126	36*						000075	
00127	37*						000102	
00130	38*						000111	
00131	39*						000115	
00132	40*						000122	
00132	41*						000122	
00133	42*						000146	
00134	43*						000152	
00135	44*						000201	


```

PV=EXP(14-328787-(7363.22/(TL-63.1713)))
HV=730.747-0.3591305*TL+1.214E-4*(TL**2.)
C
C
CALC GAS PROPERTIES
CPG=0.517
XKG=4.0E-6
XMUG=4.74E-6
XMG=20.82
C
CALC VAPOR MIX MEAN PROPERTIES
C1=(1.-PV/2./PC)
C2=PV/2./PC
XMNX=C1*XMWG+C2*XMWV
CPMX=C1*XMWG/XMWMX+CPG+C2*XMWV/XMWMX*CPV
XKMX=C1*XKG+C2*XKV
XMUMX=C1*XMUG+C2*XMUV
DMX=(.63+.001*(TAVE-2000.))*(300./F / (XMWMX**(.83-.06*( (TAVE-300
*0.)/1000.)*2.)))
RHOMX=PC*XMWMX/TAVE/18510.
RETURN
END

```

END OF COMPILATION: NO DIAGNOSTICS.

UFIN

RSS DATA STORAGE AND DATA
REDUCTION PROGRAM

D A T A (8)

```
1. SUBROUTINE DATA(INVEST,1)
2. DIMENSION IFUEL(159),XMR(159),ICOM(159),IMP(159),APC(159),XVU(159)
3. 1,XVU(159),XMR(159),XDU(159)
4. 2,ICOM(159),XPC(159),XTO(159),XTF(159),XMR(159),IMP(159),XVU(159),XMR(159)
5. 3(159),XPC(159),XTO(159),XDF(159),XVU(159),XMR(159),IMP(159),XVU(159),XMR(159)
6. 4(159),XPC(159),XTO(159),XDF(159),XVU(159),XMR(159),IMP(159),XVU(159),XMR(159)
7. 5(159),XPC(159),XTO(159),XDF(159),XVU(159),XMR(159),IMP(159),XVU(159),XMR(159)
8. 6,ICOM(159),XPC(159),XTO(159),XDF(159),XVU(159),XMR(159),IMP(159),XVU(159),XMR(159)
9. 7,IFUEL(159),XPC(159),XTO(159),XDF(159),XVU(159),XMR(159),IMP(159),XVU(159),XMR(159)
10. 8,XAT(159)
11. REAL IMP
12. REAL MR,NOM,IMP,IMP1,IMP2,IMP3,IMP4
13. COMMON/DATA/DO,DF,PC,VO,VF,TC,TF,MR,NOM,ICOM,IFUEL,IMP,RMF,SPLD
14. 1,MF,TIMP,MU,XAT
15. COMMON/BLK1/ XDU(159),XDF(159),XPC(159),XVU(159),XMR(159),XVU(159),
16. 1 XDU(159),XDF(159),XPC(159),XVU(159),XMR(159),XVU(159),
17. IFUEL(159),IMP1(159),SPACE(159),XMF(159),
18. 3 XDU(159),XDF(159),XPC(159),XVU(159),XMR(159),XVU(159),
19. XDU(159),XDF(159),XPC(159),XVU(159),XMR(159),XVU(159),
20. 4 IFUEL(159),IMP2(159),
21. IFUEL(159),IMP2(159)
22. COMMON/BLK2/IN2M(159),SGN2M(159),TASO(159),SGASO(159),TN2M(159),
23. 1SN2M(159),TMH(159),SGMH(159),SGFUEL,SGM2M
24. ***** PHYSICAL PROPERTIES OF FUEL *****
25. A-50
26. DATA TASO /0.,100.,200.,300.,400.,500.,550.,600./
27. DATA SGASO /0.93,0.890,0.835,0.780,0.710,0.620,0.565,0.480/
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115. DATA(XMFI(J),J=201,400)/
116. *0690,.0690,.0530,.0500,.0820,.0340,.0270,.0180,.0280,.0300,
117. *0290,00,.0650,.0290,.039,.033,.030,.29,.029,.030,2,.031,0.000,
118. *0230,.0100,.0100,.0260,.0298,.0350,.5,.0370,0.000,
119. *0340,.0290,.0150,.0100,.0100,.0100,.0110,.0100,.0090,.0300,.0300,
120. *0300,.0300,.0300,.0300,.0390,.0550,.0630,.0730,.0390,.0550,.0730,
121. *0630,.0740,.0630,.0630,.0630,.0630,.0630,.0630,.0630,.0630,.0630,
122. *0640,.0640,.0630,.0620,.0620,.0640,.0640,.0640,.0620,.0640,.0640,
123. *0760,.0640,.0650,.0840,.0650,.0640,.0650,.0650,.0620,.0630,.0620,
124. *0630,.0620,.0620,.0620,.0620,.0620,.0620,.0620,.0620,.0620,.0630,
125. *0630,3,.0620,240,.0580,.0680,.0580,.0750,.0760,.0750,.0900,
126. *0900,.0900,.0900,.0900,.0900,.0900,.0900,.0870,.0770,.0760,.0870,
127. *0830,.0830,.0880,.0830,.0690,.0690,.0690,.0700,.0690,.0690,.0680,
128. *0690,.0690,.0880,.0880,.0680,.0680,.0690,.0690,.0680,.0680,.0690,
129. *0690,.0690,.0600,.0800,.0760,.0760,.0620,.0620,.0620,.0620,.0620,
130. *0200,.0330,.0300,.0290,.0290,.0290,.0310,.0310,.0360,.0360,.0360,
131. *0360,.0370,.0350,.0350,.0390,.0370,.0360,.0360,.0350,.0360,.0840/
132. DATA(XMFI(J),J=401,500)/
133. *0850,.0870,.0870,.0870,.0830,.0830,.0830,.0870,.0870,.0870,.0670,
134. *0870,.0670,.0670,.0420,.0450,.0450,.0380,.0670,.0680,.0450,.0440,
135. *0350,.0350,.0340,.0170,.0340,.0350,.0350,.0240,.0240,.0230,.0220,
136. *0220,.0240,.0240,.0240,.0240,.0240,.0240,.0240,.0210,.0200,.0170,
137. *0090,.0120,.0070,.0100,.0130/
138. DATA (XVUI(J),J=1,393)/
139. *1400,23,1240,44,2940,55,240,42,0,0,41,45,340,,
140. *71,10,0,20,41,34,40,41,0,0,38,30,19,53,53,65,43,,
141. *39,61,22,0,0,23,40,20,20,18,540,47,42,25,21,27,24,,
142. *18,0,0,18,16,52,14,17,36,68,13,8,3,6,5,4,4,3,13,9,1,
143. *8,8,0,10,9,9,3,4,3,12,2,0,0,9,6,14,22,16,1,13,12,30,,
144. *21,24,24,16,14,11,0,16,23,24,23,0,0,24,15,23,240,,
145. *22,22,3,0,21,19,18,20,19,17,17,22,27,26,240,25,24,,
146. *25,24,0,0,24,19,24,23,23,20,240,31,33,33,29,28,29,,
147. *29,30,30,240,56,0,0,56,0,0,56,54,59,55,59,740,51,55,,
148. *0,0,68,50,48,56,30,22,65,74,55,55,38,0,0,57,58,61,,
149. *0,0,61,56,45,35,30,44,36,36,32,32,32,31,34,40,46,,
150. *51,56,58,58,55,49,0,0,56,58,55,42,75,0,0,61,66,0,0,,
151. *101,117,84,76,63,51,44,46,49,54,44,0,0,92,46,64,49,,
152. *49,46,36,46,47,36,0,0,29,13,16,52,49,58,58,64,,
153. *64,63,0,0,63,54,23,27,18,18,18,19,18,23,22,21,22,,
154. *21,21,27,34,40,34,26,34,36,43,45,36,41,42,41,43,,
155. *43,45,44,32,33,40,47,41,42,42,44,36,41,42,42,33,,
156. *42,40,44,44,43,45,45,45,43,41,37,34,52,51,54,54,,
157. *54,56,55,53,54,52,56,55,53,240,42,37,35,49,49,37,/
158. DATA (XVUI(J),J=394,504)/
159. *61,51,60,59,56,54,53,51,51,50,49,48,56,52,54,50,,
160. *36,34,36,40,39,44,43,41,39,37,2,340,51,7,50,3,37,2,36,2,
161. *34,9,39,7,36,3,48,7,46,7,65,4,64,4,51,0,50,1,45,2,45,2,54,1,49,5,
162. *57,4,48,8,56,5,54,1,60,50,49,3,50,6,54,1,57,2,54,2,45,7,44,4,
163. *44,9,48,8,67,7,58,5,60,5,60,5,37,9,65,1,46,4,46,6,56,5,61,3,
164. *54,9,55,9,61,9,60,4,58,2,46,6,42,6,46,6,42,6,59,5,73,5,61,3,65,4,
165. *46,5,48,4,26,5,25,4,43,1,54,1,54,1,54,1,54,1,54,1,39,4,38,0,
166. *38,6,33,2,34,9,38,3,36,6,39,6,39,6,36,30,6,30,6,33,2,33,2,33,7,
167. *20,7,21,4,41,4,40,3,47,1/
168. DATA (XVFI(J),J=1,390)/
169. *1440,35,1240,75,240,40,240,40,0,0,52,53,340,,
170. *78,1040,25,54,53,52,52,0,0,48,340,24,94,94,92,60,,
171. *56,67,29,0,0,29,64,33,36,24,340,30,30,31,24,33,33,,

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DATA (XIFI(J),J=1,381)
229. 1490.70.12.0.0.70.29.0.69.2.0.70.76.70.68.3.0.
230. 70.10.0.70.68.70.70.0.0.70.30.0.63.67.70.70.60.
231. 65.45.0.0.45.80.80.83.62.5.0.95.95.65.65.110.
232. 105.0.0.105.80.80.80.80.80.75.72.72.72.50.52.
233. 52.52.50.50.62.54.55.0.61.63.70.70.70.70.3.0.
234. 68.59.65.63.11.0.73.59.0.52.58.58.2.0.64.
235. 3.0.63.91.62.87.93.10.96.91.90.2.0.103.113.
236. 104.0.0.94.75.101.108.77.77.2.0.64.84.95.107.108.
237. 110.110.116.116.2.0.116.0.101.0.121.122.110.114.
238. 114.7.0.67.72.0.0.118.103.90.93.85.82.67.117.127.
239. 122.54.0.0.41.62.60.0.70.74.76.77.86.103.120.120.
240. 104.99.84.88.97.97.93.93.88.86.86.129.0.0.111.
241. 124.120.117.70.0.0.70.70.0.0.74.73.98.78.86.86.
242. 57.55.101.71.0.0.62.64.53.54.55.41.40.41.43.0.0.
243. 37.42.44.32.59.59.78.66.61.55.55.0.0.51.51.51.
244. 55.59.61.51.72.51.48.55.53.53.61.64.63.61.
245. 84.66.89.97.117.107.107.118.41.61.76.80.83.91.91.
246. 72.72.70.49.47.49.51.51.47.53.55.53.53.63.63.
247. 80.70.69.64.63.55.64.64.54.64.82.89.82.105.107.
248. DATA (XIFI(J),J=382,509)
249. 110.66.78.78.2.0.51.51.52.52.60.62.53.45.62.70.
250. 76.84.87.99.103.103.88.66.66.66.66.85.95.97.89.
251. 105.108.110.116.116.116.3.0.112.114.89.81.5.81.108.
252. 114.75.93.101.106.114.106.86.5.77.5.105.112.
253. 102.97.116.125.108.122.122.116.116.118.106.108.
254. 105.132.134.137.1.0.149.147.64.66.66.64.69.5.64.52.5.
255. 52.5.110.116.119.92.72.70.62.64.64.66.66.71.5.64.
256. 132.141.60.5.60.5.60.5.60.5.60.5.62.5.62.5.60.5.60.5.
257. 62.64.5.66.66.66.64.5.66.74.69.74.70.92.72.71.
258. 70.66.0/
259. DATA (XIFI(J),J=51,538)
260. 1490.70.12.0.0.70.29.0.69.2.0.70.76.70.68.3.0.
261. 3.0.1.29.10.0.1.11.1.09.0.91.0.02.1.08.0.0.1.02.3.0.1.10.0.82.
262. 0.82.1.02.1.04.1.31.1.31.1.08.0.0.1.11.0.89.0.84.0.80.1.10.
263. 5.0.1.69.2.0.1.18.1.18.1.18.1.04.0.96.0.0.0.96.0.96.1.07.1.23.
264. 1.12.1.09.1.06.1.16.0.95.0.89.0.90.1.10.1.11.0.96.1.18.1.27.1.63.
265. 1.19.1.00.0.71.1.17.1.09.0.0.1.13.0.98.1.20.1.22.1.07.1.18.3.0.
266. 1.71.1.21.1.20.1.21.0.1.11.0.1.07.1.44.1.47.1.35.0.1.31.1.45.
267. 1.25.2.0.1.20.1.3.0.1.30.1.55.1.23.1.47.0.94.1.83.1.41.1.31.
268. 1.41.1.24.2.0.1.42.1.27.1.14.1.13.0.0.1.09.0.72.1.25.1.03.1.04.
269. 0.92.2.0.1.23.1.36.1.59.1.23.0.94.1.19.1.12.1.24.1.24.2.0.1.17.
270. 0.0.1.17.0.0.1.27.1.13.1.24.1.16.1.52.7.0.1.10.1.18.0.0.1.12.
271. 1.05.1.06.1.14.1.05.1.19.1.17.1.56.1.14.1.14.0.87.0.0.1.31.1.25.
272. 1.34.0.0.1.29.1.23.1.20.1.29.1.33.1.32.1.26.1.20.1.22.1.22.1.24.
273. 1.18.1.18.1.10.1.24.1.17.1.16.1.22.1.44.1.49.0.97.0.0.1.26.1.31.
274. 1.70.1.17.1.45.0.0.1.19.1.29.0.0.1.13.1.54.1.22.1.16.1.16.1.14.
275. 1.25.1.69.1.32.1.03.1.13.0.1.09.1.24.1.44.1.15.1.23.1.20.0.93.
276. 1.11.1.16.1.07.0.0.1.04.1.00.1.26.1.50.1.30.1.25.1.25.1.25.
277. 1.25.1.29.0.0.1.44.1.45.1.15.1.56.1.00.1.40.1.29.1.46.1.57.1.37.
278. 1.30.1.30.1.1.1.27.1.27.1.24.1.11.1.13.0.62.1.23.1.11.0.89.1.24/
279. DATA (XIFI(J),J=339,509)
280. 1.11.1.05.1.16.1.21.1.16.1.24.1.26.1.28.1.25.0.93.0.93.1.12.1.35.
281. 1.21.1.23.1.19.1.23.1.03.1.21.1.19.1.19.079.1.02.1.12.1.23.1.23.
282. 1.22.1.25.1.25.1.29.1.29.1.21.1.06.0.98.1.23.1.16.1.27.1.29.1.27.
283. 1.31.1.29.1.28.1.27.1.21.1.32.1.26.1.26.1.26.1.26.1.26.1.26.1.26.
284. 1.16.0.0.1.21.1.02.1.20.1.19.1.35.1.09.1.06.1.02.1.00.1.16.1.16.

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993.102,1.14,1.11,1.09,946.377,919.103,1.01,1.14,1.13,1.05,
90.93,0.99,330.1,35,1.32,0.59,0.97,0.91,1.04,0.93,1.19,1.14,1.26,
91.23,1.20,1.18,1.08,0.08,1.52,1.32,1.37,1.30,1.58,1.45,1.60,1.24,
91.24,1.08,1.16,1.16,1.13,0.13,0.97,0.98,1.00,1.41,1.25,1.29,1.33,
91.34,0.84,1.14,1.00,0.96,1.17,1.26,1.19,1.21,1.25,1.25,1.21,1.26,
91.15,1.26,1.16,1.09,1.28,1.05,1.33,1.25,1.28,1.06,1.03,0.94,1.20,
91.23,2.31,1.22,1.21,1.21,1.31,1.25,1.37,1.16,1.16,1.26,1.21,1.27,
91.27,1.31,1.14,1.27,1.23,1.54,1.71,1.40,2.20,1.48,1.27/
DATA (XROM1(J),J=1,339)/
1480.1,62,120.2,05,290.0,47,20.0,64,0.38,1.10,
91.02,30.0,0.86,10.0,1.16,1.22,1.75,1.70,1.20,0.0,1.08,340.1,18,
92.18,2.18,1.30,1.31,0.83,0.83,1.23,0.0,1.17,1.80,2.20,2.28,1.20,
95.0,0.41,0.91,1.06,0.76,1.07,1.33,1.59,0.0,1.58,1.58,1.28,0.95,
91.13,1.21,1.28,1.06,1.06,1.81,1.77,1.17,1.15,1.36,1.01,0.88,0.43,
91.00,1.52,2.81,1.05,1.54,0.0,1.11,1.47,0.99,0.95,1.25,1.03,30.0,
90.49,0.97,0.9A,1.29,1.40,1.10,1.25,0.69,0.65,0.79,0.0,0.84,0.68,
90.92,2.0,0.99,0.99,30.0,0.84,1.07,0.94,0.66,1.61,0.43,0.72,0.84,
92.92,2.0,0.71,0.89,1.1,1.12,0.0,1.21,2.79,92.1,36,1.32,1.68,
92.0,0.95,1.56,1.59,1.23,1.44,0.99,1.13,0.93,0.93,20.0,0.65,0.0,
90.61,0.0,0.79,0.88,0.81,0.86,0.80,7.0,1.18,1.00,0.0,0.88,0.95,
90.92,0.94,0.95,1.00,0.76,0.59,1.11,1.11,1.89,0.0,0.82,0.92,0.79,
90.0,0.89,0.90,0.87,0.85,0.59,0.82,0.92,0.98,0.95,0.89,0.92,1.01,
91.04,1.18,0.93,1.03,0.86,0.95,0.68,0.67,1.52,0.0,0.87,0.82,0.49,
91.03,0.68,0.0,0.98,1.16,0.0,1.12,0.87,0.94,1.06,1.06,1.00,0.92,
90.60,0.92,0.78,1.03,0.0,1.20,0.94,0.78,1.09,1.01,0.96,1.59,1.05,
91.05,1.74,0.0,1.41,1.27,0.80,0.59,0.86,0.86,1.17,1.17,0.80,0.80,
90.80,0.0,0.70,0.71,1.09,0.77,0.76,0.78,0.80,0.70,0.62,0.78,0.65,
90.84,0.83,0.90,0.90,0.92,1.16,1.12,2.08,1.57,1.15,1.78,0.94,1.16/
DATA (XROM1(J),J=340,509)/
91.31,1.06,0.99,1.06,0.81,0.90,0.88,0.92,1.72,1.94,1.13,7.62,0.98,
90.98,1.02,0.93,1.35,0.97,1.02,1.02,1.52,1.19,1.14,0.93,0.95,0.96,
90.92,0.93,0.85,0.93,0.98,1.23,1.45,1.03,0.99,0.79,0.65,0.88,0.83,
90.86,0.93,0.86,1.08,0.81,0.99,0.91,2.0,0.848,1.10,0.484,1.04,
91.08,1.84,997.1,37,991.52,931.19,1.24,1.36,1.32,1.05,682,
91.48,0.98,1.09,1.08,1.21,62,1.85,1.69,1.34,1.39,1.09,1.11,1.13,
91.43,1.49,30.0,0.78,0.62,1.45,1.15,1.72,1.33,1.59,1.00,1.10,0.89,
90.91,1.06,1.02,1.23,1.23,0.62,0.81,0.76,0.84,0.58,0.68,0.55,0.93,
90.96,1.22,1.06,1.06,1.06,2.57,1.51,1.47,1.43,712.913,652,601,
907.2,15,789,1.42,1.54,1.04,0.89,1.00,0.97,0.79,0.91,0.98,0.90,
91.07,0.90,1.06,1.20,0.87,1.29,807.905,0.86,1.24,1.33,1.30,1.986,
936.471,852,970.970,830.910,0.71,0.66,1.04,0.90,0.98,0.89,
90.89,0.83,0.83,1.09,0.88,0.93,598.488,721.246,00.65,0.89/
DATA (XROM1(J),J=340,509)/
INVESTIGATUM *** NURICK *** (INV#2) *****
DATA XPL2 /#0,0,22*13,7,0,00,2*13,7,10*0,0,9*13,7,5*0,0,
1 DATA XVU2/ 235.225.225.220./
0 80.0,4*33,0,3*30,0,2*28,0,30,0,32,0,2*33,0,38,0,51,0,
1 2*47,0,66,0,41,6,40,3,40,5,42,3,00,0,41,3,39,2,10*0,0,
2 38,0,40,2,40,2,40,2,50,0,2*39,8,2*40,4/
1 DATA XVF2/
1 80.0,2*43,0,2*41,0,2*36,0,40,0,41,0,5*43,0,50,0,63,0,
2 2*56,0,54,0,56,2,51,9,51,9,48,5,00,0,50,1,43,2,10*0,00,
3 52,0,51,3,51,6,51,4,50,0,64,5,55,0,53,0,52,0/
DATA XTU2 /

DATA (8)

343. 1 800.0,3055.0,52.0,9055.0,3045.0,4000.0,0.0,2000.0,100.0,
344. 2 000.0,500.0,400.0/
345. DATA XTER /

346. 1 000.0,00.0,02.0,50.0,67.0,3055.0,0005.0,5055.0,4000.0,
347. 2 000.0,400.0,1000.0,4050.0,50.0,4045.0/
348. DATA XMH2 /500.0/
349. C

350. DATA XMUN2/
351. 1 000.0,200.00,0.01,0.93,201.00,0.05,0.09,0.63,0.63,0.88,
352. 2 20.91,0.05,0.94,201.00,1.03,1.26,1.15,1.10,0.90,0.00,
353. 3 1.02,1.09,1000.0,1.26,1.04,1.05,1.04,500.0,51.1,0.09,
354. 4 0.96/
355. DATA ICOM2/
356. 1 000.1503,301,403,0.203,1000,403,500,403/
357. DATA IFUEL2/ 0301,1302/
358. C

359. DATA IMH2 /
360. 1 000.0,13045.0,900.0,0.200.1000.0,4000.500.4000./
361. DATA XDL2 /
362. 1 000.0,1300.173,200.072,300.030,400.173,0.000,200.173,
363. 2 1000.0,400.173,500.0,400.173/
364. DATA KOF2 /
365. 1 000.0,1300.173,200.072,300.030,400.173,0.000,200.173,
366. 2 1000.0,400.173,500.0,400.173/
367. C

368. INVESTIGATOR *** HOUSEMAN *** INV = 3 ***
369. C
370. C

371. DATA ICUM3 /
372. 1 003,10203,1,3,1,3,3,201,3,1,003,1,3,3,1,203,1,3,3,300,
373. 2 1,3,3,401,203,201,903,201,3/
374. DATA XPC3 /
375. 1 10.2,16.0,02.0,14.2,18.0,56.0,14.2,23.0,73.0,14.2,29.0,
376. 2 14.2,52.0,120.0,14.2,28.0,68.0,14.2,19.0,42.0,110.0,14.2,
377. 3 23.0,52.0,150.0,14.2,35.0,250.0,54.0,110.0,14.2,450.0,14.2,
378. 4 94.0,250.0,420.0,503.0,552.0,3070.0,4000.0,30120.0/
379. DATA XVD3 /
380. 1 38.2,36.1,36.8,46.3,45.1,46.3,58.2,55.9,60.8,73.0,70.8,
381. 2 127.8,123.9,115.2,19.1,19.8,19.0,28.5,29.6,27.9,28.3,
382. 3 37.3, 30.6,39.9,37.8,66.3,62.9,63.2,117.4,145.0,40.0,32.3,
383. 4 21.6,20.5,18.7,10.24,0.20,5.400,37.28,28.8,8.0,120.0,
384. 5 40.0,40.0,40.0,10.0,61.0,51.0,20.91,0. 80.0,21.0,16.0,
385. 6 115.0,145.0,17.0/
386. DATA XDU3 /
387. 1 100.073,100.100,700.073,0.100,173,200.073,200.073,0.100,
388. 2 0.173,200.073,200.100,0.173,0.073,200.100,0.073,0.100,
389. 3 200.173,0.073,0.100,0.173/
390. DATA IMP3/50045.0/
391. DATA XMH3 /5001.20/
392. DATA IFUEL3/5001/
393. DATA XMUM3/5001.0/
394. DATA XTU3 / 50070.0/
395. C

396. C
397. C
398. C
399. C

***** INVESTIGATION *** LAWVER *** (INV04) *****
TEST DATA (A=AREA)

NEW
1
6
1

D A T A (8)

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DATA ICDA4 /
1 102,1,3,1,0,2,1,5,0,2,5,2,2,3,2,4,5,5,3,2,3,0,4,1,2,0,
2 3,2,2,1,3,2,3,0,1,1,5,0,6,0,1,0,1,6,0,1,5,0,1,3,2,0,2,1,
3 35,2,1,1,3,5,2,2,1,6,0,5,0,1,2,3,0,1,5,0,2,2,4,1,2,2,2,1,
4 0,2,5,0,5,1,1,2,2,2,5,3,1,2,1,5,3,2,1,4,0,6,3,1,5,2,
5 53,1,5,2,2,1,5,5,2,
DATA XPC4 /
1 308,308,309,311,507,1000,300,203,197,156,100,
2 89,101,162,203,269,301,310,510,1003,160,338,
3 334,955,995,298,203,256,191,152,95,114,411,
4 97,116,150,188,243,103,281,119,129,154,152,
5 190,258,102,99,297,484,958,98,100,102,107,
6 102,101,99,101,2,102,98,100,2,95,2,99,95,89,
7 95,196,198,201,192,197,97,202,295,182,196,
8 259,289,282,293,300,295,99,98,98,99,99,202,
9 2,202,92,94,99,97,81,102,95,201,199,193,295,
A 291,293,290,290,289,293,292,489,491,135,124,
B 98,81,148,194,122,123,122,122,144,119,122,
C 148,145,124,123,0,102,81,144,186,164,160,166,
D 192,123,117,119,139,137,147,129,125,99,80,
E 146,199,119,116,113,131,137,140,126,129,101,81,
F 157,210,129,124,132,130,156,150,124,98,78,78,
G 147,198,123,123,121,145,144,
DATA XTU4 /
1 4,88,2,89,86,2,84,83,84,76,88,82,81,80,74,
2 61,82,82,81,81,77,82,2,83,84,4,85,86,83,77,
3 78,77,76,79,61,64,66,68,2,67,4,68,69,2,70,
4 78,55,2,54,2,53,5,58,5,57,3,56,2,55,7,44,50,48,
5 47,64,66,67,75,76,71,70,69,68,70,73,64,62,
6 71,78,80,82,124,112,53,74,65,68,69,72,70,
7 66,77,80,70,63,61,62,78,76,75,71,68,64,65,
8 72,69,62,76,79,77,80,93,116,127,109,120,79,
9 136,162,153,73,0,71,74,78,79,74,77,2,78,79,
A 85,136,132,153,154,73,70,68,72,72,71,77,76,
B 140,120,153,140,59,78,3,71,72,80,2,85,130,131,
C 145,72,73,75,74,75,76,75,80,116,131,152,
DATA XTP4 /
1 4,87,2,88,85,2,84,2,83,76,87,82,78,77,76,81,
2 82,81,2,80,77,82,81,82,81,2,83,82,2,83,83,
3 197,3,199,195,59,62,64,6,67,68,3,69,74,56,
4 55,3,54,6,58,3,57,2,55,7,44,50,2,48,68,71,73,
5 78,77,3,70,69,70,72,62,93,149,177,200,190,
6 178,89,80,85,75,73,74,72,70,68,69,81,74,63,
7 64,68,84,79,77,75,70,68,70,73,71,68,75,97,
8 141,164,229,242,249,245,268,260,291,285,76,
9 0,72,76,82,79,74,79,71,75,170,239,288,283,288,
A 287,2,78,76,69,178,244,312,300,301,4,0,71,
B 77,80,77,75,73,150,226,250,287,296,37,82,
C 72,76,2,75,76,165,249,240,290,294,
DATA XMH4 /
1 1,66,1,57,1,62,1,62,1,70,1,57,1,71,65,1,64,1,61,6,
2 1,59,1,63,1,60,1,65,1,67,1,64,1,61,64,1,63,1,62,1,65,
3 1,65,1,61,1,61,1,59,2,61,58,1,61,56,1,57,1,58,1,66,1,74,
4 1,72,2,61,74,1,61,4,55,1,61,1,61,59,1,42,1,53,1,62,1,57,
5 1,61,1,49,1,61,62,1,64,1,62,1,37,1,60,1,71,52,1,61,
6 1,94,1,68,1,65,1,54,1,62,1,63,1,65,1,61,3,61,63,1,60,1,62,
7 1,63,1,64,1,66,1,62,3,61,63,1,65,1,67,1,7,1,6,2,1,62,

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DATA (8)

1.57,1.64,1.51,1.69,1.58,1.57,2.1.64,1.69,1.7,1.73,1.77,
2.1.69,1.64,1.58,1.61,1.59,1.61,2.1.63,1.61,1.67,1.61,1.6
1.57,1.62,1.47,1.61,2.1.61,1.59,2.1.61,1.67,1.7,1.63,1.66
1.69,1.7,1.77,1.62,1.61,1.6,2.1.75,1.76,1.72,1.69,
1.68,1.61,69,2.1.68,1.65,1.64,1.66,1.67,1.64,2.1.72,1.7,
3.1.69,2.15,2.37,2.33,2.41,2.13,2.28,2.23,2.27,1.90,2.31
2.3,14.0,1.76,1.66,1.67,1.64,1.63,2.1.64,1.66,1.66,1.67,
1.66,1.63,1.62,1.54,1.62,1.66,1.65,1.68,1.74,1.65,1.71,
1.65/
DATA IMP4 /
1 11.60,12.60,9.60,6.60,14.60,26.60,19.60,20.60,.
2 16.1,16.90,11.32,14.0,12.32,11.60,/
DATA X004 /
1 11.024,12.024,9.024,6.024,14.024,26.024,19.024,
2 20.024,16.024,16.024,11.0405,14.0,12.0336,11.024/
DATA X014 /
1 11.02,12.02,9.02,6.02,14.02,26.02,19.02,20.02,
2 16.02,16.02,11.02,9.02,14.02,12.02,11.02,/
DATA X004 /
1 0.246,0.258,0.144,0.255,0.239,0.255,0.234,0.173,0.243,
2 0.24,0.246,0.249,0.204,0.212,0.195,2.0246,0.15,0.237,
3 0.256,0.24,0.248,0.255,0.552,0.261,0.264,0.26,0.275,
4 0.262,0.102,0.277,0.275,0.259,0.092,0.102,0.25,0.256,
5 0.24,0.105,0.254,0.255,0.270,0.286,0.0831,0.0843,0.1056,0.1416,
6 0.24,0.105,0.254,0.255,0.270,0.286,0.0831,0.0843,0.1056,0.1416,
7 0.1537,0.0826,0.0808,0.1121,0.1083,0.1184,0.1674,0.1434,0.2072,
8 0.02637,0.1469,0.1478,0.1523,0.13254,0.3212,0.1295,0.1278,0.1539,
9 0.2655,0.1503,0.1508,0.1456,0.1485,0.1403,0.1431,0.3066
0 0.2869,0.2041,0.1948,0.1428,0.1254,0.1526,0.2531,0.1442
1 0.148,0.1465,0.1448,0.1487,0.1398,0.1339,0.1305,0.105,
2 0.127,0.143,0.149,0.226,0.255,0.315,0.117,0.45,0.32,
3 0.127,0.124,0.12,0.22,0.151,0.209,0.256,0.319,205,
4 0.263,0.125,0.142,0.113,0.093,0.168,0.222,0.1422,0.144,
5 0.142,0.141,0.165,0.45,0.141,2.0166,0.14,0.138,0,
6 0.107,0.0896,0.168,0.225,0.15,0.156,0.15,0.222,0.1,
7 0.144,0.1399,0.169,0.1699,0.161,0.578,0.554,0.464,0.81,
8 0.099,0.021,0.023,0.644,0.641,0.752,0.72,14.0,0.262,0.259,0.2
9 0.169,0.3,0.4,0.266,0.267,2.0259,0.308,0.318,0.139,0.111
DATA X014 /
1 0.0916,0.0937,0.166,0.222,2.0143,0.144,0.173,0.175/
2 0.152,0.156,0.153,0.152,0.148,0.163,2.0147,0.161,0.153,
3 0.16,0.0905,0.156,0.144,0.154,0.14,0.105,0.151,0.147,
4 0.15,0.153,0.123,0.128,0.12,0.154,0.156,0.0952,0.149,
5 0.16,0.153,0.157,0.161,0.212,0.15,0.153,0.15,0.158,0.164
6 0.07,0.171,0.171,0.163,0.0846,0.0644,0.153,0.162,0.148,
7 0.007,0.159,0.157,0.164,0.176,0.0604,0.0503,0.0617,0.0926,0.0951,
8 0.00427,0.0515,0.0678,0.0698,0.0847,0.102,0.0864,0.126,0.161,0.089
9 0.00897,0.0087,0.198,0.1945,0.0695,0.0758,0.0937,0.1243,0.0911,
0 0.1296,0.1187,0.0944,0.0744,0.0953,
1 0.1611,0.088,0.0903,0.0867,0.0851,0.0862,0.0791,0.079,
2 0.00796,0.00837,0.0811,0.0891,0.0938,0.14,0.156,0.193,0.0611,0.0867,
3 0.196,0.00792,0.0739,0.0829,0.0934,0.13,0.16,0.02,0.128,0.164,
4 0.0745,0.0832,0.069,0.0554,0.0945,0.13,0.0802,0.0738,0.078,0.091
5 0.0829,0.0806,0.0945,0.0903,0.0826,0.082,0.0632,0.0513,0.1,
6 0.136,0.0917,0.0938,0.0897,0.135,0.0818,0.0838,0.0821,2.01,0.0953
7 0.0268,0.252,0.149,0.156,0.327,0.403,0.279,0.283,0.336,0.327,0.312
8 0.140,0.116,0.156,0.119,0.0917,0.184,0.244,0.162,0.16,0.154,0.155,
9 0.184,0.192,0.0857,0.0684,0.0594,0.0576,0.1,0.135,0.085,0.0827,
10 0.184,0.192,0.0857,0.0684,0.0594,0.0576,0.1,0.135,0.085,0.0827,

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514. 3.00075,.01011,.0106/
515. DATA IFUELE/
516. 1 1103,1202,903,603,1003,2003,1903,2002,1003,1003,1103,
517. 2 00,1203,1103/
518. DATA SPACE/
519. 1 11020,.12020,9020,.0020,10020,.20020,19020,29020,,
520. 2 100,1100,1,1100,1000,1200,1100,
521. DATA TIME/
522. 1 0100,200,100,100,200,200,300,100,230,700,,
523. 2 1009,1016,171,1213,220,2100,229,190,203,109,,
524. 3 203,152,140,109,136,2120,119,100,100,150,120,111,,
525. 4 015,160,113,130,117,113,101,201,207,200,211,,
526. 5 211,171,160,170,212,143,142,109,204,210,250,,
527. 6 207,204,171,170,100,216,219,221,225,240,252,,
528. 7 203,0,0,0,300,0,75,301,0,0,0,450,0,870,0,830,200,,
529. 8 202,250,233,100,320,209,301,311,0,0,310,310,,
530. 9 320,331,315,271,0,200,250,0,0,263,261,260,,
531. 0 309,203,302,317,327,331,331,110,0,100,120,0,
532. 1100,0/
533. DATA XAT/
534. 1 10055,0055,0275,20055,2,1006,30,2009,20,1006,
535. 2 0009,0552,0200,20,200,20,1006,0055,30,2009,40,1006,
536. 3 0055,20,2009,0055,0552,0275,20,1006,30,2009,20,1006,
537. 4 100,409,0055,1006,40,2009,40,0055,20,0552,0133,
538. 5 300,50,2009,30,0055,0216,20,0302,0350,0005,0001,
539. 6 00071,0131,0177,0430,40,01,0123,0177,0216,0201,
540. 7 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
541. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
542. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
543. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
544. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
545. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
546. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
547. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
548. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
549. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
550. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
551. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
552. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
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555. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
556. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
557. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
558. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
559. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
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566. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
567. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
568. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
569. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/
570. 00131,0216,310,0201,110,1006,100,120,0500,110,0201/

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571. 177.109.121.05.0.113.0.136.120.136.30.114.
572. 0.117.119.0.116.125.129.124.107.0.107.121.
573. 119.240.131.147.240.164.153.0.152.2595.90.96.
574. 230.087.92.97.50.101.110.123.110.125.137.
575. 114.132.143.137.159.172.400.270.81.76.81.78.
576. 79.70.90.94.100.78.81.79.78.63.82.83.84.
577. 102.116.123.130.140.140.137.143.142.142.152.
578. 155.168.0.155.169.157.172.167.175.
579. DATA NMRS/
580. 0.0.1.49.1.53.60.1.75.1.94.1.75.1.79.1.66.1.65.
581. 1.5.1.57.40.1.52.1.69.1.81.1.72.0.2.1.66.1.82.1.84.
582. 1.69.3.0.1.80.0.1.51.1.94.0.1.53.1.75.1.71.2.22.2.08.
583. 0.2.41.242.29.20.2.09.2.37.20.2.22.2.36.0.2.33.
584. 1.64.2.12.2.97.2.44.340.2.23.2.22.2.44.50.2.27.2.36.
585. 8.42.2.91.1.93.2.18.2.27.2.15.2.25.2.22.40.1.61.1.84.
586. 1.54.1.56.1.73.1.70.1.76.1.61.1.76.1.79.10.0.1.67.1.78.
587. 1.74.1.73.1.74.1.75.1.66.1.64.1.65.1.66.1.81.1.68.1.73.
588. 1.81.2.1.71.59.1.77.1.73.1.68.1.71.0.1.65.1.66.1.81.
589. 1.84.1.72.1.73/
590. DATA IMPS/
591. 145.60./

592. DATA ADUS/
593. 42.024.24.04.27.024.52.04/
594. DATA XDFS/
595. 42.02.24.033.27.02.52.033/
596. DATA XVFS/
597. 0.0.66.8.05.1.60.48.5.44.3.56.1.57.8.73.2.54.5.71.6.
598. 50.1.46.5.40.57.5.65.5.71.1.05.3.0.70.9.56.1.71.7.
599. 50.5.69.9.340.48.2.0.58.5.65.5.0.57.6.69.8.73.6.48.7.
600. 44.6.0.34.4.37.36.3.22.0.47.1.60.4.20.63.2.44.1.0.
601. 62.1.54.8.59.4.34.1.53.9.340.41.7.40.1.52.1.50.0.2.9.
602. 39.7.30.2.47.4.0.48.1.43.2.24.1.60.4.54.54.3.40.
603. 58.8.57.8.59.9.61.4.55.1.54.6.75.6.58.4.53.3.53.2.10.0.
604. 57.8.54.8.57.4.54.3.77.1.53.5.36.7.77.6.55.4.56.9.54.3.
605. 55.3.74.1.50.9.52.74.3.55.8.54.3.55.54.9.75.5.0.55.2.
606. 54.2.51.1.50.2.54.2.53.4/
607. DATA SP.CES/
608. 145.12./

609. DATA IFUELS/
610. 145.3/
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DATA XH06/0301..0304..0293..0291..0320..0293..0302..0236..02.
0349..0288..0283..0280..0288..0170..0133..0129..0108.
0191..0168..0164..0162..0198..0252..0258..0201..0194.
0296..0178..0161..0269..0271..0334..0390..0308..0289.
0472..0178..0399..0411..0411..0445..0357..0297..0546.
0456..0468..0468..0531..0459..0520..0446..0354..0290.
0506..0366..0454..0453..0455..0458..0547..045..0444.
0445..0327..0299..0153..0153..0153..0153..0154..0155.
0440..0458..0383..031..0352..047..0471..0562..0468.
0471..0367..0321..0253..037..0391..0335..0302..0241.
0445..0366..0367..0366..0353..037..0446..040..0454..0471.
0378..0299..0564..0476..0475..0484..0477.
0567..0562..040..047..032..0318..2..0313..2..0465/
DATA XH06/0301..0304..0293..0291..0320..0293..0302..0236..02.
0213..0178..0167..0178..0166..0166..0166..0166..0116.
0099..0099..0097..0115..0117..0155..0126..0104..0183.
0182..0153..0154..0153..0166..0271..0214..0171..0319.
0267..0278..0273..0293..0289..0218..0177..0316..0286.
2..027..0264..0327..0276..0225..0179..031..0218.
02..2..0266..0..027..0284..0316..0257..2..0259..2..0169.
010..24..0095..0093..24..0094..0095..4740..0273..0223..0182.
0329..0274..0277..0274..0325..0271..0212..0269..0219.
0177..0316..0264..0..0206..0172..043..0256..0212..0213.
0214..0211..0215..0258..0..0274..027..022..0174..0327.
A 0272..0276..0273..0276..027..2..0327..060..0274..0276.
0..018..0182..0184..0183..0153..0175..0275/
DATA XH06/0301..0304..0293..0291..0320..0293..0302..0236..02.
00..24..04..02..05..90..120..138..66..70..80..81..86..
86..25..87..127..136..70..74..77..76..81..82..120..126..
74..76..2480..72..74..123..126..2485..80..82..83..2482..
81..74..0..76..126..129..76..77..78..82..483..84..5..85.
4740..76..77..78..83..80..125..127..89..78..279..78..24.
75..0..2377..2480..89..117..124..060..365..2484..61..84..87.
06..87..124..127..125..060..2489..0..90..2489..3488../
DATA XH06/0301..0304..0293..0291..0320..0293..0302..0236..02.
77..01..84..86..89..84..162..224..241..292..71..77..89.
80..81..84..179..234..227..291..76..77..78..76..207..
240..248..312..80..81..82..78..191..252..263..311..95..
170..62..84..86..85..87..83..183..183..225..240..248..24.
5 92..89..91..89..86..2489..94..96..97..99..4740..75..76..2477..
6 189..232..242..275..185..95..64..83..81..80..79..0..54.
7 77..177..214..223..225..279..060..85..386..85..173..94..240.
8 229..229..231..280..276..060..2490..0..90..89..90..3489../
DATA IFUEL6/181-3/
DATA SPACE6/181-3/
1 108..540..1367..060..249..0..069../
DATA XH06/0301..0304..0293..0291..0320..0293..0302..0236..02.
1..71..421..661..571..61..01..01..03..1..62..1..69..1..64..1..60.
2 1..68..1..461..671..591..571..621..671..71..751..781..69.
3 241..44..1..461..481..491..431..51..41..66..631..661..73.
4 1..71..721..731..721..621..591..61..571..61..571..61..661..68.
5 1..731..71..01..60..241..731..751..71..71..721..71..721..71..73.
6 241..01..341..641..65..4740..1..681..721..731..241..721..71..721..71..72.
7 1..441..471..491..1..51..1..48..0..1..71..751..691..731..241..72.
8 1..71..1..731..741..751..601..1..661..74..341..721..731..741..72.
9 1..741..751..721..731..721..601..1..711..741..0..1..781..741..1..71.
1..741..767

DATA (8)

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685. DATA XMP/180,90,101,81,81,20,101,70,07,0,10,0,5,0,11,1,
686. 60,1301,60,90,
687. DATA XDL/18,027,90,024,10,027,0,025,0,027,20,047,10,027,
688. 7,028,07,0,10,047,5,022,11,027,0,0,13,030,0,0,
689. 9,087,
690. DATA XDF/18,025,90,020,10,025,10,025,8,025,2,042,10,025,
691. 7,021,47,0,10,047,0,022,11,025,0,0,13,028,0,0,
692. 9,042.
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GO TO (101,102,103,104,105,106 IVEST
IF INVEST = 1 ZUNG IF INVEST = 2 MURICK
IF INVEST = 3 MOUSEMAN
IF INVEST = 4 LAWVER
IF INVEST = 5 ROCKETDYNE
IF INVEST = 6 R(OMS)

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101 DO = XDC1(I)
DF = XDC1(I)
PC = XPC1(I)
VO = XVO1(I)
VF = XVF1(I)
YO = XYO1(I)
YF = XYF1(I)
MH = XMH1(I)
MOM = XMOM1(I)
ICOM = XICOM1(I)
IFUEL = IFUEL1(I)
IMPR = IMPR1(I)
MF = XMF1(I)
SPLO = SPLO1(I)

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```

IF (IFUEL.EQ.1) SGFUEL=INTP(TF,1,0,TN2M4,SGM2M4)
IF (IFUEL.EQ.2) SGFUEL=INTP(TF,1,0,TN50,SGA50)
IF (IFUEL.EQ.3) SGFUEL=INTP(TF,1,0,TNHH,SGHH)
SGN2U4 = 1.53-(12.5E-4*TO) + (11.7E-6*100.)
GO TO 150

```

```

102 DO = XUO2(I)
DF = XDF2(I)
PC = XPC2(I)
VO = XVO2(I)
VF = XVF2(I)
YO = XYO2(I)
YF = XYF2(I)
MH = XMH2(I)
MOM = XMOM2(I)
ICOM = XICOM2(I)
IFUEL = IFUEL2(I)
IMPR = IMPR2(I)

```

```

SGN2U4 = 1.53-(12.5E-4*TO) + (11.7E-6*100.)
IF (IFUEL.EQ.1) SGFUEL=INTP(TF,1,0,TN2M4,SGM2M4)
IF (IFUEL.EQ.2) SGFUEL=INTP(TF,1,0,TN50,SGA50)
IF (IFUEL.EQ.3) SGFUEL=INTP(TF,1,0,TNHH,SGHH)
GO TO 150

```

```

103 DO = XUO3(I)
DF = XDF3(I)
PC = XPC3(I)
VO = XVO3(I)

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DATA ()

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782. TO = XT03(I)
783. TP = XT03(I)
784. MR = XMR3(I)
785. MOM = XMOM3(I)
786. ICOM = ICOM3(I)
787. IFUEL = IFUEL3(I)
788. IMP = IMP3(I)
789. SGN204 = 1.53 - ((1.7E-04 * TO) + ((1.7E-04 * 100.)
790. IF (IFUEL.EQ.1) JELMINTP(TF,1.0,TM2M,SGN2M4)
791. IF (IFUEL.EQ.2) JELMINTP(TF,1.0,TM2M,SGN2M4)
792. IF (IFUEL.EQ.3) SGN204 = ((1.53 - ((1.7E-04 * TO) + ((1.7E-04 * 100.)
793. NO = (SGN204 * (DU**2 + 0.785)) * VO / 144.
794. VF = NO / MR
795. VF = (MF / SGNFUEL * (DF**2 + 0.785)) * 144.
796. GO TO 150
797. DO = XD04(I)
798. DF = XDF4(I)
799. PC = XPC4(I)
800. YD = XT04(I)
801. YF = XT04(I)
802. IFUEL = IFUEL4(I)
803. SGN204 = 1.53 - ((1.5E-04 * XT04(I)) * ((1.7E-04 * 100.)
804. IF (IFUEL.EQ.2) SGNFUEL = MINTP(TF,1.0,TM2M,SGN2M4)
805. IF (IFUEL.EQ.3) SGNFUEL = MINTP(TF,1.0,TM2M,SGN2M4)
806. VUE = (XU04(I) / ((1.62 * SGN204) * ((1.53 - ((1.7E-04 * TO) + ((1.7E-04 * 100.)
807. MR = XMR4(I)
808. MOM = ((1.62 * SGNFUEL) * VF**2) / ((1.62 * SGN204) * VU**2)
809. ICOM = ICOM4(I)
810. IMP = IMP4(I)
811. IFUEL = IFUEL4(I)
812. SPLD = SPLD4(I)
813. XAT = XAT4(I)
814. GU TO 150
815. DU = XD05(I)
816. DF = XDF5(I)
817. PC = XPC5(I)
818. TU = XT05(I)
819. TF = XT05(I)
820. IFUEL = IFUEL5(I)
821. VF = XV05(I)
822. SGNFUEL = MINTP(TF,1.0,TM2M,SGN2M4)
823. SGN204 = 1.53 - ((1.5E-04 * XT05(I)) * ((1.7E-04 * 100.)
824. MR = (SGNFUEL * (DF**2 + 0.785)) * VF / 144.
825. MR = XMR5(I)
826. VU = MR * VF
827. VU = (NO / SGN204 * (DU**2 + 0.785)) * 144.
828. ICOM = ICOM5(I)
829. IMP = IMP5(I)
830. SPLD = SPLD5(I)
831. GU TO 150
832. DU = XD06(I)
833. DF = XDF6(I)
834. PC = XPC6(I)
835. TU = XT06(I)
836. TF = XT06(I)

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799.  VD = XN06(I)
800.  MF = XNF6(I)
801.  IFUEL = IFUEL6(I)
802.  SCN204 = 1.53 * ((12.5E-8 * TO) * ((11.7E-6 * 100.)
803.  SCFUEL = WINTP(TF, 1.8, TMM, 80MMH)
804.  VD = (XN06(I) / ((62.8 * SCN204) * ((3.1416 * DO * 2) / 4))) * 148.
805.  VF = (XNF6(I) / ((62.8 * SCFUEL) * ((3.1416 * DF * 2) / 4))) * 148.
806.  MR = XNN6(I)
807.  MOM = ((62.8 * 9.8 * UEL) * VF * 2) / ((62.8 * SCN204) * VO * 2)
808.  ICON = ICON6(I)
809.  IFUEL = IFUEL6(I)
810.  IMP = IMP6(I)
811.  SPLD = SPACE6(I)
812.  150 CONTINUE
813.  RETURN
814.  END

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10. DIMENSION TITLE(20), ITEST(200), INV(600), XLDI(20),
11. *FASO(12), VAPASO(12), TFN2H(13), VAPN2H(13), TFMMH(11), VAPMMH(11),
12. *TON2H(13), VAPN2H(13), IINJT(20),
13. 1 IFUEL(13), ICOMH(6), IINH(6), INJECT(40), POUT2(41,19), IPUT2(41,3)
14. 2 *X(300,3,6), Y(300,3,6), ISEP(4,6), OFP(4)
15. 3 *TOVIS(6), VIS(6), TOST(9), SURTEN(9)
16. 4 *DOT(20), DOVIS(20), FOT(12), EOVIS(12), FOT(19), POSTF(18)
17. 5 *AT(20)
18. REAL IMP, MOM, MR
19. COMMON/DATAI/DO, DF, PC, VU, VF, TO, TF, MK, MOM, ICOM, IFUEL, IMP, RMF, SPLD
20. 1, MF, IIMP, MU, XAT
21. NAMELIST/DUMP/SGN204, SGFUEL, VISF, VISOI
22. COMMON/BLK2/IN2H(6), SGN2H(6), TASO(6), SGASO(6), TN204(10),
23. 1BN204(10), THMH(8), SGMH(8), SGFUEL, SGN204
24. DATA INVA, ZUNG, MURICK, MOUSEH, LAVERI, RCKTD, UHS=75,
25. DATA ICOMH, MIX, SEP, POP, UNDEF, M/S, M/P,
26. DATA IFUEL, IN2H, A=50, MMH,
27. DATA INJECT, UNLI, ME=DOT, UBLET,
28. *SPL, TASH, PLATE,
29. *PLET, V, DOU, BLET,
30. *YDOT, L=3,60, LUL, CURE, LU DP, L-U, L BA,
31. *RIER, L=U, L CU, ME, VOT, A CU, ME, LAM,
32. *L-U, L CURE, VOT, A BAT, RIER, T=LOL, CU, MI DP,
33. *T=LOL-B, CURE,
34. DATA TFASO/50, 60, 100, 120, 140, 160, 180, 200, 240, 280, 300,
35. 340,
36. DATA VAPASO/95, 13, 3, 6, 0, 8, 75, 13, 19, 26, 48, 80, 105, 165,
37. DATA TFN2H/40, 60, 80, 100, 120, 140, 160, 180, 200, 220, 260,
38. 300, 340,
39. DATA VAPN2H/075, 17, 32, 65, 115, 19, 3, 2, 4, 6, 7, 3, 11, 23, 43, 77,
40. DATA TFMMH/40, 65, 90, 115, 140, 165, 190, 215, 240, 290, 340,
41. DATA VAPMMH/29, 67, 142, 8, 5, 9, 15, 2, 25, 35, 5, 73, 5, 135,
42. DATA TON2H/32, 53, 70, 80, 90, 100, 110, 120, 130, 140, 150,
43. 160, 170,
44. DATA VAPN2U/5, 08, 16, 56, 14, 78, 18, 98, 21, 30, 69, 38, 62, 48, 24, 59, 98,
45. 74, 12, 91, 06, 111, 24, 135, 147
46. NAMELIST / INPUT/ITEST, NSETS, INV, IPLUT, OFP, NPLOT, IPRINT, IPLT, XLDI
47. 1 , IFUEL, IINJT, XXDF, AT, IBILL, NIINJT, NPP
48. READ(5,1000) TITLE
49. READ(5, INPUT)
50. HERE WE SET THE CASES TO BE PROCESSED
51. NSET = NO. OF GROUPS OF SEQUENTIAL TESTS
52. ITEST = RANGE OF TESTS IN SEQUENCE ITEST(1)-ITEST(N)
53. INV = INVESTIGATORS FLAG 1 ZUNG, 2 MURICK
54. IINJT = INJECTOR TYPE OF EACH GROUP OF TESTS
55. AT = THROT AREA FOR EACH SET OR GROUP
56. IBILL = WHICH PAGE OF OUTPUT; 0= MHVP, 1= C
57. NIINJT = INJECTION TO BE PLOTTED
58. NPP = FUEL TO BE PLOTTED
59. WRITE (6, INPUT)
60. IPAGE0

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M A I N (10)

58. K=0
59. J=1
60. SYMBOL COUNTERS ZEROED FOR ZETA PLOTTER BELOW
61. IMSP(1,1) = 0
62. IMSP(2,1) = 0
63. IMSP(3,1) = 0
64. IMSP(4,1) = 0
65. IMSP(1,2) = 0
66. IMSP(2,2) = 0
67. IMSP(3,2) = 0
68. IMSP(4,2) = 0
69. IMSP(1,3) = 0
70. IMSP(2,3) = 0
71. IMSP(3,3) = 0
72. IMSP(4,3) = 0
73. IMSP(1,4) = 0
74. IMSP(2,4) = 0
75. IMSP(3,4) = 0
76. IMSP(4,4) = 0
77. IMSP(1,5) = 0
78. IMSP(2,5) = 0
79. IMSP(3,5) = 0
80. IMSP(4,5) = 0
81. IMSP(1,6) = 0
82. IMSP(2,6) = 0
83. IMSP(3,6) = 0
84. IMSP(4,6) = 0
85. DU 1 I=1,NSETS
86. INJT = IINJT(I)
87. NTEST=IEST(J+1)-IEST(J)+1
88. IPRUC=IEST+IEST(J)-1
89. INDEX=IEST(J)
90. J=J+2
91.
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HERE WE START PROCESSING EACH TEST OF NSETS

DU 2 I=INDEX,IPROC
CHECK FOR INVESTIGATORS

IPAGE=IPAGE+1
IPAGE2 = IPAGE2 + 1

K=K+1

INVEST=INV(K)

HERE WE GO TO THE SUBROUTINES

CALL DATA(INVEST,I)

KLD = SPLD

IF (SPLD.LT. 0.01) KLD = KLD(11)

IF (SPLD.LT. 0.01) SPLD = KLD(11)

IF (VU .LE. 0.) IPAGE=IPAGE+1

IF (VU.LE.0.) IPAGE2 = IPAGE2 + 1

IF (VU .LE. 0.) GO TO 2

KLD=IPAGE2/2./SIN(0.5*IMP/57.3)

KLI = UP * KLDIMP

DIV = KLI/2./VU

FNF=MF*VF*SIN(IMP/2./57.3)/52.174

A = 3.14159*OF**2/4./SIN(IMP/2./57.3)

DPF=FNF/A

ORIGINAL PAGE IS
OF POOR QUALITY

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110 DATA DOT40,80,100,110,115,160,200,222,263,300,400,
111 50,500,550,600,620,690,695,700,715,
112 DATA OV18,82,85,88,90,95,98,100,105,110,115,120,125,130,135,140,145,150,155,160,165,170,175,180,185,190,195,200,205,210,215,220,225,230,235,240,245,250,255,260,265,270,275,280,285,290,295,300,305,310,315,320,325,330,335,340,345,350,355,360,365,370,375,380,385,390,395,400,405,410,415,420,425,430,435,440,445,450,455,460,465,470,475,480,485,490,495,500,505,510,515,520,525,530,535,540,545,550,555,560,565,570,575,580,585,590,595,600,605,610,615,620,625,630,635,640,645,650,655,660,665,670,675,680,685,690,695,700,705,710,715,720,725,730,735,740,745,750,755,760,765,770,775,780,785,790,795,800,805,810,815,820,825,830,835,840,845,850,855,860,865,870,875,880,885,890,895,900,905,910,915,920,925,930,935,940,945,950,955,960,965,970,975,980,985,990,995,1000,1005,1010,1015,1020,1025,1030,1035,1040,1045,1050,1055,1060,1065,1070,1075,1080,1085,1090,1095,1100,1105,1110,1115,1120,1125,1130,1135,1140,1145,1150,1155,1160,1165,1170,1175,1180,1185,1190,1195,1200,1205,1210,1215,1220,1225,1230,1235,1240,1245,1250,1255,1260,1265,1270,1275,1280,1285,1290,1295,1300,1305,1310,1315,1320,1325,1330,1335,1340,1345,1350,1355,1360,1365,1370,1375,1380,1385,1390,1395,1400,1405,1410,1415,1420,1425,1430,1435,1440,1445,1450,1455,1460,1465,1470,1475,1480,1485,1490,1495,1500,1505,1510,1515,1520,1525,1530,1535,1540,1545,1550,1555,1560,1565,1570,1575,1580,1585,1590,1595,1600,1605,1610,1615,1620,1625,1630,1635,1640,1645,1650,1655,1660,1665,1670,1675,1680,1685,1690,1695,1700,1705,1710,1715,1720,1725,1730,1735,1740,1745,1750,1755,1760,1765,1770,1775,1780,1785,1790,1795,1800,1805,1810,1815,1820,1825,1830,1835,1840,1845,1850,1855,1860,1865,1870,1875,1880,1885,1890,1895,1900,1905,1910,1915,1920,1925,1930,1935,1940,1945,1950,1955,1960,1965,1970,1975,1980,1985,1990,1995,2000,2005,2010,2015,2020,2025,2030,2035,2040,2045,2050,2055,2060,2065,2070,2075,2080,2085,2090,2095,2100,2105,2110,2115,2120,2125,2130,2135,2140,2145,2150,2155,2160,2165,2170,2175,2180,2185,2190,2195,2200,2205,2210,2215,2220,2225,2230,2235,2240,2245,2250,2255,2260,2265,2270,2275,2280,2285,2290,2295,2300,2305,2310,2315,2320,2325,2330,2335,2340,2345,2350,2355,2360,2365,2370,2375,2380,2385,2390,2395,2400,2405,2410,2415,2420,2425,2430,2435,2440,2445,2450,2455,2460,2465,2470,2475,2480,2485,2490,2495,2500,2505,2510,2515,2520,2525,2530,2535,2540,2545,2550,2555,2560,2565,2570,2575,2580,2585,2590,2595,2600,2605,2610,2615,2620,2625,2630,2635,2640,2645,2650,2655,2660,2665,2670,2675,2680,2685,2690,2695,2700,2705,2710,2715,2720,2725,2730,2735,2740,2745,2750,2755,2760,2765,2770,2775,2780,2785,2790,2795,2800,2805,2810,2815,2820,2825,2830,2835,2840,2845,2850,2855,2860,2865,2870,2875,2880,2885,2890,2895,2900,2905,2910,2915,2920,2925,2930,2935,2940,2945,2950,2955,2960,2965,2970,2975,2980,2985,2990,2995,3000,3005,3010,3015,3020,3025,3030,3035,3040,3045,3050,3055,3060,3065,3070,3075,3080,3085,3090,3095,3100,3105,3110,3115,3120,3125,3130,3135,3140,3145,3150,3155,3160,3165,3170,3175,3180,3185,3190,3195,3200,3205,3210,3215,3220,3225,3230,3235,3240,3245,3250,3255,3260,3265,3270,3275,3280,3285,3290,3295,3300,3305,3310,3315,3320,3325,3330,3335,3340,3345,3350,3355,3360,3365,3370,3375,3380,3385,3390,3395,3400,3405,3410,3415,3420,3425,3430,3435,3440,3445,3450,3455,3460,3465,3470,3475,3480,3485,3490,3495,3500,3505,3510,3515,3520,3525,3530,3535,3540,3545,3550,3555,3560,3565,3570,3575,3580,3585,3590,3595,3600,3605,3610,3615,3620,3625,3630,3635,3640,3645,3650,3655,3660,3665,3670,3675,3680,3685,3690,3695,3700,3705,3710,3715,3720,3725,3730,3735,3740,3745,3750,3755,3760,3765,3770,3775,3780,3785,3790,3795,3800,3805,3810,3815,3820,3825,3830,3835,3840,3845,3850,3855,3860,3865,3870,3875,3880,3885,3890,3895,3900,3905,3910,3915,3920,3925,3930,3935,3940,3945,3950,3955,3960,3965,3970,3975,3980,3985,3990,3995,4000,4005,4010,4015,4020,4025,4030,4035,4040,4045,4050,4055,4060,4065,4070,4075,4080,4085,4090,4095,4100,4105,4110,4115,4120,4125,4130,4135,4140,4145,4150,4155,4160,4165,4170,4175,4180,4185,4190,4195,4200,4205,4210,4215,4220,4225,4230,4235,4240,4245,4250,4255
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172. IF(IIFUEL.EQ.2) PPF = WINTP(TF,1,12,TF450,VAPASO)
173. IF(IIFUEL.EQ.3) PPF = WINTP(TF,1,11,TF450,VAPMMH)
174. PPD = WINTP(TO,1,13,TON204,VAPN20)
175. IF(IIFUEL.EQ.1) XMF = 32.0
176. IF(IIFUEL.EQ.2) XMF = 42.0
177. IF(IIFUEL.EQ.3) XMF = 46.0
178. XMRVP = PFC92.0*(TF460.) /PPF/XMF/(TC460.)
179. XF = PPF /PC
180. XD = PPU/PC
181. REGF = (.000373*PC*VF*DT)/(.000053*12.)
182. S9P = MEZ/(REGF*0.5)
183. MT = MU + MF
184. CSTAR = PC *XAT = 32.14 / MT
185. XP = (XP*XD)**.5
186. RESIDUE TIME IS DF/VF AND WILL BE CALLED RESID
187. RESID = DF/VF/12
188. REACT=RF*XP
189. THE DIMENSION OF REACT IS IN SECONDS
190. *****
191. IPUT2(IPAGE2,1) = IFUELN(IFUEL)
192. IPUT2(IPAGE2,3) = ICUMN(ICUM)
193. POUT2(IPAGE2,4) = PC
194. POUT2(IPAGE2,5) = VAVG
195. POUT2(IPAGE2,6) = MEF
196. POUT2(IPAGE2,7) = MEO
197. POUT2(IPAGE2,8) = REF
198. POUT2(IPAGE2,9) = NEQ
199. POUT2(IPAGE2,10) = DELT1
200. POUT2(IPAGE2,11) = HELF
201. POUT2(IPAGE2,12) = HELU
202. POUT2(IPAGE2,13) = PPF
203. POUT2(IPAGE2,14) = PPU
204. POUT2(IPAGE2,15) = XMRVP
205. POUT2(IPAGE2,16) = XF
206. POUT2(IPAGE2,17) = XD
207. POUT2(IPAGE2,18) = XP
208. POUT2(IPAGE2,19) = RESID
209. IF(PC .LE. 0.)WRITE(6,99) DIV,I,TEST
210. DIV = LE, 0.)WRITE(6,99) DIV,I,TEST
211. IF(PC .LE. 0.) GO TO 40
212. I = DELETED, TF(DIV,LE,0) GO TO 40
213. 88 FORMAT(5X,1VE 'F0.1', ITEST = '18',)
214. 89 FORMAT(5X,1DIV 'E10.5', ITEST = '14',)
215. THE FOLLOWING STATEMENT DETERMINES INJECTOR TO BE PLOTTED
216. IF(INJT.ME.NINJT) GO TO 10
217. IF(IIFUEL.EQ.1) N = 1
218. IF(IIFUEL.EQ.2) N = 2 FUEL USED IS MZM4
219. IF(IIFUEL.EQ.3) N = 3 FUEL USED IS A-50
220. IF(IIFUEL.EQ.3) N = 3 FUEL USED IS MMH
221. DELT=DF, IPUT2(GT,1) GO TO 40
222. IFUEL = IFUEL
223. IF(IPLUT .EQ. 0) GO TO 40
224. IF(ICUM.EQ.5) ICUM = 4
225. IF(ICUM.EQ.6) ICUM = 1
226. IF(ICUM.LT. 5)ICUM = ICUM
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M A I N (10)

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286. IF(INVEST.EQ.1) CALL SYMBOL(3.5,9.9,1,1,10INVESTIGATOR ZUNG,0.0,10)
287. IF(INVEST.EQ.2) CALL SYMBOL(3.5,9.9,1,1,20INVESTIGATOR NURICK,0.0,
288. *20)
289. IF(INVEST.EQ.3) CALL SYMBOL(3.5,9.9,1,1,22INVESTIGATOR HOUSEMAN,0.0,
290. *22)
291. IF(INVEST.EQ.4) CALL SYMBOL(3.5,9.9,1,1,20INVESTIGATOR LAWYER,0.0,
292. *20)
293. IF(INVEST.EQ.5) CALL SYMBOL(3.5,9.9,1,1,24INVESTIGATOR ROCKETOYNE,
294. *0.24)
295. IF(INVEST.EQ.6) CALL SYMBOL(3.5,9.9,1,1,20INVESTIGATOR QNS=75.0,
296. *20)
297. CALL SYMBOL(3.5,0.5,1,1,11INJECTOR = ,0.0,11)
298. CALL SYMBOL(999.999,1,1,INJECT(INIINT),0.0,5)
299. CALL SYMBOL(999.999,1,1,INJECT(INIINT+1),0.0,5)
300. CALL SYMBOL(999.999,1,1,INJECT(INIINT+2),0.0,5)
301. CALL SYMBOL(3.5,0.0,1,1,SHDP = ,0.0,5)
302. CALL NUMBER(999.999,1,1,2POF,0.0,3)
303. CALL SYMBOL(3.5,7.5,1,1,TFUEL = ,0.0,7)
304. IF(NPLOT.EQ.NFP) CALL SYMBOL(4.5,7.5,1,1,FUEL(NFP),0.0,4)
305. IF(NPLOT.EQ.1.AND.NFP.EQ.2) CALL SYMBOL(0.5,7.5,1,1,13H2M4 AND A=5
306. *0.0,13)
307. IF(NPLOT.EQ.1.AND.NFP.EQ.3) CALL SYMBOL(0.5,7.5,1,1,12H2M4 AND MMH
308. *0.0,12)
309. IF(NPLOT.EQ.2.AND.NFP.EQ.3) CALL SYMBOL(0.5,7.5,1,1,12MA=50 AND MMH
310. *0.0,12)
311. IF(NPLOT.EQ.1.AND.NFP.EQ.3) CALL SYMBOL(0.5,7.5,1,1,10HMMH, A=50,
312. *ND N2M4,0.0,19)
313. DO 601 K=NPLUT,NFP
314. THE ABOVE WAS, K = NPLUT,3
315. IF NPLUT = 3 (ONLY MMH FUEL PLOTTED)
316. IF NPLUT = 2 (BOTH MMH & A=50 FUELS PLOTTED)
317. IF NPLUT = 1 (MMH, A=50, & N2M4 FUELS PLOTTED)
318.
319.
320.
321.
322.
323. IF(K.EQ.1.AND.1.EQ.1) ISYMB = 10
324. IF(K.EQ.1.AND.1.EQ.2) ISYMB = 11
325. IF(K.EQ.1.AND.1.EQ.3) ISYMB = 6
326. IF(K.EQ.1.AND.1.EQ.4) ISYMB = 12
327. IF(K.EQ.2.AND.1.EQ.1) ISYMB = 1
328. IF(K.EQ.2.AND.1.EQ.1) ISYMB = 4
329. IF(K.EQ.2.AND.1.EQ.3) ISYMB = 7
330. IF(K.EQ.2.AND.1.EQ.4) ISYMB = 5
331. IF(K.EQ.3.AND.1.EQ.1) ISYMB = 0
332. IF(K.EQ.3.AND.1.EQ.2) ISYMB = 3
333. IF(K.EQ.3.AND.1.EQ.3) ISYMB = 2
334. IF(K.EQ.3.AND.1.EQ.4) ISYMB = 9
335. IF (IMSP(I,K).EQ.0) GO TO 601
336. CALL PCATA((1,1,K),Y(1,1,K),IMSP(1,K),1,1,ISYMB,XM,DX,1.0,2.0,
337. *0.05)
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PAGE 7

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343. 60 CONTINUE.
344. 50 CONTINUE
345. 1000 FORMAT(20A6)
346. 3000 FORMAT(//40X,20A6,/)
347. 5000 FORMAT(//120, 'PAGE 1 OF 2')
348. 5002 FORMAT(//120, 'PAGE 2 OF 2')
349. 4001 FORMAT(//140, 'HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION',
350. 1//754, 'INVESTIGATOR', A6, A2,
351. 2 //7144, 'IMP', /72, 'FUEL TEST',
352. 3LE PC VO VF TO TF INJECTOR DU OF L/O ANG
353. 4//72,
354. 4//TYPE NO. TYPE (IN) (IN) (OEG) (PSIA) (FT/S) (FT/
355. 5S) (F) (F) (SEC),/,)
356. 3001 FORMAT(12, A6, I3, 3A5, 2F5, 3, F5, 0, F6, 0, F7, 0, 2F7, 1, 2F6, 0, F6, 2, F6, 3, A7,
357. *E8, 3)
358. 7002 FORMAT(12, A6, I3, A6, 2F6, 9, 2F6, 1, 2E8, 3, F6, 0, F9, 2, E8, 2, 3F6, 1, 2F5, 2,
359. *E8, 2, I8, 2)
360. 4002 FORMAT(//140, 'HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION',
361. 1//754, 'INVESTIGATOR', A6, A2,
362. 2//72, 'FUEL TEST MODE', PC VAVG MEF MEU RCF MEO OE
363. 3LTI RELF RELO PPF PPO MRVP XF XO XP RESID, /
364. 4//72, 'TYPE NO. (PSIA)
365. 5) (PSIA)(PSIA),/,)
366. END

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PAGE 1

M I N T P

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1. FUNCTION MINTP(U,NU,NV,V1,V2)
2. DIMENSION V1(1),V2(1)
3. IF (NU-NV)2,13,13
4. 2 IF ((U-V1(NU))/(V1(NV)-V1(NU))) 3,13,4
5. 3 SPECIAL FLAG TO WARN OF EXCEEDING LOWER TABLE RANGE
6. 3 IF(ABS(U-U1) .LE. .00001) GO TO 13
7. U1=U
8. 13 MINTP=V2(NU)
9. RETURN
10. 4 IF((U-V1(NV))/(V1(NV)-V1(NU))) 6,15,5
11. 5 SPECIAL FLAG TO WARN OF EXCEEDING UPPER TABLE RANGE
12. 5 IF(ABS(U-U1) .LE. .00001) GO TO 15
13. U1=U
14. 15 MINTP=V2(NV)
15. RETURN
16. 6 N=NU+1
17. DO 7 I=N,NV
18. IF((U-V1(I))/(V1(NV)-V1(NU))) 6,6,7
19. 7 CONTINUE
20. GO TO 9
21. 8 J=1
22. MINTP = (U-V1(J))/(V1(I)-V1(J))
23. MINTP = V2(J)+V2(I)-V2(J))*MINTP
24. RETURN
25. 9 CONTINUE
26. STOP
27. END
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APPENDIX E
TASK III AND IV DATA SUMMARIES

List of Appendix E Symbols
Test Condition Log (Table E-I
Test Result Summary Impingement Data
Compilation (Table E-III)
List of Appendix E Data Sources

APPENDIX E LIST OF SYMBOLS

D_f, D_o	Fuel and oxidizer orifice diameter, in.
DELTI	Impingement point temperature rise, °F
Imp. Angle	Propellant Stream Impingement angle, °
L/D	Orifice length to diameter ratio
ME/MO	Fuel to oxidizer momentum ratio
MR	Oxidizer to fuel mixture ratio
MRVP	Vapor phase mixture ratio
P_c	Chamber pressure, psia
PFJ, POJ	Fuel and oxidizer manifold pressure, psia
PFT, POT	Fuel and oxidizer tank pressure, psia
PPF, PPO	Fuel and oxidizer partial pressures, psia
$\Delta PFJ, \Delta POJ$	Fuel and oxidizer injector pressure drop, psid
PN2	Nitrogen Gas Pressure, psia
NOZ	Chamber throat diameter, in.
REACT	Reactivity, $Re_f \times XP$
REF, REO	Fuel and oxidizer orifice Reynolds number based on dia.
RELF, RELO	Fuel and oxidizer orifice Reynolds no, based on length
RESID	Propellant stream contact time, sec.
TF (TFJ)	Fuel temperature, °F
TO (TOJ)	Oxidizer temperature, °F
VANG	Average of fuel and oxidizer injection velocity, ft/sec
V_f, V_o	Fuel and oxidizer injection velocity, ft/se

List of Symbols (cont.)

WEF, WEO	Fuel and oxidizer Weber no.
\dot{W}_F, \dot{W}_O	Fuel and oxidizer weight flow rate, lbm/sec
XF, XO	Fuel and oxidizer mole fraction
XP	Product of fuel and oxidizer mole fraction ** 1/2

C-2

HIGH PERFORMANCE N_2O_4 /AMINE ELEMENTS

Table E.1 TEST CONDITION LOG

Test No.	D_f	Fuel	P_c	MR	T_F	T_o	ΔP_f	Moze	P_{N_2} (PSIG)	P_{OT} (PSIG)	P_{FT} (PSIG)	DATE	f-Stop	FR/Rate (PPS)	Process	Light Meter W/2 N.D.F.	Remarks
9C-27-101	.020	MMH	300	1.6	Amb	Amb	100	0.330	1350	385	385	10/1/74	16	400	Normal	.6	Long Impingement Element
-102	.020	MMH	300	1.6	Amb	Amb	100	0.330	1350	380	390	10/1/74	11	400	Normal	18	Long Impingement Element
-103	.020	MMH	300	1.6	Amb	Amb	100	0.330	1350	385	385	10/1/74	22	400	Normal	18	Long Impingement Element
-104	.020	MMH	300	1.6	Amb	Amb	100	0.330	1350	385	385	10/1/74	3.3	8000	Normal	18	Long Impingement Element
-105	.020	MMH	500	1.6	Amb	Amb	100	0.265	1463	585	585	10/1/74	3.3	8000	1 Stop	18	Long Impingement Element
-106	.020	MMH	1000	1.6	Amb	Amb	100	0.187	1463	1085	1085	10/1/74	3.3	8000	Normal	18	Long Impingement Element Poloroid Filter Paper & Ground Glass Diffuser Installed Between Backlite & Window
-107	.020	MMH	300	1.6	Amb	Amb	100	0.330	1350	385	385	10/7/74	3	400	Normal	15	Long Impingement Element
-108	.020	MMH	250	1.6	Amb	Amb	100	0.330	1095	335	335	10/7/74	8	400	Normal	15	Long Impingement Element
-109	.020	MMH	200	1.6	Amb	Amb	100	0.432	1568	285	285	10/7/74	8	400	Normal	15	Long Impingement Element
-110	.020	MMH	150	1.6	Amb	Amb	100	0.432	1129	235	235	10/7/74	8	400	Normal	15	Long Impingement Element
-111	.020	MMH	100	1.6	Amb	Amb	100	0.563	1350	185	185	10/7/74	3.3	8000	Normal	15	Long Impingement Element
-112	.020	A-50	100	1.6	Amb	Amb	20	0.563	1533	105	105	10/8/74	3.3	8000	1 Stop	15	Long Impingement Element Moved camera to opposite window & added front lighting
-113	.020	A-50	100	1.6	Amb	Amb	100	0.563	1350	185	185	10/12/74	5.6	400	Normal	14	Long Impingement Element

Test No.	D _f	Fuel	P _C	MR	T _F	T _O	ΔP _f	Noz	P _{N2} (PSIG)	P _{OT} (PSIG)	P _{FT} (PSIG)	DATE	f-Stop	FR/Rate (PPS)	Process	Light Meter W/2 N.D.F.	REMARKS
OC-27-114	.020	A-50	150	1.6	Amb	Amb	100	0.432	1129	235	235	10/12/74	5.6	400	Normal	14	Long Impingement Element
-115	.020	A-50	200	1.6	Amb	Amb	100	0.432	1568	285	285	10/12/74	5.6	400	Normal	14	Long Impingement Element
-116	.020	A-50	250	1.6	Amb	Amb	100	0.330	1095	335	335	10/12/74	5.6	400	Normal	14	Long Impingement Element
-117	.020	A-50	300	1.6	Amb	Amb	20	0.330	1646	305	305	10/12/74	5.6	400	Normal	14	Long Impingement Element
-118	.020	A-50	300	1.6	Amb	Amb	100	0.330	1350	385	385	10/12/74	5.6	400	Normal	14	Long Impingement Element
-119	.020	A-50	500	1.6	Amb	Amb	100	0.265	1463	585	585	10/12/74	5.6	400	Normal	14	Long Impingement Element
-120	.020	A-50	1000	1.6	Amb	Amb	100	0.187	1463	1085	1085	10/12/74	5.6	400	Normal	14	Long Impingement Element
-121	.020	A-50	150	1.6	Amb	Amb	100	0.432	1129	235	235	10/12/74	5.6	400	Normal	14	Rerun of #114 Reduced window purge from 10X to 1X
-122	.020	A-50	300	1.6	Amb	Amb	100	0.187	360	385	385	10/12/74	3.3	4000	Normal	14	Long Impingement Element
-123	.020	A-50	300	1.6	Amb	Amb	100	0.187	360	385	385	10/12/74	3.3	4000	Normal	14	Repeat of 122
-124	.020	HHH	1000	1.6	Amb	Amb	100	0.187	1463	1085	1085	10/16/74	5.6	400	Push 1 Stop	14	Increased purge flow to 10X Short Impingement Element
-125	.020	HHH	500	1.6	Amb	Amb	100	0.265	1463	585	585	10/16/74	5.6	400	Push 1 Stop	14	Short Impingement Element
-126	.020	HHH	300	1.6	Amb	Amb	100	0.330	1350	385	385	10/16/74	5.6	400	Push 1 Stop	14	Short Impingement Element
-127	.020	HHH	300	1.6	Amb	Amb	20	0.330	1646	305	305	10/16/74	5.6	400	Push 1 Stop	14	Short Impingement Element
-128	.020	HHH	250	1.6	Amb	Amb	100	0.330	1095	335	335	10/16/74	5.6	400	Push 1 Stop	14	Short Impingement Element
-129	.020	HHH	200	1.6	Amb	Amb	100	0.432	1568	285	285	10/16/74	5.6	400	Push 1 Stop	14	Short Impingement Element
-130	.020	HHH	150	1.6	Amb	Amb	100	0.432	1129	235	235	10/16/74	5.6	400	Push 1 Stop	14	Short Impingement Element

Test No.	D _f	Fuel	P _c	MR	T _F	T _O	ΔP _f	Noz	P _{N2} (PSIG)	P _{OT} (PSIG)	P _{FT} (PSIG)	DATE	f-Stop	FR/Rate (PPS)	Process	W/2 N.D.F.	REMARKS
OC-27-131	.020	MMH	100	1.6	Amb	Amb	100	0.563	1350	185	185	10/16/74	5.6	400	Push 1 Stop	14	Short Impingement Element
-132	.020	MMH	125	1.6	Amb	Amb	100	0.563	1732	210	210	10/16/74	3.3	4000	Push 1 Stop	14	Short Impingement Element
-133	.020	MMH	500	1.6	Amb	Amb	100	0.265	1463	585	585	10/17/74	Still Photo				Long Impingement Element
-134	.020	MMH	100	1.6	200	Amb	100	0.563	1350	185	185	10/23/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-135	.020	MMH	125	1.6	200	Amb	100	0.563	1923	210	210	10/23/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-136	.020	MMH	150	1.6	200	Amb	100	0.432	1129	235	235	10/23/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-137	.020	MMH	200	1.6	200	Amb	100	0.432	1568	285	285	10/23/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-138	.020	MMH	250	1.6	200	Amb	100	0.330	1095	335	335	10/23/74	5.6	400	Push 1 Stop	14	Long Impingement Element Heater Pump Seal Half installed T/C @ Imping. Pt.
-139	.020	MMH	100	1.6	Amb	Amb	20	0.563	2110	101	105	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-140	.020	MMH	100	1.6	Amb	Amb	100	0.563	1350	181	185	10/25/74	5.6	400	Push 1 Stop	14	Changed T/C from 0.01" Dia. to 0.02" Dia. prior to test 142
-141	.020	MMH	100	1.6	Amb	Amb	100	0.563	1350	181	185	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-142	.020	MMH	125	1.6	Amb	Amb	100	0.563	1923	206	210	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-143	.020	MMH	125	1.6	Amb	Amb	20	0.432	1516	126	130	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-144	.020	MMH	150	1.6	Amb	Amb	20	0.432	1839	151	155	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-145	.020	MMH	150	1.6	Amb	Amb	100	0.432	1129	231	235	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-146	.020	MMH	200	1.6	Amb	Amb	100	0.432	1568	281	285	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element

Test No.	D _f	Fuel	P _c	PR	T _F	T _O	ΔP _f	Noz	(PSIG)	(PSIG)	(PSIG)	DATE	f-Stop	FR/Rate (PPS)	Process	W/2 N.D.F.	REMARKS
OC-27-147	.020	MMH	250	1.6	Amb	Amb	100	0.330	1095	331	335	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-148	.020	MMH	100	1.6	Amb	Amb	20	0.563	2110	101	105	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-149	.020	MMH	100	1.6	Amb	Amb	100	0.563	1350	181	185	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-150	.020	MMH	300	1.6	Amb	Amb	100	0.330	1350	381	385	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-151	.020	MMH	500	1.6	Amb	Amb	100	0.265	1463	581	585	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element
-152	.020	MMH	1000	1.6	Amb	Amb	100	0.187	1463	1081	1085	10/25/74	5.6	400	Push 1 Stop	14	Long Impingement Element

HIGH PERFORMANCE N₂O₄/AMINE ELEMENTS

Table E-I - TEST CONDITION LOG

Test No.	D _f	Fuel	P _c	MR	T _F	T _O	ΔP _f	N ₂	P _{N₂}	P _{OT}	P _{FT}	Date	f-Stop	FR/Rate (PPS)	Process.	Light Meter W/2 N.D.F.	Film Roll No.	Remarks
OC-27-153	0.020	MMH	100	1.6	Amb.	Amb.	10	0.432	1238	94	95	11/22/74	5.6	400	Normal	14-1/2	20	Added Lens Tube to In- crease Magni- fication
154	0.020	MMH	100	1.6	Amb.	Amb.	10	0.432	1238	94	91	11/22/74	5.6	400	Normal	14-1/2		
155	0.020	MMH	100	1.6	Amb.	Amb.	20	0.563	2110	104	102	11/22/74	5.6	400	Normal	14-1/2		
156	0.020	MMH	100	1.6	Amb.	Amb.	40	0.563	2000	123	125	11/22/74	5.6	400	Normal	14-1/2		
157	0.020	MMH	100	1.6	Amb.	Amb.	40	0.563	1900	123	123	11/22/74	5.6	400	Normal	14-1/2		
158	0.020	MMH	100	1.6	Amb.	Amb.	10	0.432	1238	94	91	11/25/74	5.6	400	Normal	14-1/2	21	
159	0.020	MMH	100	1.6	Amb.	Amb.	10	0.432	1238	94	94	11/25/74	5.6	400	Normal	14-1/2		
160	0.020	MMH	100	1.6	Amb.	Amb.	20	0.563	2110	104	104	11/25/74	5.6	400	Normal	14-1/2		
161	0.020	MMH	100	1.6	Amb.	Amb.	20	0.563	2110	104	107	11/25/74	5.6	400	Normal	14-1/2		
162	0.020	MMH	100	1.6	Amb.	Amb.	20	0.563	2110	107	103	11/25/74	5.6	400	Normal	14-1/2		
163	0.020	MMH	100	1.6	Amb.	Amb.	40	0.563	1800	125	125	11/25/74	5.6	400	Normal	14-1/2	22	
164	0.020	MMH	100	1.6	Amb.	Amb.	30	0.563	1900	115	115	11/25/74	5.6	400	Normal	14-1/2		
165	0.020	MMH	100	1.6	Amb.	Amb.	60	0.563	1600	145	145	11/25/74	5.6	400	Normal	14-1/2		
166	0.020	MMH	100	1.6	Amb.	Amb.	100	0.563	1350	185	185	11/25/74	5.6	400	Normal	14-1/2		
167	0.020	MMH	100	1.6	Amb.	Amb.	30	0.563	1900	115	115	12/6/74	3.3	4000	Push 1 Stop	14-1/2	23	
168	0.020	MMH	100	1.6	Amb.	Amb.	30	0.563	1900	115	115	12/6/74	3.3	4000	Push 1 Stop	14-1/2	24	
169	0.020	MMH	100	1.6	Amb.	Amb.	30	0.563	1900	115	115	12/9/74	-	-	-	-		Repeat of #168 - No Film
170	0.020	MMH	100	1.6	Amb.	Amb.	150	0.563	1100	235	235	12/9/74	3.3	4000	Push 1 Stop	14-1/2	25	
171	0.020	MMH	100	1.6	Amb.	Amb.	150	0.563	1264	235	235	12/9/74	5.0	400	Normal	14-1/2	26	Repeat of #170
172	0.020	MMH	200	1.6	Amb.	Amb.	20	0.330	1268	205	205	12/9/74	5.6	400	Normal	14-1/2	26	
173	0.020	MMH	200	1.6	Amb.	Amb.	20	0.330	1208	205	210	12/9/74	5.6	400	Normal	14-1/2	26	Repeat of -172

Test No.	n_f	Fuel	P_c	MR	T_F	T_o	ΔP_f	P_{N_2} (psig)	P_{CT} (psig)	P_{FT} (psig)	Date	f-Stop	FR/Rate (PPS)	Process.	Light Meter W/2 N.D.F.	Film Roll No.	Remarks
OC-27-174	0.020	MMH	200	1.6	Amb.	Amb.	40	0.330 1184	220	225	12/9/74	5.6	400	Normal	14-1/2	26	
175	0.020	MMH	200	1.6	Amb.	Amb.	60	0.432 1896	240	245	12/9/74	5.6	400	Normal	14-1/2	26	
176	0.020	MMH	100	1.6	Amb.	Amb.	30	0.563 1900	115	116	12/13/74	3.3	4000	Push 1 Stop	14-1/2	27	Changed Lense to 75 MM Repeat of 169
177	0.020	MMH	100	1.6	Amb.	Amb.	30	0.563 1900	115	116	12/13/74	3.3	10,000	Normal	14-1/2	28	Changed Shut- ter to 1/20th
178	0.020	MMH	100	1.6	Amb.	Amb.	30	0.563 1900	115	116	12/13/74	3.3	10,000	Normal	14-1/2	29	Repeat of 177
179	0.020	MMH	100	1.6	Amb.	Amb.	30	0.563 1900	115	116	12/17/74	5.6	400	Normal	15	30	
180	0.020	MMH	200	1.6	Amb.	Amb.	30	0.330 1200	215	215	12/17/74	5.6	400	Normal	15	30	
181	0.020	MMH	300	1.6	Amb.	Amb.	30	0.330 1850	310	315	12/17/74	5.6	400	Normal	15	30	
182	0.020	MMH	300	1.6	Amb.	Amb.	30	0.330 700	310	315	12/17/74	5.6	400	Normal	15	30	
183	0.020	MMH	300	1.6	Amb.	Amb.	30	0.330 1500	310	315	12/17/74	5.6	400	Normal	15	30	
184	0.020	MMH	300	1.6	Amb.	Amb.	30	0.265 930	310	315	12/18/74	5.6	400	Normal	15	31	
185	0.020	MMH	300	1.6	Amb.	Amb.	30	0.265 1300	310	315	12/18/74	5.6	400	Normal	15	31	
186	0.020	MMH	300	1.6	Amb.	Amb.	30	0.130 0.0	280	285	12/18/74	5.6	400	Normal	15	31	
187	0.020	MMH	300	1.6	Amb.	Amb.	20	0.330 1942	304	305	12/18/74	5.6	400	Normal	15	31	
188	0.020	MMH	300	1.6	Amb.	Amb.	40	0.330 1783	320	325	12/18/74	5.6	400	Normal	15	31	
189	0.020	MMH	300	1.6	Amb.	Amb.	100	0.330 1350	380	385	12/18/74	5.6	400	Normal	15	32	
190	0.020	MMH	100	1.6	100	Amb.	30	0.563 1900	114	116	12/18/74	5.6	400	Normal	15	32	
191	0.020	MMH	100	1.6	100	Amb.	30	0.563 1900	114	116	12/18/74	5.6	400	Normal	15	32	

Test No.	D _f	Fuel	P _c	MR	T _F	T _O	ΔP _f	Noz	P _{N₂} (psig)	P _{OT} (psig)	P _{FT} (psig)	Date	f-Stop	FR/Rate (PPS)	Process.	Light Meter W/2 N.D.F.	Film Roll No.	Remarks
OC-27-192	0.020	MMH	100	1.6	150	Amb.	30	0.563	1900	114	116	12/18/74	5.6	400	Normal	15	32	
193	0.020	MMH	100	1.6	175	Amb.	30	0.563	1900	114	116	12/18/74	5.6	400	Normal	15	32	
194	0.020	MMH	100	1.6	200	Amb.	30	0.563	1900	114	116	12/18/74	5.6	400	Normal	15	33	
195	0.020	MMH	200	1.6	200	Amb.	30	0.330	1200	214	215	12/18/74	5.6	400	Normal	15	33	
196	0.020	MMH	200	1.6	175	125	30	0.330	1200	214	215	12/18/74	5.6	400	Normal	15	33	
197	0.020	MMH	200	1.6	Amb.	150	30	0.330	1200	214	215	12/18/74	5.6	400	Normal	15	33	
198	0.020	A-50	100	1.65	Amb	Amb	10	0.166	N/A	94	95	1/17/75	4	800	Normal	13	34	Changed Shut- ter to 1/50- no purge
199	0.020	A-50	100	1.65	Amb	Amb	20	0.196	N/A	104	105	1/17/75	4	800	Normal	13	34	Film Broke
200	0.020	A-50	100	1.65	Amb	Amb	20	0.196	N/A	105	105	1/20/75	4	800	Normal	13	35	Repeat of #199
201	0.020	A-50	100	1.65	Amb	Amb	30	0.213	N/A	115	115	1/20/75	4	800	Normal	13	35	
202	0.020	A-50	100	1.65	Amb	Amb	60	0.257	N/A	145	145	1/20/75	4	800	Normal	13	35	
203	0.020	A-50	100	1.65	Amb	Amb	100	0.290	N/A	185	185	1/20/75	4	800	Normal	13	35	
204	0.020	A-50	100	1.65	Amb	Amb	150	0.333	N/A	235	235	1/20/75	4	800	Normal	13	35	
205	0.020	A-50	200	1.65	Amb	Amb	20	0.129	N/A	205	205	1/20/75	4	800	Normal	13	36	
206	0.020	A-50	200	1.65	Amb	Amb	30	0.150	N/A	215	215	1/20/75	3.3	800	Normal	13	36	
207	0.020	A-50	200	1.65	Amb	Amb	150	0.234	N/A	335	335	1/20/75	4	800	Normal	13	36	
208	0.020	A-50	300	1.65	Amb	Amb	20	0.113	N/A	305	305	1/20/75	4	800	Normal	13	36	T/C Probe Malfunction
209	0.020	A-50	300	1.65	Amb	Amb	20	0.113	N/A	305	305	1/20/75	3.3	800	Normal	N/A	36	Strobe Light on Bottom
210	0.020	A-50	300	1.65	Amb	Amb	20	0.113	N/A	305	305	1/20/75	4.0	800	Normal	N/A	36	Strobe Light in back

Test No.	D _f	Fuel	P _c	MR	T _F	T _O	ΔP _f	P _{N₂} (psig)	P _{O₂} (psig)	P _{FT} (psig)	Date	f-Stop	FR/Rate (pps)	Process.	Light Meter W/2 N.D.F.	Film Roll No.	Remarks	
OC-27-211	0.020	A-50	300	1.65	Amb	Amb	20	0.113	N/A	305	305	1/21/75	4.0	800	Normal	13	37	
	0.020	A-50	300	1.65	Amb	Amb	30	0.125	N/A	315	315	1/21/75	4.0	800	Normal	13	37	T/C shows bi-level mode
213	0.020	A-50	300	1.65	Amb	Amb	60	0.150	N/A	345	345	1/21/75	4.0	800	Normal	13	37	
214	0.020	A-50	300	1.65	Amb	Amb	100	0.166	N/A	385	385	1/21/75	4.0	800	Normal	13	37	
215	0.020	A-50	300	1.65	Amb	Amb	150	0.189	N/A	435	435	1/21/75	4.0	800	Normal	13	37	
216	0.020	A-50	500	1.65	Amb	Amb	60	0.113	N/A	545	545	1/21/75	4.0	800	Normal	13	38	
217	0.020	A-50	500	1.65	Amb	Amb	100	0.129	N/A	585	585	1/21/75	4.0	800	Normal	13	38	
218	0.021	MHH	125	1.65	Amb	Amb	50	0.166	N/A	150	160	1/31/75	4.0	800	Normal	13	39	XDT-1 Plate-let inj.
219	0.021	MHH	125	1.65	Amb	Amb	50	0.180	N/A	160	160	1/31/75	4.0	800	Normal	13	39	
220	0.021	MHH	100	1.65	Amb	Amb	30	0.189	N/A	116	116	1/31/75	4.0	800	Normal	13	39	
221	0.021	MHH	80	1.65	Amb	Amb	20	0.189	N/A	85	85	1/31/75	4.0	800	Normal	13	39	
222	0.021	MHH	150	1.65	Amb	Amb	70	0.189	N/A	206	206	1/31/75	4.0	800	Normal	13	39	
223	0.021	MHH	200	1.65	Amb	Amb	125	0.189	N/A	311	311	1/31/75	4.0	800	Normal	13	39	
224	0.021	MHH	125	1.65	170	Amb	50	0.189	N/A	160	160	2/4/75	3.3	800	Normal	13	40	XDT-1
225	0.021	MHH	125	1.65	170	Amb	50	0.189	N/A	160	160	2/4/75	3.3	800	Normal	13	40	Repeat of #224
226	0.021	MHH	125	1.65	240	Amb	50	0.189	N/A	160	160	2/4/75	3.3	800	Normal	13	40	
227	0.021	MHH	125	1.65	250	125	50	0.189	N/A	160	160	2/4/75	3.3	800	Normal	13	40	Windows not cleaned - T ₁ Malf.
228	0.021	MHH	150	1.65	250	125	50	0.189	N/A	206	206	2/4/75	3.3	800	Normal	13	40	Windows not cleaned
229	0.021	MHH	125	1.65	290	125	50	0.189	N/A	160	160	2/6/75	4.0	800	Normal	13	41	XDT-1
230	0.021	MHH	125	1.65	290	125	50	0.189	N/A	160	160	2/6/75	4.0	800	Normal	13	41	
231	0.021	MHH	150	1.65	290	125	50	0.189	N/A	160	160	2/6/75	4.0	800	Normal	13	41	Windows not cleaned

Test No.	D _f	Fuel	P _c	MR	T _F	T _O	ΔP _f	P _{N₂} (psia)	P _{OT} (psig)	P _{FT} (psig)	Date	r-Stop	FR/Rate (FPS)	Process.	Light Meter w/2 N.O.F.	Film Roll No	Remarks
OC-27-232	0.021	MMH	150	1.65	290	150	50	0.189	N/A	160	160	2/6/75	4.0	800	Normal	13	41 Windows not cleaned
	233 0.021	MMH	150	1.65	290	150	50	0.189	N/A	160	160	2/6/75	4.0	800	Normal	13	41 Windows not cleaned
	234 0.021	MMH	125	1.65	Amb	Amb	65	0.189	N/A	175	175	2/6/75	4.0	800	Normal	13	42 Splash Plate - Spikes
	235 0.021	MMH	100	1.65	Amb	Amb	42	0.189	N/A	127	127	2/6/75	4.0	800	Normal	13	42 Instru. Malf. Spikes
	236 0.021	MMH	100	1.65	Amb	Amb	42	0.189		127	127	2/6/75	4.0	800	Normal	13	42 Repeat of #235
	237 0.021	MMH	80	1.65	Amb	Amb	27	0.189	N/A	92	92	2/6/75	4.0	800	Normal	13	42
	238 0.021	MMH	150	1.65	Amb	Amb	95	0.189	N/A	230	230	2/6/75	4.0	800	Normal	13	42 T ₁ Malf.
	239 0.021	MMH	200	1.65	Amb	Amb	168	0.189	N/A	353	353	2/6/75	4.0	800	Normal	13	42 Window not cleaned
	240 0.021	MMH	150	1.65	Amb	Amb	95	0.189	N/A	230	230	2/10/75	4.0	800	Normal	13	43 Manifold & Pc Spikes
	241 0.021	MMH	150	1.65	Amb	Amb		0.189	N/A	230	230	2/10/75	5.6	800	Normal	13	43 No Spikes
	242 0.021	MMH	150	1.65	Amb	Amb	95	0.189	N/A	230	230	2/10/75	8.0	800	Normal	13	43
	243 0.021	MMH	200	1.65	Amb	Amb	168	0.189	N/A	353	353	2/10/75	4.0	800	Normal	13	43
	244 0.021	MMH	125	1.65	170	Amb	65	0.189	N/A	75	75	2/10/75	4.0	800	Normal	13	43 Spike on ox manifold
	245 0.021	MMH	125	1.65	240	Amb	65	0.189	N/A	175	175	2/10/75	4.0	800	Normal	13	44
	246 0.021	MMH	125	1.65	290	125	65	0.189	N/A	175	175	2/10/75	4.0	800	Normal	13	44
	247 0.021	MMH	150	1.65	290	125	95	0.189	N/A	230	230	2/10/75	4.0	800	Normal	13	44 Spikes in manifolds
	248 0.021	MMH	150	1.65	290	150	95	0.189	N/A	230	230	2/10/75	4.0	800	Normal	13	43 Spikes in manifolds
	249 0.021	MMH	150	1.65	290	150	95	0.189	N/A	230	230	2/10/75	8.0	800	Normal	13	43 Spikes in manifolds

Test No.	U_f	Fuel	P_c	MR	T_F	T_o	ΔP_f	Noz	P_{N_2} (psig)	P_{OT} (psig)	P_{FT} (psig)	Date	f-Stop	FR/Rate (pps)	Process.	Light Meter W/2 N.D.F.	Film Roll No.	Remarks
OC-27-250	0.0295	HH	125	1.65	Amb	Amb	45	0.365	N/A	160	155	2/14/75	4.0	800	Normal	13	45	Triplet-No O-Graph
251	0.0295	HH	125	1.65	Amb	Amb	45	0.365	N/A	160	155	2/14/75	4.0	800	Normal	13	45	Repeat of 250
252	0.0295	HH	100	1.65	Amb	Amb	27	0.365	N/A	114	112	2/14/75	4.0	800	Normal	13	45	
253	0.0295	HH	80	1.65	Amb	Amb	17	0.365	N/A	84	82	2/14/75	4.0	800	Normal	13	45	
254	0.0295	HH	150	1.65	Amb	Amb	61	0.365	N/A	202	196	2/14/75	4.0	800	Normal	13	45	
255	0.0295	HH	200	1.65	Amb	Amb	108	0.365	N/A	303	293	2/14/75	4.0	800	Normal	13	45	
256	0.0295	HH	125	1.65	170	Amb	45	0.365	N/A	160	155	2/14/75	4.0	800	Normal	13	46	
257	0.0295	HH	125	1.65	240	Amb	45	0.365	N/A	160	155	2/14/75	4.0	800	Normal	13	46	
258	0.0295	HH	125	1.65	290	125	45	0.365	N/A	160	155	2/14/75	4.0	800	Normal	13	46	
259	0.0295	HH	150	1.65	290	125	61	0.365	N/A	202	196	2/14/75	4.0	800	Normal	13	46	
260	0.0295	HH	150	1.65	290	150	61	0.365	N/A	202	196	2/14/75	4.0	800	Normal	13	46	
261	0.020	Water	14.7	1.65	Amb	Amb	10	N/A	200	20	10	2/21/75	4.0	800	Normal	13	47	Water/Freon Cold Flow
262	0.020	Water	14.7	1.65	Amb	Amb	20	N/A	200	20	20	2/21/75	4.0	800	Normal	13	47	with unlike doublet
263	0.020	Water	14.7	1.65	Amb	Amb	40	N/A	200	40	40	2/21/75	4.0	800	Normal	13	48	
264	0.020	Water	14.7	1.65	Amb	Amb	60	N/A	200	60	60	2/21/75	4.0	800	Normal	13	48	
265	0.020	Water	14.7	1.65	Amb	Amb	80	N/A	200	80	80	2/21/75	4.0	800	Normal	13	48	
266	0.020	Water	14.7	1.65	Amb	Amb	100	N/A	200	100	100	2/21/75	4.0	800	Normal	13	48	
267	0.020	Water	14.7	1.65	Amb	Amb	150	N/A	200	150	150	2/21/75	4.0	800	Normal	13	48	
268	0.020	Water	14.7	1.65	Amb	Amb	200	N/A	200	200	200	2/21/75	4.0	800	Normal	13	48	
269	0.020	Water	14.7	1.65	Amb	Amb	20	N/A	200	20	20	2/21/75	4.0	2000	Normal	N/A	49	Changed Shut- ter to 1/10th
270	0.020	Water	14.7	1.65	Amb	Amb	40	N/A	200	40	40	2/21/75	4.0	2000	Normal	N/A	49	with strobe backlight
271	0.020	Water	14.7	1.65	Amb	Amb	60	N/A	200	60	60	2/21/75	4.0	2000	Normal	N/A	49	
272	0.020	Water	14.7	1.65	Amb	Amb	100	N/A	200	100	100	2/21/75	4.0	2000	Normal	N/A	50	
273	0.020	Water	14.7	1.65	Amb	Amb	150	N/A	200	150	150	2/21/75	4.0	2000	Normal	N/A	50	
274	0.020	Water	14.7	1.65	Amb	Amb	100	N/A	200	100	100	2/21/75	3.3	2000	Normal	N/A	50	

Test No.	D _f	Fuel	P _c	MR	T _F	T _O	ΔP _f	Noz	P _{N₂} (psig)	P _{OT} (psig)	P _{FY} (psig)	Date	f-Stop	rR/Rate (PPS)	Process.	Light Meter W/2 N.D.F.	Film Roll No.	Remarks
OC-27-275	0.020	HH	125	1.65	70	70	50	0.269	N/A	160	160	2/27/75	4.00	800	Normal	13	51	Small Dia. Triplet
276	0.020	HH	125	1.65	70	70	50	0.269	N/A	156	162	2/27/75	4.00	800	Normal	13	51	Repeat of 275
277	0.020	HH	100	1.65	70	70	30	0.269	N/A	113	115	2/27/75	4.00	800	Normal	13	51	
278	0.020	HH	80	1.65	70	70	20	0.269	N/A	84	85	2/27/75	4.00	800	Normal	13	51	
279	0.020	HH	150	1.65	70	70	70	0.269	N/A	200	209	2/27/75	4.00	800	Normal	13	51	
280	0.020	HH	200	1.65	70	70	125	0.269	N/A	300	310	2/27/75	4.00	800	Normal	13	51	
281	0.020	HH	125	1.65	170	70	50	0.269	N/A	156	162	2/27/75	4.00	800	Normal	13	52	
282	0.020	HH	125	1.65	240	70	50	0.269	N/A	156	162	2/27/75	4.00	800	Normal	13	52	
283	0.020	HH	125	1.65	290	125	50	0.259	N/A	156	162	2/27/75	4.00	800	Normal	13	52	Repeat of 281
284	0.020	HH	150	1.65	290	125	70	0.269	N/A	200	210	2/27/75	4.00	800	Normal	13	52	
285	0.020	HH	150	1.65	290	150	70	0.269	N/A	200	210	2/27/75	4.00	800	Normal	13	52	
286	0.020	HH	125	1.65	Amb	Amb	30	0.189	N/A	140	140	3/4/75	4.00	800	Normal	13	53	Unlike Doub.
288	0.020	HH	100	1.65	Amb	Amb	18	0.189	N/A	103	103	3/4/75	4.00	800	Normal	13	53	
289	0.020	HH	80	1.65	Amb	Amb	12	0.189	N/A	77	77	3/4/75	4.00	800	Normal	13	53	
290	0.020	HH	80	1.65	Amb	Amb	12	0.189	N/A	77	77	3/4/75	4.00	800	Normal	13	53	Repeat of 289
291	0.020	HH	150	1.65	Amb	Amb	40	0.189	N/A	175	175	3/4/75	4.00	800	Normal	13	53	
292	0.020	HH	200	1.65	Amb	Amb	75	0.189	N/A	260	260	3/4/75	4.00	800	Normal	13	53	
293	0.020	HH	125	65	170	Amb	30	0.189	N/A	140	140	3/4/75	4.00	800	Normal	13	54	
294	0.020	HH	125	1.65	240	Amb	30	0.189	N/A	140	140	3/4/75	4.00	800	Normal	13	54	
295	0.020	HH	125	1.65	240	125	30	0.189	N/A	140	140	3/4/75	4.00	800	Normal	13	54	
296	0.020	HH	150	1.65	290	125	40	0.189	N/A	175	175	3/4/75	4.00	800	Normal	13	54	
297	0.020	HH	150	1.65	290	150	40	0.189	N/A	175	175	3/4/75	4.00	800	Normal	13	54	

HIGH PERFORMANCE N_2O_4 /AMINE ELEMENTS

TABLE E-II- TEST RESULT SUMMARY

Test No.	P_c (psia)	P_{OJ} (psia)	P_{FJ} (psia)	ΔP_{OJ} (psi)	ΔP_{FJ} (psi)	\dot{W}_o (lb/sec)	\dot{W}_f (lb/sec)	T_{OJ} °F	T_{FJ} °F	MR
OC-27-101	308	406	401	98	93	.0252	.0152	88	87	1.66
-102	308	400	406	92	98	.0245	.0156	88	87	1.57
-103	309	404	404	95	95	.0249	.0153	88	87	1.62
-104	311	404	404	93	93	.0246	.0152	88	87	1.62
-105	507	605	595	98	88	.0252	.0142	89	88	1.70
-106	1000	1101	1108	101	108	.0256	.0163	89	88	1.57
-107	308	404	395	96	87	.0250	.0147	86	85	1.7
-108	263	353	350	90	87	.0242	.0147	84	84	1.65
-109	197	305	302	108	105	.0265	.0161	84	84	1.64
-110	158	251	253	93	95	.0246	.0153	83	83	1.60
-111	100	202	204	102	104	.0258	.0160	84	83	1.60
-112	89	121	122	32	33	.0144	.00905	76	76	1.59
-113	101	201	197	100	97	.0255	.0156	88	87	1.63
-114	162	250	245	88	3	.0239	.0144	82	82	1.66
-115	203	303	298	100	95	.0255	.0154	81	78	1.65
-116	269	353	347	84	78	.0234	.0140	80	77	1.67
-117	301	347	345	46	44	.0173	.0105	78	76	1.64
-118	310	401	401	91	91	.0243	.0157	81	81	1.60
-119	510	599	596	89	86	.0240	.0147	82	82	1.64
-120	1003	1096	1093	93	90	.0246	.0150	82	81	1.63

Test No.	P _c (psia)	P _{OJ} (psia)	P _{FJ} (psia)	ΔP _{OJ} (psi)	ΔP _{FJ} (psi)	W _o (lb/sec)	W _f (lb/sec)	T _{OJ} °F	T _{FJ} °F	MR
OC-27-121	160	255	254	95	94	.0249	.0153	81	80	1.62
-122	338	402	399	64	61	.0204	.0123	81	80	1.65
-123	334	403	400	69	66	.0212	.0128	77	77	1.65
-124	955	1016	1021	66	61	.0195	.0120	82	82	1.61
-125	495	601	595	106	100	.0248	.0154	83	81	1.60
-126	298	404	400	106	102	.0248	.0156	83	82	1.59
-127	283	322	321	39	38	.00950	.00952	84	81	1.58
-128	256	353	350	97	94	.0237	.0149	85	83	1.58
-129	191	304	299	113	108	.0256	.0160	85	83	1.60
-130	152	252	251	100	99	.0240	.0153	85	82	1.56
-131	95	201	199	106	104	.0248	.0157	85	83	1.57
-132	114	226	223	112	109	.0255	.0161	86	83	1.58
-133	411	602	595	191	184	.0352	.0212	83	83	1.66
-134	97	202	197	105	100	.0261	.0150	77	197	1.74
-135	119	226	223	107	104	.0264	.0153	78	199	1.72
-136	150	254	249	104	99	.0260	.0150	77	199	1.74
-137	188	304	298	116	110	.0275	.0158	76	199	1.74
-138	243	349	353	106	110	.0262	.0164	79	195	1.60
-139	103	119	123	16	20	.0102	.0070	61	59	1.455
-140	81	199	201	118	120	.0277	.0171	64	62	1.61
-141	81	197	201	116	120	.0275	.0171	66	64	1.60
-142	119	222	227	103	108	.0259	.0163	68	67	1.59
-143	129	142	146	13	17	.00920	.00646	67	67	1.42
-144	154	170	172	16	18	.0102	.00664	67	67	1.53
-145	152	248	248	96	96	.0250	.0153	68	67	1.62

Table E-II (cont.)

Test No.	P_c (psia)	P_{OJ} (psia)	P_{FJ} (psia)	ΔP_{OJ} (psi)	ΔP_{FJ} (psi)	\dot{W}_o (lb/sec)	\dot{W}_f (lb/sec)	T_{OJ} °F	T_{FJ} °F	MR
OC-27-146	190	298	298	108	108	.0256	.0162	68	67	1.57
-147	258	347	348	89	90	.0240	.0148	68	67	1.61
-148	102	119	122	17	20	.0105	.00700	68	68	1.49
-149	99	198	202	99	103	.0254	.0159	69	69	1.60
-150	297	397	398	100	101	.0255	.0157	70	69	1.52
-151	484	596	594	112	110	.0270	.0164	70	69	1.64
-152	958	1084	1085	126	127	.0286	.0176	78	74	1.62

Table E-II (cont.)

Test No.	P _c (psia)	P _{OJ} (psia)	P _{FJ} (psia)	ΔP _{OJ} (psi)	ΔP _{FJ} (psi)	W _o (lb/sec)	W _f (lb/sec)	T _{OJ} (°F)	T _{FJ} (°F)	MR
0C-27-153	98.3	108.9	113.0	10.6	14.7	0.00831	0.00604	55	56	1.38
154	99.5	110.4	109.7	10.9	10.2	0.00843	0.00503	54	55	1.68
155	102.0	119.1	117.4	17.0	15.4	0.01056	0.00617	54	54	1.72
156	106.8	137.4	141.3	30.6	34.5	0.01416	0.00926	53	54	1.53
157	101.7	137.7	138.1	36.1	36.5	0.01537	0.00951	53	54	1.62
158	100.6	111.1	108.0	10.5	7.4	0.00826	0.00427	58	58	1.94
159	98.7	110.2	109.3	11.5	10.7	0.00868	0.00515	57	58	1.69
160	100.8	120.0	119.4	19.2	18.5	0.01121	0.00678	57	58	1.66
161	101.8	119.7	121.4	18.0	19.6	0.01083	0.00698	58	58	1.55
162	102.4	123.8	119.2	21.5	16.9	0.01184	0.00647	57	58	1.83
163	98.3	141.1	140.1	42.9	41.9	0.01674	0.01020	58	58	1.65
164	100.1	131.6	130.2	31.5	30.1	0.01434	0.00864	56	57	1.66
165	95.0	160.6	161.0	65.7	66.0	0.02072	0.01280	56	57	1.62
166	94.9	201.3	199.4	106.4	104.5	0.02637	0.01610	56	57	1.64
167	98.2	132.1	131.1	33.0	31.9	0.01463	0.00890	55	55	1.65
168	98.6	132.0	130.9	33.4	32.4	0.01478	0.00897	55	55	1.65
169	95.2	130.4	128.1	35.2	33.0	0.01523	0.00907	44	44	1.68
170	88.7	249.4	246.2	157.6	160.7	0.03254	0.01983	44	44	1.64
171	94.8	251.4	246.5	156.6	151.7	0.03212	0.01945	44	44	1.65
172	196.5	221.9	215.8	25.4	19.3	0.01295	0.00695	44	44	1.87
173	197.7	222.5	220.7	24.8	23.0	0.01278	0.00758	44	44	1.69
174	201.2	236.9	236.4	35.2	35.2	0.01534	0.00937	43	44	1.64
175	192.2	256.2	254.1	64.1	61.9	0.02055	0.01243	43	44	1.66
176	96.7	131.2	130.0	34.5	33.3	0.01505	0.00911	50	50	1.66
177	97.2	131.9	129.6	34.6	32.3	0.01508	0.00897	48	49	1.68
178	95.7	128.0	126.7	32.2	31.0	0.01456	0.00878	47	48	1.66
179	97.2	131.1	129.5	33.9	32.3	0.01485	0.00893	64	68	1.67
180	201.9	232.2	229.7	30.4	27.8	0.01403	0.00828	66	71	1.70
181	295.4	327.0	327.8	31.6	32.4	0.01431	0.00892	67	73	1.60
182	181.8	327.6	327.0	145.8	145.2	0.03066	0.01885	75	78	1.62
183	197.9	327.5	327.8	129.5	129.8	0.02889	0.01784	76	77	1.62

Table E-II (cont.)

Test No.	P _c (psia)	P _{OJ} (psia)	P _{FJ} (psia)	ΔP_{OJ} (psi)	ΔP_{FJ} (psi)	\dot{W}_O (lb/sec)	\dot{W}_f (lb/sec)	T _{OJ} (°F)	T _{FJ} (°F)	MR
OC-27-184	259.0	323.3	327.1	64.3	68.0	0.02041	0.01296	71	70	1.57
185	269.3	327.8	326.4	58.5	57.1	0.01948	0.01187	70	70	1.64
186	261.9	293.3	298.1	31.4	36.2	0.01428	0.00944	69	70	1.51
187	292.9	317.1	315.3	24.2	22.4	0.01254	0.00744	68	69	1.69
188	300.0	335.9	337.6	35.9	37.6	0.01526	0.00963	70	70	1.58
189	294.9	394.0	400.2	99.1	105.3	0.02531	0.01611	73	72	1.57
190	99.2	131.1	130.8	31.9	31.6	0.01442	0.00880	64	82	1.64
191	97.9	131.4	131.4	33.5	33.5	0.01460	0.00903	62	93	1.64
192	98.4	131.5	130.4	33.1	32.0	0.01465	0.00867	71	149	1.69
193	98.0	131.5	130.2	32.6	31.3	0.01448	0.00851	78	177	1.70
194	99.2	133.7	131.8	34.5	32.7	0.01487	0.00862	80	200	1.73
195	201.5	232.1	229.0	30.5	27.4	0.01398	0.00791	82	196	1.77
196	202.4	231.5	229.4	29.1	27.0	0.01339	0.00790	124	178	1.69
197	202.4	231.3	228.3	28.9	26.0	0.01345	0.00796	112	89	1.69
198	92.4	109.4	108.4	17.0	16.0	0.0105	0.00637	53	60	1.64
199	94.3	119.7	120.2	25.4	25.9	0.0128	0.00811	74	85	1.58
200	88.5	120.	119.8	31.5	31.3	0.0143	0.00891	65	75	1.60
201	96.8	131.4	131.5	34.6	34.7	0.0149	0.00938	68	73	1.59
202	81.0	159.7	159.5	78.7	78.2	0.0226	0.0140	669	74	1.61
203	101.8	202.6	198.8	100.8	97.0	0.0255	0.0156	72	72	1.63
204	94.8	247.8	242.8	153.	148.0	0.0315	0.0193	80	70	1.63
205	201.2	222.5	215.9	21.3	1477	0.0117	0.00611	66	68	1.01
206	198.5	231.2	228.1	32.7	29.6	0.0145	0.00867	77	69	1.67
207	193.0	353.0	348.	160	155	0.0320	0.0198	80	81	1.61
208	295.0	320.2	319.7	25.2	24.7	0.0127	0.00792	70	74	1.60
209	291	314.8	315.6	23.8	24.6	0.0124	0.00790	63	63	1.57
210	293	315.3	314.5	22.3	21.5	0.0120	0.00739	61	64	1.62
211	290	313.6	317.1	23.0	27.1	0.0122	0.00829	62	68	1.47
212	290	325.6	324.4	35.6	34.4	0.0151	0.00934	78	84	1.61
213	289	357.1	356.0	68.1	67.0	0.0209	0.0130	76	79	1.60
214	293	395	395	102	102	0.0256	0.0160	75	77	1.60
215	292	447	450	155	158	0.0319	0.020	71	75	1.59
216	489	554.9	553.7	65.9	64.7	0.0205	0.0128	68	70	1.60
217	491	598.7	597.8	107.7	106.8	0.0263	0.0164	68	68	1.60

Table E-II (cont.)

<u>t No.</u>	<u>P_c</u> (psia)	<u>P_{OJ}</u> (psia)	<u>P_{FJ}</u> (psia)	<u>ΔP_{OJ}</u> (psi)	<u>ΔP_{FJ}</u> (psi)	<u>W_O</u> (lb/sec)	<u>W_f</u> (lb/sec)	<u>T_{OJ}</u> (°F)	<u>T_{FJ}</u> (°F)	<u>MR</u>
OC-27-218	135	176.7	175.7	41.7	40.7	.0125	.00745	65	70	1.67
219	124	177.5	174.8	53.5	50.8	.0142	.00832	72	73	1.70
220	98	132.1	133.0	34.1	35.0	.0113	.0059	69	71	1.63
221	81	103.3	103.5	22.7	22.9	.0093	.00559	62	68	1.66
222	148	233.1	220.8	75.1	72.8	.0168	.00995	76	75	1.69
223	194	326.0	32.10	132.0	127.0	.0222	.0130	79	97	1.71
224	122	175.7	171.1	53.7	49.1	.01422	.00802	77	141	1.77
225	123	178.1	172.1	55.4	49.4	.0144	.00798	80	164	1.80
226	122	176.6	171.4	54.3	49.1	.0142	.0078	93	229	1.82
227	122	176.8	171.1	55.3	49.6	.0141	.0078	116	242	1.81
228	144	219.6	212.1	75.0	68.4	.0165	.00914	127	249	1.80
229	119	177.0	175.4	57.8	56.2	.0145	.00829	109	245	1.75
230	122	177.1	175.5	55.5	53.9	.0141	.00806	120	268	1.75
231	146	224.4	220.9	78.2	74.7	.0166	.00945	136	280	1.76
232	145	224.9	222.8	80.3	78.2	.0166	.00963	162	291	1.72
233	124	180.5	181.5	56.3	57.3	.0140	.00826	153	285	1.69
234	123	190.9	186.9	67.9	63.9	.0138	.0082	73	76	1.68
236	102	142.7	139.5	41.1	37.9	.0107	.00632	71	72	1.69
237	81	109.9	108.4	28.5	27.0	.00896	.00533	74	76	1.68
238	144	244.3	240.2	100.7	96.6	.0168	.0100	78	82	1.68
239	186	367.6	362.4	181.6	176.4	.0225	.0136	79	79	1.65
240	164	245.0	243.9	80.7	79.6	.0150	.00917	74	74	1.64
241	160	246.5	243.3	86.6	83.4	.0156	.00938	77	79	1.66
242	166	246.2	242.2	80.0	76.0	.0150	.00897	78	71	1.67
243	192	368.8	365.5	176.7	173.4	.0222	.0135	78	75	1.64
244	123	193.1	189.9	70.5	67.3	.0140	.00818	79	170	1.72
245	117	192.0	191.2	74.6	73.8	.0144	.00838	85	239	1.72
246	119	192.6	192.0	73.9	73.3	.01399	.00821	136	288	1.70
247	139	245.9	247.3	107.2	108.6	.0169	.0100	132	283	1.69
248	137	248.0	246.0	111.1	109.1	.01699	.0100	153	288	1.69
249	147	247.5	246.0	100.1	98.6	.0161	.00953	154	287	1.69
250	125	174.5	171.1	46.0	42.6	.0578	.0268	73	75	2.15
251	125	174.1	168.2	49.2	43.3	.0598	.0252	70	75	2.37
252	99	128.4	125.5	29.6	26.7	.0464	.0199	68	78	2.33

Table E-II (cont.)

t No.	P _c (psia)	P _{OJ} (psia)	P _{FJ} (psia)	ΔP _{OJ} (psi)	ΔP _{FJ} (psi)	W _o (lb/sec)	W _f (lb/sec)	T _{OJ} (°F)	T _{FJ} (°F)	MR
OC-27-253	80	100.3	97.9	19.9	17.5	.0381	.0158	72	78	2.41
254	146	212.8	207.7	66.9	61.8	.0698	.0327	72	76	2.13
255	194	310.6	303.3	116.6	109.3	.0921	.0403	71	69	2.28
256	119	172.9	169.6	53.6	50.3	.0523	.0279	77	178	2.23
257	116	173.2	168.9	57.1	52.8	.0644	.0283	76	244	2.27
258	113	173.4	168.7	60.2	55.5	.0641	.0336	140	312	1.90
259	131	213.7	210.0	82.6	78.9	.0752	.0327	120	300	2.31
260	137	213.9	208.5	77.4	72.0	.0720	.0312	153	301	2.30
275	126	174.1	170.9	48.2	44.9	.0282	.0160	59	71	1.76
276	129	169.9	171.5	41.4	42.6	.0259	.0156	78	77	1.66
277	101	125.3	125.9	24.5	25.0	.0200	.0119	71	80	1.67
278	81	98.1	95.3	17.5	14.7	.0169	.00917	71	77	1.84
279	157	211.9	216.3	54.9	59.3	.0300	.0184	71	75	1.63
280	210	303.2	314.5	97.8	104	.0400	.0244	72	73	1.64
281	129	172.3	177.2	43.5	48.4	.0266	.0162	80	150	1.64
282	129	172.6	178.3	44.0	49.6	.0267	.0160	85	226	1.66
283	132	173.1	178.1	41.7	46.7	.0259	.0154	85	250	1.68
284	130	173.1	78.1	43.4	48.4	.0259	.0155	130	287	1.66
285	156	217.3	224.3	61.5	68.4	.0308	.0184	131	296	1.67
286	150	217.2	224.7	66.7	74.3	.0318	.0192	145	297	1.66
287	124	154.3	154	30.1	29.9	.0139	.00857	72	82	1.63
288	98	117.3	117.3	19.0	19.0	.0111	.00684	73	72	1.62
289	78	91.2	92.6	13.0	14.3	.00916	.00594	75	76	1.54
290	78	91.2	91.1	13.6	13.5	.00937	.00576	74	75	1.62
291	147	190.3	188.3	43.0	41.0	.0166	.0100	75	75	1.66
292	198	274.8	272.1	76.8	74.1	.0222	.0135	76	76	1.65
293	123	154.9	154.7	31.7	31.5	.0143	.0085	75	185	1.68
294	123	154.9	153.6	32.2	30.9	.0143	.00827	80	249	1.74
295	121	154.9	155.7	33.7	34.5	.0144	.00875	116	240	1.65
296	145	194.0	192.6	49.1	47.7	.0173	.01011	131	290	1.71
297	144	195.2	196.5	51.5	52.8	.0175	.0106	152	294	1.65

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Table E-III

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR LANVER

FUEL TEST TYPE NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR MF/MO	MODE	REACT (SEC)
MMH 101	UNLIKE-DOUBLET	.024	.020	24	60	308	90.5	129.1	83	87	1.66	1.239	SEP .102-04
MMH 102	UNLIKE-DOUBLET	.024	.020	24	60	308	87.9	132.5	88	87	1.57	1.381	SEP .104-04
MMH 103	UNLIKE-DOUBLET	.024	.020	24	60	309	89.4	129.9	88	87	1.62	1.286	SEP .102-04
MMH 104	UNLIKE-DOUBLET	.024	.020	24	60	311	88.3	129.1	88	87	1.62	1.301	SEP .102-04
MMH 105	UNLIKE-DOUBLET	.024	.020	24	60	507	90.5	125.7	89	88	1.70	1.175	SEP .101-04
MMH 106	UNLIKE-DOUBLET	.024	.020	24	60	1000	92.0	138.5	89	88	1.57	1.381	SEP .112-04
MMH 107	UNLIKE-DOUBLET	.024	.020	24	60	308	89.6	124.7	86	85	1.70	1.178	SEP .934-05
MMH 108	UNLIKE-DOUBLET	.024	.020	24	60	283	86.6	124.6	84	84	1.65	1.259	SEP .899-05
MMH 109	UNLIKE-DOUBLET	.024	.020	24	60	197	94.8	136.5	84	84	1.64	1.260	SEP .985-05
MMH 110	UNLIKE-DOUBLET	.024	.020	24	60	158	87.9	129.6	83	83	1.60	1.320	SEP .911-05
MMH 111	UNLIKE-DOUBLET	.024	.020	24	60	100	92.3	135.5	84	83	1.60	1.312	MIX .964-05
A-50 112	UNLIKE-DOUBLET	.024	.020	24	60	89	51.1	73.9	76	76	1.59	1.308	MIX .716-05
A-50 113	UNLIKE-DOUBLET	.024	.020	24	60	101	91.5	128.0	88	87	1.63	1.232	MIX .168-04
A-50 114	UNLIKE-DOUBLET	.024	.020	24	60	162	85.3	117.9	88	82	1.66	1.199	MIX .133-04
A-50 115	UNLIKE-DOUBLET	.024	.020	24	60	203	91.0	125.9	81	78	1.65	1.203	SEP .133-04
A-50 116	UNLIKE-DOUBLET	.024	.020	24	60	249	83.4	114.4	88	77	1.67	1.181	SEP .117-04
A-50 117	UNLIKE-DOUBLET	.024	.020	24	60	301	61.6	65.7	73	76	1.64	1.217	SEP .851-05
A-50 118	UNLIKE-DOUBLET	.024	.020	24	60	310	86.7	123.6	81	81	1.60	1.276	SEP .135-04
A-50 119	UNLIKE-DOUBLET	.024	.020	24	60	510	85.7	120.4	82	82	1.64	1.239	SEP .136-04
A-50 120	UNLIKE-DOUBLET	.024	.020	24	60	1003	87.8	122.8	82	81	1.63	1.227	SEP .136-04
A-50 121	UNLIKE-DOUBLET	.024	.020	24	60	160	88.8	125.2	81	80	1.62	1.247	SEP .135-04
A-50 122	UNLIKE-DOUBLET	.024	.020	24	60	338	72.8	100.6	81	80	1.65	1.200	SEP .108-04
A-50 123	UNLIKE-DOUBLET	.024	.020	24	60	339	75.4	104.6	77	77	1.65	1.206	SEP .104-04
MMH 124	UNLIKE-DOUBLET	.024	.020	24	60	955	69.6	101.6	82	82	1.61	1.293	SEP .695-05
MMH 125	UNLIKE-DOUBLET	.024	.020	24	60	495	87.6	130.3	83	81	1.60	1.315	SEP .890-05
MMH 126	UNLIKE-DOUBLET	.024	.020	24	60	298	71.6	132.1	83	82	1.59	1.350	SEP .915-05
MMH 127	UNLIKE-DOUBLET	.024	.020	24	60	283	53.7	80.5	88	81	1.58	1.372	M/S .557-05
MMH 128	UNLIKE-DOUBLET	.024	.020	24	60	256	84.8	126.2	85	83	1.56	1.347	SEP .909-05
MMH 129	UNLIKE-DOUBLET	.024	.020	24	60	191	91.6	135.5	85	83	1.60	1.331	SEP .976-05
MMH 130	UNLIKE-DOUBLET	.024	.020	24	60	152	85.9	129.5	85	82	1.56	1.384	M/S .920-05
MMH 131	UNLIKE-DOUBLET	.024	.020	24	60	95	88.6	131.0	85	83	1.57	1.366	MIX .958-05
MMH 132	UNLIKE-DOUBLET	.024	.020	24	60	114	91.4	136.4	88	83	1.58	1.357	MIX .994-05
MMH 133	UNLIKE-DOUBLET	.024	.020	24	60	411	125.8	179.6	83	83	1.66	1.238	UNDEF .126-04
MMH 134	UNLIKE-DOUBLET	.024	.020	24	60	97	92.8	136.3	77	197	1.74	1.216	M/S .422-04
MMH 135	UNLIKE-DOUBLET	.024	.020	24	60	119	93.9	139.2	77	199	1.72	1.237	M/S .447-04
MMH 136	UNLIKE-DOUBLET	.024	.020	24	60	150	92.4	136.5	77	199	1.74	1.227	SEP .434-04
MMH 137	UNLIKE-DOUBLET	.024	.020	24	60	188	97.7	143.8	76	199	1.74	1.218	SEP .451-04
MMH 138	UNLIKE-DOUBLET	.024	.020	24	60	243	93.3	148.9	79	195	1.60	1.438	SEP .459-04
MMH 139	UNLIKE-DOUBLET	.024	.020	24	60	103	35.8	58.4	61	59	1.45	1.615	UNDEF .209-05
MMH 140	UNLIKE-DOUBLET	.024	.020	24	60	81	97.4	143.0	64	62	1.61	1.306	UNDEF .553-05

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR LANVER

FUEL TEST MODE TYPE-NO.	PC (PSIA)	WAVG NEF	WEO	REF	HEU	DELTA (DEG F)	REL	PPF (PSIA)	PPU	MRVP	XF	XU	XP	RESID
NNH 101 SEP	308.	105.	43.7	36.0	245+05	0.	59+06	16+07	1.3	23.2	34.8	00	08	18-01 .13-04
NNH 102 SEP	308.	105.	46.0	34.0	252+05	0.	60+06	15+07	1.3	23.2	34.8	00	08	18-01 .13-04
NNH 103 SEP	309.	105.	48.4	35.2	247+05	0.	59+06	15+07	1.3	23.2	34.8	00	07	18-01 .13-04
NNH 104 SEP	311.	104.	48.1	34.6	285+05	0.	59+06	15+07	1.3	23.2	34.8	00	07	18-01 .13-04
NNH 105 SEP	307.	104.	68.3	59.6	281+05	0.	58+06	16+07	1.4	23.7	34.8	00	05	11-01 .13-04
NNH 106 SEP	1000.	110.	163.5	121.4	265+05	0.	54+06	16+07	1.4	23.7	34.8	00	02	57-02 .12-04
NNH 107 SEP	308.	103.	40.6	34.9	213+05	0.	56+06	15+07	1.3	22.1	34.8	00	07	17-01 .13-04
NNH 108 SEP	283.	101.	34.6	27.5	231+05	0.	55+06	15+07	1.2	21.1	34.8	00	08	19-01 .13-04
NNH 109 SEP	197.	111.	31.1	24.7	253+05	0.	61+06	16+07	1.2	21.1	34.0	01	11	26-01 .12-04
NNH 110 SEP	158.	104.	22.4	17.0	238+05	0.	57+06	15+07	1.2	20.5	34.0	01	13	32-01 .13-04
NNH 111 MIX	100.	109.	15.5	11.9	249+05	0.	60+06	16+07	1.2	21.1	34.8	01	21	50-01 .12-04
A-50 112 MIX	89.	60.	4.7	3.1	112+05	0.	27+06	02+06	2.1	17.3	17.9	02	19	68-01 .23-03
A-50 113 MIX	101.	105.	16.1	12.1	211+05	0.	51+06	16+07	2.8	23.2	17.9	03	23	80-01 .13-04
A-50 114 MIX	162.	98.	21.6	16.3	186+05	0.	45+06	14+07	2.4	20.0	17.9	02	12	43-01 .14-04
A-50 115 SEP	203.	108.	30.9	23.1	193+05	0.	46+06	15+07	2.2	19.5	19.2	01	10	32-01 .13-04
A-50 116 SEP	289.	95.	33.8	25.6	174+05	0.	42+06	14+07	2.2	19.0	19.1	01	07	24-01 .15-04
A-50 117 SEP	301.	71.	21.2	15.4	129+05	0.	31+06	10+07	2.1	18.1	18.7	01	06	21-01 .19-04
A-50 118 SEP	310.	101.	45.7	32.0	194+05	0.	47+06	14+07	2.4	19.5	18.0	01	06	22-01 .13-04
A-50 119 SEP	510.	99.	71.4	51.8	190+05	0.	46+06	14+07	2.4	20.0	17.9	00	04	14-01 .14-04
A-50 120 SEP	1003.	101.	145.9	107.0	193+05	0.	46+06	15+07	2.4	20.0	18.4	00	02	69-02 .14-04
A-50 121 MIX	100.	103.	24.2	17.4	195+05	0.	47+06	15+07	2.3	19.5	18.5	01	12	82-01 .13-04
A-50 122 SEP	334.	83.	33.0	24.6	157+05	0.	38+06	12+07	2.3	19.5	18.5	01	06	20-01 .17-04
A-50 123 SEP	334.	86.	35.1	25.6	159+05	0.	38+06	12+07	2.2	17.7	17.9	01	05	19-01 .16-04
NNH 124 SEP	955.	87.	63.2	64.0	185+05	0.	44+06	12+07	1.2	20.0	33.9	00	02	51-02 .16-04
NNH 125 SEP	495.	105.	70.6	54.0	235+05	0.	56+06	15+07	1.1	20.5	35.6	00	04	98-02 .13-04
NNH 126 SEP	294.	105.	83.4	32.5	241+05	0.	35+06	15+07	1.2	20.5	34.8	00	07	17-01 .13-04
NNH 127 M/S	283.	84.	15.5	11.4	145+05	0.	35+06	15+07	1.1	21.1	36.4	00	07	17-01 .13-04
NNH 128 SEP	250.	101.	34.5	25.9	232+05	0.	56+06	14+07	1.2	21.6	35.6	00	08	20-01 .13-04
NNH 129 SEP	191.	109.	29.7	22.5	249+05	0.	60+06	16+07	1.2	21.6	35.6	01	11	27-01 .12-04
NNH 130 M/S	152.	193.	21.5	15.8	236+05	0.	57+06	15+07	1.2	21.6	36.4	01	14	33-01 .13-04
NNH 131 MIX	95.	100.	14.2	10.5	244+05	0.	59+06	15+07	1.2	21.6	35.6	01	23	54-01 .13-04
NNH 132 MIX	411.	106.	112.1	90.4	251+05	0.	60+06	16+07	1.2	22.1	36.4	01	19	45-01 .12-04
NNH 133 MIX	411.	106.	112.1	90.4	251+05	0.	60+06	16+07	1.2	22.1	36.4	01	19	45-01 .12-04
NNH 134 M/S	97.	109.	14.4	11.3	558+05	0.	13+07	15+07	12.9	17.7	2.4	18	18	18+00 .12-04
NNH 135 M/S	119.	113.	24.1	14.2	577+05	0.	14+07	15+07	18.1	17.7	2.4	18	18	18+00 .12-04
NNH 136 SEP	140.	100.	24.2	17.3	266+05	0.	14+07	15+07	18.1	17.7	2.3	18	18	18+00 .12-04
NNH 137 SEP	140.	115.	30.7	24.0	546+05	0.	14+07	16+07	18.7	17.3	2.3	18	18	18+00 .12-04
NNH 138 SEP	243.	115.	55.9	24.8	561+05	0.	14+07	15+07	17.2	16.6	2.6	01	08	73-01 .11-04
NNH 139 MIX	103.	45.	2.0	1.6	600+05	0.	21+06	55+06	.6	12.0	41.2	01	12	26-01 .29-04
NNH 140 MIX	81.	115.	13.5	9.7	217+05	0.	52+06	15+07	.6	12.9	41.2	01	16	35-01 .12-04

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HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

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FUEL TEST TYPE NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TD (F)	TF (F)	MR MF/MO	MODE	REACT (SEC)
MMH 141	UNLIKE-DOUBLET	.024	.020	24	60	81	96.8	143.2	66	64	1.60	1.324	UNDEF .582-05
MMH 142	UNLIKE-DOUBLET	.024	.020	24	60	119	91.4	136.7	68	67	1.59	1.357	MIX .604-05
MMH 143	UNLIKE-DOUBLET	.024	.020	24	60	129	32.4	54.2	67	67	1.42	1.690	UNDEF .237-05
MMH 144	UNLIKE-DOUBLET	.024	.020	24	60	154	35.9	55.7	67	67	1.53	1.453	UNDEF .243-05
MMH 145	UNLIKE-DOUBLET	.024	.020	24	60	152	88.2	128.4	68	67	1.62	1.283	SEP .567-05
MMH 146	UNLIKE-DOUBLET	.024	.020	24	60	190	90.3	135.9	68	67	1.57	1.372	SEP .600-05
MMH 147	UNLIKE-DOUBLET	.024	.020	24	60	256	84.7	124.2	68	67	1.61	1.302	SEP .548-05
MMH 148	UNLIKE-DOUBLET	.024	.020	24	60	102	37.0	58.8	68	68	1.49	1.523	MIX .265-05
MMH 149	UNLIKE-DOUBLET	.024	.020	24	60	99	89.7	133.6	69	69	1.60	1.343	MIX .622-05
MMH 150	UNLIKE-DOUBLET	.024	.020	24	60	297	90.1	131.9	70	69	1.62	1.298	SEP .621-05
MMH 151	UNLIKE-DOUBLET	.024	.020	24	60	484	95.4	137.8	70	69	1.64	1.263	SEP .648-05
MMH 152	UNLIKE-DOUBLET	.024	.020	24	60	958	101.8	148.3	78	74	1.62	1.291	SEP .850-05
MMH 153	UNLIKE-DOUBLET	.024	.020	24	60	98	29.0	50.3	55	56	1.37	1.818	M/P .158-05
MMH 154	UNLIKE-DOUBLET	.024	.020	24	60	100	29.4	41.9	54	55	1.66	1.225	M/P .127-05
MMH 155	UNLIKE-DOUBLET	.024	.020	24	60	102	36.8	51.4	54	54	1.70	1.174	M/P .153-05
MMH 156	UNLIKE-DOUBLET	.024	.020	24	60	107	49.3	77.1	53	54	1.52	1.472	MIX .228-05
MMH 157	UNLIKE-DOUBLET	.024	.020	24	60	102	51.5	79.2	53	54	1.61	1.316	MIX .233-05
MMH 158	UNLIKE-DOUBLET	.024	.020	24	60	101	28.9	35.6	58	58	1.68	1.211	M/P .143-05
MMH 159	UNLIKE-DOUBLET	.024	.020	24	60	99	30.3	43.0	57	58	1.65	1.258	M/P .188-05
MMH 160	UNLIKE-DOUBLET	.024	.020	24	60	101	39.2	56.6	57	58	1.65	1.258	M/P .188-05
MMH 161	UNLIKE-DOUBLET	.024	.020	24	60	102	37.8	58.2	57	58	1.54	1.829	M/P .194-05
MMH 162	UNLIKE-DOUBLET	.024	.020	24	60	102	41.4	54.0	57	58	1.62	1.577	M/P .180-05
MMH 163	UNLIKE-DOUBLET	.024	.020	24	60	98	58.5	85.1	57	58	1.63	1.577	MIX .283-05
MMH 164	UNLIKE-DOUBLET	.024	.020	24	60	100	50.1	72.1	56	57	1.65	1.249	MIX .233-05
MMH 165	UNLIKE-DOUBLET	.024	.020	24	60	95	72.3	106.7	56	57	1.61	1.313	MIX .345-05
MMH 166	UNLIKE-DOUBLET	.024	.020	24	60	95	92.1	134.3	56	57	1.63	1.282	MIX .434-05
MMH 167	UNLIKE-DOUBLET	.024	.020	24	60	99	51.2	74.1	55	55	1.63	1.262	MIX .229-05
MMH 168	UNLIKE-DOUBLET	.024	.020	24	60	99	51.6	74.7	55	55	1.63	1.267	MIX .230-05
MMH 169	UNLIKE-DOUBLET	.024	.020	24	60	95	52.6	75.1	44	44	1.66	1.223	UNDEF .160-05
MMH 170	UNLIKE-DOUBLET	.024	.020	24	60	89	112.4	164.1	44	44	1.62	1.281	MIX .350-05
MMH 171	UNLIKE-DOUBLET	.024	.020	24	60	95	111.0	161.0	44	44	1.63	1.265	MIX .343-05
MMH 172	UNLIKE-DOUBLET	.024	.020	24	60	196	44.8	57.5	44	44	1.64	.993	M/P .123-05
MMH 173	UNLIKE-DOUBLET	.024	.020	24	60	198	44.2	62.7	44	44	1.66	1.213	M/P .134-05
MMH 174	UNLIKE-DOUBLET	.024	.020	24	60	201	53.0	77.5	44	44	1.62	1.287	MIX .165-05
MMH 175	UNLIKE-DOUBLET	.024	.020	24	60	192	71.0	102.9	44	44	1.63	1.262	M/S .219-05
MMH 176	UNLIKE-DOUBLET	.024	.020	24	60	97	52.3	75.7	50	50	1.63	1.262	MIX .197-05
MMH 177	UNLIKE-DOUBLET	.024	.020	24	60	97	52.3	74.4	48	48	1.63	1.219	MIX .162-05
MMH 178	UNLIKE-DOUBLET	.024	.020	24	60	97	50.4	72.6	47	48	1.65	1.254	MIX .176-05
MMH 179	UNLIKE-DOUBLET	.024	.020	24	60	97	52.2	75.0	64	68	1.67	1.244	MIX .323-05
MMH 180	UNLIKE-DOUBLET	.024	.020	24	60	202	49.4	69.6	66	71	1.70	1.198	MIX .326-05

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGULIC STREAM IMPINGMENT DATA COMPILATION

FUEL TEST		MODE		PC		VAVG		MEF		MED		REF		REU		DELTI		RELF		RELO		PPF		PPU		MRVP		XF		XC		XP		RESID	
TYPE/NO.				(PSIA)												(DEG F)						(PSIA)													
MMH	181	UNDEF	81.	115.	10.3	13.6	9.7	221+05	612+05	0.	53+06	15+07	7	13.5	41.2	01	17	37-01	12-04																
MMH	182	MIX	119.	109.	10.3	12.8	12.8	218+05	583+05	201.	52+06	14+07	7	14.2	38.7	01	12	27-01	12-04																
MMH	183	UNDEF	129.	41.	3.1	1.7	883+04	206+05	81.	21+06	49+06	7	13.8	37.9	00	11	25-01	13-04																	
MMH	184	UNDEF	158.	44.	3.9	2.6	887+04	228+05	78.	49+06	14+07	7	14.2	38.7	00	09	21-01	13-04																	
MMH	185	SEP	158.	104.	20.6	15.2	204+05	583+05	182.	52+06	14+07	7	14.2	38.7	00	07	17-01	13-04																	
MMH	186	SEP	158.	108.	28.9	20.0	216+05	577+05	198.	47+06	13+07	7	14.2	38.7	00	05	12-01	13-04																	
MMH	187	SEP	250.	100.	32.8	25.8	198+05	591+05	326.	23+06	57+06	8	14.2	37.3	00	14	32-01	12-04																	
MMH	188	MIX	102.	46.	2.9	1.8	944+04	237+05	114.	52+06	14+07	8	14.8	37.3	00	05	12-01	13-04																	
MMH	189	MIX	107.	107.	14.8	10.3	217+05	576+05	101.	51+06	14+07	8	14.8	37.3	00	03	11-02	12-04																	
MMH	190	SEP	297.	106.	42.7	31.4	214+05	582+05	679.	50+06	15+07	8	14.8	37.3	00	02	10-01	11-04																	
MMH	191	SEP	484.	111.	75.9	57.3	223+05	618+05	940.	60+06	17+07	9	18.1	38.3	00	10	24-01	13-04																	
MMH	192	SEP	958.	120.	175.5	134.3	252+05	688+05	1342.	17+06	41+06	5	10.1	38.0	00	10	24-01	13-04																	
MMH	193	M/P	98.	38.	2.0	1.0	719+04	173+05	115.	14+06	52+06	5	9.8	37.9	00	10	23-01	12-04																	
MMH	194	M/P	100.	34.	1.4	1.0	592+04	174+05	158.	14+06	52+06	5	9.8	37.9	00	10	23-01	12-04																	
MMH	195	M/P	102.	42.	2.2	1.7	718+04	218+05	170.	17+06	52+06	5	9.8	37.9	00	10	23-01	12-04																	
MMH	196	MIX	107.	60.	5.1	3.1	108+05	291+05	86.	26+06	70+06	5	9.5	37.8	00	09	20-01	22-04																	
MMH	197	MIX	102.	63.	5.2	3.5	111+05	316+05	86.	27+06	70+06	5	9.5	37.8	00	09	20-01	22-04																	
MMH	198	M/P	101.	31.	1.9	1.0	519+04	175+05	171.	12+06	42+06	6	11.0	38.2	00	11	25-01	17-04																	
MMH	199	M/P	99.	35.	1.5	1.1	626+04	193+05	136.	15+06	44+06	6	10.7	38.2	00	11	25-01	17-04																	
MMH	200	M/P	101.	40.	2.6	1.0	824+04	236+05	145.	20+06	57+06	6	10.7	38.2	00	11	25-01	17-04																	
MMH	201	M/P	102.	46.	2.8	1.6	849+04	248+05	131.	20+06	55+06	6	10.7	38.2	00	11	25-01	17-04																	
MMH	202	MIX	98.	48.	2.4	2.1	787+04	249+05	145.	19+06	60+06	6	10.7	38.2	00	11	25-01	17-04																	
MMH	203	MIX	98.	69.	5.8	4.1	124+05	332+05	94.	30+06	65+06	6	10.7	38.2	00	11	25-01	17-04																	
MMH	204	MIX	100.	59.	4.2	3.0	164+05	300+05	85.	25+06	72+06	5	10.4	38.1	00	11	25-01	17-04																	
MMH	205	MIX	95.	86.	8.8	6.0	154+05	434+05	72.	37+06	10+07	5	10.4	38.1	00	11	25-01	17-04																	
MMH	206	MIX	95.	108.	15.9	9.8	194+05	552+05	79.	48+06	13+07	5	10.4	38.1	00	11	25-01	17-04																	
MMH	207	MIX	99.	80.	4.9	3.1	105+05	308+05	65.	25+06	73+06	5	10.1	38.1	00	10	23-01	17-04																	
MMH	208	MIX	99.	60.	4.5	3.2	106+05	307+05	65.	25+06	74+06	5	10.1	38.1	00	10	23-01	17-04																	
MMH	209	UNDEF	95.	81.	4.3	3.0	941+04	308+05	75.	23+06	74+06	4	7.4	42.2	00	08	17-01	10-04																	
MMH	210	MIX	132.	14.0	1.0	13.0	206+05	658+05	102.	49+06	16+07	4	7.4	42.2	00	08	17-01	10-04																	
MMH	211	MIX	130.	14.6	1.0	13.5	202+05	659+05	102.	49+06	16+07	4	7.4	42.2	00	08	17-01	10-04																	
MMH	212	M/P	146.	44.	5.2	4.5	721+04	282+05	78.	17+06	63+06	4	7.4	42.2	00	04	62-02	29-04																	
MMH	213	M/P	146.	51.	6.2	4.5	786+04	258+05	67.	19+06	62+06	4	7.4	42.2	00	04	61-02	27-04																	
MMH	214	MIX	201.	62.	9.6	6.5	972+04	310+05	71.	23+06	74+06	4	7.4	42.2	00	04	60-02	21-04																	
MMH	215	M/S	192.	63.	10.1	11.2	129+05	415+05	116.	31+06	10+07	4	7.4	42.2	00	04	58-02	16-04																	
MMH	216	MIX	97.	61.	4.5	3.1	101+05	304+05	83.	24+06	73+06	4	6.6	38.7	00	09	50-01	22-04																	
MMH	217	MIX	97.	61.	4.5	3.1	975+04	304+05	82.	23+06	73+06	4	6.6	38.7	00	09	50-01	22-04																	
MMH	218	MIX	97.	59.	4.1	2.9	955+04	294+05	69.	23+06	71+06	4	6.0	38.9	00	08	49-01	23-04																	
MMH	219	MIX	97.	61.	4.5	3.3	120+05	326+05	68.	29+06	78+06	6	12.9	34.2	00	13	32-01	22-04																	
MMH	220	MIX	202.	57.	6.1	6.3	115+05	312+05	110.	29+06	75+06	8	13.5	32.2	00	07	17-01	24-04																	

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR LANVER

FUEL TEST TYPE NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TU (F)	TF (F)	MR	MF/MD	MODE	REACT (SEC)
MMH 181	UNLIKE-DOUBLET	.024	.020	24	60	295	50.4	75.1	67	73	1.60	1.337	MIX	.370-05
MMH 182	UNLIKE-DOUBLET	.024	.020	24	60	182	108.8	159.2	75	78	1.62	1.295	SEP	.941-05
MMH 183	UNLIKE-DOUBLET	.024	.020	24	60	198	102.6	150.6	76	77	1.62	1.305	SEP	.887-05
MMH 184	UNLIKE-DOUBLET	.024	.020	24	60	259	72.2	106.9	71	70	1.57	1.380	SEP	.530-05
MMH 185	UNLIKE-DOUBLET	.024	.020	24	60	269	68.8	99.8	70	70	1.64	1.272	UNDEF	.479-05
MMH 186	UNLIKE-DOUBLET	.024	.020	24	60	282	50.4	79.3	69	70	1.51	1.496	MIX	.377-05
MMH 187	UNLIKE-DOUBLET	.024	.020	24	60	293	44.2	62.5	68	69	1.69	1.207	MIX	.288-05
MMH 188	UNLIKE-DOUBLET	.024	.020	24	60	300	53.9	80.9	70	70	1.58	1.364	M/S	.380-05
MMH 189	UNLIKE-DOUBLET	.024	.020	24	60	295	89.7	135.6	73	72	1.57	1.386	SEP	.705-05
MMH 190	UNLIKE-DOUBLET	.024	.020	24	60	99	50.7	74.5	64	82	1.64	1.292	MIX	.809-05
MMH 191	UNLIKE-DOUBLET	.024	.020	24	60	98	51.9	77.0	62	93	1.64	1.302	MIX	.887-05
MMH 192	UNLIKE-DOUBLET	.024	.020	24	60	98	51.8	76.5	71	109	1.69	1.258	M/S	.121-04
MMH 193	UNLIKE-DOUBLET	.024	.020	24	60	99	51.5	76.4	78	177	1.70	1.258	M/S	.188-04
MMH 194	UNLIKE-DOUBLET	.024	.020	24	60	99	53.0	78.5	80	200	1.73	1.236	M/S	.261-04
MMH 195	UNLIKE-DOUBLET	.024	.020	24	60	202	49.9	71.8	82	196	1.77	1.173	SEP	.233-04
MMH 196	UNLIKE-DOUBLET	.024	.020	24	60	202	49.9	70.9	124	178	1.69	1.218	SEP	.302-04
MMH 197	UNLIKE-DOUBLET	.024	.020	24	60	202	49.3	67.7	112	89	1.69	1.169	MIX	.723-05
MMH 198	UNLIKE-DOUBLET	.024	.020	24	60	92	36.6	51.6	53	60	1.64	1.234	M/P	.300-05
MMH 199	UNLIKE-DOUBLET	.024	.020	24	60	94	45.4	66.5	74	85	1.58	1.337	UNDEF	.712-05
MMH 200	UNLIKE-DOUBLET	.024	.020	24	60	89	50.3	72.7	65	75	1.60	1.297	MIX	.609-05
MMH 201	UNLIKE-DOUBLET	.024	.020	24	60	97	52.5	76.5	68	73	1.59	1.319	MIX	.647-05
MMH 202	UNLIKE-DOUBLET	.024	.020	24	60	81	79.8	114.2	69	74	1.61	1.277	MIX	.989-05
MMH 203	UNLIKE-DOUBLET	.024	.020	24	60	102	90.3	127.2	72	72	1.63	1.241	MIX	.112-04
MMH 204	UNLIKE-DOUBLET	.024	.020	24	60	95	111.3	157.2	70	70	1.63	1.246	MIX	.131-04
MMH 205	UNLIKE-DOUBLET	.024	.020	24	60	201	41.2	49.7	66	68	1.91	.907	M/P	.366-05
MMH 206	UNLIKE-DOUBLET	.024	.020	24	60	199	51.5	70.6	77	69	1.67	1.179	MIX	.635-05
MMH 207	UNLIKE-DOUBLET	.024	.020	24	60	193	114.1	162.0	80	81	1.61	1.266	SEP	.175-04
MMH 208	UNLIKE-DOUBLET	.024	.020	24	60	295	43.6	64.6	70	74	1.60	1.293	MIX	.565-05
MMH 209	UNLIKE-DOUBLET	.024	.020	24	60	291	43.6	64.1	63	63	1.57	1.351	MIX	.449-05
MMH 210	UNLIKE-DOUBLET	.024	.020	24	60	293	42.1	60.0	61	47	1.62	1.285	MIX	.415-05
MMH 211	UNLIKE-DOUBLET	.024	.020	24	60	290	42.8	67.5	62	68	1.47	1.542	UNDEF	.499-05
MMH 212	UNLIKE-DOUBLET	.024	.020	24	60	290	53.7	76.5	78	84	1.61	1.289	UNDEF	.848-05
MMH 213	UNLIKE-DOUBLET	.024	.020	24	60	289	74.2	106.3	76	79	1.60	1.282	UNDEF	.107-04
MMH 214	UNLIKE-DOUBLET	.024	.020	24	60	293	90.8	130.7	73	77	1.60	1.295	UNDEF	.127-04
MMH 215	UNLIKE-DOUBLET	.024	.020	24	60	292	112.8	163.2	71	75	1.59	1.306	UNDEF	.147-04
MMH 216	UNLIKE-DOUBLET	.024	.020	24	60	489	72.3	104.2	68	70	1.60	1.296	SEP	.809-05
MMH 217	UNLIKE-DOUBLET	.024	.020	24	60	491	92.8	133.4	68	66	1.60	1.292	SEP	.106-04
MMH 218	UNLIKE-DOUBLET	.024	.021	0	0	135	44.0	56.8	65	70	1.67	1.005	MIX	.284-05
MMH 219	UNLIKE-DOUBLET	.024	.021	0	0	124	50.3	63.5	72	73	1.70	.947	MIX	.366-05
MMH 220	UNLIKE-DOUBLET	.024	.021	0	0	98	39.9	52.6	69	71	1.63	1.052	MIX	.281-05

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR LAWVER

FUEL TYPE	TEST NO.	MODE	PC (PSIA)	VAVG	MEF	WEN	REF	REO	DELTI (DEG F)	RELF	RELU (PSIA)	PPF	PPD	MRVP	XF	YU	XP	WESID
MMH	181	MIX	295.	60.	13.8	9.6	126+05	320+05	128.	30+06	.77+06	.9	13.8	30.8	.00	.05	.12-01	.22-04
MMH	182	SEP	182.	128.	38.7	28.7	280+05	723+05	129.	.67+06	.17+07	1.1	16.9	32.0	.01	.09	.23-01	.10-04
MMH	183	SEP	198.	121.	37.6	27.9	263+05	686+05	129.	.63+06	.16+07	1.0	17.3	33.7	.01	.09	.21-01	.11-04
MMH	184	SEP	259.	86.	25.4	17.6	178+05	469+05	131.	.43+06	.11+07	.8	15.2	37.0	.00	.06	.14-01	.15-04
MMH	185	UNDEF	269.	81.	22.2	16.6	163+05	444+05	141.	.39+06	.11+07	.8	14.8	36.0	.00	.05	.13-01	.17-04
MMH	186	MIX	262.	62.	13.7	8.6	130+05	324+05	101.	.31+06	.74+06	.8	14.5	35.4	.00	.06	.15-01	.21-04
MMH	187	MIX	293.	51.	9.5	7.4	101+05	282+05	95.	.24+06	.65+06	.8	14.2	35.9	.00	.05	.1-01	.27-04
MMH	188	M/S	300.	64.	16.3	11.3	132+05	346+05	108.	.32+06	.84+06	.8	14.8	36.0	.00	.05	.12-01	.21-04
MMH	189	SEP	295.	108.	45.0	31.3	226+05	589+05	140.	.54+06	.14+07	.9	16.0	36.4	.00	.05	.13-01	.22-04
MMH	190	MIX	99.	60.	4.6	3.2	136+05	317+05	61.	.33+06	.76+06	1.2	12.9	22.6	.01	.13	.37-01	.22-04
MMH	191	MIX	98.	61.	5.0	3.3	154+05	321+05	49.	.37+06	.77+06	1.6	12.3	16.4	.02	.13	.45-01	.22-04
MMH	192	M/S	96.	-1.	5.4	3.4	229+05	336+05	40.	.55+06	.61+06	6.5	15.2	5.3	.07	.16	.10+00	.22-04
MMH	193	M/S	99.	-1.	5.6	3.6	275+05	348+05	27.	.60+06	.84+06	12.0	18.1	3.6	.12	.18	.15+00	.22-04
MMH	194	M/S	99.	62.	6.4	3.8	327+05	363+05	18.	.79+06	.87+06	19.1	19.0	2.4	.19	.19	.19+00	.21-04
MMH	195	SEP	202.	58.	10.8	7.0	292+05	346+05	54.	.70+06	.83+06	17.6	20.0	2.8	.09	.10	.93-01	.23-04
MMH	196	SEP	202.	58.	10.2	6.8	292+05	347+05	69.	.62+06	.10+07	12.2	52.9	9.5	.06	.26	.13+00	.23-04
MMH	197	MIX	202.	56.	7.9	8.0	131+05	408+05	115.	.31+06	.98+06	1.4	40.5	56.0	.01	.20	.37-01	.25-04
A-50	198	P/P	92.	42.	2.3	1.5	695+04	216+05	111.	.17+06	.52+06	1.4	9.5	15.1	.02	.10	.40-01	.32-04
A-50	199	UNDEF	94.	54.	4.0	2.6	108+05	300+05	89.	.26+06	.72+06	2.7	16.5	13.8	.03	.18	.71-01	.25-04
A-50	200	MIX	89.	59.	4.5	2.9	109+05	316+05	109.	.26+06	.76+06	2.1	13.2	14.2	.02	.15	.59-01	.23-04
A-50	201	MIX	97.	62.	5.4	3.5	113+05	316+05	133.	.27+06	.81+06	2.0	14.2	15.8	.02	.15	.55-01	.22-04
A-50	202	MIX	61.	93.	10.1	6.7	170+05	512+05	195.	.41+06	.12+07	2.0	14.5	15.8	.03	.18	.67-01	.15-04
A-50	203	MIX	102.	104.	15.8	10.9	186+05	599+05	149.	.54+06	.14+07	1.9	15.6	17.6	.02	.15	.54-01	.13-04
A-50	204	MIX	95.	129.	22.4	15.3	227+05	719+05	155.	.54+06	.17+07	1.9	14.8	17.5	.02	.16	.55-01	.11-04
A-50	205	P/P	201.	43.	4.7	4.4	707+04	260+05	112.	.17+06	.62+06	1.3	13.5	16.9	.01	.07	.24-01	.34-04
A-50	206	MIX	194.	50.	9.5	7.1	101+05	367+05	183.	.24+06	.83+06	1.8	17.7	21.2	.01	.09	.28-01	.24-04
A-50	207	SEP	193.	132.	48.9	34.4	254+05	781+05	202.	.61+06	.19+07	2.4	19.0	17.5	.01	.10	.35-01	.10-04
A-50	208	MIX	245.	52.	11.8	7.7	950+04	290+05	202.	.23+06	.70+06	2.0	14.8	16.1	.01	.05	.19-01	.26-04
A-50	209	MIX	241.	52.	11.4	6.9	891+04	271+05	202.	.21+06	.65+06	1.5	12.6	14.0	.01	.04	.15-01	.26-04
A-50	210	MIX	243.	40.	10.1	6.5	830+04	259+05	282.	.20+06	.62+06	1.6	12.0	16.7	.01	.04	.15-01	.28-04
A-50	211	P/P	240.	53.	12.6	6.7	980+04	285+05	272.	.23+06	.64+06	1.8	12.3	15.5	.01	.06	.16-01	.25-04
A-50	212	MIX	240.	62.	16.4	11.3	123+05	363+05	391.	.30+06	.87+06	2.6	18.1	15.5	.01	.06	.24-01	.22-04
A-50	213	P/P	269.	67.	31.4	21.3	164+05	496+05	292.	.39+06	.12+07	2.3	17.3	16.9	.01	.06	.22-01	.16-04
A-50	214	UNDEF	293.	106.	48.1	32.3	199+05	604+05	332.	.48+06	.14+07	2.2	16.9	17.1	.01	.06	.21-01	.16-04
A-50	215	UNDEF	292.	132.	74.8	48.6	244+05	733+05	379.	.59+06	.18+07	2.1	15.2	16.2	.01	.05	.19-01	.10-04
A-50	216	MIX	434.	89.	50.7	32.9	150+05	482+05	403.	.36+06	.11+07	1.9	14.2	16.8	.00	.03	.10-01	.16-04
A-50	217	SEP	421.	108.	43.5	54.4	190+05	592+05	762.	.46+06	.14+07	1.8	14.2	17.6	.00	.03	.10-01	.12-04
MMH	218	MIX	124.	55.	4.9	3.8	976+04	276+05	214.	.94+03	.28+04	.4	13.2	32.4	.01	.10	.24-01	.31-04
MMH	219	MIX	124.	55.	4.4	4.1	112+05	318+05	269.	.11+04	.33+04	.9	15.6	34.4	.01	.13	.30-01	.28-04
MMH	220	MIX	124.	45.	2.4	2.0	913+04	256+05	183.	.91+03	.26+04	.8	14.5	34.2	.01	.15	.36-01	.35-04

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HIGH PERFORMANCE M204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR LAWVER

FUEL TEST TYPE NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR HF/MO	MODE	REACT (SEC)
MMH 221	XDT1	.024	.021	0.	0.	81.	32.6	42.6	62.	68.	1.66	1.024	MIX .197-05
MMH 222	XDT1	.024	.021	0.	0.	148.	59.7	74.1	76.	75.	1.69	.986	SEP .478-15
MMH 223	XDT1	.024	.021	0.	0.	194.	79.1	100.7	79.	97.	1.71	.975	SEP .928-05
MMH 224	XDT1	.024	.021	0.	0.	122.	50.6	63.6	71.	141.	1.77	.930	MIX .106-04
MMH 225	XDT1	.024	.021	0.	0.	121.	51.3	64.4	80.	164.	1.80	.909	MIX .150-04
MMH 226	XDT1	.024	.021	0.	0.	122.	51.2	66.0	93.	229.	1.82	.925	SEP .572-04
MMH 227	XDT1	.024	.021	0.	0.	122.	51.9	66.7	116.	242.	1.81	.929	SEP .821-04
MMH 228	XDT1	.024	.021	0.	0.	148.	61.3	78.7	121.	249.	1.80	.928	SEP .582-04
MMH 229	XDT1	.024	.021	0.	0.	119.	53.0	71.1	109.	245.	1.75	1.001	SEP .826-04
MMH 230	XDT1	.024	.021	0.	0.	122.	52.1	70.5	120.	268.	1.75	1.011	SEP .129-03
MMH 231	XDT1	.024	.021	0.	0.	145.	62.2	83.5	136.	280.	1.76	.999	SEP .190-03
MMH 232	XDT1	.024	.021	0.	0.	145.	63.7	86.0	162.	291.	1.72	1.022	SEP .141-03
MMH 233	XDT1	.024	.021	0.	0.	122.	53.3	73.5	153.	285.	1.69	1.060	SEP .386-05
MMH 234	SPLASH PLATE	.024	.021	0.	90.	125.	47.9	62.7	73.	76.	1.68	.996	M/S .254-05
MMH 235	SPLASH PLATE	.024	.021	0.	90.	102.	37.8	48.2	71.	72.	1.69	.983	M/S .551-05
MMH 236	SPLASH PLATE	.024	.021	0.	90.	81.	31.8	40.8	74.	76.	1.68	.997	MIX .724-05
MMH 237	SPLASH PLATE	.024	.021	0.	90.	148.	59.8	76.8	78.	82.	1.68	.999	SEP .488-05
MMH 238	SPLASH PLATE	.024	.021	0.	90.	186.	60.1	104.2	79.	79.	1.65	1.027	SEP .609-05
MMH 239	SPLASH PLATE	.024	.021	0.	90.	169.	53.2	70.1	74.	74.	1.64	1.052	SEP .664-05
MMH 240	SPLASH PLATE	.024	.021	0.	90.	160.	55.5	71.9	77.	79.	1.66	1.018	SEP .166-04
MMH 241	SPLASH PLATE	.024	.021	0.	90.	169.	53.4	68.4	78.	71.	1.67	1.001	SEP .412-04
MMH 242	SPLASH PLATE	.024	.021	0.	90.	192.	79.0	103.2	78.	75.	1.64	1.038	SEP .121-03
MMH 243	SPLASH PLATE	.024	.021	0.	90.	121.	49.9	66.3	79.	170.	1.72	1.015	SEP .135-03
MMH 244	SPLASH PLATE	.024	.021	0.	90.	117.	51.6	71.5	85.	239.	1.72	1.054	SEP .175-03
MMH 245	SPLASH PLATE	.024	.021	0.	90.	119.	54.4	73.1	136.	280.	1.70	1.069	SEP .124-04
MMH 246	SPLASH PLATE	.024	.021	0.	90.	139.	61.1	88.6	132.	283.	1.69	1.066	M/S .910-05
MMH 247	SPLASH PLATE	.024	.021	0.	90.	131.	64.7	89.0	132.	288.	1.69	1.058	MIX .759-05
MMH 248	SPLASH PLATE	.024	.021	0.	90.	147.	61.4	84.8	154.	267.	1.69	1.068	MIX .152-04
MMH 249	TRIPLT	.050	.029	4.	32.	125.	48.1	103.8	73.	75.	2.15	2.816	M/S .162-04
MMH 250	TRIPLT	.050	.029	4.	32.	125.	49.7	97.6	70.	75.	2.17	2.332	M/S .133-03
MMH 251	TRIPLT	.050	.029	4.	32.	98.	38.5	77.2	68.	78.	2.33	2.424	M/S .561-03
MMH 252	TRIPLT	.050	.029	4.	32.	80.	31.7	61.3	72.	78.	2.41	2.259	SEP .459-03
MMH 253	TRIPLT	.050	.029	4.	32.	140.	58.1	126.8	72.	76.	2.13	2.879	SEP .625-03
MMH 254	TRIPLT	.050	.029	4.	32.	194.	76.6	155.6	71.	69.	2.28	2.503	MIX .416-04
MMH 255	TRIPLT	.050	.029	4.	32.	119.	42.1	115.2	77.	176.	2.23	2.789	M/S .133-03
MMH 256	TRIPLT	.050	.029	4.	32.	116.	53.6	122.9	76.	204.	2.27	2.828	M/S .561-03
MMH 257	TRIPLT	.050	.029	4.	32.	113.	56.7	155.1	140.	312.	1.90	4.039	SEP .459-03
MMH 258	TRIPLT	.050	.029	4.	32.	131.	65.3	149.1	120.	300.	2.31	2.796	SEP .625-03
MMH 259	TRIPLT	.050	.029	4.	32.	137.	64.4	142.4	153.	301.	2.30	2.697	M/S .578-05
MMH 260	TRIPLT	.050	.029	4.	32.	126.	50.4	134.6	58.	71.	1.76	4.279	M/S .578-05
MMH 275	SMALL TRIPLT	.034	.020	4.	32.	126.	50.4	134.6	58.	71.	1.76	4.279	M/S .578-05

HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR LAWVER

FUEL TEST TYPE NO.	INJECTOR TYPE	DD (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VD (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	MODE	REACT (SEC)
MMH 276	SMALL TRIPLET	.034	.020	4.	32.	129.	47.0	131.7	78.	77.	1.66	4.760	MIX	.794-05
MMH 277	SMALL TRIPLET	.034	.020	4.	32.	101.	36.1	100.6	71.	80.	1.67	4.682	MIX	.582-05
MMH 278	SMALL TRIPLET	.034	.020	4.	32.	81.	30.5	77.4	71.	77.	1.64	3.887	MIX	.427-05
MMH 279	SMALL TRIPLET	.034	.020	4.	32.	157.	54.1	155.1	71.	75.	1.63	4.960	M/S	.828-05
MMH 280	SMALL TRIPLET	.034	.020	4.	32.	210.	72.2	205.4	72.	73.	1.64	4.896	SEP	.107-04
MMH 281	SMALL TRIPLET	.034	.020	4.	32.	129.	48.4	142.9	80.	150.	1.64	5.079	MIX	.255-04
MMH 282	SMALL TRIPLET	.034	.020	4.	32.	129.	48.8	148.9	85.	226.	1.66	5.163	MIX	.695-04
MMH 283	SMALL TRIPLET	.034	.020	4.	32.	132.	47.3	146.2	85.	250.	1.68	5.118	MIX	.868-04
MMH 284	SMALL TRIPLET	.034	.020	4.	32.	130.	48.3	152.0	130.	287.	1.66	5.213	M/S	.211-03
MMH 285	SMALL TRIPLET	.034	.020	4.	32.	156.	58.6	181.9	131.	296.	1.67	5.232	SEP	.279-03
MMH 286	SMALL TRIPLET	.034	.020	4.	32.	150.	61.3	190.0	145.	297.	1.66	5.280	SEP	.341-03
MMH 287	UNLIKE-DOUBLET	.024	.020	24.	60.	124.	49.2	72.5	72.	82.	1.63	1.309	MIX	.438-05
MMH 288	UNLIKE-DOUBLET	.024	.020	24.	60.	98.	39.3	57.6	73.	72.	1.62	1.299	M/P	.299-05
MMH 289	UNLIKE-DOUBLET	.024	.020	24.	60.	78.	32.5	50.1	75.	76.	1.54	1.439	M/P	.287-05
MMH 290	UNLIKE-DOUBLET	.024	.020	24.	60.	78.	33.2	48.2	74.	75.	1.62	1.294	M/P	.270-05
MMH 291	UNLIKE-DOUBLET	.024	.020	24.	60.	147.	58.9	84.3	75.	75.	1.66	1.241	MIX	.474-05
MMH 292	UNLIKE-DOUBLET	.024	.020	24.	60.	198.	78.9	113.9	76.	76.	1.65	1.265	M/S	.600-05
MMH 293	UNLIKE-DOUBLET	.024	.020	24.	60.	123.	50.7	76.7	75.	185.	1.68	1.293	SEP	.198-04
MMH 294	UNLIKE-DOUBLET	.024	.020	24.	60.	123.	51.0	78.5	80.	249.	1.74	1.282	SEP	.432-04
MMH 295	UNLIKE-DOUBLET	.024	.020	24.	60.	121.	53.0	82.4	116.	240.	1.65	1.360	SEP	.622-04
MMH 296	UNLIKE-DOUBLET	.024	.020	24.	60.	145.	64.5	99.4	31.	290.	1.71	1.296	SEP	.143-03
MMH 297	UNLIKE-DOUBLET	.024	.020	24.	60.	144.	66.6	104.6	152.	294.	1.65	1.371	SEP	.195-03

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGULIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR LAWVER

FUEL TEST TYPE NO.	MODE	PC (PSIA)	WAVG	WEF	WED	REF	RED	DELTA (DEG F)	RELF	RELO (PSIA)(PSIA)	PPF	PPU	HRVP	MF	XU	XP	RESID
MMH 276	MIX	129.	79.	16.7	5.4	230+05	405+05	44.	.92+05	.18+06	1.0	18.1	55.2	.01	.14	.34+01	.13+04
MMH 277	MIX	101.	60.	6.6	2.4	150+05	328+05	44.	.72+05	.13+06	1.1	15.2	27.6	.01	.15	.41+01	.17+04
MMH 278	MIX	81.	47.	4.1	1.4	135+05	277+05	44.	.54+05	.11+06	1.0	15.2	29.8	.01	.19	.49+01	.22+04
MMH 279	M/S	157.	93.	31.5	8.4	266+05	492+05	44.	.11+06	.20+06	1.0	15.2	31.6	.01	.10	.24+01	.11+04
MMH 280	SEP	210.	123.	73.7	20.2	336+05	600+05	44.	.14+06	.28+06	.9	15.6	34.4	.00	.07	.18+01	.81+05
MMH 281	MIX	129.	84.	25.1	5.8	432+05	463+05	44.	.17+06	.19+06	6.7	19.0	6.4	.05	.15	.87+01	.12+04
MMH 282	MIX	129.	86.	31.7	6.0	727+05	482+05	44.	.29+06	.19+06	29.6	21.6	1.8	.23	.17	.20+01	.11+04
MMH 283	M/S	130.	88.	33.0	5.8	626+05	487+05	44.	.33+06	.19+06	43.1	21.6	1.3	.33	.14	.23+06	.11+04
MMH 284	M/S	130.	88.	33.3	8.1	107+06	627+05	44.	.43+06	.25+06	71.2	60.0	2.1	.55	.46	.50+00	.11+04
MMH 285	SEP	156.	105.	67.4	13.9	136+06	751+05	44.	.54+06	.30+06	80.9	61.4	1.9	.52	.39	.45+00	.92+05
MMH 286	SEP	150.	110.	70.8	16.1	143+06	855+05	44.	.57+06	.34+06	82.1	82.6	2.5	.55	.55	.55+00	.88+05
MMH 287	MIX	124.	58.	5.5	3.9	132+05	321+05	44.	.37+06	.77+06	1.2	15.6	27.0	.01	.13	.35+01	.23+04
MMH 288	M/P	98.	46.	2.7	2.0	900+04	258+05	44.	.23+06	.62+06	.9	16.0	36.4	.01	.16	.38+01	.24+04
MMH 289	M/P	78.	39.	1.8	1.1	806+04	218+05	44.	.21+06	.52+06	1.0	16.9	33.8	.01	.22	.53+01	.33+04
MMH 290	M/P	78.	39.	1.5	1.1	832+04	219+05	44.	.20+06	.53+06	1.0	16.5	34.0	.01	.21	.51+01	.34+04
MMH 291	MIX	147.	64.	8.7	6.8	144+05	391+05	44.	.35+06	.94+06	1.0	16.9	34.8	.01	.11	.28+01	.20+04
MMH 292	M/S	196.	92.	21.5	16.5	197+05	527+05	44.	.47+06	.13+07	1.0	17.3	34.6	.01	.09	.21+01	.15+04
MMH 293	SEP	123.	60.	7.4	4.2	291+05	337+05	44.	.70+06	.81+06	14.0	16.9	2.9	.11	.14	.12+00	.22+04
MMH 294	SEP	123.	61.	6.6	4.4	441+05	349+05	44.	.11+07	.84+06	42.3	19.0	1.2	.34	.15	.23+00	.21+04
MMH 295	SEP	121.	64.	9.4	5.7	439+05	447+05	44.	.17+07	.14+07	55.5	44.4	3.0	.29	.37	.33+00	.20+04
MMH 296	SEP	145.	77.	18.4	11.2	716+05	590+05	44.	.19+07	.17+07	73.5	61.4	2.1	.51	.42	.40+00	.17+04
MMH 297	SEP	144.	81.	20.5	13.7	772+05	693+05	44.	.19+07	.17+07	78.4	95.1	3.0	.54	.66	.60+00	.16+04

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR RCKTD

FUEL TEST TYPE NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR MF/MD	MODE	REACT (SEC)
MMH	9 UNLIKE-DOUBLET	.024	.020	12.	60.	63.	41.7	66.8	50.	52.	1.49	MIX	.180-05
MMH	10 UNLIKE-DOUBLET	.024	.020	12.	60.	74.	41.7	65.1	49.	52.	1.53	MIX	.173-05
MMH	17 UNLIKE-DOUBLET	.024	.020	12.	60.	62.	35.6	48.5	52.	55.	1.75	MIX	.142-05
MMH	18 UNLIKE-DOUBLET	.024	.020	12.	60.	61.	36.0	44.3	50.	56.	1.94	MIX	.128-05
MMH	19 UNLIKE-DOUBLET	.024	.020	12.	60.	67.	42.2	56.1	100.	84.	1.75	MIX	.488-05
MMH	20 UNLIKE-DOUBLET	.024	.020	12.	60.	69.	45.0	57.8	114.	84.	1.79	MIX	.592-05
MMH	21 UNLIKE-DOUBLET	.024	.020	12.	60.	67.	52.6	73.2	109.	86.	1.66	MIX	.727-05
MMH	22 UNLIKE-DOUBLET	.024	.020	12.	60.	114.	38.7	54.5	84.	88.	1.68	MIX	.415-05
MMH	23 UNLIKE-DOUBLET	.024	.020	12.	60.	115.	50.0	71.6	88.	90.	1.65	MIX	.585-05
MMH	24 UNLIKE-DOUBLET	.024	.020	12.	60.	166.	32.3	50.1	104.	88.	1.50	MIX	.483-05
MMH	25 UNLIKE-DOUBLET	.024	.020	12.	60.	214.	31.3	46.5	103.	92.	1.57	MIX	.473-05
MMH	30 UNLIKE-DOUBLET	.024	.020	12.	60.	109.	37.2	57.5	84.	77.	1.52	MIX	.374-05
MMH	31 UNLIKE-DOUBLET	.024	.020	12.	60.	119.	46.1	65.5	81.	109.	1.69	MIX	.669-05
MMH	32 UNLIKE-DOUBLET	.024	.020	12.	60.	114.	47.1	71.1	77.	121.	1.61	MIX	.825-05
MMH	33 UNLIKE-DOUBLET	.024	.020	12.	60.	171.	32.6	45.3	76.	95.	1.72	MIX	.353-05
MMH	35 UNLIKE-DOUBLET	.024	.020	12.	60.	69.	48.5	70.9	74.	115.	1.66	MIX	.716-05
MMH	36 UNLIKE-DOUBLET	.024	.020	12.	60.	112.	38.4	56.1	78.	113.	1.66	MIX	.553-05
MMH	37 UNLIKE-DOUBLET	.024	.020	12.	60.	117.	53.0	71.7	72.	136.	1.62	SEP	.956-05
MMH	38 UNLIKE-DOUBLET	.024	.020	12.	60.	162.	38.0	50.5	73.	128.	1.84	MIX	.618-05
MMH	39 UNLIKE-DOUBLET	.024	.020	12.	60.	185.	48.1	69.9	75.	136.	1.69	SEP	.969-05
MMH	43 UNLIKE-DOUBLET	.040	.033	12.	60.	67.	36.7	48.2	89.	114.	1.86	MIX	.157-04
MMH	45 UNLIKE-DOUBLET	.040	.033	12.	60.	71.	36.1	58.5	90.	117.	1.51	MIX	.202-04
MMH	46 UNLIKE-DOUBLET	.040	.033	12.	60.	71.	51.9	65.5	89.	119.	1.94	SEP	.232-04
MMH	48 UNLIKE-DOUBLET	.040	.033	12.	60.	122.	36.0	57.8	84.	116.	1.53	MIX	.183-04
MMH	50 UNLIKE-DOUBLET	.040	.033	12.	60.	126.	49.5	69.8	85.	125.	1.75	SEP	.259-04
MMH	51 UNLIKE-DOUBLET	.040	.033	12.	60.	122.	50.9	73.6	85.	129.	1.71	SEP	.288-04
MMH	52 UNLIKE-DOUBLET	.040	.033	12.	60.	62.	40.7	44.7	97.	124.	2.22	SEP	.108-04
MMH	54 UNLIKE-DOUBLET	.040	.033	12.	60.	113.	38.5	44.6	98.	107.	2.08	MIX	.148-04
MMH	55 UNLIKE-DOUBLET	.040	.033	12.	60.	67.	54.3	54.4	98.	107.	2.41	SEP	.177-04
MMH	56 UNLIKE-DOUBLET	.040	.033	12.	60.	114.	53.8	57.0	101.	121.	2.29	SEP	.240-04
MMH	59 UNLIKE-DOUBLET	.040	.033	12.	60.	164.	53.1	69.1	99.	119.	2.29	MIX	.224-04
MMH	60 UNLIKE-DOUBLET	.040	.033	12.	60.	64.	40.3	47.1	100.	131.	2.09	SEP	.226-04
MMH	63 UNLIKE-DOUBLET	.040	.033	12.	60.	113.	57.9	60.4	98.	147.	2.37	SEP	.351-04
MMH	64 UNLIKE-DOUBLET	.040	.033	12.	60.	64.	56.2	63.2	99.	164.	2.62	SEP	.456-04
MMH	66 UNLIKE-DOUBLET	.040	.033	12.	60.	114.	42.0	44.1	99.	153.	2.36	SEP	.280-04
MMH	67 UNLIKE-DOUBLET	.024	.020	12.	60.	111.	57.5	62.1	110.	192.	2.33	SEP	.722-04
MMH	68 UNLIKE-DOUBLET	.024	.020	12.	60.	62.	38.5	54.8	103.	95.	1.64	MIX	.590-05
MMH	69 UNLIKE-DOUBLET	.024	.020	12.	60.	61.	53.7	59.4	98.	95.	2.12	MIX	.610-05
MMH	70 UNLIKE-DOUBLET	.024	.020	12.	60.	104.	37.4	34.1	98.	90.	2.57	MIX	.307-05
MMH						104.	55.9	53.9	98.	96.	2.44	MIX	.544-05

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HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR RCKTD

FUEL TYPE	TEST NO.	MODE	PC (PSIA)	VAVE	WEF	WEO	REF	HEU	DELTI (DEG F)	RELF	RELO (PSIA)	PPF (PSIA)	MRVP	XF	XU	XP	RESID	
MMH	9	MIX	63.	52.	2.3	1.3	914+04	243+05	0.	11+06	29+06	.5	8.6	36.4	.01	.14	.32-01	.25-04
MMH	10	MIX	74.	51.	2.5	1.5	891+04	243+05	0.	11+06	29+06	.5	8.4	35.6	.01	.11	.27-01	.26-04
MMH	17	MIX	62.	40.	1.2	.9	685+04	209+05	0.	82+05	25+06	.5	9.2	35.7	.01	.15	.35-01	.34-04
MMH	18	MIX	61.	39.	1.0	.9	632+04	209+05	0.	76+05	25+06	.5	8.6	34.5	.01	.14	.35-01	.34-04
MMH	19	MIX	67.	47.	1.8	1.8	104+05	329+05	0.	12+06	39+06	1.2	30.7	48.1	.02	.46	.92-01	.30-04
MMH	20	MIX	69.	50.	2.0	2.3	107+05	376+05	0.	13+06	45+06	1.2	42.5	64.9	.02	.82	.11+00	.29-04
MMH	21	MIX	67.	60.	3.1	3.0	138+05	428+05	0.	17+06	51+06	1.3	37.8	55.8	.02	.56	.11+00	.23-04
MMH	22	MIX	114.	45.	2.9	2.4	104+05	271+05	0.	13+06	33+06	1.4	21.1	31.2	.01	.18	.47-01	.31-04
MMH	23	MIX	115.	58.	5.0	4.1	140+05	360+05	0.	17+06	43+06	1.4	23.2	38.7	.01	.20	.50-01	.23-04
MMH	24	MIX	166.	39.	3.6	2.7	960+04	257+05	0.	12+06	31+06	1.4	33.9	48.4	.01	.20	.61-01	.33-04
MMH	25	MIX	214.	37.	4.0	3.3	921+04	248+05	0.	11+06	30+06	1.5	33.1	42.4	.01	.15	.33-01	.36-04
MMH	30	MIX	109.	45.	3.1	2.1	100+05	261+05	0.	12+06	31+06	1.0	21.1	40.4	.01	.19	.43-01	.29-04
MMH	31	MIX	119.	53.	4.5	3.5	148+05	317+05	0.	18+06	38+06	2.5	19.5	11.6	.02	.16	.58-01	.25-04
MMH	32	MIX	114.	56.	5.2	3.4	176+05	317+05	0.	21+06	38+06	3.4	17.7	11.4	.03	.16	.68-01	.23-04
MMH	33	MIX	171.	37.	3.0	2.4	914+04	218+05	0.	11+06	26+06	1.7	17.3	21.1	.01	.10	.32-01	.37-04
MMH	35	MIX	69.	57.	3.1	2.2	168+05	329+05	0.	20+06	38+06	2.8	16.5	12.7	.04	.24	.98-01	.24-04
MMH	36	MIX	112.	45.	3.1	2.2	131+05	254+05	0.	16+06	30+06	2.7	16.5	13.1	.02	.15	.59-01	.30-04
MMH	37	S/P	117.	60.	5.6	4.3	197+05	346+05	0.	24+06	42+06	4.8	15.6	7.5	.04	.13	.74-01	.23-04
MMH	38	MIX	162.	42.	3.8	3.1	131+05	249+05	0.	16+06	30+06	4.0	16.0	7.8	.02	.10	.50-01	.35-04
MMH	39	S/P	165.	56.	6.4	5.7	192+05	329+05	0.	23+06	38+06	4.8	16.9	7.9	.03	.09	.49-01	.24-04
MMH	43	MIX	69.	41.	2.4	2.2	187+05	493+05	0.	22+06	53+06	2.7	23.7	18.0	.04	.34	.12+00	.57-04
MMH	45	MIX	71.	45.	3.6	2.2	232+05	439+05	0.	28+06	53+06	3.0	24.2	17.0	.04	.34	.12+00	.47-04
MMH	46	S/P	71.	57.	4.5	4.6	263+05	629+05	0.	32+06	75+06	3.2	23.7	15.7	.04	.33	.12+00	.42-04
MMH	48	MIX	122.	45.	6.0	3.7	227+05	421+05	0.	27+06	51+06	2.9	21.1	15.4	.02	.17	.64-01	.39-04
MMH	49	S/P	122.	57.	8.9	7.0	293+05	582+05	0.	35+06	70+06	3.7	21.6	12.4	.03	.18	.74-01	.39-04
MMH	50	S/P	122.	59.	10.0	7.4	318+05	599+05	0.	38+06	72+06	4.1	21.6	11.3	.03	.18	.77-01	.37-04
MMH	51	S/P	62.	42.	1.9	2.6	186+05	518+05	0.	22+06	42+06	3.6	28.7	16.5	.04	.46	.17+00	.62-04
MMH	52	MIX	113.	40.	3.3	4.2	164+05	493+05	0.	20+06	59+06	2.4	28.4	25.3	.02	.26	.74-01	.62-04
MMH	54	S/P	67.	54.	2.9	4.9	200+05	689+05	0.	24+06	62+06	2.4	28.1	24.3	.04	.42	.12+00	.51-04
MMH	55	S/P	114.	55.	5.5	8.4	232+05	703+05	0.	28+06	64+06	3.4	31.5	19.6	.03	.28	.90-01	.48-04
MMH	56	MIX	164.	54.	7.7	11.7	226+05	685+05	0.	27+06	62+06	3.2	30.0	19.6	.02	.18	.60-01	.49-04
MMH	59	S/P	64.	42.	2.4	2.6	208+05	524+05	0.	25+06	63+06	4.3	30.7	15.1	.07	.48	.18+00	.56-04
MMH	60	S/P	113.	54.	6.4	9.5	293+05	742+05	0.	35+06	69+06	6.2	29.4	10.3	.06	.26	.12+00	.46-04
MMH	65	S/P	64.	58.	4.1	5.1	343+05	726+05	0.	41+06	87+06	8.8	30.0	7.6	.14	.47	.25+00	.44-04
MMH	66	S/P	114.	43.	3.5	5.1	244+05	542+05	0.	27+06	65+06	7.2	30.0	9.2	.06	.26	.13+00	.62-04
MMH	67	MIX	111.	59.	7.5	9.9	496+05	784+05	0.	49+06	94+06	10.0	38.6	5.5	.14	.35	.22+00	.44-04
MMH	67	MIX	62.	45.	1.6	1.4	111+05	504+05	0.	13+06	37+06	1.7	33.1	38.4	.03	.53	.12+00	.30-04
MMH	68	MIX	61.	56.	1.9	2.7	121+05	416+05	0.	14+06	50+06	1.7	30.0	35.2	.03	.49	.12+00	.28-04
MMH	69	MIX	104.	36.	1.0	2.2	664+04	284+05	0.	60+05	34+06	1.4	28.1	39.1	.01	.27	.61-01	.49-04
MMH	70	MIX	104.	55.	1.6	4.8	110+05	424+05	0.	13+06	51+06	1.8	28.1	32.1	.02	.27	.67-01	.31-04

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HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR RCKTD

FUEL TEST TYPE NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (P)	TF (P)	MR MF/MO	MODE	REACT (SEC)
MMH 74	UNLIKE-DOUBLET	.024	.020	12	60	59	39.8	41.7	97	87	2.23	MIX	.366-05
MMH 75	UNLIKE-DOUBLET	.024	.020	12	60	106	37.8	40.1	92	92	2.22	MIX	.358-05
MMH 76	UNLIKE-DOUBLET	.024	.020	12	60	104	53.6	52.1	88	97	2.44	MIX	.486-05
MMH 82	UNLIKE-DOUBLET	.024	.020	12	60	62	41.2	42.9	93	101	2.27	UNDEF	.454-05
MMH 83	UNLIKE-DOUBLET	.024	.020	12	60	108	39.1	39.7	86	110	2.36	MIX	.438-05
MMH 84	UNLIKE-DOUBLET	.024	.020	12	60	154	38.4	38.2	88	123	2.42	MIX	.523-05
MMH 85	UNLIKE-DOUBLET	.024	.020	12	60	58	38.3	47.4	101	110	1.91	MIX	.623-05
MMH 86	UNLIKE-DOUBLET	.024	.020	12	60	58	38.2	47.8	98	125	1.91	UNDEF	.759-05
MMH 87	UNLIKE-DOUBLET	.024	.020	12	60	58	38.5	48.1	97	137	1.93	MIX	.860-05
MMH 88	UNLIKE-DOUBLET	.024	.020	12	60	58	39.5	43.2	94	117	2.18	MIX	.550-05
MMH 89	UNLIKE-DOUBLET	.024	.020	12	60	64	38.6	41.0	94	132	2.27	MIX	.683-05
MMH 90	UNLIKE-DOUBLET	.024	.020	12	60	64	38.5	41.0	97	143	2.27	MIX	.815-05
MMH 91	UNLIKE-DOUBLET	.024	.020	12	60	59	53.7	60.4	94	137	2.15	MIX	.107-04
MMH 92	UNLIKE-DOUBLET	.024	.020	12	60	62	50.0	54.0	102	159	2.25	MIX	.141-04
MMH 93	UNLIKE-DOUBLET	.024	.020	12	60	62	49.2	54.3	103	172	2.22	MIX	.169-04
MMH 98	UNLIKE-DOUBLET	.040	.033	12	60	25	39.1	58.8	73	78	1.61	MIX	.923-05
MMH 99	UNLIKE-DOUBLET	.040	.033	12	60	31	43.9	57.8	74	78	1.84	MIX	.919-05
MMH 100	UNLIKE-DOUBLET	.040	.033	12	60	51	37.9	59.9	69	81	1.54	UNDEF	.934-05
MMH 101	UNLIKE-DOUBLET	.040	.033	12	60	31	39.5	61.4	71	76	1.56	MIX	.908-05
MMH 102	UNLIKE-DOUBLET	.040	.033	12	60	74	39.3	55.1	74	81	1.73	MIX	.917-05
MMH 103	UNLIKE-DOUBLET	.040	.033	12	60	108	40.1	54.6	73	78	1.78	MIX	.857-05
MMH 104	UNLIKE-DOUBLET	.040	.033	12	60	113	54.9	75.6	74	79	1.76	MIX	.122-04
MMH 105	UNLIKE-DOUBLET	.040	.033	12	60	165	38.7	58.4	71	78	1.61	MIX	.892-05
MMH 106	UNLIKE-DOUBLET	.040	.033	12	60	34	38.4	53.3	73	90	1.74	MIX	.987-05
MMH 107	UNLIKE-DOUBLET	.040	.033	12	60	54	38.9	53.2	71	94	1.79	MIX	.104-04
MMH 118	UNLIKE-DOUBLET	.040	.033	12	60	36	40.4	57.8	89	78	1.67	MIX	.110-04
MMH 119	UNLIKE-DOUBLET	.040	.033	12	60	51	40.7	54.8	88	81	1.78	MIX	.108-04
MMH 120	UNLIKE-DOUBLET	.040	.033	12	60	69	41.8	57.4	89	79	1.74	MIX	.111-04
MMH 121	UNLIKE-DOUBLET	.040	.033	12	60	109	39.6	54.3	96	78	1.73	MIX	.113-04
MMH 122	UNLIKE-DOUBLET	.040	.033	12	60	108	56.3	77.1	96	83	1.73	MIX	.172-04
MMH 123	UNLIKE-DOUBLET	.040	.033	12	60	160	39.3	53.5	96	62	1.75	SEP	.118-04
MMH 124	UNLIKE-DOUBLET	.040	.033	12	60	38	39.6	56.7	97	83	1.66	SEP	.128-04
MMH 125	UNLIKE-DOUBLET	.040	.033	12	60	45	53.5	77.6	99	84	1.64	SEP	.182-04
MMH 126	UNLIKE-DOUBLET	.040	.033	12	60	42	37.9	55.8	94	102	1.65	MIX	.168-04
MMH 127	UNLIKE-DOUBLET	.040	.033	12	60	51	38.9	56.9	97	116	1.66	SEP	.211-04
MMH 128	UNLIKE-DOUBLET	.040	.033	12	60	69	40.3	54.3	98	123	1.81	SEP	.228-04
MMH 129	UNLIKE-DOUBLET	.040	.033	12	60	108	38.0	55.3	98	130	1.68	SEP	.256-04
MMH 130	UNLIKE-DOUBLET	.040	.033	12	60	108	52.0	74.1	101	148	1.73	SEP	.451-04
MMH 131	UNLIKE-DOUBLET	.040	.033	12	60	100	37.5	50.9	100	140	1.61	SEP	.271-04
MMH 132	UNLIKE-DOUBLET	.040	.033	12	60	38	36.0	52.0	99	137	1.70	MIX	.262-04

HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR RCKTD

FUEL TEST TYPE NO.	MODE	PC (PSIA)	VAVG	WEF	WEO	REF	REO	DELTI (DEG F)	HELF	RELU	OPF (PSIA)	PPU	MRVP	XF	XQ	XP	RESID
MMH 74	MIX	59.	40.	1.9	1.4	793+04	304+05	0.	.95+05	36+06	1.3	28.7	42.5	.02	.49	.10+00	.40-04
MMH 75	MIX	106.	39.	1.5	2.2	794+04	274+05	0.	.95+05	33+06	1.5	25.5	33.3	.01	.24	.59-01	.42-04
MMH 76	MIX	104.	53.	2.4	4.3	107+05	386+05	0.	.13+06	46+06	1.8	23.2	26.1	.02	.22	.62-01	.32-04
MMH 82	UNDEF	62.	42.	1.0	1.5	913+04	506+05	0.	.11+06	37+06	2.0	26.2	26.2	.03	.42	.12+00	.39-04
MMH 83	MIX	108.	39.	1.5	2.2	905+04	278+05	0.	.11+06	33+06	2.5	22.1	17.3	.02	.21	.12-01	.42-04
MMH 84	MIX	154.	36.	2.0	3.2	957+04	276+05	0.	.11+06	33+06	3.6	23.2	13.9	.02	.15	.59-01	.44-04
MMH 85	MIX	56.	41.	1.2	1.3	108+05	300+05	0.	.13+06	36+06	2.5	31.5	25.3	.04	.54	.15+00	.35-04
MMH 86	UNDEF	58.	42.	1.2	1.3	122+05	294+05	0.	.15+06	35+06	3.7	29.4	16.5	.06	.51	.14+00	.35-04
MMH 87	MIX	58.	41.	1.2	1.3	133+05	294+05	0.	.16+06	35+06	4.9	28.7	12.7	.08	.50	.20+00	.35-04
MMH 88	MIX	59.	41.	1.0	1.4	101+05	296+05	0.	.12+06	35+06	2.7	26.8	20.2	.05	.45	.15+00	.39-04
MMH 89	MIX	64.	39.	1.0	1.4	109+05	269+05	0.	.13+06	35+06	4.4	26.8	13.0	.07	.42	.17+00	.41-04
MMH 90	MIX	64.	39.	1.0	1.4	118+05	294+05	0.	.14+06	35+06	5.6	28.7	11.1	.09	.45	.20+00	.41-04
MMH 91	MIX	59.	56.	2.0	2.5	167+05	402+05	0.	.20+06	48+06	4.9	26.4	11.9	.08	.45	.14+00	.26-04
MMH 92	MIX	62.	51.	1.7	2.4	173+05	393+05	0.	.21+06	47+06	8.1	32.3	8.8	.13	.52	.26+00	.31-04
MMH 93	MIX	62.	51.	1.8	2.3	189+05	389+05	0.	.23+06	47+06	10.7	33.1	6.9	.17	.53	.30+00	.31-04
MMH 96	MIX	25.	47.	1.2	.8	171+05	428+05	0.	.20+06	51+06	1.1	16.0	30.5	.04	.64	.16+00	.47-04
MMH 99	MIX	51.	49.	1.4	1.3	168+05	484+05	0.	.20+06	58+06	1.1	16.5	31.3	.03	.53	.13+00	.48-04
MMH 100	UNDEF	51.	47.	2.5	1.6	179+05	405+05	0.	.21+06	49+06	1.1	14.5	25.7	.02	.24	.40-01	.48-04
MMH 101	MIX	31.	48.	1.6	1.1	175+05	428+05	0.	.21+06	51+06	1.1	15.2	31.7	.03	.44	.13+00	.45-04
MMH 102	MIX	74.	45.	3.1	2.5	166+05	433+05	0.	.20+06	52+06	1.1	16.5	30.5	.01	.22	.54-01	.50-04
MMH 103	MIX	108.	45.	4.5	3.8	159+05	439+05	0.	.19+06	53+06	1.1	16.0	30.5	.01	.15	.38-01	.50-04
MMH 104	MIX	113.	62.	9.0	7.5	221+05	605+05	0.	.27+06	73+06	1.1	16.5	30.5	.01	.15	.37-01	.36-04
MMH 105	MIX	165.	46.	7.8	5.4	170+05	419+05	0.	.20+06	50+06	1.1	15.2	29.1	.01	.04	.24-01	.47-04
MMH 106	MIX	34.	44.	1.4	1.1	171+05	421+05	0.	.21+06	51+06	1.4	16.0	23.3	.04	.47	.14+00	.50-04
MMH 107	MIX	54.	44.	2.2	1.8	177+05	421+05	0.	.21+06	50+06	1.6	15.2	19.3	.03	.28	.42-01	.52-04
MMH 110	MIX	36.	47.	1.7	1.4	164+05	487+05	0.	.20+06	58+06	1.1	23.7	43.8	.03	.66	.14+00	.48-04
MMH 119	MIX	51.	46.	2.1	2.0	163+05	488+05	0.	.20+06	59+06	1.1	23.2	39.8	.02	.45	.10+00	.50-04
MMH 120	MIX	69.	47.	3.2	2.9	168+05	504+05	0.	.20+06	60+06	1.1	23.7	42.7	.02	.34	.14+00	.50-04
MMH 121	MIX	109.	45.	4.4	4.2	158+05	500+05	0.	.19+06	60+06	1.1	28.1	51.3	.01	.26	.40-01	.51-04
MMH 122	MIX	104.	64.	9.0	6.5	234+05	712+05	0.	.26+06	85+06	1.2	28.1	45.4	.03	.26	.44-01	.51-04
MMH 123	MIX	101.	44.	0.4	6.2	161+05	497+05	0.	.19+06	60+06	1.2	28.1	46.4	.01	.14	.36-01	.51-04
MMH 124	MIX	34.	46.	1.7	1.5	172+05	503+05	0.	.21+06	60+06	1.2	28.7	48.3	.03	.76	.16+00	.49-04
MMH 125	MIX	45.	63.	3.6	3.3	237+05	491+05	0.	.26+06	83+06	1.2	30.0	47.2	.03	.67	.14+00	.35-04
MMH 126	MIX	114	45.	1.9	1.5	196+05	480+05	0.	.24+06	58+06	2.1	28.1	27.3	.05	.67	.14+00	.50-04
MMH 127	MIX	51.	46.	2.9	1.9	224+05	495+05	0.	.27+06	59+06	2.9	28.7	20.5	.06	.56	.14+00	.48-04
MMH 129	MIX	74.	45.	3.0	2.8	225+05	517+05	0.	.27+06	62+06	3.6	29.4	17.3	.05	.44	.15+00	.51-04
MMH 130	MIX	108.	44.	5.6	5.9	220+05	487+05	0.	.29+06	58+06	4.2	29.4	14.8	.04	.27	.10+00	.50-04
MMH 131	MIX	104.	66.	9.1	7.3	264+05	674+05	0.	.44+06	81+06	6.4	31.5	10.7	.06	.30	.13+00	.47-04
MMH 132	MIX	102.	62.	8.4	5.7	237+05	487+05	0.	.28+06	66+06	5.1	30.7	12.4	.03	.19	.29-01	.54-04
MMH 133	MIX	34.	42.	1.0	1.2	237+05	461+05	0.	.28+06	55+06	4.9	29.4	12.9	.13	.77	.31+00	.53-04

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR RCKTD

FUEL TEST TYPE NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR WF/MO	MODE	REACT (SEC)
MMH 133	UNLIKE-DOUBLET	.040	.033	12.	60.	45.	51.4	74.3	101.	143.	1.70	.000	MIX .421-04
MMH 134	UNLIKE-DOUBLET	.040	.033	12.	60.	37.	35.9	55.8	95.	142.	1.59	.000	SEP .290-04
MMH 135	UNLIKE-DOUBLET	.040	.033	12.	60.	56.	37.5	54.3	98.	142.	1.70	.000	SEP .292-04
MMH 136	UNLIKE-DOUBLET	.040	.033	12.	60.	69.	38.5	55.0	102.	152.	1.73	.000	SEP .558-04
MMH 137	UNLIKE-DOUBLET	.040	.033	12.	60.	110.	37.3	54.9	103.	155.	1.68	.000	SEP .375-04
MMH 138	UNLIKE-DOUBLET	.040	.033	12.	60.	106.	52.0	75.5	107.	168.	1.71	.000	SEP .033-04
MMH 140	UNLIKE-DOUBLET	.040	.033	12.	60.	40.	35.6	55.2	66.	155.	1.65	.000	SEP .242-04
MMH 141	UNLIKE-DOUBLET	.040	.033	12.	60.	39.	34.6	54.2	55.	169.	1.66	.000	SEP .244-04
MMH 142	UNLIKE-DOUBLET	.040	.033	12.	60.	52.	36.0	51.1	60.	157.	1.81	.000	SEP .213-04
MMH 143	UNLIKE-DOUBLET	.040	.033	12.	60.	58.	35.4	50.2	54.	172.	1.84	.000	SEP .232-04
MMH 144	UNLIKE-DOUBLET	.040	.033	12.	60.	71.	35.9	54.2	57.	167.	1.72	.000	SEP .244-04
MMH 145	UNLIKE-DOUBLET	.040	.033	12.	60.	72.	35.3	53.4	53.	175.	1.73	.000	SEP .252-04

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR RCKTD

FUEL TEST TYPE NO.	MODE	PC (PSIA)	WVG	WEP	WEO	REF	WEO	DELTI (DEG F)	REL F	RELO (PSIA)(PSIA)	PPF	P2U	MRVP	XF	XII	XP	WESIU
MNH 133	MIX	45.	60.	3.8	3.0	.353+05	.67+05	0.	.42+06	.80+06	5.6	31.5	12.1	.12	.70	.30+00	.37+04
MNH 134	SEP	37.	44.	1.8	1.2	.263+05	.451+05	0.	.32+06	.54+06	5.5	27.5	10.9	.15	.74	.33+00	.49+04
MNH 135	SEP	56.	44.	2.6	2.0	.256+05	.483+05	0.	.33+06	.53+06	5.5	29.4	11.6	.10	.52	.23+00	.51+04
MNH 136	SEP	69.	45.	3.3	2.6	.278+05	.505+05	0.	.33+06	.61+06	7.0	32.3	10.0	.10	.47	.22+00	.50+04
MNH 137	SEP	110.	44.	5.3	4.0	.283+05	.492+05	0.	.34+06	.59+06	7.5	33.1	9.7	.07	.30	.14+00	.50+04
MNH 138	SEP	106.	41.	9.8	7.6	.423+05	.696+05	0.	.51+06	.84+06	9.7	36.2	8.2	.09	.34	.18+00	.36+04
MNH 140	SEP	40.	43.	1.9	1.1	.284+05	.375+05	0.	.34+06	.45+06	7.5	33.5	4.2	.14	.34	.25+00	.50+04
MNH 141	SEP	39.	42.	1.9	.9	.316+05	.344+05	0.	.37+06	.41+06	10.0	10.1	2.5	.26	.26	.26+00	.51+04
MNH 142	SEP	52.	41.	2.2	1.4	.267+05	.367+05	0.	.32+06	.44+06	7.8	11.7	3.6	.15	.22	.18+00	.54+04
MNH 143	SEP	52.	41.	2.1	1.3	.289+05	.350+05	0.	.35+06	.42+06	10.7	9.8	2.2	.21	.19	.20+00	.55+04
MNH 144	SEP	71.	43.	3.4	1.9	.302+05	.361+05	0.	.36+06	.43+06	9.5	10.7	2.7	.13	.12	.14+00	.51+04
MNH 145	SEP	72.	42.	3.4	1.8	.313+05	.368+05	0.	.38+06	.42+06	11.5	9.5	2.0	.16	.13	.14+00	.51+04

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HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TEST TYPE NO.	INJECTION TYPE	DI (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TD (F)	TF (F)	MR MF/MU	MUDE	REACT (SEC)
N2H4	15 UNLINF-DOUBLET	.027	.027	4	60	15	23.0	35.0	70	70	.94	1.620	SEP .849-06
N2H4	20 UNLINF-DOUBLET	.027	.027	4	60	15	49.0	75.0	70	70	.84	2.050	SEP .182-05
N2H4	58 UNLINF-DOUBLET	.027	.027	4	60	15	55.0	46.0	69	69	1.78	.470	MIX .108-05
N2H4	61 UNLINF-DOUBLET	.027	.027	4	60	15	42.0	40.0	70	70	1.50	.640	SEP .970-06
N2H4	63 UNLINF-DOUBLET	.027	.027	4	60	15	41.0	52.0	67	70	1.14	1.100	MIX .122-05
N2H4	64 UNLINF-DOUBLET	.027	.027	4	60	15	45.0	53.0	65	68	1.21	1.020	MIX .117-05
N2H4	68 UNLINF-DOUBLET	.027	.027	4	60	15	71.0	78.0	70	70	1.29	.860	SEP .189-05
N2H4	70 UNLINF-DOUBLET	.027	.027	4	60	15	20.0	25.0	70	70	1.11	1.160	SEP .608-06
N2H4	80 UNLINF-DOUBLET	.027	.027	4	60	15	41.0	54.0	68	68	1.09	1.220	MIX .124-05
N2H4	81 UNLINF-DOUBLET	.027	.027	4	60	15	34.0	53.0	70	70	.91	1.750	SEP .129-05
N2H4	82 UNLINF-DOUBLET	.027	.027	4	60	15	40.0	52.0	70	70	.02	1.700	SEP .126-05
N2H4	83 UNLINF-DOUBLET	.027	.027	4	60	15	41.0	52.0	70	70	1.08	1.200	SEP .126-05
N2H4	85 UNLINF-DOUBLET	.027	.027	4	60	15	38.0	48.0	70	70	1.02	1.080	SEP .118-05
N2H4	89 UNLINF-DOUBLET	.027	.027	4	60	15	53.0	94.0	67	68	1.10	1.180	MIX .544-06
N2H4	90 UNLINF-DOUBLET	.027	.027	4	60	15	53.0	94.0	67	67	.82	2.180	MIX .209-05
N2H4	91 UNLINF-DOUBLET	.027	.027	4	60	15	53.0	94.0	70	70	.82	2.180	SEP .228-05
N2H4	92 UNLINF-DOUBLET	.027	.027	4	60	15	65.0	92.0	70	70	1.02	1.380	SEP .223-05
N2H4	93 UNLINF-DOUBLET	.027	.027	4	60	15	43.0	60.0	60	60	1.04	1.310	MIX .106-05
N2H4	94 UNLINF-DOUBLET	.027	.027	4	60	15	39.0	56.0	45	45	1.31	.830	MIX .589-06
N2H4	95 UNLINF-DOUBLET	.027	.027	4	60	15	61.0	67.0	45	45	1.31	.830	MIX .704-06
N2H4	96 UNLINF-DOUBLET	.027	.027	4	60	15	22.0	29.0	45	45	1.08	1.230	MIX .305-06
N2H4	98 UNLINF-DOUBLET	.027	.027	4	60	15	24.0	29.0	45	45	1.11	1.170	MIX .305-06
N2H4	99 UNLINF-DOUBLET	.027	.027	4	60	15	40.0	64.0	80	80	.89	1.800	SEP .203-05
N2H4	100 UNLINF-DOUBLET	.027	.027	4	60	15	20.0	33.0	75	80	.84	2.280	SEP .998-06
N2H4	101 UNLINF-DOUBLET	.027	.027	4	60	15	20.0	36.0	79	83	.80	2.280	SEP .123-05
N2H4	102 UNLINF-DOUBLET	.027	.027	4	60	15	16.0	24.0	78	82	1.10	1.200	UNDEF .793-06
N2H4	108 UNLINF-DOUBLET	.027	.027	4	60	15	47.0	36.0	70	95	1.89	.410	SEP .140-05
N2H4	109 UNLINF-DOUBLET	.027	.027	4	60	15	42.0	30.0	70	95	2.01	.410	SEP .117-05
N2H4	110 UNLINF-DOUBLET	.027	.027	4	60	15	25.0	31.0	70	95	1.18	1.060	SEP .120-05
N2H4	111 UNLINF-DOUBLET	.027	.027	4	60	15	23.0	24.0	65	65	1.38	.760	MIX .502-06
N2H4	112 UNLINF-DOUBLET	.027	.027	4	60	15	27.0	33.0	65	65	1.16	1.070	MIX .690-06
N2H4	113 UNLINF-DOUBLET	.027	.027	4	60	15	24.0	33.0	70	110	1.04	1.330	SEP .167-05
N2H4	114 UNLINF-DOUBLET	.027	.027	4	60	15	18.0	27.0	70	105	.96	1.540	SEP .125-05
N2H4	116 UNLINF-DOUBLET	.027	.027	4	60	15	18.0	22.0	70	105	.96	1.580	UNDEF .102-05
N2H4	117 UNLINF-DOUBLET	.027	.027	4	60	15	16.0	25.0	70	80	.96	1.580	UNDEF .707-06
N2H4	118 UNLINF-DOUBLET	.060	.060	4	60	15	52.0	70.0	70	80	1.07	1.280	UNDEF .978-05
N2H4	119 UNLINF-DOUBLET	.060	.060	4	60	15	14.0	17.0	70	80	1.23	.950	SEP .237-05
N2H4	120 UNLINF-DOUBLET	.060	.060	4	60	15	17.0	22.0	70	80	1.12	1.150	UNDEF .307-05
N2H4	121 UNLINF-DOUBLET	.060	.060	4	60	15	36.0	48.0	70	80	1.09	1.210	UNDEF .671-05
N2H4	122 UNLINF-DOUBLET	.060	.060	4	60	15	66.0	93.0	70	80	1.06	1.280	UNDEF .130-04

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR ZUNG

BURL TYPE	TEST NO.	MODE	PC (PSIA)	YAVG	MEF	WEO	REF	RED	DELTI (DEG F)	RELF	RELU (PSIA)	PPF	MRVP	XF	XU	XP	RESID	
N2H4	15	SEP	15	29	.1	.1	754+04	167+05	0	.30+05	.67+05	.2	14.8	173.4	.02	1.01	.13+00	.64+04
N2H4	20	SEP	15	61	.5	.4	162+05	320+05	0	.65+05	.13+06	.2	14.8	173.4	.02	1.01	.13+00	.30+04
N2H4	50	MIX	15	52	.2	.6	983+04	397+05	0	.39+05	.16+06	.2	14.5	175.2	.02	.98	.13+00	.49+04
N2H4	61	SEP	15	41	.1	.4	862+04	305+05	0	.34+05	.12+06	.2	14.8	173.4	.02	1.01	.13+00	.56+04
N2H4	63	MIX	15	46	.2	.4	112+05	293+05	0	.45+05	.12+06	.2	13.8	163.4	.02	.94	.13+00	.43+04
N2H4	64	MIX	15	49	.2	.4	112+05	316+05	0	.45+05	.13+06	.2	13.2	166.3	.02	.90	.12+00	.42+04
N2H4	68	SEP	15	74	.5	1.1	168+05	516+05	0	.67+05	.21+06	.2	14.8	173.4	.02	1.01	.13+00	.29+04
N2H4	79	SEP	15	22	.1	.1	539+04	145+05	0	.22+05	.58+05	.2	14.8	173.4	.02	1.01	.13+00	.90+04
N2H4	80	MIX	15	47	.2	.4	115+05	295+05	0	.46+05	.12+06	.2	14.2	177.0	.02	.96	.12+00	.42+04
N2H4	81	SEP	15	44	.2	.2	114+05	247+05	0	.46+05	.12+06	.2	14.8	173.4	.02	1.01	.13+00	.42+04
N2H4	82	SEP	15	52	.2	.3	112+05	291+05	0	.45+05	.12+06	.2	14.8	173.4	.02	1.01	.13+00	.43+04
N2H4	83	SEP	15	46	.2	.4	112+05	298+05	0	.45+05	.12+06	.2	14.8	173.4	.02	1.01	.13+00	.43+04
N2H4	85	SEP	15	43	.2	.3	103+05	276+05	0	.41+05	.11+06	.2	14.8	173.4	.02	1.01	.13+00	.47+04
N2H4	89	MIX	15	21	.0	.1	509+04	136+05	0	.20+05	.54+05	.2	13.8	173.4	.02	.94	.12+00	.94+04
N2H4	90	MIX	15	76	.7	.6	198+05	379+05	0	.79+05	.15+06	.2	13.8	178.9	.02	.94	.12+00	.24+04
N2H4	91	SEP	15	76	.7	.6	202+05	395+05	0	.81+05	.15+06	.2	14.8	173.4	.02	1.01	.13+00	.24+04
N2H4	92	SEP	15	74	.7	.9	198+05	472+05	0	.79+05	.19+06	.2	14.8	173.4	.02	1.01	.13+00	.24+04
N2H4	93	MIX	15	51	.3	.4	120+05	286+05	0	.48+05	.12+06	.2	11.7	197.4	.01	.79	.96+01	.37+04
N2H4	94	MIX	15	46	.2	.3	101+05	256+05	0	.40+05	.10+06	.1	7.6	221.1	.01	.52	.59+01	.40+04
N2H4	95	MIX	15	64	.4	.7	121+05	401+05	0	.48+05	.16+06	.1	7.6	221.1	.01	.52	.59+01	.34+04
N2H4	96	MIX	15	25	.1	.1	523+04	145+05	0	.21+05	.56+05	.1	7.6	221.1	.01	.52	.59+01	.78+04
N2H4	98	MIX	15	26	.1	.1	523+04	151+05	0	.21+05	.60+05	.1	7.6	221.1	.01	.52	.59+01	.78+04
N2H4	99	SEP	15	53	.3	.4	150+05	304+05	0	.60+05	.12+06	.3	19.0	170.5	.02	1.29	.17+00	.35+04
N2H4	100	SEP	15	27	.1	.1	712+04	149+05	0	.31+05	.60+05	.3	16.9	153.1	.02	1.15	.16+00	.66+04
N2H4	101	SEP	15	24	.1	.1	603+04	153+05	0	.35+05	.61+05	.4	18.6	145.5	.03	1.26	.18+00	.63+04
N2H4	102	UNDEF	15	21	.0	.1	571+04	137+05	0	.23+05	.55+05	.4	18.1	146.8	.02	1.23	.17+00	.94+04
N2H4	106	SEP	15	43	.1	.5	956+04	341+05	0	.38+05	.14+06	.6	14.8	78.4	.04	1.01	.20+00	.63+04
N2H4	109	SEP	15	34	.1	.4	786+04	305+05	0	.32+05	.12+06	.6	14.8	78.4	.04	1.01	.20+00	.75+04
N2H4	110	SEP	15	24	.1	.1	423+04	142+05	0	.33+05	.73+05	.6	14.8	78.4	.04	1.01	.20+00	.73+04
N2H4	111	MIX	15	23	.1	.1	497+04	163+05	0	.20+05	.65+05	.2	13.2	183.2	.01	.90	.11+00	.94+04
N2H4	112	MIX	15	30	.1	.2	684+04	191+05	0	.27+05	.76+05	.2	13.2	183.2	.01	.90	.11+00	.68+04
N2H4	113	SEP	15	26	.1	.1	968+04	174+05	0	.39+05	.70+05	.9	14.8	50.6	.06	1.01	.25+00	.68+04
N2H4	114	SEP	15	25	.1	.1	771+04	131+05	0	.31+05	.52+05	.8	14.8	58.4	.05	1.01	.23+00	.83+04
N2H4	116	UNDEF	15	20	.0	.1	628+04	131+05	0	.25+05	.52+05	.8	14.8	58.4	.05	1.01	.23+00	.83+04
N2H4	117	UNDEF	15	21	.0	.1	585+04	116+05	0	.23+05	.46+05	.3	14.8	135.3	.02	1.01	.15+00	.90+04
N2H4	118	UNDEF	15	61	.9	1.3	364+05	840+05	0	.35+06	.34+06	.3	14.8	135.3	.02	1.01	.15+00	.71+04
N2H4	119	SEP	15	15	.1	.1	444+04	226+05	0	.35+05	.90+05	.3	14.8	135.3	.02	1.01	.15+00	.29+03
N2H4	120	UNDEF	15	14	.1	.1	114+05	274+05	0	.46+05	.11+06	.3	14.8	135.3	.02	1.01	.15+00	.23+03
N2H4	121	UNDEF	15	42	.4	.6	230+05	541+05	0	.10+06	.23+06	.3	14.8	135.3	.02	1.01	.15+00	.10+03
N2H4	122	UNDEF	15	40	1.0	2.2	444+05	1110+06	0	.19+06	.44+06	.3	14.8	135.3	.02	1.01	.15+00	.54+04

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGEOMETRIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TEST TYPE NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IPP ANGLE (DEG)	PC (PSIA)	VU (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR MF/M ²	MODE	REACT (SEC)
N2H4 123	UNLINE-DUUBLET	.060	.060	4.	60.	15.	13.0	16.0	75.	75.	1.16	1.060	SEP .250-05
N2H4 124	UNLINE-DUUBLET	.060	.060	4.	60.	15.	8.3	13.0	71.	72.	.95	1.060	SEP .163-05
N2H4 125	UNLINE-DUUBLET	.060	.060	4.	60.	15.	6.6	11.0	71.	72.	.89	1.810	SEP .138-05
N2H4 126	UNLINE-DUUBLET	.060	.060	4.	60.	15.	5.4	8.6	72.	72.	.90	1.770	SEP .110-05
N2H4 127	UNLINE-DUUBLET	.060	.060	4.	60.	15.	4.3	5.6	72.	72.	1.10	1.170	SEP .713-06
N2H4 128	UNLINE-DUUBLET	.060	.060	4.	60.	15.	13.0	16.0	40.	50.	1.11	1.150	POP .672-06
N2H4 129	UNLINE-DUUBLET	.060	.060	4.	60.	15.	9.1	14.0	42.	52.	.96	1.560	POP .818-06
N2H4 130	UNLINE-DUUBLET	.060	.060	4.	60.	15.	8.3	10.0	42.	52.	1.18	1.010	POP .595-06
N2H4 131	UNLINE-DUUBLET	.060	.060	4.	60.	15.	8.0	9.0	42.	52.	1.27	.880	POP .526-06
N2H4 132	UNLINE-DUUBLET	.060	.060	4.	60.	15.	10.0	7.9	45.	52.	1.83	.430	POP .480-06
N2H4 133	UNLINE-DUUBLET	.060	.060	4.	60.	15.	9.9	12.0	45.	50.	1.18	1.000	POP .770-06
N2H4 134	UNLINE-DUUBLET	.060	.060	4.	60.	15.	9.3	13.0	47.	50.	1.00	1.420	POP .778-06
N2H4 135	UNLINE-DUUBLET	.060	.060	4.	60.	15.	4.3	4.6	63.	62.	.71	2.810	POP .814-06
N2H4 136	UNLINE-DUUBLET	.060	.060	4.	60.	15.	12.0	14.0	54.	54.	1.17	1.050	POP .101-05
N2H4 137	UNLINE-DUUBLET	.060	.060	4.	60.	15.	8.2	12.0	55.	55.	1.04	1.540	POP .892-06
N2H4 139	UNLINE-DUUBLET	.060	.060	4.	60.	15.	9.6	12.0	61.	61.	1.13	1.110	POP .108-05
N2H4 140	UNLINE-DUUBLET	.060	.060	4.	60.	15.	14.0	20.0	63.	63.	.98	1.470	POP .193-05
N2H4 141	UNLINE-DUUBLET	.060	.060	4.	60.	15.	22.0	26.0	66.	70.	1.20	.990	POP .132-05
N2H4 142	UNLINE-DUUBLET	.060	.060	4.	60.	15.	18.0	21.0	70.	70.	1.22	.950	SEP .112-05
N2H4 143	UNLINE-DUUBLET	.060	.060	4.	60.	15.	13.0	17.0	70.	70.	1.07	1.250	SEP .905-06
N2H4 148	UNLINE-DUUBLET	.060	.060	4.	60.	15.	12.0	14.0	70.	70.	1.18	1.030	SEP .745-06
N2H4 149	UNLINE-DUUBLET	.060	.060	4.	60.	15.	21.0	17.0	72.	92.	1.71	.490	SEP .142-05
N2H4 150	UNLINE-DUUBLET	.060	.060	4.	60.	15.	24.0	26.0	68.	68.	1.21	.970	MIX .141-05
N2H4 151	UNLINE-DUUBLET	.060	.060	4.	60.	15.	24.0	26.0	67.	59.	1.20	.980	POP .116-05
N2H4 152	UNLINE-DUUBLET	.060	.060	4.	60.	15.	16.0	19.0	62.	65.	1.22	1.290	POP .840-06
N2H4 154	UNLINE-DUUBLET	.060	.060	4.	60.	149.	14.0	19.0	63.	63.	1.01	1.400	MIX .816-06
N2H4 165	UNLINE-DUUBLET	.060	.060	4.	60.	189.	23.0	23.0	69.	73.	1.07	1.250	MIX .122-05
N2H4 166	UNLINE-DUUBLET	.060	.060	4.	60.	187.	24.0	23.0	66.	73.	1.44	.690	MIX .123-05
N2H4 167	UNLINE-DUUBLET	.060	.060	4.	60.	189.	23.0	25.0	47.	59.	1.35	.790	MIX .786-06
N2H4 169	UNLINE-DUUBLET	.060	.060	4.	60.	209.	24.0	27.0	40.	52.	1.31	.840	UNDEF .682-06
N2H4 170	UNLINE-DUUBLET	.060	.060	4.	60.	111.	15.0	15.0	40.	58.	1.45	.680	MIX .423-06
N2H4 171	UNLINE-DUUBLET	.060	.060	4.	60.	191.	23.0	27.0	43.	58.	1.25	.920	MIX .793-06
N2H4 174	UNLINE-DUUBLET	.060	.060	4.	60.	172.	22.0	27.0	54.	64.	1.20	.990	MIX .105-05
N2H4 175	UNLINE-DUUBLET	.060	.060	4.	60.	187.	22.0	27.0	57.	65.	1.20	.990	MIX .112-05
N2H4 179	UNLINE-DUUBLET	.060	.060	4.	60.	163.	21.0	23.0	56.	63.	1.30	.840	MIX .699-06
N2H4 180	UNLINE-DUUBLET	.060	.060	4.	60.	165.	19.0	24.0	76.	81.	1.55	1.070	MIX .166-05
N2H4 181	UNLINE-DUUBLET	.060	.060	4.	60.	140.	18.0	20.0	77.	82.	1.23	.940	MIX .143-05
N2H4 182	UNLINE-DUUBLET	.060	.060	4.	60.	259.	20.0	19.0	81.	87.	1.47	.660	UNDEF .160-05
N2H4 183	UNLINE-DUUBLET	.060	.060	4.	60.	157.	19.0	29.0	85.	93.	.94	1.610	MIX .289-05

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	NUDE	PC (PSIA)	VAVG	REF	WED	RFF	MEU	DELTI (DEG F)	RELF	RELO	PPF (PSIA)	MRVP	XF	XU	AP	RESID	
N2H4	123	SEP	15	15	1	1	697+04	216+05	0	36+05	86+05	3	16.9	171.8	02	1.15	15+00	23-03
N2H4	124	SEP	15	11	0	0	632+04	135+05	0	25+05	54+05	3	15.2	168.4	02	1.03	14+00	38-03
N2H4	125	SEP	15	9	0	0	535+04	107+05	0	21+05	43+05	3	15.2	168.4	02	1.03	14+00	45-03
N2H4	126	SEP	15	7	0	0	418+04	882+04	0	17+05	35+05	3	15.6	172.7	02	1.06	14+00	56-03
N2H4	127	SEP	15	5	0	0	272+04	702+04	0	11+05	28+05	3	15.6	172.7	02	1.06	14+00	64-03
N2H4	128	PUP	15	14	0	1	662+04	191+05	0	26+05	76+05	1	6.6	156.6	01	0.45	61-01	31-03
N2H4	129	PUP	15	12	0	0	567+04	133+05	0	23+05	53+05	1	7.0	155.8	01	0.48	65-01	56-03
N2H4	130	PUP	15	9	0	0	420+04	122+05	0	17+05	49+05	1	7.0	155.8	01	0.48	65-01	50-03
N2H4	131	PUP	15	8	0	0	378+04	117+05	0	15+05	47+05	1	7.0	155.8	01	0.48	65-01	50-03
N2H4	132	PUP	15	9	0	0	331+04	146+05	0	13+05	38+05	1	7.6	167.7	01	0.52	66-01	61-03
N2H4	133	PUP	15	11	0	0	497+04	145+05	0	20+05	59+05	1	7.6	167.7	01	0.52	66-01	61-03
N2H4	134	PUP	15	11	0	0	538+04	136+05	0	22+05	54+05	1	8.0	168.4	01	0.54	67-01	42-03
N2H4	135	PUP	15	7	0	0	387+04	668+04	0	15+05	27+05	2	12.6	188.2	01	0.82	99-01	42-03
N2H4	136	PUP	15	13	0	1	596+04	178+05	0	24+05	71+05	1	9.8	198.2	01	0.67	80-01	36-03
N2H4	137	PUP	15	10	0	0	514+04	122+05	0	21+05	49+05	1	10.1	198.8	01	0.69	83-01	42-03
N2H4	139	PUP	15	11	0	0	536+04	148+05	0	21+05	59+05	2	12.0	198.8	01	0.82	99-01	42-03
N2H4	140	PUP	15	17	1	1	907+04	218+05	0	36+05	87+05	2	12.6	188.2	01	0.86	11+00	25-03
N2H4	141	PUP	15	24	1	1	230+04	232+05	0	33+05	93+05	2	13.5	160.0	02	1.02	12+00	13-03
N2H4	142	SEP	15	19	1	1	670+04	194+05	0	27+05	77+05	2	14.8	173.4	02	1.01	13+00	16-03
N2H4	143	SEP	15	15	0	1	543+04	140+05	0	22+05	56+05	2	14.8	173.4	02	1.01	13+00	20-03
N2H4	144	SEP	15	13	0	0	447+04	129+05	0	18+05	42+05	2	14.8	173.4	02	1.01	13+00	24-03
N2H4	145	SEP	15	20	0	1	651+04	229+05	0	35+05	100+05	3	15.6	90.0	04	1.06	19+00	20-03
N2H4	149	MIX	15	26	1	2	840+04	255+05	0	35+05	100+05	2	14.2	171.0	02	0.96	12+00	12-03
N2H4	150	PUP	15	20	1	2	623+04	254+05	0	33+05	100+05	2	13.8	237.3	01	0.94	10+00	12-03
N2H4	151	PUP	15	17	0	1	543+04	165+05	0	33+05	86+05	2	12.3	171.3	01	0.84	11+00	16-03
N2H4	152	MIX	15	10	0	1	575+04	145+05	0	23+05	58+05	2	12.6	188.2	01	0.86	11+00	16-03
N2H4	154	MIX	149	10	0	8	719+04	171+05	0	29+05	68+05	3	14.5	156.7	00	1.0	13+00	15-03
N2H4	155	MIX	149	23	0	8	752+04	243+05	0	30+05	97+05	3	13.8	150.5	00	0.7	10-01	14-03
N2H4	156	MIX	147	24	0	2	752+04	251+05	0	30+05	100+05	3	13.5	147.4	00	0.7	10-01	14-03
N2H4	157	MIX	149	24	1	1	734+04	224+05	0	29+05	49+05	2	8.0	142.1	00	0.4	61-02	13-03
N2H4	159	MIX	209	25	1	2	755+04	235+05	0	30+05	94+05	1	6.6	147.8	00	0.4	61-02	13-03
N2H4	171	MIX	111	15	2	5	337+04	147+05	0	17+05	39+05	2	6.6	125.0	00	0.6	80-02	12-03
N2H4	171	MIX	191	25	1	1	747+04	224+05	0	31+05	90+05	2	7.2	132.9	00	0.4	93-02	22-03
N2H4	172	MIX	172	24	1	1	743+04	218+05	0	33+05	87+05	2	9.8	142.7	00	0.4	56-02	12-03
N2H4	175	MIX	147	1	1	1	629+04	221+05	0	33+05	88+05	2	10.7	151.1	00	0.6	81-02	12-03
N2H4	174	MIX	165	22	7	1	596+04	210+05	0	15	39+05	2	10.4	147.8	00	0.6	80-02	12-03
N2H4	181	MIX	165	21	0	1	339+04	212+05	0	24+05	84+05	3	17.3	149.2	00	1.0	15-01	14-03
N2H4	181	MIX	146	17	5	1	704+04	222+05	0	28+05	81+05	4	17.7	149.7	00	1.3	18-01	17-03
N2H4	182	MIX	259	24	0	2	697+04	229+05	0	28+05	82+05	4	19.5	130.2	00	0.8	11-01	18-03
N2H4	183	MIX	157	24	1	1	512+05	223+05	0	45+05	89+05	5	21.6	117.9	00	1.4	22-01	11-03

HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGULIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VU (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MU	MODE	REACT (SEC)
N2M4	184	UNLINE-DUUBLET	.040	.040	4.	60.	119.	17.0	14.0	95.	101.	1.83	.430	MIX	.180-05
N2M4	185	UNLINE-DUUBLET	.040	.040	4.	60.	131.	17.0	17.0	107.	96.	1.41	.720	MIX	.231-05
N2M4	186	UNLINE-DUUBLET	.040	.040	4.	60.	155.	22.0	24.0	77.	91.	1.31	.840	MIX	.209-05
N2M4	187	UNLINE-DUUBLET	.040	.040	4.	60.	184.	27.0	27.0	91.	91.	1.51	.720	MIX	.278-05
N2M4	188	UNLINE-DUUBLET	.040	.040	4.	60.	194.	26.0	29.0	86.	90.	1.24	.920	MIX	.277-05
N2M4	191	UNLINE-DUUBLET	.040	.040	4.	60.	170.	25.0	25.0	96.	103.	1.02	.710	MIX	.338-05
N2M4	192	UNLINE-DUUBLET	.040	.040	4.	60.	152.	24.0	27.0	104.	113.	1.27	.890	MIX	.475-05
N2M4	193	UNLINE-DUUBLET	.040	.040	4.	60.	174.	25.0	31.0	94.	99.	1.14	1.100	MIX	.379-05
N2M4	194	UNLINE-DUUBLET	.040	.040	4.	60.	169.	24.0	30.0	100.	104.	1.13	1.120	MIX	.432-05
N2M4	196	UNLINE-DUUBLET	.040	.040	4.	60.	187.	24.0	32.0	76.	94.	1.09	1.210	MIX	.290-05
N2M4	198	UNLINE-DUUBLET	.040	.040	4.	60.	187.	16.0	32.0	106.	76.	.72	2.790	MIX	.291-05
N2M4	199	UNLINE-DUUBLET	.040	.040	4.	60.	164.	24.0	27.0	91.	101.	1.25	.920	MIX	.330-05
N2M4	200	UNLINE-DUUBLET	.040	.040	4.	60.	181.	23.0	32.0	93.	108.	1.03	1.360	MIX	.457-05
N2M4	201	UNLINE-DUUBLET	.040	.040	4.	60.	203.	23.0	32.0	84.	77.	1.04	1.320	MIX	.178-05
N2M4	204	UNLINE-DUUBLET	.040	.040	4.	60.	178.	20.0	32.0	62.	77.	.92	1.680	MIX	.174-05
N2M4	205	UNLINE-DUUBLET	.040	.040	4.	60.	110.	31.0	36.0	52.	64.	1.23	.950	MIX	.135-05
N2M4	206	UNLINE-DUUBLET	.040	.040	4.	60.	115.	33.0	35.0	76.	84.	1.36	1.360	MIX	.260-05
N2M4	207	UNLINE-DUUBLET	.040	.040	4.	60.	115.	33.0	30.0	91.	95.	1.59	1.590	MIX	.332-05
N2M4	208	UNLINE-DUUBLET	.040	.040	4.	60.	110.	24.0	34.0	110.	107.	1.23	1.230	MIX	.580-05
N2M4	209	UNLINE-DUUBLET	.040	.040	4.	60.	110.	28.0	40.0	102.	108.	.99	1.440	MIX	.634-05
N2M4	210	UNLINE-DUUBLET	.040	.040	4.	60.	110.	29.0	35.0	105.	110.	1.19	.990	MIX	.594-05
N2M4	211	UNLINE-DUUBLET	.040	.040	4.	60.	110.	30.0	37.0	121.	.0.	1.12	1.130	MIX	.750-05
N2M4	212	UNLINE-DUUBLET	.040	.040	4.	60.	110.	30.0	35.0	121.	118.	1.24	.930	MIX	.777-05
N2M4	215	UNLINE-DUUBLET	.040	.040	4.	60.	110.	56.0	68.0	108.	118.	1.17	.850	MIX	.134-04
N2M4	219	UNLINE-DUUBLET	.040	.040	4.	60.	110.	56.0	68.0	93.	101.	1.17	.850	MIX	.852-05
N2M4	220	UNLINE-DUUBLET	.040	.040	4.	60.	110.	56.0	63.0	123.	121.	1.27	.790	MIX	.154-04
N2M4	221	UNLINE-DUUBLET	.040	.040	4.	60.	110.	59.0	64.0	117.	122.	1.33	.880	MIX	.158-04
N2M4	222	UNLINE-DUUBLET	.040	.040	4.	60.	110.	59.0	68.0	103.	110.	1.24	.810	MIX	.113-04
N2M4	223	UNLINE-DUUBLET	.040	.040	4.	60.	110.	55.0	67.0	119.	114.	1.16	.860	MIX	.143-04
N2M4	231	UNLINE-DUUBLET	.040	.040	4.	60.	350.	51.0	66.0	79.	67.	1.10	1.180	UNDEF	.373-05
N2M4	232	UNLINE-DUUBLET	.040	.040	4.	60.	380.	55.0	66.0	85.	72.	1.18	1.000	SEP	.439-05
N2M4	234	UNLINE-DUUBLET	.040	.040	4.	60.	353.	66.0	68.0	118.	118.	1.12	.880	SEP	.150-04
N2M4	235	UNLINE-DUUBLET	.040	.040	4.	60.	353.	50.0	68.0	84.	103.	1.05	.950	SEP	.796-05
N2M4	236	UNLINE-DUUBLET	.040	.040	4.	60.	346.	48.0	63.0	71.	90.	1.08	.920	SEP	.499-05
N2M4	237	UNLINE-DUUBLET	.040	.040	4.	60.	362.	56.0	68.0	63.	93.	1.18	.840	SEP	.518-05
N2M4	239	UNLINE-DUUBLET	.040	.040	4.	60.	301.	50.0	65.0	66.	85.	1.05	.950	SEP	.458-05
N2M4	240	UNLINE-DUUBLET	.040	.040	4.	60.	145.	22.0	27.0	76.	82.	1.19	1.000	MIX	.182-05
N2M4	240	UNLINE-DUUBLET	.040	.040	4.	60.	234.	65.0	68.0	63.	67.	1.17	.760	UNDEF	.317-05

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATION										ZUNG								
FUEL TYPE	TEST NO.	MODE	PC (PSIA)	WAVG	MEF	NEO	REF	REO (DEG F)	DELTI	RELIF	PELO	PPF (PSIA)	MRVP	XF	KU	XP	RESID	
N2H4	184	MIX	119.	16.	.2	.8	.500+04	.213+05	0.	.23+05	.85+05	.7	27.5	118.2	.01	.23	.36-01	.24-03
N2H4	185	MIX	131.	17.	.3	1.0	.675+04	.228+05	0.	.27+05	.91+05	.6	36.2	175.0	.00	.28	.25-01	.20-03
N2H4	186	MIX	155.	23.	.8	1.7	.911+04	.246+05	0.	.36+05	.99+05	.5	17.7	104.2	.00	.11	.19-01	.14-03
N2H4	187	MIX	192.	27.	1.2	3.4	.102+05	.330+05	0.	.41+05	.13+06	.5	24.9	142.5	.00	.13	.18-01	.12-03
N2H4	188	MIX	194.	27.	1.4	3.1	.102+05	.308+05	0.	.44+05	.12+06	.5	22.1	132.1	.00	.11	.17-01	.11-03
N2H4	191	MIX	170.	25.	1.0	2.0	.105+05	.316+05	0.	.42+05	.13+06	.7	28.1	112.4	.00	.17	.27-01	.15-03
N2H4	192	MIX	152.	25.	1.0	2.3	.119+05	.318+05	0.	.43+05	.13+06	1.0	33.9	101.4	.01	.22	.34-1	.12-03
N2H4	193	MIX	174.	28.	1.5	2.7	.126+05	.312+05	0.	.51+05	.12+06	.6	26.8	122.7	.00	.15	.24-01	.11-03
N2H4	194	MIX	169.	27.	1.4	2.5	.126+05	.312+05	0.	.50+05	.12+06	.8	30.7	118.5	.00	.14	.24-01	.11-03
N2H4	196	MIX	194.	28.	1.8	2.5	.125+05	.287+05	0.	.50+05	.11+06	.6	17.3	93.3	.00	.09	.16-01	.10-03
N2H4	197	MIX	187.	25.	1.6	1.3	.107+05	.214+05	0.	.43+05	.06+05	.3	35.4	332.8	.00	.19	.17-01	.10-03
N2H4	198	MIX	164.	25.	1.1	2.3	.112+05	.293+05	0.	.45+05	.12+06	.7	24.9	137.8	.00	.15	.25-01	.12-03
N2H4	199	MIX	181.	27.	1.7	2.3	.134+05	.285+05	0.	.55+05	.11+06	.9	26.2	90.9	.00	.14	.26-01	.10-03
N2H4	200	MIX	203.	27.	1.8	2.3	.108+05	.240+05	0.	.43+05	.09+05	.3	12.3	127.9	.00	.00	.57-02	.10-03
N2H4	201	MIX	176.	26.	1.6	1.5	.106+05	.206+05	0.	.43+05	.02+05	.3	12.3	124.2	.00	.07	.11-01	.10-03
N2H4	202	MIX	110.	33.	1.2	2.1	.110+05	.304+05	0.	.44+05	.12+06	.2	9.2	135.1	.00	.08	.12-01	.03-04
N2H4	205	MIX	115.	34.	1.2	2.8	.125+05	.368+05	0.	.50+05	.15+06	.4	17.3	130.8	.00	.15	.22-01	.05-04
N2H4	206	MIX	115.	32.	.9	3.0	.118+05	.403+05	0.	.47+05	.16+06	.6	24.9	126.8	.00	.24	.33-01	.11-03
N2H4	207	MIX	110.	31.	1.2	2.5	.115+05	.395+05	0.	.58+05	.16+06	.8	38.6	133.9	.01	.35	.51-01	.18-04
N2H4	208	MIX	110.	32.	1.2	2.4	.152+05	.366+05	0.	.61+05	.15+06	.9	32.3	110.3	.01	.29	.46-01	.03-04
N2H4	209	MIX	110.	32.	1.2	2.4	.151+05	.418+05	0.	.61+05	.15+06	.9	32.7	111.7	.01	.32	.51-1	.05-04
N2H4	210	MIX	110.	33.	1.4	2.7	.151+05	.418+05	0.	.61+05	.17+06	.9	49.4	154.9	.01	.45	.61-01	.00-04
N2H4	211	MIX	110.	32.	1.2	2.9	.157+05	.433+05	0.	.63+05	.17+06	1.0	49.4	134.1	.01	.45	.65-01	.05-04
N2H4	212	MIX	110.	32.	1.3	2.9	.159+05	.433+05	0.	.64+05	.17+06	1.1	49.4	128.5	.01	.45	.67-01	.05-04
N2H4	215	MIX	110.	32.	4.7	9.2	.369+05	.736+05	0.	.12+06	.30+06	1.1	37.0	98.5	.01	.34	.58-01	.09-04
N2H4	217	MIX	110.	32.	4.6	8.4	.262+05	.617+05	0.	.11+06	.28+06	1.2	26.2	113.0	.01	.24	.34-01	.09-04
N2H4	219	MIX	110.	34.	4.1	10.1	.292+05	.617+05	0.	.12+06	.33+06	1.2	51.8	124.9	.01	.47	.71-01	.53-04
N2H4	220	MIX	110.	31.	4.4	9.0	.317+05	.763+05	0.	.13+06	.31+06	1.2	45.4	107.4	.01	.41	.68-01	.09-04
N2H4	221	MIX	110.	33.	4.6	4.4	.296+05	.778+05	0.	.12+06	.31+06	.9	33.1	137.0	.01	.30	.56-01	.09-04
N2H4	222	MIX	110.	31.	4.7	9.5	.302+05	.765+05	0.	.12+06	.31+06	1.0	47.3	154.8	.01	.43	.61-01	.09-04
N2H4	223	MIX	110.	33.	4.7	10.9	.302+05	.842+05	0.	.12+06	.33+06	1.0	47.3	134.8	.01	.43	.63-01	.09-04
N2H4	231	MIX	155.	58.	12.7	20.7	.206+05	.574+05	0.	.85+05	.23+06	.2	18.6	234.5	.00	.05	.58-02	.51-04
N2H4	232	MIX	155.	58.	13.9	20.9	.214+05	.607+05	0.	.86+05	.23+06	.3	21.6	233.1	.00	.06	.62-02	.51-04
N2H4	233	MIX	155.	59.	12.2	20.2	.209+05	.566+05	0.	.85+05	.23+06	1.1	46.3	121.1	.00	.13	.20-01	.09-04
N2H4	235	MIX	155.	58.	12.7	20.5	.205+05	.584+05	0.	.81+06	.23+06	.7	21.1	166.5	.00	.06	.11-01	.09-04
N2H4	237	MIX	155.	58.	12.7	17.4	.237+05	.520+05	0.	.95+05	.21+06	.5	15.2	93.3	.00	.04	.74-02	.55-04
N2H4	238	MIX	155.	59.	12.7	23.4	.205+05	.566+05	0.	.81+06	.23+06	.5	12.6	11.7	.00	.03	.72-02	.09-04
N2H4	240	MIX	155.	59.	12.4	19.2	.205+05	.526+05	0.	.98+05	.21+06	.4	13.5	100.2	.00	.04	.55-02	.09-04
N2H4	241	MIX	155.	60.	12.4	1.5	.931+04	.219+05	0.	.38+05	.09+05	.4	15.6	129.6	.00	.11	.16-01	.12-03
N2H4	242	MIX	155.	60.	9.2	21.5	.212+05	.219+05	0.	.85+05	.27+06	.2	12.6	164.1	.00	.05	.70-02	.09-04

HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	INJECTOR TYPE	DD (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VD (FT/S)	VF (FT/S)	TO (F)	IF (F)	MR WF/PO	MODE	REACT (SEC)
N2H4	241	UNLIKE-DOUBLET	.040	.040	4.	60.	230.	74.0	68.0	115.	117.	1.56	MIX	.144-04
N2H4	242	UNLIKE-DOUBLET	.040	.040	4.	60.	176.	55.0	69.0	116.	127.	1.14	MIX	.238-04
N2H4	243	UNLIKE-DOUBLET	.040	.040	4.	60.	176.	55.0	69.0	117.	122.	1.14	MIX	.161-04
N2H4	244	UNLIKE-DOUBLET	.040	.040	4.	60.	338.	34.0	65.0	44.	54.	.67	SEP	.180-05
N2H4	245	UNLIKE-DOUBLET	.040	.040	4.	60.	400.	57.0	61.0	39.	41.	1.31	SEP	.116-05
N2H4	246	UNLIKE-DOUBLET	.040	.040	4.	60.	215.	58.0	66.0	51.	62.	1.25	MIX	.233-05
N2H4	247	UNLIKE-DOUBLET	.040	.040	4.	60.	115.	61.0	65.0	51.	60.	1.34	MIX	.219-05
N2H4	248	UNLIKE-DOUBLET	.040	.040	4.	60.	213.	61.0	68.0	58.	70.	1.29	MIX	.313-05
N2H4	249	UNLIKE-DOUBLET	.040	.040	4.	60.	304.	56.0	65.0	62.	74.	1.23	SEP	.337-05
N2H4	250	UNLIKE-DOUBLET	.040	.040	4.	60.	307.	45.0	50.0	65.	76.	1.28	UNDEF	.277-05
N2H4	251	UNLIKE-DOUBLET	.040	.040	4.	60.	507.	35.0	38.0	68.	77.	1.29	SEP	.221-05
N2H4	252	UNLIKE-DOUBLET	.040	.040	4.	60.	520.	36.0	28.0	77.	86.	1.33	SEP	.221-05
N2H4	253	UNLIKE-DOUBLET	.040	.040	4.	60.	84.	49.0	52.0	106.	103.	1.32	MIX	.790-05
N2H4	254	UNLIKE-DOUBLET	.040	.040	4.	60.	100.	56.0	65.0	126.	120.	1.26	MIX	.161-04
N2H4	255	UNLIKE-DOUBLET	.040	.040	4.	60.	122.	56.0	66.0	124.	120.	1.20	MIX	.160-04
N2H4	256	UNLIKE-DOUBLET	.040	.040	4.	60.	423.	32.0	37.0	93.	104.	1.22	SEP	.492-05
N2H4	257	UNLIKE-DOUBLET	.040	.040	4.	60.	461.	32.0	37.0	94.	99.	1.22	SEP	.453-05
N2H4	258	UNLIKE-DOUBLET	.040	.040	4.	60.	461.	32.0	37.0	74.	84.	1.24	SEP	.268-05
N2H4	259	UNLIKE-DOUBLET	.040	.040	4.	60.	115.	31.0	37.0	79.	88.	1.18	MIX	.311-05
N2H4	260	UNLIKE-DOUBLET	.040	.040	4.	60.	130.	34.0	41.0	92.	97.	1.18	MIX	.475-05
N2H4	261	UNLIKE-DOUBLET	.040	.040	4.	60.	149.	40.0	52.0	90.	97.	1.10	MIX	.586-05
N2H4	262	UNLIKE-DOUBLET	.040	.040	4.	60.	169.	46.0	53.0	90.	97.	1.24	MIX	.598-05
N2H4	263	UNLIKE-DOUBLET	.040	.040	4.	60.	188.	51.0	61.0	84.	93.	1.17	MIX	.600-05
N2H4	264	UNLIKE-DOUBLET	.040	.040	4.	60.	203.	56.0	69.0	83.	93.	1.16	MIX	.671-05
N2H4	265	UNLIKE-DOUBLET	.040	.040	4.	60.	388.	58.0	68.0	81.	88.	1.22	SEP	.586-05
N2H4	266	UNLIKE-DOUBLET	.040	.040	4.	60.	353.	58.0	57.0	78.	86.	1.44	SEP	.454-05
N2H4	267	UNLIKE-DOUBLET	.040	.040	4.	60.	353.	58.0	52.0	77.	86.	1.49	SEP	.410-05
N2H4	268	UNLIKE-DOUBLET	.040	.040	4.	60.	176.	49.0	71.0	138.	129.	.97	MIX	.232-04
N2H4	269	UNLIKE-DOUBLET	.040	.040	4.	60.	179.	56.0	63.0	106.	111.	1.28	MIX	.110-04
N2H4	270	UNLIKE-DOUBLET	.040	.040	4.	60.	176.	58.0	63.0	120.	124.	1.31	MIX	.157-04
N2H4	271	UNLIKE-DOUBLET	.040	.040	4.	60.	158.	55.0	46.0	117.	120.	1.70	MIX	.103-04
N2H4	272	UNLIKE-DOUBLET	.040	.040	4.	60.	138.	42.0	51.0	114.	117.	1.17	MIX	.106-04
N2H4	273	UNLIKE-DOUBLET	.040	.040	4.	60.	279.	75.0	73.0	70.	70.	1.45	UNDEF	.388-05
N2H4	274	UNLIKE-DOUBLET	.040	.040	4.	60.	237.	61.0	73.0	70.	70.	1.19	MIX	.389-05
N2H4	275	UNLIKE-DOUBLET	.040	.040	4.	60.	250.	66.0	73.0	70.	70.	1.29	MIX	.389-05
N2H4	276	UNLIKE-DOUBLET	.040	.040	4.	60.	100.	101.0	127.0	69.	74.	1.13	MIX	.714-05
N2H4	277	UNLIKE-DOUBLET	.040	.040	4.	60.	115.	117.0	127.0	69.	73.	1.54	MIX	.703-05
N2H4	278	UNLIKE-DOUBLET	.040	.040	4.	60.	326.	84.0	98.0	84.	98.	1.22	SEP	.105-04
N2H4	279	UNLIKE-DOUBLET	.040	.040	4.	60.	282.	76.0	93.0	54.	74.	1.16	SEP	.431-05
N2H4	280	UNLIKE-DOUBLET	.040	.040	4.	60.	228.	63.0	78.0	54.	86.	1.16	MIX	.457-05

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGULIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TEST TYPE NO.	MODE	PC (PSIA)	VAVG	MEF	MEQ	REF	HEQ	DELTI (DEG F)	RELF	RELU (PSIA)	PPF (PSIA)	PPD (PSIA)	MRVP	XF	XU	XP	RESID
N2H4 241	MIX	230.	72.	9.9	35.0	308+05	103+00	0.	12+06	41+06	1.1	43.4	116.6	.00	.19	.30-01	.49-04
N2H4 242	MIX	176.	62.	7.9	18.2	331+05	919+05	0.	13+06	37+06	1.4	88.3	166.2	.01	.48	.62-01	.48-04
N2H4 243	MIX	176.	62.	7.9	15.0	321+05	777+05	0.	13+06	31+06	1.2	45.4	107.4	.01	.26	.42-01	.48-04
N2H4 244	SEP	336.	53.	11.6	9.9	184+05	380+05	0.	74+05	15+06	.1	7.4	153.3	.00	.02	.30-02	.51-04
N2H4 246	SEP	400.	59.	11.8	24.4	159+05	558+05	0.	64+05	22+06	.1	6.4	232.9	.00	.02	.18-02	.55-04
N2H4 247	MIX	215.	62.	7.7	14.3	198+05	585+05	0.	79+05	23+06	.2	8.9	140.8	.00	.04	.60-02	.51-04
N2H4 248	MIX	115.	63.	4.0	8.5	192+05	594+05	0.	77+05	24+06	.2	8.4	152.7	.00	.08	.11-01	.51-04
N2H4 250	MIX	218.	64.	8.5	16.6	217+05	616+05	0.	87+05	25+06	.2	11.0	132.6	.00	.05	.75-02	.49-04
N2H4 251	SEP	384.	60.	13.7	25.1	234+05	771+05	0.	86+05	23+06	.3	12.3	131.5	.00	.03	.48-02	.51-04
N2H4 252	UNDEF	307.	47.	6.5	13.2	168+05	471+05	0.	67+05	19+06	.3	13.2	133.9	.00	.04	.64-02	.67-04
N2H4 253	SEP	507.	36.	6.2	13.3	128+05	373+05	0.	51+05	15+06	.3	14.2	139.2	.00	.03	.40-02	.88-04
N2H4 254	SEP	520.	33.	3.5	15.1	102+05	403+05	0.	41+05	16+06	.4	17.7	123.6	.00	.03	.52-02	.12-03
N2H4 255	MIX	48.	50.	2.0	5.3	218+05	653+05	0.	87+05	26+06	.7	35.4	139.8	.01	.42	.60-01	.64-04
N2H4 256	MIX	100.	60.	3.9	9.3	299+05	830+05	0.	12+06	33+06	1.2	55.3	136.8	.01	.55	.80-01	.51-04
N2H4 257	MIX	122.	61.	5.0	11.2	304+05	82+05	0.	12+06	33+06	1.2	52.9	131.4	.01	.43	.64-01	.51-04
N2H4 258	SEP	423.	34.	5.2	10.6	156+05	395+05	0.	62+05	16+06	.8	26.2	102.3	.00	.06	.10-01	.90-04
N2H4 259	SEP	401.	34.	5.6	11.6	151+05	399+05	0.	60+05	16+06	.6	26.8	122.7	.00	.06	.89-02	.90-04
N2H4 260	SEP	401.	34.	5.4	10.4	132+05	352+05	0.	53+05	14+06	.4	16.5	124.9	.00	.04	.55-02	.90-04
N2H4 261	MIX	115.	34.	1.4	2.5	137+05	351+05	0.	55+05	14+06	.5	18.6	120.0	.00	.16	.25-01	.90-04
N2H4 262	MIX	130.	37.	1.9	3.7	164+05	418+05	0.	64+05	17+06	.6	25.5	123.2	.00	.20	.30-01	.81-04
N2H4 263	MIX	147.	46.	3.6	5.7	208+05	486+05	0.	83+05	19+06	.6	24.2	117.4	.00	.16	.26-01	.64-04
N2H4 264	MIX	169.	49.	4.2	8.6	212+05	559+05	0.	85+05	22+06	.6	24.2	117.4	.00	.14	.23-01	.63-04
N2H4 265	MIX	184.	56.	6.2	11.4	236+05	596+05	0.	11+06	26+06	.5	21.1	115.2	.00	.11	.18-01	.55-04
N2H4 266	MIX	293.	62.	4.5	14.7	267+05	650+05	0.	94+05	24+06	.5	20.5	112.6	.00	.10	.18-01	.48-04
N2H4 267	SEP	384.	63.	15.2	24.9	252+05	667+05	0.	11+06	27+06	.5	19.5	125.	.00	.05	.77-02	.49-04
N2H4 268	SEP	391.	58.	10.7	24.0	207+05	654+05	0.	63+05	26+06	.4	16.1	126.3	.00	.05	.72-02	.58-04
N2H4 269	SEP	353.	54.	8.3	24.0	189+05	616+05	0.	76+05	25+06	.4	17.7	123.6	.00	.05	.77-02	.64-04
N2H4 270	MIX	176.	60.	4.4	13.7	345+05	778+05	0.	14+06	31+06	1.5	71.3	135.7	.01	.41	.59-01	.47-04
N2H4 271	MIX	179.	59.	6.5	14.8	275+05	749+05	0.	11+06	30+06	.9	35.4	111.1	.01	.20	.32-01	.53-04
N2H4 273	MIX	176.	60.	6.0	17.0	297+05	677+05	0.	12+06	33+06	1.3	48.2	107.4	.01	.27	.45-01	.53-04
N2H4 274	MIX	176.	52.	3.7	15.0	212+05	771+05	0.	85+05	31+06	1.2	45.4	114.0	.01	.26	.41-01	.72-04
N2H4 275	MIX	134.	46.	3.3	6.7	231+05	584+05	0.	92+05	33+06	1.1	42.5	114.2	.01	.31	.49-01	.65-04
N2H4 276	MIX	219.	72.	16.5	34.0	233+05	807+05	0.	93+05	32+06	.2	14.8	173.4	.00	.05	.68-02	.48-04
N2H4 277	MIX	250.	64.	10.6	19.1	233+05	657+05	0.	93+05	26+06	.2	14.8	173.4	.00	.06	.80-02	.46-04
N2H4 278	MIX	134.	46.	3.3	6.7	231+05	584+05	0.	92+05	33+06	1.2	45.4	114.0	.01	.26	.41-01	.72-04
N2H4 279	MIX	219.	72.	16.5	34.0	233+05	807+05	0.	93+05	32+06	.2	14.8	173.4	.00	.05	.68-02	.48-04
N2H4 280	MIX	250.	64.	10.6	19.1	233+05	657+05	0.	93+05	26+06	.2	14.8	173.4	.00	.06	.80-02	.46-04
N2H4 281	MIX	134.	46.	3.3	6.7	231+05	584+05	0.	92+05	33+06	1.2	45.4	114.0	.01	.26	.41-01	.72-04
N2H4 282	MIX	219.	72.	16.5	34.0	233+05	807+05	0.	93+05	32+06	.2	14.8	173.4	.00	.05	.68-02	.48-04
N2H4 283	MIX	250.	64.	10.6	19.1	233+05	657+05	0.	93+05	26+06	.2	14.8	173.4	.00	.06	.80-02	.46-04
N2H4 284	MIX	134.	46.	3.3	6.7	231+05	584+05	0.	92+05	33+06	1.2	45.4	114.0	.01	.26	.41-01	.72-04
N2H4 285	MIX	219.	72.	16.5	34.0	233+05	807+05	0.	93+05	32+06	.2	14.8	173.4	.00	.05	.68-02	.48-04
N2H4 286	MIX	250.	64.	10.6	19.1	233+05	657+05	0.	93+05	26+06	.2	14.8	173.4	.00	.06	.80-02	.46-04
N2H4 287	MIX	134.	46.	3.3	6.7	231+05	584+05	0.	92+05	33+06	1.2	45.4	114.0	.01	.26	.41-01	.72-04
N2H4 288	MIX	219.	72.	16.5	34.0	233+05	807+05	0.	93+05	32+06	.2	14.8	173.4	.00	.05	.68-02	.48-04
N2H4 289	MIX	250.	64.	10.6	19.1	233+05	657+05	0.	93+05	26+06	.2	14.8	173.4	.00	.06	.80-02	.46-04
N2H4 290	MIX	134.	46.	3.3	6.7	231+05	584+05	0.	92+05	33+06	1.2	45.4	114.0	.01	.26	.41-01	.72-04
N2H4 291	MIX	219.	72.	16.5	34.0	233+05	807+05	0.	93+05	32+06	.2	14.8	173.4	.00	.05	.68-02	.48-04
N2H4 292	MIX	250.	64.	10.6	19.1	233+05	657+05	0.	93+05	26+06	.2	14.8	173.4	.00	.06	.80-02	.46-04
N2H4 293	MIX	134.	46.	3.3	6.7	231+05	584+05	0.	92+05	33+06	1.2	45.4	114.0	.01	.26	.41-01	.72-04
N2H4 294	MIX	219.	72.	16.5	34.0	233+05	807+05	0.	93+05	32+06	.2	14.8	173.4	.00	.05	.68-02	.48-04
N2H4 295	MIX	250.	64.	10.6	19.1	233+05	657+05	0.	93+05	26+06	.2	14.8	173.4	.00	.06	.80-02	.46-04
N2H4 296	MIX	134.	46.	3.3	6.7	231+05	584+05	0.	92+05	33+06	1.2	45.4	114.0	.01	.26	.41-01	.72-04
N2H4 297	MIX	219.	72.	16.5	34.0	233+05	807+05	0.	93+05	32+06	.2	14.8	173.4	.00	.05	.68-02	.48-04
N2H4 298	MIX	250.	64.	10.6	19.1	233+05	657+05	0.	93+05	26+06	.2	14.8	173.4	.00	.06	.80-02	.46-04
N2H4 299	MIX	134.	46.	3.3	6.7	231+05	584+05	0.	92+05	33+06	1.2	45.4	114.0	.01	.26	.41-01	.72-04
N2H4 300	MIX	219.	72.	16.5	34.0	233+05	807+05	0.	93+05	32+06	.2	14.8	173.4	.00	.05	.68-02	.48-04

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STRAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TEST TYPE NO.	INJECTION TYPE	NO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TU (F)	TF (F)	MR	MF/MO	MODE	REACT (SEC)
N204	286 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	192.	1.0	63.0	54.	86.	1.14	1.060	MIX	.369-05
N204	287 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	182.	44.0	51.0	55.	86.	1.25	.920	MIX	.303-05
N204	288 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	186.	46.0	52.0	62.	57.	1.49	.600	MIX	.196-05
N204	289 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	174.	49.0	53.0	62.	55.	1.32	.820	MIX	.193-05
N204	290 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	186.	54.0	56.0	98.	101.	1.03	.780	MIX	.727-05
N204	291 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	186.	44.0	54.0	106.	71.	1.13	1.090	MIX	.453-05
N204	292 UNLIKE-DUUBLET	.060 .060	.060 .060	4.	60.	362.	92.0	121.0	60.	62.	1.09	1.200	SEP	.110-04
N204	293 UNLIKE-DUUBLET	.060 .060	.060 .060	4.	60.	180.	46.0	54.0	60.	64.	1.24	.940	MIX	.513-05
N204	294 UNLIKE-DUUBLET	.060 .060	.060 .060	4.	60.	15.	64.0	73.0	59.	53.	1.44	.780	PUP	.554-05
N204	295 UNLIKE-DUUBLET	.060 .060	.060 .060	4.	60.	15.	49.0	61.0	57.	54.	1.15	1.090	PUP	.459-05
N204	296 UNLIKE-DUUBLET	.060 .060	.060 .060	4.	60.	68.	49.0	61.0	57.	55.	1.23	1.010	PUP	.467-05
N204	297 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	76.	46.0	53.0	52.	41.	1.20	.960	MIX	.120-05
N204	298 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	108.	36.0	53.0	53.	40.	.93	1.590	MIX	.118-05
N204	299 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	126.	46.0	57.0	53.	41.	1.11	1.050	MIX	.851-06
N204	300 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	100.	47.0	57.0	53.	41.	1.16	1.050	MIX	.131-05
N204	301 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	192.	36.0	57.0	53.	43.	1.07	1.740	MIX	.144-05
N204	302 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	29.0	42.0	37.	37.	1.04	1.410	PUP	.742-06
N204	303 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	13.0	16.0	35.	42.	1.00	1.270	MIX	.331-06
N204	304 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	16.0	16.0	35.	44.	1.26	.800	PUP	.331-06
N204	305 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	52.0	49.0	32.	32.	1.58	.590	MIX	.786-06
N204	306 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	49.0	55.0	58.	59.	1.30	.860	PUP	.203-05
N204	307 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	58.0	66.0	58.	59.	1.25	.860	PUP	.244-05
N204	308 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	58.0	68.0	70.	78.	1.25	1.170	MIX	.410-05
N204	309 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	64.0	68.0	60.	66.	1.25	1.170	MIX	.299-05
N204	310 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	64.0	68.0	60.	61.	1.25	.800	MIX	.290-05
N204	311 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	63.0	68.0	53.	55.	1.25	.800	MIX	.218-05
N204	312 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	63.0	68.0	53.	55.	1.29	.800	MIX	.212-05
N204	313 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	54.0	55.0	45.	51.	1.44	.700	MIX	.173-05
N204	314 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	27.0	28.0	48.	51.	1.15	1.090	SEP	.770-06
N204	315 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	18.0	19.0	45.	51.	1.16	.770	SEP	.742-06
N204	316 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	18.0	19.0	55.	55.	1.40	.760	PUP	.628-06
N204	317 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	18.0	19.0	57.	59.	1.40	.760	PUP	.693-06
N204	318 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	18.0	19.0	58.	61.	1.29	.620	MIX	.731-06
N204	319 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	19.0	19.0	50.	51.	1.46	.700	PUP	.534-06
N204	320 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	18.0	18.0	50.	51.	1.57	.620	SEP	.506-06
N204	321 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	23.0	24.0	60.	72.	1.37	.780	SEP	.122-05
N204	322 UNLIKE-DUUBLET	.040 .040	.040 .040	4.	60.	15.	22.0	24.0	58.	51.	1.30	.850	PUP	.173-05
N204	323 UNLIKE-DUUBLET	.060 .060	.060 .060	4.	60.	15.	21.0	23.0	48.	48.	1.30	.840	PUP	.153-05
N204	324 UNLIKE-DUUBLET	.060 .060	.060 .060	4.	60.	15.	22.0	24.0	47.	55.	1.31	.830	PUP	.156-05

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TEST TYPE NO.	MODE	PC (PSIA)	WAVG	MEF	WEO	REF	WEO	DELTI (DEG F)	HELF	RELO	PPF (PSIA)	PPU	MHVP	XF	XU	XP	RESID
N2H8 286	MIX	192	57	6.6	10.0	.229+05	.504+05	0	.92+05	.20+06	.4	9.8	71.5	.00	.05	.11-01	.53-04
N2H8 287	MIX	162	47	4.1	7.1	.186+05	.437+05	0	.74+05	.17+06	.4	10.1	73.6	.00	.06	.11-01	.65-04
N2H8 288	MIX	186	48	4.1	8.2	.151+05	.474+05	0	.60+05	.14+06	.2	12.3	224.7	.00	.07	.74-02	.64-04
N2H8 289	MIX	174	51	4.0	6.7	.151+05	.505+05	0	.61+05	.20+06	.1	12.3	238.4	.00	.07	.77-02	.63-04
N2H8 290	MIX	186	55	5.2	13.5	.232+05	.683+05	0	.93+05	.27+06	.7	28.1	120.8	.00	.15	.23-01	.60-04
N2H8 291	MIX	162	49	4.0	6.2	.174+05	.588+05	0	.69+05	.24+06	.3	35.4	378.1	.00	.22	.11-01	.62-04
N2H8 293	SEP	362	106	85.7	95.0	.545+05	.141+06	0	.22+06	.56+06	.2	11.7	182.1	.00	.93	.41-02	.01-04
N2H8 294	MIX	180	50	6.5	11.8	.247+05	.704+05	0	.99+05	.28+06	.2	11.7	169.0	.00	.06	.45-02	.93-04
N2H8 295	PUP	15	68	1.0	1.9	.308+05	.974+05	0	.12+06	.39+06	.1	11.4	236.0	.01	.77	.85-01	.68-04
N2H8 296	PUP	15	55	1.7	1.1	.259+05	.738+05	0	.10+06	.30+06	.1	10.7	210.3	.00	.16	.18-01	.62-04
N2H8 297	PUP	66	54	3.1	5.0	.261+05	.758+05	0	.10+06	.30+06	.1	10.7	210.3	.00	.16	.18-01	.62-04
N2H8 298	MIX	76	49	1.7	3.2	.138+05	.450+05	0	.55+05	.18+06	.1	9.2	323.9	.00	.12	.11-01	.63-04
N2H8 299	MIX	108	45	2.4	2.7	.137+05	.354+05	0	.55+05	.18+06	.1	9.5	334.7	.00	.09	.40-02	.63-04
N2H8 300	MIX	122	42	1.3	5.2	.963+04	.453+05	0	.39+05	.18+06	.1	9.5	334.2	.00	.08	.71-02	.90-04
N2H8 301	MIX	100	52	2.6	4.4	.148+05	.463+05	0	.59+05	.19+06	.1	9.5	334.2	.00	.09	.87-02	.58-04
N2H8 302	MIX	192	46	5.0	5.0	.150+05	.358+05	0	.60+05	.14+06	.1	10.1	318.2	.00	.05	.49-02	.56-04
N2H8 304	PUP	15	35	.2	.2	.109+05	.284+05	0	.44+05	.11+06	.1	6.0	231.8	.01	.41	.46-01	.79-04
N2H8 305	MIX	15	14	.0	.0	.419+04	.127+05	0	.17+05	.51+05	.1	7.2	244.7	.01	.49	.53-01	.21-03
N2H8 306	PUP	15	16	.0	.1	.425+04	.157+05	0	.17+05	.51+05	.1	6.4	188.7	.01	.44	.53-01	.21-03
N2H8 307	MIX	15	51	.3	.7	.127+05	.512+05	0	.51+05	.20+06	.1	5.1	194.7	.01	.35	.42-01	.68-04
N2H8 308	SEP	15	52	.4	.7	.162+05	.495+05	0	.65+05	.20+06	.2	11.0	192.6	.01	.75	.92-01	.61-04
N2H8 309	SEP	15	62	.5	1.0	.194+05	.585+05	0	.78+05	.23+06	.2	14.8	141.4	.02	1.01	.14+00	.49-04
N2H8 310	MIX	15	62	.6	1.1	.232+05	.624+05	0	.93+05	.25+06	.3	14.8	141.4	.02	1.01	.14+00	.49-04
N2H8 311	MIX	15	62	.6	1.0	.210+05	.591+05	0	.84+05	.24+06	.2	11.7	151.9	.01	.74	.11+00	.49-04
N2H8 312	MIX	15	66	.6	1.3	.203+05	.674+05	0	.81+05	.27+06	.2	13.5	217.2	.01	.92	.11+00	.49-04
N2H8 313	MIX	15	66	.6	1.2	.194+05	.630+05	0	.78+05	.25+06	.1	9.5	187.3	.01	.65	.40-01	.49-04
N2H8 314	MIX	15	65	.6	1.2	.194+05	.620+05	0	.78+05	.25+06	.1	9.5	187.3	.01	.65	.40-01	.49-04
N2H8 316	MIX	15	63	.5	1.1	.175+05	.612+05	0	.70+05	.24+06	.1	8.2	185.8	.01	.56	.69-01	.53-04
N2H8 317	PUP	15	54	.4	.8	.153+05	.526+05	0	.61+05	.21+06	.1	7.6	173.6	.01	.52	.67-01	.61-04
N2H8 319	SEP	15	25	.1	.2	.778+04	.263+05	0	.31+05	.11+06	.1	4.2	185.8	.01	.56	.69-01	.61-04
N2H8 321	PUP	15	27	.1	.2	.778+04	.263+05	0	.31+05	.11+06	.1	7.6	173.6	.01	.52	.67-01	.61-04
N2H8 322	MIX	15	18	.0	.1	.543+04	.179+05	0	.22+05	.72+05	.1	10.1	198.4	.01	.69	.83-01	.18-03
N2H8 323	PUP	15	18	.0	.1	.558+04	.181+05	0	.22+05	.72+05	.2	10.7	187.5	.01	.73	.91-01	.18-03
N2H8 324	MIX	15	14	.0	.1	.508+04	.182+05	0	.23+05	.73+05	.2	11.0	180.0	.01	.75	.95-01	.18-03
N2H8 325	PUP	15	14	.0	.1	.508+04	.182+05	0	.23+05	.73+05	.1	8.6	193.8	.01	.58	.71-01	.18-03
N2H8 326	SEP	15	23	.0	.1	.500+04	.175+05	0	.20+05	.70+05	.1	8.6	193.8	.01	.58	.71-01	.18-03
N2H8 327	PUP	15	23	.1	.2	.778+04	.238+05	0	.31+05	.95+05	.3	12.6	181.8	.02	.86	.12+00	.14-03
N2H8 328	PUP	15	23	.1	.2	.100+05	.333+05	0	.40+05	.13+06	.1	11.0	246.2	.01	.75	.41-01	.21-03
N2H8 329	PUP	15	22	.1	.2	.940+04	.308+05	0	.38+05	.12+06	.1	8.2	207.9	.01	.56	.65-01	.22-03
N2H8 330	PUP	15	22	.1	.2	.103+05	.321+05	0	.41+05	.13+06	.1	8.0	159.3	.01	.54	.73-01	.21-03

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HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VD (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR MF/MO	MODE	REACT (SEC)
N2H4	329	UNLIKE-DUUBLET	.060	.060	4	60	15	21.0	24.0	47	53	1.27	PUP	.153-05
N2H4	330	UNLIKE-DUUBLET	.060	.060	4	60	15	21.0	24.0	48	53	1.27	PUP	.154-05
N2H4	331	UNLIKE-DUUBLET	.060	.060	4	60	15	27.0	31.0	64	61	1.24	PUP	.290-05
N2H4	332	UNLIKE-DUUBLET	.060	.060	4	60	75	34.0	43.0	63	64	1.11	PUP	.425-05
N2H4	333	UNLIKE-DUUBLET	.060	.060	4	60	80	40.0	50.0	63	64	1.13	PUP	.494-05
N2H4	334	UNLIKE-DUUBLET	.060	.060	4	60	95	34.0	57.0	61	53	.82	PUP	.537-05
N2H4	335	UNLIKE-DUUBLET	.060	.060	4	60	96	26.0	31.0	64	61	1.23	PUP	.290-05
N2H4	336	UNLIKE-DUUBLET	.060	.060	4	60	70	34.0	44.0	63	64	1.11	PUP	.414-05
N2H4	337	UNLIKE-DUUBLET	.060	.060	4	60	60	36.0	57.0	61	66	.89	PUP	.571-05
N2H4	338	UNLIKE-DUUBLET	.060	.060	4	60	85	45.0	50.0	86	89	1.24	PUP	.105-04
N2H4	339	UNLIKE-DUUBLET	.060	.060	4	60	125	43.0	58.0	108	97	1.11	PUP	.182-04
N2H4	340	UNLIKE-DUUBLET	.060	.060	4	60	75	36.0	50.0	108	117	1.05	PUP	.219-04
N2H4	341	UNLIKE-DUUBLET	.060	.060	4	60	75	41.0	50.0	100	107	1.16	PUP	.171-04
N2H4	342	UNLIKE-DUUBLET	.060	.060	4	60	145	42.0	50.0	99	107	1.21	PUP	.169-04
N2H4	343	UNLIKE-DUUBLET	.060	.060	4	60	145	41.0	50.0	103	118	1.16	PUP	.210-04
N2H4	344	UNLIKE-DUUBLET	.060	.060	4	60	145	43.0	50.0	80	81	1.24	PUP	.813-05
N2H4	345	UNLIKE-DUUBLET	.060	.060	4	60	165	45.0	49.0	78	76	1.26	PUP	.779-05
N2H4	346	UNLIKE-DUUBLET	.060	.060	4	60	165	45.0	50.0	76	76	1.28	PUP	.714-05
N2H4	347	UNLIKE-DUUBLET	.060	.060	4	60	165	44.0	49.0	78	80	1.25	PUP	.758-05
N2H4	348	UNLIKE-DUUBLET	.060	.060	4	60	165	32.0	50.0	83	83	.91	PUP	.890-05
N2H4	349	UNLIKE-DUUBLET	.060	.060	4	60	165	33.0	51.0	90	91	.93	PUP	.117-04
N2H4	350	UNLIKE-DUUBLET	.060	.060	4	60	185	40.0	51.0	94	91	1.12	PUP	.123-04
N2H4	351	UNLIKE-DUUBLET	.060	.060	4	60	185	47.0	50.0	70	72	1.35	PUP	.619-05
N2H4	352	UNLIKE-DUUBLET	.060	.060	4	60	185	41.0	49.0	68	72	1.21	UNDEF	.594-05
N2H4	353	UNLIKE-DUUBLET	.060	.060	4	60	185	42.0	49.0	66	70	1.23	PUP	.562-05
N2H4	354	UNLIKE-DUUBLET	.060	.060	4	60	185	42.0	51.0	50	49	1.19	PUP	.309-05
N2H4	355	UNLIKE-DUUBLET	.060	.060	4	60	175	44.0	51.0	47	47	1.23	UNDEF	.285-05
N2H4	356	UNLIKE-DUUBLET	.060	.060	4	60	165	36.0	51.0	50	49	1.03	PUP	.309-05
N2H4	357	UNLIKE-DUUBLET	.060	.060	4	60	165	41.0	49.0	50	51	1.21	PUP	.310-05
N2H4	358	UNLIKE-DUUBLET	.060	.060	4	60	165	42.0	51.0	50	51	1.19	PUP	.323-05
N2H4	359	UNLIKE-DUUBLET	.060	.060	4	60	165	42.0	51.0	48	47	1.19	PUP	.289-05
N2H4	360	UNLIKE-DUUBLET	.060	.060	4	60	165	33.0	60.0	57	53	.08	PUP	.443-05
N2H4	361	UNLIKE-DUUBLET	.060	.060	4	60	165	42.0	51.0	55	55	1.02	PUP	.379-05
N2H4	362	UNLIKE-DUUBLET	.060	.060	4	60	165	40.0	52.0	55	53	1.12	PUP	.372-05
N2H4	363	UNLIKE-DUUBLET	.060	.060	4	60	165	44.0	51.0	53	53	1.23	PUP	.354-05
N2H4	364	UNLIKE-DUUBLET	.060	.060	4	60	165	44.0	52.0	60	63	1.23	PUP	.484-05
N2H4	365	UNLIKE-DUUBLET	.060	.060	4	60	165	45.0	51.0	60	63	1.22	PUP	.474-05
N2H4	366	UNLIKE-DUUBLET	.060	.060	4	60	165	43.0	52.0	60	63	1.25	PUP	.484-05
N2H4	367	UNLIKE-DUUBLET	.060	.060	4	60	165	45.0	52.0	76	80	1.25	PUP	.786-05
N2H4	368	UNLIKE-DUUBLET	.060	.060	4	60	165	45.0	49.0	68	70	1.29	PUP	.574-05

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

FUEL TEST TYPE NO.	MODE	PC (PSIA)	VAVG	WEF	MED	REF	REU	DEL(I (DEG F)	HELF	RELO	PPF (PSIA)(PSTA)	PPU	MRVP	KF	XU	XP	MESLD
N2M4 329	POP	15.	22.	.1	.2	.101+05	.306+05	0.	.41+05	.12+06	.1	8.0	169.8	.01	.54	.71-01	.21-03
N2M4 330	POP	15.	22.	.1	.2	.101+05	.306+05	0.	.41+05	.12+06	.1	8.2	173.5	.01	.56	.72-01	.21-03
N2M4 331	POP	15.	29.	.2	.3	.139+05	.422+05	0.	.55+05	.17+06	.2	12.9	208.0	.01	.48	.10+00	.16-03
N2M4 332	POP	75.	38.	1.7	2.7	.197+05	.528+05	0.	.79+05	.21+06	.2	12.6	181.5	.00	.17	.21-01	.12-03
N2M4 333	PUP	80.	45.	2.5	4.0	.239+05	.622+05	0.	.91+05	.25+06	.2	12.6	181.5	.00	.16	.20-01	.10-03
N2M4 334	PUP	95.	47.	3.8	3.4	.239+05	.523+05	0.	.10+06	.21+06	.2	12.0	179.6	.00	.13	.16-01	.88-04
N2M4 335	PUP	96.	28.	1.1	2.0	.139+05	.406+05	0.	.55+05	.16+06	.2	12.9	208.0	.00	.13	.16-01	.16-03
N2M4 336	POP	70.	39.	1.7	2.5	.201+05	.528+05	0.	.80+05	.21+06	.2	12.6	181.5	.00	.18	.23-01	.11-03
N2M4 337	POP	80.	47.	3.2	3.2	.265+05	.554+05	0.	.11+06	.22+06	.2	12.0	161.7	.00	.15	.20-01	.88-04
N2M4 338	POP	85.	46.	2.8	5.5	.280+05	.713+05	0.	.11+06	.31+06	.5	22.1	136.5	.01	.26	.34-01	.10-03
N2M4 339	POP	125.	51.	5.6	10.1	.348+05	.911+05	0.	.14+06	.36+06	.6	37.0	173.9	.00	.30	.38-01	.86-04
N2M4 340	POP	75.	43.	2.6	3.9	.339+05	.729+05	0.	.14+06	.29+06	1.1	37.0	100.6	.01	.49	.84-01	.10-03
N2M4 341	PUP	75.	45.	2.5	4.8	.321+05	.799+05	0.	.13+06	.32+06	.8	30.7	108.3	.01	.41	.67-01	.10-03
N2M4 342	PUP	145.	46.	4.9	9.7	.321+05	.813+05	0.	.13+06	.33+06	.8	30.0	106.2	.01	.21	.34-01	.10-03
N2M4 343	PUP	145.	45.	5.0	9.4	.381+05	.811+05	0.	.14+06	.32+06	1.1	33.1	88.7	.01	.23	.42-01	.10-03
N2M4 344	POP	145.	46.	4.7	9.2	.232+05	.736+05	0.	.10+06	.29+06	.3	19.0	162.5	.00	.13	.17-01	.10-03
N2M4 345	POP	165.	46.	5.1	10.3	.257+05	.727+05	0.	.10+06	.29+06	.3	18.1	155.8	.00	.11	.15-01	.10-03
N2M4 346	PUP	165.	47.	5.3	11.2	.251+05	.752+05	0.	.10+06	.30+06	.3	17.3	171.5	.00	.10	.14-01	.10-03
N2M4 347	PUP	165.	46.	5.1	10.8	.255+05	.784+05	0.	.10+06	.30+06	.3	18.1	163.0	.00	.11	.15-01	.10-03
N2M4 348	PUP	165.	41.	5.3	5.9	.286+05	.558+05	0.	.11+06	.22+06	.4	20.5	159.9	.00	.12	.17-01	.10-03
N2M4 349	PUP	165.	42.	5.6	6.5	.290+05	.611+05	0.	.12+06	.24+06	.5	24.2	139.0	.00	.15	.21-01	.98-04
N2M4 350	PUP	185.	45.	6.3	10.9	.290+05	.748+05	0.	.12+06	.30+06	.5	26.8	152.8	.00	.14	.20-01	.98-04
N2M4 351	PUP	185.	44.	5.8	13.3	.293+05	.759+05	0.	.97+05	.30+06	.3	14.8	164.0	.00	.08	.11-01	.10-03
N2M4 352	Unit	185.	45.	5.0	4.9	.238+05	.655+05	0.	.95+05	.26+06	.3	14.2	157.7	.00	.09	.12-01	.10-03
N2M4 353	PUP	165.	45.	5.0	7.3	.235+05	.663+05	0.	.94+05	.27+06	.2	13.5	160.0	.00	.08	.11-01	.10-03
N2M4 354	PUP	165.	46.	5.2	6.6	.210+05	.611+05	0.	.84+05	.24+06	.1	8.6	208.6	.00	.05	.53-02	.98-04
N2M4 355	Unit	175.	47.	5.5	9.9	.207+05	.642+05	0.	.83+05	.26+06	.1	8.0	211.9	.00	.05	.53-02	.98-04
N2M4 356	PUP	165.	43.	5.2	6.3	.210+05	.524+05	0.	.84+05	.21+06	.1	8.6	208.6	.00	.05	.61-02	.98-04
N2M4 357	PUP	165.	45.	4.4	4.2	.204+05	.596+05	0.	.82+05	.24+06	.1	8.6	193.8	.00	.05	.63-02	.10-03
N2M4 358	PUP	165.	46.	5.2	6.6	.213+05	.611+05	0.	.85+05	.24+06	.1	8.6	193.8	.00	.05	.63-02	.98-04
N2M4 359	PUP	165.	46.	5.2	4.5	.207+05	.612+05	0.	.83+05	.24+06	.1	8.2	216.6	.00	.05	.57-02	.98-04
N2M4 360	PUP	165.	56.	7.2	5.5	.253+05	.497+05	0.	.10+06	.20+06	.1	10.7	224.0	.00	.07	.73-02	.98-04
N2M4 361	PUP	165.	46.	5.2	6.8	.218+05	.626+05	0.	.87+05	.25+06	.1	10.1	198.8	.00	.06	.74-02	.98-04
N2M4 362	PUP	165.	46.	3.5	5.1	.220+05	.596+05	0.	.88+05	.24+06	.1	10.1	211.8	.00	.10	.11-01	.96-04
N2M4 363	PUP	175.	47.	3.3	6.1	.215+05	.650+05	0.	.86+05	.24+06	.1	9.5	199.6	.00	.09	.11-01	.98-04
N2M4 364	PUP	165.	44.	3.5	6.3	.236+05	.673+05	0.	.94+05	.27+06	.2	11.7	175.3	.00	.11	.14-01	.96-04
N2M4 365	PUP	165.	47.	3.4	6.0	.231+05	.658+05	0.	.93+05	.26+06	.2	11.7	175.3	.00	.11	.14-01	.96-04
N2M4 366	PUP	165.	46.	3.5	6.6	.236+05	.688+05	0.	.94+05	.28+06	.2	11.7	175.3	.00	.11	.14-01	.96-04
N2M4 367	PUP	165.	44.	3.6	7.1	.270+05	.752+05	0.	.11+06	.30+06	.3	17.3	156.6	.00	.16	.23-01	.98-04
N2M4 368	PUP	185.	47.	5.6	12.1	.235+05	.718+05	0.	.94+05	.24+06	.2	14.2	166.8	.00	.04	.10-01	.10-03

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HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TEST TYPE NO.	INJECTOR TYPE	DN (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MD	MODE	REACT (SEC)
N2M4 369	UNLINE-DUUBLET	.060	.060	4.	60.	185.	43.0	50.0	70.	69.	1.24	.930	PUP	.588-05
N2M4 370	UNLINE-DUUBLET	.060	.060	4.	60.	185.	41.0	49.0	65.	64.	1.21	.980	PUP	.496-05
N2M4 371	UNLINE-DUUBLET	.060	.060	4.	60.	185.	37.0	49.0	63.	63.	1.06	1.230	PUP	.474-05
N2M4 372	UNLINE-DUUBLET	.060	.060	4.	60.	185.	34.0	49.0	55.	55.	.98	1.450	PUP	.364-05
N2M4 373	UNLINE-DUUBLET	.055	.055	4.	60.	95.	52.0	60.0	63.	64.	1.23	1.030	PUP	.498-05
N2M4 374	UNLINE-DUUBLET	.055	.055	4.	60.	95.	51.0	60.0	63.	64.	1.19	.990	PUP	.498-05
N2M4 375	UNLINE-DUUBLET	.055	.055	4.	60.	95.	54.0	60.0	63.	64.	1.27	.790	PUP	.498-05
N2M4 376	UNLINE-DUUBLET	.055	.055	4.	60.	165.	54.0	60.0	63.	64.	1.29	.850	PUP	.498-05
N2M4 377	UNLINE-DUUBLET	.055	.055	4.	60.	175.	54.0	60.0	65.	62.	1.27	.880	PUP	.703-05
N2M4 378	UNLINE-DUUBLET	.055	.055	4.	60.	175.	56.0	61.0	66.	69.	1.31	.830	PUP	.864-05
N2M4 379	UNLINE-DUUBLET	.055	.055	4.	60.	175.	55.0	61.0	104.	82.	1.29	.860	PUP	.114-04
N2M4 380	UNLINE-DUUBLET	.055	.055	4.	60.	175.	53.0	61.0	114.	105.	1.24	.930	PUP	.199-04
N2M4 381	UNLINE-DUUBLET	.055	.055	4.	60.	175.	54.0	61.0	114.	107.	1.27	.880	PUP	.206-04
N2M4 382	UNLINE-DUUBLET	.055	.055	4.	60.	175.	52.0	61.0	111.	110.	1.21	1.080	PUP	.209-04
N2M4 383	UNLINE-DUUBLET	.055	.055	4.	60.	175.	56.0	60.0	65.	66.	1.32	.810	PUP	.531-05
N2M4 384	UNLINE-DUUBLET	.055	.055	4.	60.	175.	53.0	60.0	88.	78.	1.29	.990	PUP	.857-05
N2M4 385	UNLINE-DUUBLET	.055	.055	4.	60.	175.	53.0	60.0	76.	78.	1.26	.910	PUP	.741-05
N2M4 386	UNLINE-DUUBLET	.055	.055	4.	60.	175.	42.0	46.0	48.	51.	1.29	.848	PUP	.239-05
N2M4 389	UNLINE-DUUBLET	.055	.055	4.	60.	175.	34.0	46.0	48.	51.	1.13	1.100	PUP	.239-05
N2M4 390	UNLINE-DUUBLET	.055	.055	4.	60.	175.	33.0	46.0	48.	51.	1.71	.484	PUP	.239-05
N2M4 391	UNLINE-DUUBLET	.055	.055	4.	60.	195.	49.0	60.0	48.	52.	1.17	1.040	PUP	.318-05
N2M4 392	UNLINE-DUUBLET	.055	.055	4.	60.	195.	49.0	60.0	48.	52.	1.16	1.080	PUP	.318-05
N2M4 393	UNLINE-DUUBLET	.055	.055	4.	60.	415.	37.0	59.0	55.	60.	.88	1.840	SEP	.401-05
N2M4 394	UNLINE-DUUBLET	.060	.060	4.	60.	235.	61.0	71.0	56.	62.	1.21	.967	MIX	.611-05
N2M4 395	UNLINE-DUUBLET	.060	.060	4.	60.	225.	51.0	71.0	45.	53.	1.02	1.370	MIX	.440-05
N2M4 396	UNLINE-DUUBLET	.060	.060	4.	60.	235.	60.0	71.0	38.	45.	1.20	.991	MIX	.334-05
N2M4 397	UNLINE-DUUBLET	.060	.060	4.	60.	225.	59.0	71.0	63.	62.	1.19	1.520	MIX	.672-05
N2M4 398	UNLINE-DUUBLET	.060	.060	4.	60.	235.	56.0	71.0	71.	70.	1.35	.931	MIX	.862-05
N2M4 399	UNLINE-DUUBLET	.060	.060	4.	60.	235.	54.0	71.0	79.	76.	1.09	1.190	MIX	.105-04
N2M4 400	UNLINE-DUUBLET	.060	.060	4.	60.	235.	53.0	71.0	78.	84.	1.06	1.240	MIX	.122-04
N2M4 401	UNLINE-DUUBLET	.060	.060	4.	60.	235.	51.0	71.0	83.	87.	1.02	1.360	MIX	.138-04
N2M4 402	UNLINE-DUUBLET	.060	.060	4.	60.	235.	51.0	69.0	83.	99.	1.60	1.320	MIX	.166-04
N2M4 403	UNLINE-DUUBLET	.060	.060	4.	60.	255.	50.0	61.0	98.	103.	1.16	1.050	MIX	.190-04
N2M4 404	UNLINE-DUUBLET	.060	.060	4.	60.	215.	49.0	50.0	97.	108.	1.16	.882	MIX	.168-04
N2M4 405	UNLINE-DUUBLET	.060	.060	4.	60.	235.	46.0	69.0	99.	48.	.99	1.440	MIX	.766-05
N2M4 406	UNLINE-DUUBLET	.060	.060	4.	60.	235.	56.0	66.0	63.	66.	1.02	.980	MIX	.678-05
N2M4 407	UNLINE-DUUBLET	.060	.060	4.	60.	235.	52.0	65.0	63.	66.	1.14	1.090	MIX	.668-05
N2M4 408	UNLINE-DUUBLET	.060	.060	4.	60.	235.	54.0	65.0	63.	66.	1.11	1.080	MIX	.668-05
N2M4 409	UNLINE-DUUBLET	.060	.060	4.	60.	235.	50.0	65.0	63.	66.	1.09	1.210	MIX	.668-05
N2M4 410	UNLINE-DUUBLET	.060	.060	4.	60.	185.	36.0	54.0	76.	85.	.95	1.620	PUP	.925-05

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERCOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TEST TYPE NO.	PC (PSIA)	WAVG WET	WED WET	REF WET	REG (DELTA F)	DELTA (DELTA F)	REL (PSIA)	PPF (PSIA)	MRVP	XF	XI	XP	RESIDU			
N2H4 369	185	46	5.8	11.1	237+05	694+05	0	.95+05	.26+06	.2	14.8	178.6	.00	.08	.10-01	.10-03
N2H4 370	185	45	5.5	9.9	224+05	644+05	0	.90+05	.26+06	.2	13.2	189.7	.00	.07	.88-02	.10-03
N2H4 371	135	43	5.5	8.0	222+05	575+05	0	.89+05	.23+06	.2	12.6	188.2	.00	.07	.84-02	.10-03
N2H4 372	185	42	5.4	6.5	210+05	507+05	0	.84+05	.20+06	.1	10.1	198.8	.00	.05	.66-02	.10-03
N2H4 373	95	56	3.9	7.4	251+05	741+05	0	.10+06	.30+06	.2	12.6	181.5	.00	.13	.17-01	.76-04
N2H4 374	95	55	3.9	7.1	251+05	727+05	0	.10+06	.29+06	.2	12.6	181.5	.00	.13	.17-01	.76-04
N2H4 375	95	57	3.9	8.0	251+05	769+05	0	.10+06	.31+06	.2	12.6	181.5	.00	.13	.17-01	.76-04
N2H4 376	165	57	6.8	13.9	251+05	769+05	0	.10+06	.31+06	.2	12.6	181.5	.00	.08	.96-02	.76-04
N2H4 377	175	57	7.2	14.8	291+05	778+05	0	.12+06	.31+06	.4	13.2	111.2	.00	.08	.12-01	.76-04
N2H4 378	175	58	7.8	16.2	313+05	820+05	0	.13+06	.33+06	.5	14.2	98.3	.00	.08	.15-01	.75-04
N2H4 379	175	58	7.7	14.9	295+05	100+06	0	.12+06	.40+06	.4	33.9	265.0	.00	.19	.20-01	.75-04
N2H4 380	175	57	8.1	18.6	355+05	101+06	0	.14+06	.41+06	.8	42.5	155.1	.00	.24	.33-01	.75-04
N2H4 381	175	57	8.1	19.4	359+05	103+06	0	.14+06	.41+06	.8	42.5	155.1	.00	.24	.34-01	.75-04
N2H4 382	175	56	8.2	17.6	365+05	980+05	0	.15+06	.39+06	.9	39.6	126.2	.01	.23	.34-01	.75-04
N2H4 383	175	58	7.2	16.0	255+05	806+05	0	.10+06	.32+06	.2	13.2	177.2	.00	.03	.96-02	.76-04
N2H4 384	175	57	7.4	17.3	281+05	907+05	0	.11+06	.36+06	.3	23.2	214.4	.00	.13	.15-01	.76-04
N2H4 385	175	56	7.4	15.1	281+05	612+05	0	.11+06	.32+06	.3	17.3	63.7	.00	.10	.13-01	.76-04
N2H4 386	175	44	4.1	8.3	176+05	561+05	0	.70+05	.22+06	.1	8.2	185.8	.00	.05	.58-02	.10-03
N2H4 387	175	41	4.1	8.4	176+05	494+05	0	.70+05	.20+06	.1	8.2	185.8	.00	.05	.58-02	.10-03
N2H4 388	175	39	4.1	5.8	176+05	468+05	0	.70+05	.19+06	.1	8.2	185.8	.00	.05	.58-02	.10-03
N2H4 389	195	54	7.6	12.6	231+05	655+05	0	.92+05	.26+06	.1	8.2	179.4	.00	.04	.53-02	.76-04
N2H4 390	195	54	7.8	12.6	231+05	655+05	0	.92+05	.26+06	.1	8.2	179.4	.00	.04	.53-02	.76-04
N2H4 391	415	46	10.5	15.8	240+05	506+05	0	.96+05	.20+06	.2	10.1	172.7	.00	.02	.52-02	.78-04
N2H4 392	415	66	14.7	26.6	320+05	914+05	0	.13+06	.37+06	.2	10.4	163.9	.00	.04	.59-02	.78-04
N2H4 393	235	61	13.8	16.9	300+05	745+05	0	.12+06	.30+06	.1	7.6	162.2	.00	.03	.45-02	.70-04
N2H4 394	235	65	14.5	23.8	644+05	882+05	0	.11+06	.35+06	.1	6.2	184.2	.00	.03	.33-02	.70-04
N2H4 395	235	64	14.1	24.6	320+05	917+05	0	.13+06	.37+06	.2	12.6	195.5	.00	.06	.68-02	.70-04
N2H4 396	235	62	14.9	24.1	340+05	909+05	0	.14+06	.36+06	.2	15.2	178.0	.00	.06	.82-02	.70-04
N2H4 397	235	62	15.1	23.3	357+05	918+05	0	.14+06	.37+06	.3	18.6	183.0	.00	.08	.94-02	.70-04
N2H4 398	235	62	15.3	22.7	381+05	896+05	0	.15+06	.36+06	.4	18.1	156.6	.00	.08	.11-01	.70-04
N2H4 399	235	61	15.4	21.2	422+05	869+05	0	.16+06	.36+06	.4	20.5	136.7	.00	.09	.15-01	.70-04
N2H4 400	235	54	15.9	21.2	422+05	869+05	0	.17+06	.36+06	.6	20.5	96.0	.00	.09	.15-01	.72-04
N2H4 401	235	55	17.8	24.0	363+05	961+05	0	.15+06	.38+06	.7	29.4	117.6	.00	.12	.16-01	.82-04
N2H4 402	235	49	7.3	18.5	322+05	936+05	0	.13+06	.37+06	.9	28.7	99.1	.00	.13	.23-01	.10-03
N2H4 403	235	54	13.5	20.5	282+05	929+05	0	.11+06	.37+06	.1	30.0	694.6	.00	.13	.76-02	.72-04
N2H4 404	235	61	12.4	23.2	306+05	870+05	0	.12+06	.35+06	.2	12.6	169.5	.00	.05	.70-02	.76-04
N2H4 405	235	58	12.4	21.0	302+05	806+05	0	.12+06	.32+06	.2	12.6	169.5	.00	.05	.70-02	.77-04
N2H4 406	235	54	12.4	21.5	302+05	839+05	0	.12+06	.34+06	.2	12.6	169.5	.00	.05	.70-02	.77-04
N2H4 407	235	57	12.4	18.5	302+05	777+05	0	.12+06	.31+06	.2	12.6	169.5	.00	.05	.70-02	.77-04
N2H4 408	235	57	12.4	18.5	302+05	777+05	0	.12+06	.31+06	.2	12.6	169.5	.00	.05	.70-02	.77-04
N2H4 409	235	57	12.4	18.5	302+05	777+05	0	.12+06	.31+06	.2	12.6	169.5	.00	.05	.70-02	.77-04
N2H4 410	185	45	7.9	6.0	292+05	601+05	0	.12+06	.24+06	.4	17.3	125.6	.00	.09	.14-01	.93-04

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TYPE	TEST NO.	INJECTOR TYPE	DU OF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VD (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR MF/MO	MODE	REACT (SEC)
N2H4	411	UNLINE-DUUBLET	.060 .060	4.	60.	175.	34.0	55.0	89.	95.	.88	1.850	PUP .134-04
N2H4	412	UNLINE-DUUBLET	.060 .060	4.	60.	190.	36.0	55.0	87.	97.	.92	1.690	PUP .135-04
N2H4	413	UNLINE-DUUBLET	.060 .060	4.	60.	90.	40.0	55.0	83.	89.	1.03	1.340	PUP .112-04
N2H4	414	UNLINE-DUUBLET	.060 .060	4.	60.	80.	39.0	54.0	94.	105.	1.01	1.390	PUP .167-04
N2H4	415	UNLINE-DUUBLET	.060 .060	4.	60.	85.	44.0	54.0	98.	108.	1.14	1.090	PUP .164-04
N2H4	416	UNLINE-DUUBLET	.060 .060	4.	60.	80.	43.0	54.0	101.	110.	1.13	1.110	PUP .197-04
N2H4	417	UNLINE-DUUBLET	.060 .060	4.	60.	80.	41.0	55.0	109.	116.	1.05	1.130	PUP .240-04
N2H4	418	UNLINE-DUUBLET	.060 .060	4.	60.	80.	39.0	55.0	102.	116.	.95	1.430	PUP .222-04
N2H4	419	UNLINE-DUUBLET	.060 .060	4.	60.	80.	37.2	54.3	102.	116.	.98	1.490	PUP .219-04
N2H4	423	UNLINE-DUUBLET	.060 .060	4.	60.	105.	51.7	54.3	109.	112.	1.35	.780	PUP .224-04
N2H4	425	UNLINE-DUUBLET	.060 .060	4.	60.	95.	50.3	54.3	103.	114.	1.32	.820	PUP .216-04
N2H4	426	UNLINE-DUUBLET	.060 .060	4.	60.	95.	37.2	53.5	75.	89.	.99	1.450	PUP .985-05
N2H4	427	UNLINE-DUUBLET	.060 .060	4.	60.	95.	36.2	53.5	68.	81.	.97	1.150	PUP .732-05
N2H4	428	UNLINE-DUUBLET	.060 .060	4.	60.	95.	34.9	54.7	79.	82.	.91	1.720	PUP .901-05
N2H4	429	UNLINE-DUUBLET	.060 .060	4.	60.	92.	39.7	54.7	104.	108.	1.04	1.330	PUP .200-04
N2H4	430	UNLINE-DUUBLET	.060 .060	4.	60.	95.	36.3	54.7	106.	114.	.95	1.590	PUP .225-04
N2H4	431	UNLINE-DUUBLET	.055 .055	4.	60.	175.	48.7	56.2	68.	75.	1.19	1.000	PUP .622-05
N2H4	432	UNLINE-DUUBLET	.055 .055	4.	60.	175.	46.7	56.5	81.	43.	1.14	1.100	PUP .105-04
N2H4	433	UNLINE-DUUBLET	.055 .055	4.	60.	215.	65.4	74.0	92.	101.	1.26	.690	MIX .174-04
N2H4	434	UNLINE-DUUBLET	.055 .055	4.	60.	215.	64.4	73.5	97.	106.	1.23	.910	MIX .202-04
N2H4	435	UNLINE-DUUBLET	.055 .055	4.	60.	175.	51.0	64.5	101.	114.	1.20	1.060	PUP .210-04
N2H4	436	UNLINE-DUUBLET	.055 .055	4.	60.	75.	50.1	60.4	96.	106.	1.18	1.020	PUP .163-04
N2H4	437	UNLINE-DUUBLET	.055 .055	4.	60.	95.	45.2	59.8	72.	86.	1.08	1.230	PUP .851-05
N2H4	438	UNLINE-DUUBLET	.040 .040	4.	60.	95.	45.2	59.8	68.	77.	1.08	1.230	PUP .663-05
N2H4	439	UNLINE-DUUBLET	.040 .040	4.	60.	45.	54.1	50.8	99.	105.	1.52	.620	MIX .737-05
N2H4	440	UNLINE-DUUBLET	.040 .040	4.	60.	45.	49.5	53.4	94.	112.	1.32	.810	MIX .873-05
N2H4	441	UNLINE-DUUBLET	.040 .040	4.	60.	45.	57.4	59.7	97.	102.	1.37	.760	MIX .805-05
N2H4	442	UNLINE-DUUBLET	.040 .040	4.	60.	45.	48.8	53.4	91.	97.	1.30	.640	MIX .610-05
N2H4	443	UNLINE-DUUBLET	.040 .040	4.	60.	45.	56.5	53.4	107.	116.	1.58	.580	MIX .610-05
N2H4	444	UNLINE-DUUBLET	.040 .040	4.	60.	45.	54.1	53.4	114.	125.	1.45	.680	SEP .102-04
N2H4	445	UNLINE-DUUBLET	.040 .040	4.	60.	45.	60.0	53.4	108.	108.	1.40	.550	SEP .127-04
N2H4	446	UNLINE-DUUBLET	.040 .040	4.	60.	45.	50.0	57.4	114.	122.	1.29	.930	SEP .124-04
N2H4	447	UNLINE-DUUBLET	.040 .040	4.	60.	45.	49.3	57.4	116.	126.	1.24	.960	SEP .146-04
N2H4	448	UNLINE-DUUBLET	.040 .040	4.	60.	45.	50.6	66.6	108.	122.	1.08	1.220	MIX .140-04
N2H4	449	UNLINE-DUUBLET	.040 .040	4.	60.	55.	54.1	66.6	104.	114.	1.16	1.060	MIX .119-04
N2H4	450	UNLINE-DUUBLET	.040 .040	4.	60.	55.	57.2	66.7	104.	116.	1.16	1.060	MIX .125-04
N2H4	451	UNLINE-DUUBLET	.040 .040	4.	60.	45.	54.2	66.6	104.	118.	1.16	1.060	MIX .126-04
N2H4	452	UNLINE-DUUBLET	.040 .040	4.	60.	45.	45.7	68.2	101.	106.	1.13	2.570	MIX .103-04
N2H4	453	UNLINE-DUUBLET	.040 .040	4.	60.	45.	44.4	65.1	97.	108.	.97	1.510	MIX .980-05
N2H4	454	UNLINE-DUUBLET	.040 .040	4.	60.	45.	44.9	65.1	91.	105.	.98	1.470	MIX .860-05

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HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TEST TYPE NO.	MODE	PC (PSIA)	VAVG	MEF	WED	REF	KEO	DELTA (DEG F)	REL	PPF (PSIA)	MRVP	XF	KU	XP	RESID
N2M4 411	PUP	175.	45.	7.0	7.3	124+05	615+05	0.	13+06	25+06	23.7	121.3	.00	.14	.21-01
N2M4 412	PUP	190.	48.	7.6	8.7	330+05	643+05	0.	13+06	26+06	22.6	110.4	.00	.12	.19-01
N2M4 413	PUP	90.	47.	3.6	5.0	308+05	697+05	0.	12+06	28+06	5	20.5	127.5	.01	.23
N2M4 414	PUP	80.	46.	3.2	4.5	343+05	729+05	0.	14+06	29+06	8	26.8	101.4	.01	.35
N2M4 415	PUP	85.	49.	3.4	6.2	348+05	846+05	0.	14+06	34+06	9	29.4	101.2	.01	.35
N2M4 416	PUP	80.	48.	3.2	5.7	352+05	842+05	0.	15+06	33+06	1.0	37.8	104.8	.01	.47
N2M4 417	PUP	80.	48.	3.4	5.4	371+05	834+05	0.	15+06	31+06	1.0	32.3	90.6	.01	.40
N2M4 418	PUP	80.	47.	3.4	4.7	371+05	764+05	0.	15+06	29+06	1.0	32.3	90.6	.01	.40
N2M4 419	PUP	80.	46.	3.3	4.3	366+05	752+05	0.	14+06	42+06	1.0	37.8	115.1	.01	.36
N2M4 420	PUP	95.	52.	3.9	9.3	362+05	995+05	0.	14+06	40+06	1.0	33.1	96.9	.01	.35
N2M4 421	PUP	95.	45.	3.0	4.4	299+05	618+05	0.	12+06	25+06	5	16.9	106.3	.00	.18
N2M4 422	PUP	95.	45.	3.5	3.9	282+05	567+05	0.	12+06	23+06	3	13.1	112.5	.00	.14
N2M4 423	PUP	95.	47.	3.7	3.9	289+05	594+05	0.	12+06	24+06	4	18.6	152.7	.00	.20
N2M4 424	PUP	95.	47.	3.9	5.8	353+05	789+05	0.	14+06	32+06	9	33.9	115.3	.01	.36
N2M4 425	PUP	95.	47.	3.9	4.9	365+05	728+05	0.	15+06	29+06	1.0	35.4	103.4	.01	.37
N2M4 426	PUP	175.	53.	7.3	13.0	266+05	713+05	0.	11+06	29+06	3	14.2	146.0	.00	.08
N2M4 427	PUP	175.	52.	7.2	12.0	311+05	757+05	0.	12+06	29+06	5	19.5	107.2	.00	.11
N2M4 428	PUP	215.	60.	14.5	30.8	421+05	111+06	0.	17+06	44+06	7	25.8	111.7	.00	.12
N2M4 429	PUP	215.	55.	14.4	30.7	430+05	113+06	0.	17+06	45+06	8	29.1	106.1	.00	.14
N2M4 430	PUP	175.	57.	9.2	16.0	394+05	916+05	0.	16+06	37+06	1.0	31.5	92.6	.01	.18
N2M4 431	PUP	75.	55.	3.4	6.4	353+05	871+05	0.	14+06	35+06	8	28.1	102.8	.01	.37
N2M4 432	PUP	95.	52.	4.1	5.9	300+05	674+05	0.	12+06	27+06	4	15.8	109.3	.00	.17
N2M4 433	PUP	95.	52.	4.1	5.7	279+05	662+05	0.	11+06	26+06	3	14.2	137.5	.00	.15
N2M4 434	PUP	95.	53.	1.0	3.3	215+05	698+05	0.	8+05	25+06	8	30.0	112.6	.02	.67
N2M4 435	PUP	95.	51.	1.2	2.7	235+05	617+05	0.	9+05	25+06	1.0	26.8	83.7	.02	.60
N2M4 436	PUP	95.	52.	1.4	3.7	248+05	733+05	0.	9+05	29+06	7	29.1	120.4	.02	.65
N2M4 437	PUP	95.	51.	1.1	2.6	214+05	596+05	0.	8+05	24+06	6	24.9	120.3	.01	.55
N2M4 438	PUP	95.	55.	1.2	3.8	240+05	754+05	0.	9+05	30+06	1.0	36.2	100.4	.02	.41
N2M4 439	PUP	95.	52.	1.2	3.6	253+05	753+05	0.	10+06	30+06	1.3	42.5	95.0	.03	.94
N2M4 440	PUP	95.	52.	1.2	4.3	267+05	810+05	0.	12+06	32+06	9	37.0	125.3	.02	.82
N2M4 441	PUP	95.	51.	1.4	3.1	267+05	696+05	0.	11+06	28+06	1.2	42.5	101.1	.03	.94
N2M4 442	PUP	95.	53.	1.1	3.1	277+05	693+05	0.	11+06	28+06	1.5	44.4	99.9	.03	.99
N2M4 443	PUP	95.	51.	1.4	3.1	310+05	693+05	0.	12+06	27+06	1.2	37.0	99.1	.03	.82
N2M4 444	PUP	95.	52.	2.2	4.2	266+05	717+05	0.	12+06	29+06	1.0	33.9	90.1	.02	.62
N2M4 445	PUP	95.	52.	2.3	4.7	306+05	765+05	0.	12+06	31+06	1.0	35.4	98.8	.02	.64
N2M4 446	PUP	95.	52.	2.3	4.2	313+05	714+05	0.	12+06	29+06	1.1	33.9	90.7	.02	.62
N2M4 447	PUP	95.	50.	1.9	4.4	294+05	597+05	0.	12+06	24+06	8	31.5	114.2	.02	.70
N2M4 448	PUP	95.	55.	1.7	2.2	280+05	567+05	0.	11+06	23+06	9	29.1	100.2	.02	.65
N2M4 449	PUP	95.	55.	1.9	2.4	275+05	544+05	0.	11+06	22+06	8	24.9	94.6	.02	.50
N2M4 450	PUP	95.	55.	1.9	2.4	275+05	544+05	0.	11+06	22+06	8	24.9	94.6	.02	.50

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HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

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FUEL TEST TYPE NO.	INJECTOR TYPE	DO OF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MD	MODE	REACT (SEC)
N2M4 454	UNLINE-DUUMLET	.040 .040	4.	60.	65.	49.8	40.9	114.	132.	1.00	1.430	MIX	.108-04
N2M4 455	UNLINE-DUUMLET	.040 .040	4.	60.	50.	67.7	68.2	122.	134.	1.41	.712	SEP	.201-04
N2M4 456	UNLINE-DUUMLET	.040 .040	4.	60.	55.	58.5	66.6	129.	137.	1.25	.913	SEP	.221-04
N2M4 457	UNLINE-DUUMLET	.040 .040	4.	60.	55.	60.5	66.3	135.	144.	1.29	.852	SEP	.262-04
N2M4 458	UNLINE-DUUMLET	.040 .040	4.	60.	55.	60.0	64.1	131.	144.	1.33	.801	SEP	.242-04
N2M4 459	UNLINE-DUUMLET	.040 .040	4.	60.	55.	58.5	62.3	137.	147.	1.34	.797	SEP	.264-04
N2M4 460	UNLINE-DUUMLET	.040 .040	4.	60.	470.	37.9	66.3	61.	64.	.81	2.150	SEP	.642-05
N2M4 461	UNLINE-DUUMLET	.040 .040	4.	60.	535.	65.1	66.9	63.	66.	1.34	.789	SEP	.663-05
N2M4 462	UNLINE-DUUMLET	.040 .040	4.	60.	535.	48.4	68.6	63.	66.	1.00	1.420	SEP	.703-05
N2M4 463	UNLINE-DUUMLET	.040 .040	4.	60.	535.	48.4	68.6	67.	64.	.96	1.540	SEP	.712-05
N2M4 464	UNLINE-DUUMLET	.040 .040	4.	60.	585.	56.5	68.8	66.	69.	1.17	1.040	SEP	.782-05
N2M4 465	UNLINE-DUUMLET	.040 .040	4.	0.	475.	61.3	68.8	67.	64.	1.26	.890	SEP	.712-05
N2M4 466	UNLINE-DUUMLET	.040 .040	4.	60.	435.	54.9	65.7	48.	52.	1.19	1.000	SEP	.411-05
N2M4 467	UNLINE-DUUMLET	.040 .040	4.	60.	455.	55.9	65.7	48.	52.	1.21	.970	SEP	.419-05
N2M4 468	UNLINE-DUUMLET	.040 .040	4.	60.	455.	61.9	65.7	108.	110.	1.35	.790	SEP	.260-04
N2M4 469	UNLINE-DUUMLET	.040 .040	4.	60.	455.	60.4	68.8	114.	116.	1.25	.910	SEP	.319-04
N2M4 470	UNLINE-DUUMLET	.040 .040	4.	60.	445.	58.2	68.8	111.	119.	1.21	.980	SEP	.320-04
N2M4 471	UNLINE-DUUMLET	.040 .040	4.	60.	100.	48.6	52.6	96.	92.	1.26	.900	PUP	.133-04
N2M4 472	UNLINE-DUUMLET	.040 .040	4.	60.	100.	42.6	52.6	64.	72.	1.15	1.070	PUP	.613-05
N2M4 473	UNLINE-DUUMLET	.040 .040	4.	60.	100.	48.6	52.6	63.	70.	1.26	.900	PUP	.582-05
N2M4 474	UNLINE-DUUMLET	.040 .040	4.	60.	100.	42.6	52.6	59.	62.	1.16	1.060	PUP	.476-05
N2M4 475	UNLINE-DUUMLET	.040 .040	4.	60.	630.	59.5	77.9	68.	69.	1.09	1.200	SEP	.362-05
N2M4 476	UNLINE-DUUMLET	.040 .040	4.	60.	80.	71.5	81.7	68.	64.	1.28	.870	MIX	.380-05
N2M4 477	UNLINE-DUUMLET	.040 .040	4.	60.	85.	61.3	83.2	61.	66.	1.05	1.290	MIX	.371-05
N2M4 478	UNLINE-DUUMLET	.040 .040	4.	60.	640.	65.4	70.0	59.	66.	1.33	.807	MIX	.306-05
N2M4 479	UNLINE-DUUMLET	.040 .040	4.	60.	100.	48.5	52.8	68.	71.	1.25	.905	PUP	.635-05
N2M4 480	UNLINE-DUUMLET	.040 .040	4.	60.	100.	48.4	53.7	88.	64.	1.27	.860	PUP	.719-05
N2M4 481	UNLINE-DUUMLET	.040 .040	4.	60.	25.	26.5	35.1	128.	132.	1.06	1.240	PUP	.242-04
N2M4 482	UNLINE-DUUMLET	.040 .040	4.	60.	147.	25.4	34.9	135.	141.	1.03	1.330	PUP	.244-04
N2M4 483	UNLINE-DUUMLET	.040 .040	4.	60.	175.	43.1	65.1	51.	60.	.95	1.600	PUP	.504-05
N2M4 484	UNLINE-DUUMLET	.040 .040	4.	60.	185.	54.1	64.1	51.	60.	1.20	.986	PUP	.396-05
N2M4 485	UNLINE-DUUMLET	.040 .040	4.	60.	220.	54.1	62.6	57.	60.	1.23	.938	UNDEF	.236-05
N2M4 486	UNLINE-DUUMLET	.040 .040	4.	60.	90.	54.1	64.1	54.	58.	2.51	.471	MIX	.223-05
N2M4 487	UNLINE-DUUMLET	.040 .040	4.	60.	100.	54.1	63.1	56.	56.	1.22	.952	MIX	.225-05
N2M4 488	UNLINE-DUUMLET	.040 .040	4.	60.	100.	54.1	63.8	56.	58.	1.21	.970	MIX	.227-05
N2M4 489	UNLINE-DUUMLET	.040 .040	4.	60.	100.	54.1	63.6	56.	58.	1.21	.970	MIX	.227-05
N2M4 490	UNLINE-DUUMLET	.040 .040	4.	60.	150.	39.8	43.4	63.	63.	1.31	.830	MIX	.183-05
N2M4 491	UNLINE-DUUMLET	.040 .040	4.	60.	30.	36.0	43.4	51.	56.	1.25	.610	MIX	.139-05
N2M4 492	UNLINE-DUUMLET	.040 .040	4.	60.	30.	36.6	41.4	64.	60.	1.37	.750	MIX	.169-05
N2M4 493	UNLINE-DUUMLET	.040 .040	4.	60.	30.	33.2	40.4	56.	60.	1.16	1.060	MIX	.149-05

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FUEL TYPE	TEST NO.	MODE	PC (PSIA)	VAVG	MEF	MEQ	REF	RED	DELTI (DEG F)	RELF	RELD (PSIA)	PP1 (PSIA)	PPU	MRVP	AF	XD	XP	WESID
N2M4	454	MIX	65.	45.	1.0	4.4	203+05	693+05	0.	81+05	28+06	1.6	42.5	76.7	.02	.65	.13+00	81-04
N2M4	455	SEP	50.	60.	2.2	6.6	343+05	982+05	0.	14+06	39+06	1.7	50.6	88.6	.03	1.01	.16+00	49-04
N2M4	456	SEP	55.	62.	2.4	5.7	341+05	892+05	0.	14+06	35+06	1.8	50.8	95.9	.03	1.07	.19+00	50-04
N2M4	457	SEP	55.	63.	2.4	6.4	353+05	944+05	0.	14+06	38+06	2.2	67.1	90.6	.04	1.22	.22+00	50-04
N2M4	458	SEP	55.	62.	2.2	6.1	341+05	915+05	0.	14+06	37+06	2.2	61.4	83.5	.04	1.12	.21+00	52-04
N2M4	459	SEP	55.	60.	2.1	6.0	337+05	924+05	0.	13+06	37+06	2.4	69.9	86.7	.04	1.27	.23+00	54-04
N2M4	460	SEP	470.	54.	25.7	21.1	303+05	584+05	0.	12+06	23+06	.2	12.1	175.3	.00	.03	.33+02	75-04
N2M4	461	SEP	535.	66.	29.9	71.1	310+05	101+06	0.	12+06	40+06	.2	12.4	167.6	.00	.02	.31+02	75-04
N2M4	462	SEP	535.	59.	31.6	39.3	319+05	750+05	0.	13+06	30+06	.2	12.4	167.6	.00	.02	.31+02	73-04
N2M4	463	SEP	535.	58.	31.5	37.2	314+05	740+05	0.	13+06	30+06	.2	13.6	197.9	.00	.03	.31+02	73-04
N2M4	464	SEP	545.	62.	34.8	59.6	328+05	892+05	0.	13+06	36+06	.2	13.5	162.4	.00	.02	.31+02	73-04
N2M4	465	SEP	475.	65.	24.0	57.2	314+05	973+05	0.	13+06	39+06	.2	13.8	197.9	.00	.03	.35+02	73-04
N2M4	466	SEP	435.	60.	22.8	36.2	277+05	801+05	0.	11+06	32+06	.1	7.9	170.7	.00	.02	.20+02	76-04
N2M4	467	SEP	455.	60.	23.9	41.7	277+05	815+05	0.	11+06	33+06	.1	8.2	176.4	.00	.02	.23+02	76-04
N2M4	468	SEP	475.	64.	26.9	69.6	428+05	125+06	0.	17+06	50+06	.9	37.0	116.7	.00	.08	.13+01	76-04
N2M4	469	SEP	455.	64.	29.9	68.7	464+05	126+06	0.	19+06	50+06	1.0	42.5	116.7	.00	.09	.15+01	73-04
N2M4	470	SFP	445.	63.	29.4	61.2	472+05	127+06	0.	19+06	48+06	1.1	39.6	102.6	.00	.09	.15+01	73-04
N2M4	471	P1P	100.	49.	3.7	8.1	303+05	884+05	0.	12+06	35+06	.5	28.1	154.8	.01	.28	.38+01	95-04
N2M4	472	P1P	100.	47.	3.5	5.7	258+05	667+05	0.	10+06	27+06	.3	13.1	146.6	.00	.13	.18+01	95-04
N2M4	473	P1P	100.	49.	3.5	6.8	252+05	722+05	0.	10+06	29+06	.2	12.6	149.9	.00	.13	.18+01	95-04
N2M4	474	P0P	100.	47.	3.4	5.7	237+05	653+05	0.	95+05	26+06	.2	11.5	179.8	.00	.12	.15+01	95-04
N2M4	475	SEP	630.	64.	31.7	47.9	237+05	633+05	0.	95+05	25+06	.2	14.2	202.0	.00	.02	.27+02	43-04
N2M4	476	P0P	60.	77.	4.4	9.3	249+05	782+05	0.	10+06	31+06	.2	14.2	202.0	.00	.18	.21+01	41-04
N2M4	477	P0P	35.	72.	4.9	6.6	257+05	628+05	0.	10+06	25+06	.2	12.0	161.7	.00	.14	.19+01	40-04
N2M4	478	P1P	640.	67.	26.1	56.4	217+05	665+05	0.	87+05	27+06	.2	11.5	155.9	.00	.02	.25+02	46-04
N2M4	479	P1P	100.	49.	3.5	7.0	258+05	742+05	0.	10+06	30+06	.3	14.2	159.9	.00	.14	.19+01	95-04
N2M4	480	P0P	100.	51.	3.6	6.4	243+05	870+05	0.	98+05	35+06	.2	23.2	318.4	.00	.23	.22+01	93-04
N2M4	481	P1P	23.	31.	4.4	7.7	261+05	598+05	0.	10+06	24+06	1.6	57.6	104.3	.07	2.51	.42+00	14+03
N2M4	482	P1P	147.	30.	2.6	4.5	272+05	593+05	0.	11+06	24+06	2.0	67.1	99.1	.11	.46	.78+01	14+03
N2M4	483	P1P	175.	54.	9.2	9.7	290+05	632+05	0.	12+06	25+06	.2	9.0	152.0	.00	.05	.72+02	77-04
N2M4	484	P1P	145.	59.	9.4	10.1	260+05	793+05	0.	11+06	32+06	.2	9.0	152.0	.00	.05	.68+02	78-04
N2M4	485	P1P	220.	54.	7.1	13.1	186+05	543+05	0.	74+05	22+06	.2	10.9	181.3	.00	.05	.63+02	53+04
N2M4	486	P1P	40.	57.	3.0	5.4	189+05	530+05	0.	75+05	21+06	.2	10.0	177.2	.00	.11	.14+01	52+04
N2M4	487	P1P	100.	54.	3.3	5.9	185+05	541+05	0.	74+05	22+06	.2	10.4	184.9	.00	.10	.13+01	53+04
N2M4	488	P1P	100.	58.	3.3	5.9	187+05	541+05	0.	75+05	22+06	.2	10.4	184.9	.00	.10	.13+01	52+04
N2M4	489	P1P	150.	41.	2.3	5.0	131+05	411+05	0.	75+05	16+06	.2	12.4	189.6	.00	.08	.10+01	77+04
N2M4	490	P1P	40.	40.	.5	9.125+05	371+05	0.	50+05	15+06	.2	9.0	170.9	.01	.30	.39+01	77+04	
N2M4	491	P1P	30.	40.	.4	9.122+05	403+05	0.	49+05	16+06	.2	13.1	219.1	.01	.44	.50+01	81+04	
N2M4	492	P1P	50.	37.	.4	7.121+05	332+05	0.	48+05	13+06	.2	10.4	177.7	.01	.35	.44+01	82+04	

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FUEL TEST TYPE NO.	INJECTOR TYPE	DN (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VD (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR MF/MD	MODE	REACT (SEC)
N2M4 494	UNLIKE-DOUBLET	.040	.040	4.	60.	30.	33.9	40.3	56.	62.	1.16	1.040	MIX .156-05
N2M4 495	UNLIKE-DOUBLET	.040	.040	4.	60.	35.	36.3	43.4	56.	64.	1.20	.900	MIX .175-05
N2M4 496	UNLIKE-DOUBLET	.040	.040	4.	60.	35.	36.6	43.4	56.	64.	1.21	.980	MIX .180-05
N2M4 497	UNLIKE-DOUBLET	.040	.040	4.	60.	30.	39.8	44.8	56.	66.	1.27	.890	MIX .186-05
N2M4 498	UNLIKE-DOUBLET	.040	.040	4.	60.	30.	39.8	44.8	56.	66.	1.27	.890	MIX .192-05
N2M4 499	UNLIKE-DOUBLET	.040	.040	4.	60.	30.	39.6	43.4	56.	64.	1.31	.830	MIX .175-05
N2M4 500	UNLIKE-DOUBLET	.040	.040	4.	60.	30.	39.6	43.4	56.	66.	1.31	.830	MIX .180-05
N2M4 501	UNLIKE-DOUBLET	.040	.040	4.	60.	30.	39.6	38.3	63.	74.	1.13	1.090	MIX .200-05
N2M4 502	UNLIKE-DOUBLET	.040	.040	4.	60.	30.	33.2	37.3	64.	69.	1.27	.860	MIX .183-05
N2M4 503	UNLIKE-DOUBLET	.040	.040	4.	60.	30.	33.2	38.3	66.	74.	1.21	.930	MIX .208-05
N2M4 504	UNLIKE-DOUBLET	.040	.040	4.	60.	30.	33.7	31.2	64.	74.	1.54	.598	MIX .167-05
N2M4 505	UNLIKE-DOUBLET	.040	.040	4.	60.	15.	20.7	17.2	72.	92.	1.71	.488	SEP .143-05
N2M4 506	UNLIKE-DOUBLET	.040	.040	4.	60.	15.	21.4	21.7	72.	72.	1.40	.721	SEP .123-05
N2M4 507	UNLIKE-DOUBLET	.027	.027	4.	60.	15.	41.4	27.4	71.	71.	2.20	.296	SEP .686-06
N2M4 508	UNLIKE-DOUBLET	.027	.027	4.	60.	15.	40.3	39.2	68.	70.	1.48	.650	MIX .931-06
N2M4 509	UNLIKE-DOUBLET	.027	.027	4.	60.	15.	41.1	53.2	68.	68.	1.27	.890	MIX .122-05

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HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATOR ZUNG

FUEL TEST TYPE NO.	MODE	PC (PSIA)	WAVG	MEF	WEU	REF	REG	DELTI (DEG F)	RELF	RELO (PSIA)	PPF (PSIA)	MRVP	XF	XU	XP	IESID	
N2M4 498	MIX	30.	37.	.4	.7	.123+05	.339+05	0.	.49+05	.14+06	.2	10.4	163.9	.01	.35	.46-01	.82-04
N2M4 495	MIX	35.	41.	.5	1.0	.133+05	.383+05	0.	.53+05	.15+06	.2	10.4	149.5	.01	.30	.42-01	.77-04
N2M4 496	MIX	35.	40.	.5	1.0	.134+05	.386+05	0.	.54+05	.15+06	.2	10.4	142.1	.01	.30	.43-01	.77-04
N2M4 497	MIX	30.	42.	.5	1.0	.139+05	.398+05	0.	.55+05	.16+06	.2	10.4	142.1	.01	.35	.50-01	.74-04
N2M4 498	MIX	30.	42.	.5	1.0	.139+05	.402+05	0.	.55+05	.16+06	.2	11.0	150.0	.01	.37	.51-01	.74-04
N2M4 499	MIX	30.	41.	.5	1.0	.133+05	.396+05	0.	.53+05	.16+06	.2	10.4	149.5	.01	.35	.49-01	.77-04
N2M4 500	MIX	30.	34.	.5	1.0	.134+05	.396+05	0.	.54+05	.16+06	.2	10.4	149.5	.01	.35	.50-01	.77-04
N2M4 501	MIX	30.	31.	.4	.6	.126+05	.316+05	0.	.50+05	.13+06	.3	12.1	.0	.01	.41	.62-01	.87-04
N2M4 502	MIX	30.	35.	.3	.7	.118+05	.347+05	0.	.47+05	.14+06	.2	13.1	.6	.01	.44	.59-01	.89-04
N2M4 503	MIX	30.	35.	.4	.7	.124+05	.350+05	0.	.50+05	.14+06	.3	13.5	143.7	.01	.45	.64-01	.87-04
N2M4 504	MIX	30.	32.	.2	.7	.103+05	.352+05	0.	.41+05	.14+06	.3	13.1	139.1	.01	.44	.63-01	.11-03
N2M4 505	SEP	15.	19.	.0	.1	.659+04	.225+05	0.	.26+05	.90+05	.3	15.6	90.0	.04	1.02	.19+00	.19-03
N2M4 506	SEP	15.	2.	.1	.1	.704+04	.233+05	0.	.28+05	.93+05	.3	15.6	172.7	.02	1.02	.14+00	.15-03
N2M4 507	SEP	15.	37.	.1	.4	.595+04	.302+05	0.	.24+05	.12+06	.3	15.2	173.1	.02	1.03	.13+00	.64-04
N2M4 508	MIX	15.	40.	.1	.3	.844+04	.290+05	0.	.34+05	.12+06	.2	14.2	166.8	.02	.51	.13+00	.57-04
N2M4 509	MIX	15.	50.	.2	.5	.113+05	.538+05	0.	.45+05	.14+06	.2	14.2	177.0	.02	.51	.12+00	.42-04

ZETA PLUT FILE HAS BEEN CREATED...

72% CHARACTERS

3F1

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OF POOR QUALITY

HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR OMS-75

FUEL TEST TYPE NO.	INJECTOR TYPE	DO (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VU (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR MF/HO	MODE	REACT (SEC)
MMH 101	VDT - 360	.027 .025		5.	0.	120.	85.0	96.5	83.	83.	1.69 .783	SEP	.106-04
MMH 102	VDT - 360	.027 .025		5.	0.	105.	66.0	111.1	83.	83.	1.68 1.016	SEP	.122-04
MMH 103	VDT - 360	.027 .025		5.	0.	115.	81.9	94.0	72.	77.	1.69 .794	SEP	.822-05
MMH 104	VDT - 360	.027 .025		5.	0.	116.	81.7	94.5	76.	76.	1.67 .811	SEP	.855-05
MMH 105	VDT - 360	.027 .025		5.	0.	133.	90.0	100.0	78.	76.	1.60 .875	SEP	.100-04
MMH 106	VDT - 360	.027 .025		5.	0.	128.	82.5	98.5	30.	80.	1.61 .865	M/S	.995-05
MMH 107	VDT - 360	.027 .025		5.	0.	127.	84.9	100.7	78.	80.	1.63 .852	M/S	.994-05
MMH 108	VDT - 360	.027 .025		5.	0.	103.	66.5	79.1	81.	82.	1.62 .858	MIX	.834-05
MMH 109	VDT - 360	.027 .025		5.	0.	81.	56.5	64.0	84.	84.	1.69 .779	MIX	.722-05
MMH 110	VDT - 360	.027 .025		5.	0.	158.	98.0	115.3	77.	80.	1.64 .837	SEP	.113-04
MMH 111	VDT - 360	.027 .025		5.	0.	172.	80.6	96.6	81.	160.	1.68 .831	SEP	.304-04
MMH 112	VDT - 360	.027 .025		5.	0.	131.	60.3	101.6	84.	251.	1.70 .848	SEP	.941-04
MMH 113	VDT - 360	.027 .025		5.	0.	132.	81.8	93.7	90.	111.	1.42 .789	SEP	.171-04
MMH 114	VDT - 360	.027 .025		5.	0.	131.	80.8	94.0	76.	77.	1.66 .819	SEP	.865-05
MMH 115	XDT-A	.024 .020		1.	0.	116.	60.6	89.7	80.	81.	1.60 1.329	M/S	.588-05
MMH 116	XDT-A	.024 .020		1.	0.	133.	56.7	84.8	84.	84.	1.63 1.269	MIX	.612-05
MMH 117	XDT-A	.024 .020		1.	0.	103.	46.1	68.7	84.	86.	1.58 1.347	MIX	.509-05
MMH 118	XDT-A	.024 .020		1.	0.	82.	38.7	56.1	86.	89.	1.63 1.276	MIX	.443-05
MMH 119	XDT-A	.024 .020		1.	0.	159.	68.2	98.3	82.	84.	1.65 1.261	SEP	.692-05
MMH 120	XDT-A	.024 .020		1.	0.	131.	58.0	88.0	85.	162.	1.63 1.335	SEP	.194-04
MMH 121	XDT-A	.024 .020		1.	0.	130.	59.0	91.0	90.	224.	1.67 1.328	SEP	.442-04
MMH 122	XDT-A	.024 .020		1.	0.	130.	59.8	91.4	120.	241.	1.67 1.317	SEP	.729-04
MMH 123	XDT-A	.024 .020		1.	0.	157.	72.8	113.3	136.	292.	1.68 1.328	SEP	.179-03
MMH 124	LUL CORE LU DP	.027 .025		5.	1.	127.	70.1	91.5	66.	71.	1.48 1.027	UNDEF	.670-05
MMH 125	LUL CORE LU DP	.027 .025		5.	1.	130.	72.0	83.7	70.	77.	1.67 .815	UNDEF	.712-05
MMH 126	LUL CORE LU DP	.027 .025		5.	1.	101.	56.6	68.6	80.	89.	1.59 .886	UNDEF	.783-05
MMH 127	LUL CORE LU DP	.027 .025		5.	1.	82.	46.2	56.3	81.	80.	1.57 .901	UNDEF	.576-05
MMH 128	LUL CORE LU DP	.027 .025		5.	1.	151.	63.6	99.1	86.	81.	1.62 .853	UNDEF	.110-04
MMH 129	LUL CORE LU DP	.027 .025		5.	1.	152.	86.1	98.7	86.	84.	1.67 .801	UNDEF	.114-04
MMH 130	LUL CORE LU DP	.027 .025		5.	1.	130.	73.7	88.0	83.	179.	1.70 .816	UNDEF	.369-04
MMH 131	LUL CORE LU DP	.027 .025		5.	1.	129.	76.2	92.3	87.	234.	1.75 .809	UNDEF	.740-04
MMH 132	LUL CORE LU DP	.027 .025		5.	1.	130.	79.8	91.2	127.	227.	1.78 .755	UNDEF	.109-03
MMH 133	LUL CORE LU DP	.027 .025		5.	1.	150.	93.0	117.2	136.	291.	1.69 .871	UNDEF	.281-03
MMH 134	L-O-L BARRIER	.025 .025		5.	1.	129.	127.0	146.3	70.	76.	1.44 .800	UNDEF	.127-04
MMH 135	L-O-L BARRIER	.025 .025		5.	1.	103.	100.6	115.6	74.	77.	1.44 .798	UNDEF	.104-04
MMH 136	L-O-L BARRIER	.025 .025		5.	1.	42.	81.6	92.4	77.	78.	1.46 .778	UNDEF	.875-03
MMH 137	L-O-L BARRIER	.025 .025		5.	1.	151.	154.5	172.2	76.	70.	1.48 .753	UNDEF	.156-04
MMH 138	L-O-L BARRIER	.025 .025		5.	1.	124.	130.9	156.5	51.	207.	1.49 .802	UNDEF	.895-04
MMH 139	L-O-L BARRIER	.025 .025		5.	1.	124.	131.3	167.5	82.	240.	1.43 .889	UNDEF	.1-03
MMH 140	L-O-L BARRIER	.025 .025		5.	1.	125.	139.9	165.6	120.	248.	1.50 .786	UNDEF	.1-03

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR UMS-75

FUEL TEST TYPE NO.	MODE	PC (PSIA)	WAVG	MEF	WED	REF	REU	DELTA (DEG F)	RELF	RELU (PSIA)	PPF (PSIA)	MRVP	XF	XO	XP	RESID
MMH 101	SEP	120.	69.	11.8	13.6	.222+05	.666+05	0.	.11+06	.33+06	1.2	20.5	34.0	.01	.17	.42-01
MMH 102	SEP	105.	95.	13.7	12.1	.255+05	.674+05	0.	.13+06	.34+06	1.2	20.5	34.0	.01	.20	.47-01
MMH 103	SEP	115.	86.	10.6	11.4	.205+05	.602+05	0.	.10+06	.30+06	1.0	15.6	32.6	.01	.14	.35-01
MMH 104	SEP	116.	86.	10.8	11.7	.204+05	.614+05	0.	.10+06	.31+06	1.0	17.3	34.6	.01	.15	.36-01
MMH 105	SEP	133.	97.	16.2	16.4	.233+05	.684+05	0.	.12+06	.34+06	1.0	18.1	36.1	.01	.15	.36-01
MMH 106	M/S	129.	89.	13.1	13.4	.250+05	.635+05	0.	.11+06	.32+06	1.1	19.0	33.9	.01	.15	.36-01
MMH 107	M/S	127.	91.	13.5	13.9	.225+05	.646+05	0.	.11+06	.32+06	1.1	18.1	32.5	.01	.14	.35-01
MMH 108	M/S	103.	71.	6.8	7.1	.180+05	.515+05	0.	.09+05	.26+06	1.2	19.5	33.1	.01	.19	.47-01
MMH 109	M/S	81.	59.	3.5	4.1	.148+05	.406+05	0.	.07+05	.22+06	1.2	21.1	34.0	.02	.26	.63-01
MMH 110	SEP	158.	105.	22.1	23.0	.258+05	.741+05	0.	.13+06	.37+06	1.1	17.7	31.6	.01	.11	.28-01
MMH 111	SEP	132.	87.	14.9	13.3	.309+05	.624+05	0.	.19+06	.31+06	8.2	19.5	5.4	.06	.15	.96-01
MMH 112	SEP	131.	84.	19.8	13.3	.722+05	.633+05	0.	.36+06	.32+06	43.9	21.1	1.3	.33	.16	.23+00
MMH 113	SEP	132.	87.	12.9	14.3	.209+05	.671+05	0.	.13+06	.34+06	2.6	24.2	19.5	.02	.18	.60-01
MMH 114	SEP	131.	86.	12.1	12.9	.205+05	.608+05	0.	.10+06	.30+06	1.0	17.3	33.7	.01	.13	.32-01
MMH 115	M/S	116.	72.	7.9	5.6	.162+05	.415+05	0.	.16+05	.41+05	1.1	19.0	33.1	.01	.16	.40-01
MMH 116	M/S	133.	95.	8.1	6.4	.157+05	.411+05	0.	.16+05	.41+05	1.2	21.1	32.5	.01	.16	.38-01
MMH 117	M/S	103.	55.	4.1	3.1	.130+05	.324+05	0.	.13+05	.32+05	1.3	21.1	32.5	.01	.20	.51-01
MMH 118	M/S	82.	45.	2.2	1.7	.108+05	.275+05	0.	.11+05	.27+05	1.4	22.1	32.0	.02	.27	.68-01
MMH 119	SEP	159.	80.	13.0	10.2	.182+05	.472+05	0.	.18+05	.47+05	1.2	20.0	32.4	.01	.13	.31-01
MMH 120	SEP	131.	69.	9.9	6.2	.288+05	.409+05	0.	.29+05	.41+05	8.5	21.6	5.6	.07	.16	.10+00
MMH 121	SEP	130.	71.	11.9	6.5	.439+05	.430+05	0.	.44+05	.43+05	28.8	24.2	2.1	.22	.19	.20+00
MMH 122	SEP	130.	72.	12.4	8.0	.489+05	.515+05	0.	.49+05	.52+05	36.3	48.2	3.2	.28	.37	.32+00
MMH 123	SEP	157.	84.	26.0	16.2	.626+05	.694+05	0.	.83+05	.69+05	76.0	71.3	2.4	.48	.45	.47+00
MMH 124	UNDEF	127.	79.	11.0	9.0	.149+05	.498+05	0.	.94+05	.25+06	.8	13.5	52.2	.01	.11	.27-01
MMH 125	UNDEF	150.	76.	9.5	9.9	.163+05	.523+05	0.	.91+05	.26+06	1.0	14.8	29.1	.01	.11	.30-01
MMH 126	UNDEF	101.	61.	5.1	5.0	.166+05	.436+05	0.	.83+05	.22+06	1.4	19.0	27.8	.01	.19	.51-01
MMH 127	UNDEF	82.	50.	2.7	2.7	.126+05	.358+05	0.	.63+05	.18+06	1.1	19.5	54.8	.01	.24	.57-01
MMH 128	UNDEF	151.	90.	15.6	16.4	.224+05	.609+05	0.	.11+06	.33+06	1.1	22.1	38.1	.01	.15	.33-01
MMH 129	UNDEF	152.	91.	15.7	17.9	.224+05	.688+05	0.	.11+06	.34+06	1.2	22.1	35.5	.01	.15	.34-01
MMH 130	UNDEF	130.	79.	12.6	11.0	.401+05	.574+05	0.	.20+06	.29+06	12.5	20.5	3.9	.10	.16	.12+00
MMH 131	UNDEF	129.	82.	15.5	12.0	.592+05	.613+05	0.	.30+06	.31+06	33.0	22.0	1.7	.26	.18	.21+00
MMH 132	UNDEF	150.	84.	15.0	16.7	.560+05	.801+05	0.	.28+06	.40+06	30.0	56.5	4.4	.23	.43	.52+00
MMH 133	UNDEF	129.	102.	33.2	27.3	.106+06	.905+05	0.	.53+06	.49+06	74.7	68.5	2.3	.50	.46	.48+00
MMH 134	UNDEF	129.	135.	25.4	27.7	.316+05	.854+05	0.	.18+06	.43+06	1.0	14.8	29.9	.01	.11	.30-01
MMH 135	UNDEF	133.	107.	14.4	14.4	.252+05	.693+05	0.	.13+06	.35+06	1.0	16.5	32.1	.01	.16	.40-01
MMH 136	UNDEF	129.	86.	7.3	7.7	.372+05	.571+05	0.	.19+06	.29+06	1.1	17.7	33.5	.01	.22	.53-01
MMH 137	UNDEF	131.	102.	46.4	50.3	.373+05	.108+06	0.	.19+06	.54+06	1.0	17.3	34.6	.01	.11	.28-01
MMH 138	UNDEF	124.	141.	46.7	50.4	.851+05	.938+05	0.	.43+06	.47+06	21.9	19.5	2.2	.18	.16	.17+00
MMH 139	UNDEF	124.	146.	44.7	50.4	.111+06	.947+05	0.	.50+06	.47+06	35.5	20.0	1.5	.29	.16	.22+00
MMH 140	UNDEF	125.	150.	49.6	43.4	.116+06	.125+06	0.	.58+06	.63+06	41.6	48.2	2.8	.33	.39	.36+00

ORIGINAL PAGE IS
OF POOR QUALITY

HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR OHS-75

FUEL TEST TYPE NO.	INJECTOR TYPE	DD (IN)	DF (IN)	L/D	IMP ANGLE (DEG)	PC (PBIA)	VO (FT/S)	VF (FT/S)	TO (F)	TF (F)	MR	MF/MO	MODE	REACT (SEC)
MMH 141	L-O-L BARRIER	.025	.025	5.	1.	122.	140.7	108.3	126.	312.	1.40	.955	UNDEF	.498-03
MMH 142	L-O-L CORE	.027	.025	5.	1.	126.	124.7	145.6	74.	86.	1.66	.823	SEP	.137-04
MMH 143	L-O-L CORE	.027	.025	5.	1.	101.	100.2	118.0	76.	81.	1.63	.839	M/S	.116-04
MMH 144	L-O-L CORE	.027	.025	5.	1.	81.	83.6	95.9	80.	82.	1.68	.797	SEP	.998-05
MMH 145	L-O-L CORE	.027	.025	5.	1.	153.	153.8	170.8	80.	76.	1.73	.750	SEP	.167-04
MMH 146	L-O-L CORE	.027	.025	5.	1.	124.	127.5	155.3	72.	191.	1.70	.835	SEP	.650-04
MMH 147	L-O-L CORE	.027	.025	5.	1.	123.	130.0	164.4	74.	252.	1.72	.858	SEP	.136-03
MMH 148	L-O-L CORE	.027	.025	5.	1.	123.	136.9	165.9	123.	263.	1.73	.815	SEP	.272-03
MMH 149	L-O-L CORE	.027	.025	5.	1.	147.	149.9	210.0	128.	311.	1.72	.866	SEP	.561-03
MMH 150	VDI-A CORE	.027	.042	9.	0.	114.	42.8	54.9	85.	95.	1.62	.994	SEP	.211-04
MMH 151	VDI-A CORE	.027	.042	9.	0.	95.	48.5	66.2	85.	176.	1.59	1.075	SEP	.715-04
MMH 152	LAM L-O-L CORE	.027	.025	5.	1.	127.	125.6	149.5	80.	82.	1.61	.859	SEP	.156-04
MMH 153	LAM L-O-L CORE	.027	.025	5.	1.	100.	99.9	122.0	82.	84.	1.57	.906	SEP	.134-04
MMH 154	LAM L-O-L CORE	.027	.025	5.	1.	80.	83.0	97.2	83.	86.	1.64	.832	M/S	.111-04
MMH 155	LAM L-O-L CORE	.027	.025	5.	1.	152.	154.0	168.1	82.	85.	1.76	.723	SEP	.187-04
MMH 156	LAM L-O-L CORE	.027	.025	5.	1.	100.	103.3	118.5	82.	81.	1.68	.797	SEP	.136-04
MMH 157	LAM L-O-L CORE	.027	.025	5.	1.	126.	128.0	142.0	81.	83.	1.73	.747	SEP	.152-04
MMH 158	LAM L-O-L CORE	.027	.025	5.	1.	126.	127.7	160.6	74.	185.	1.71	.831	SEP	.685-04
MMH 159	LAM L-O-L CORE	.027	.025	5.	1.	125.	124.7	141.2	77.	92.	1.71	.770	SEP	.168-04
MMH 160	LAM L-O-L CORE	.027	.025	5.	1.	125.	125.1	140.9	78.	89.	1.72	.765	SEP	.157-04
MMH 161	LAM L-O-L CORE	.027	.025	5.	1.	81.	86.6	92.1	82.	91.	1.82	.683	M/S	.112-04
MMH 162	LAM L-O-L CORE	.027	.025	5.	1.	79.	84.4	92.0	83.	86.	1.76	.718	M/S	.109-04
MMH 163	LAM L-O-L CORE	.027	.025	5.	1.	127.	126.5	140.3	76.	94.	1.48	1.202	UNDEF	.683-05
MMH 164	LAM L-O-L CORE	.027	.025	5.	1.	127.	128.3	140.3	76.	94.	1.61	1.087	UNDEF	.618-05
MMH 165	LAM L-O-L CORE	.027	.025	5.	1.	127.	128.3	140.3	76.	94.	1.61	1.087	M/S	.618-05
MMH 166	LAM L-O-L CORE	.027	.025	5.	1.	127.	128.3	140.3	76.	94.	1.61	1.087	UNDEF	.689-05
MMH 167	LAM L-O-L CORE	.027	.025	5.	1.	127.	128.3	140.3	76.	94.	1.61	1.087	UNDEF	.711-05
MMH 168	VDI-A CORE	.024	.021	1.	0.	127.	126.5	140.3	76.	94.	1.64	1.053	M/S	.721-05
MMH 169	VDI-A CORE	.024	.021	1.	0.	127.	126.5	140.3	76.	94.	1.64	1.053	M/S	.721-05
MMH 170	VDI-A CORE	.024	.021	1.	0.	127.	126.5	140.3	76.	94.	1.64	1.053	M/S	.721-05
MMH 171	VDI-A CORE	.024	.021	1.	0.	127.	126.5	140.3	76.	94.	1.64	1.053	M/S	.721-05
MMH 172	VDI-A CORE	.024	.021	1.	0.	127.	126.5	140.3	76.	94.	1.64	1.053	M/S	.721-05
MMH 173	VDI-A CORE	.024	.021	1.	0.	127.	126.5	140.3	76.	94.	1.64	1.053	M/S	.721-05
MMH 174	VDI-A CORE	.024	.021	1.	0.	127.	126.5	140.3	76.	94.	1.64	1.053	M/S	.721-05
MMH 222	VDI-A CORE	.027	.042	9.	0.	124.	42.4	52.2	76.	75.	1.68	.916	SEP	.131-04
MMH 223	VDI-A CORE	.027	.042	9.	0.	101.	35.5	42.7	77.	70.	1.72	.876	SEP	.110-04
MMH 224	VDI-A CORE	.027	.042	9.	0.	81.	29.1	34.8	78.	71.	1.73	.868	M/S	.926-05
MMH 225	VDI-A CORE	.027	.042	9.	0.	149.	52.4	63.0	78.	71.	1.72	.876	SEP	.167-04
MMH 226	VDI-A CORE	.027	.042	9.	0.	124.	43.8	56.2	83.	169.	1.72	.936	SEP	.742-04
MMH 227	VDI-A CORE	.027	.042	9.	0.	123.	43.7	58.7	80.	235.	1.70	.992	SEP	.120-03
MMH 228	VDI-A CORE	.027	.042	9.	0.	124.	45.6	58.6	125.	242.	1.72	.937	SEP	.221-03

HIGH PERFORMANCE N204 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGEMENT DATA COMPILATION

INVESTIGATION OMS-75

FUEL TEST TYPE NO.	MODE	PC (PSIA)	WAVG	REF	NEU	REF	REU	DELTA (DEG F)	SELF	RELU	PPF (/SIA)	HRVP	XF	XO	XP	RESU
MMH 141	UNDEF	122.	101.	73.0	44.9	193+06	130+06	0.	.97+06	.65+06	10.0	1.4	.82	.45	.61+00	.11-04
MMH 142	SEP	126.	133.	20.1	29.2	326+05	926+05	0.	.16+06	.46+06	1.1	29.7	.01	.13	.34+01	.14-04
MMH 143	M/S	101.	107.	14.8	15.3	267+05	753+05	0.	.13+06	.36+06	1.1	30.4	.01	.17	.43+01	.16-04
MMH 144	SEP	81.	86.	7.4	8.7	218+05	644+05	0.	.11+06	.32+06	1.2	32.3	.01	.23	.56+01	.22-04
MMH 145	SEP	153.	160.	46.8	55.7	376+05	118+06	0.	.19+06	.59+06	1.1	35.7	.01	.12	.29+01	.12-04
MMH 146	SEP	124.	136.	30.5	24.8	765+05	937+05	0.	.36+06	.47+06	15.6	2.5	.13	.13	.1+00	.13-04
MMH 147	SEP	123.	143.	49.7	31.0	118+06	966+05	0.	.59+06	.44+06	44.6	16.5	.10	.36	.13	.22+00
MMH 148	SEP	123.	148.	51.0	45.5	127+06	135+06	0.	.63+06	.67+06	53.0	1.4	.43	.42	.43+00	.13-04
MMH 149	SEP	127.	181.	109.4	81.4	214+06	167+06	0.	.11+07	.83+06	99.3	1.5	.66	.39	.51+00	.09-05
MMH 150	SEP	114.	47.	6.2	5.8	234+05	592+05	0.	.21+06	.53+06	1.7	25.9	.01	.19	.53+01	.64-04
MMH 151	SEP	95.	55.	8.6	6.2	479+05	671+05	0.	.43+06	.60+06	10.2	4.9	.11	.23	.16+00	.53-04
MMH 152	SEP	127.	135.	30.0	30.8	391+05	967+05	0.	.17+06	.48+06	1.2	32.3	.01	.15	.37+01	.14-04
MMH 153	SEP	100.	104.	15.8	15.5	203+05	778+05	0.	.14+06	.39+06	1.2	32.4	.01	.20	.50+01	.17-04
MMH 154	M/S	80.	88.	8.0	8.6	220+05	651+05	0.	.11+06	.33+06	1.3	31.8	.02	.26	.65+01	.21-04
MMH 155	SEP	152.	159.	45.6	56.1	353+05	120+06	0.	.20+06	.60+06	1.3	20.0	.01	.13	.53+01	.12-04
MMH 156	SEP	100.	109.	14.9	16.6	281+05	805+05	0.	.14+06	.40+06	1.3	20.0	.01	.20	.52+01	.18-04
MMH 157	SEP	127.	133.	27.1	32.2	326+05	991+05	0.	.16+06	.50+06	1.2	19.5	.01	.15	.36+01	.15-04
MMH 158	S/F	126.	137.	36.3	30.3	748+05	943+05	0.	.37+06	.47+06	15.0	32.4	.01	.15	.36+01	.15-04
MMH 160	S/P	126.	140.	44.9	31.0	975+05	960+05	0.	.49+06	.48+06	20.2	2.7	.12	.13	.12+00	.14-04
MMH 161	S/P	125.	143.	45.1	45.4	106+06	134+06	0.	.53+06	.67+06	35.5	1.5	.25	.14	.16+00	.13-04
MMH 162	SEP	148.	175.	96.5	78.6	187+06	164+06	0.	.93+06	.82+06	80.9	3.7	.28	.44	.37+00	.13-04
MMH 163	SEP	125.	131.	26.5	30.1	353+05	949+05	0.	.18+06	.47+06	1.6	17.3	.01	.14	.43+01	.15-04
MMH 164	S/P	125.	131.	26.7	29.5	350+05	943+05	0.	.17+06	.47+06	1.5	17.7	.01	.14	.42+01	.15-04
MMH 165	SEP	125.	131.	26.5	29.8	360+05	932+05	0.	.17+06	.48+06	1.4	18.1	.01	.15	.40+01	.15-04
MMH 166	M/S	61.	69.	7.4	9.5	226+05	675+05	0.	.11+06	.34+06	1.5	20.0	.02	.25	.67+01	.23-04
MMH 167	M/S	79.	87.	7.1	8.4	222+05	662+05	0.	.11+06	.33+06	1.4	20.5	.02	.26	.66+01	.23-04
MMH 168	UNDEF	127.	67.	7.7	5.6	166+05	394+05	0.	.17+05	.39+05	1.4	20.5	.01	.16	.42+01	.21-04
MMH 169	UNDEF	127.	62.	6.1	5.3	149+05	391+05	0.	.15+05	.38+05	1.4	20.5	.01	.16	.42+01	.21-04
MMH 170	M/S	127.	62.	6.1	5.3	149+05	391+05	0.	.15+05	.38+05	1.4	20.5	.01	.16	.42+01	.21-04
MMH 171	UNDEF	126.	61.	5.4	5.3	152+05	394+05	0.	.16+05	.39+05	1.6	21.1	.01	.17	.47+01	.24-04
MMH 172	M/S	126.	62.	6.1	5.4	156+05	394+05	0.	.16+05	.39+05	1.6	21.6	.01	.17	.49+01	.24-04
MMH 173	M/S	126.	62.	6.1	5.4	158+05	392+05	0.	.16+05	.39+05	1.8	21.6	.01	.17	.50+01	.24-04
MMH 174	M/S	125.	63.	6.2	5.5	162+05	394+05	0.	.16+05	.39+05	1.9	21.6	.01	.17	.51+01	.24-04
MMH 222	S/P	124.	46.	5.9	5.9	168+05	555+05	0.	.17+06	.50+06	1.0	17.3	.01	.14	.33+01	.67-04
MMH 223	S/P	111.	51.	3.2	3.4	155+05	467+05	0.	.14+06	.42+06	1.0	17.7	.01	.18	.72+01	.82-04
MMH 224	M/S	111.	51.	3.7	1.8	124+05	366+05	0.	.11+06	.35+06	1.0	18.1	.01	.22	.53+01	.10-03
MMH 225	S/P	149.	56.	10.4	10.9	231+05	694+05	0.	.21+06	.62+06	1.0	18.1	.01	.12	.29+01	.56-04
MMH 226	S/P	123.	49.	10.4	6.5	559+05	598+05	0.	.41+06	.54+06	15.0	20.5	.01	.17	.14+00	.62-04
MMH 227	S/P	123.	49.	10.4	6.5	625+05	595+05	0.	.41+06	.54+06	32.1	19.0	.01	.15	.20+00	.60-04
MMH 228	S/P	124.	50.	10.5	8.9	603+05	709+05	0.	.60+06	.71+06	37.0	54.1	.01	.44	.36+00	.60-04

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OF POOR QUALITY

HIGH PERFORMANCE N2O4 / AMINE ELEMENTS

HYPERGOLIC STREAM IMPINGMENT DATA COMPILATION

INVESTIGATOR UMS-75

FUEL TEST NO.	INJECTION TYPE	DD (IN)	DP (IN)	L/D	IMP ANGLE (DEG)	PC (PSIA)	VO (FT/S)	VF (FT/S)	TU (F)	TF (F)	MR NF/MO	MODE	REACT (SEC)
MMH 229 VOT-A	CURE	.047	.042	9.	0.	150.	54.5	71.5	127.	275.	1.73	.951	.385-03
MMH 230 VOT-A	CURE	.047	.042	9.	0.	124.	43.8	55.4	89.	185.	1.73	.917	.753-04
MMH 231 VOT-A	CURE	.047	.042	9.	0.	124.	43.7	52.6	78.	95.	1.73	.871	.185-04
MMH 232 VOT-A	HARMIER	.042	.042	10.	0.	126.	45.0	51.7	79.	14.	1.44	.799	.154-04
MMH 233 VOT-A	HARMIER	.042	.042	10.	0.	103.	37.3	42.1	76.	83.	1.87	.769	.124-04
MMH 234 VOT-A	HARMIER	.042	.042	10.	0.	80.	29.7	34.0	78.	81.	1.46	.769	.961-05
MMH 235 VOT-A	HARMIER	.042	.042	10.	0.	152.	54.8	60.6	75.	60.	1.50	.738	.163-04
MMH 236 VOT-A	HARMIER	.042	.042	10.	0.	127.	45.3	50.6	75.	79.	1.48	.754	.134-04
MMH 237 VOT-A	HARMIER	.042	.042	10.	0.	122.	100.0	112.9	77.	77.	1.70	.773	.105-04
MMH 238 VOT-A	HARMIER	.042	.042	10.	0.	102.	84.8	92.9	77.	77.	1.75	.728	.865-05
MMH 239 VOT-A	HARMIER	.042	.042	10.	0.	80.	67.7	77.2	77.	77.	1.69	.790	.719-05
MMH 240 VOT-A	HARMIER	.042	.042	10.	0.	151.	125.0	139.4	77.	77.	1.73	.754	.130-04
MMH 241 VOT-A	HARMIER	.042	.042	10.	0.	125.	102.8	114.5	77.	77.	1.72	.753	.107-04
MMH 242 VOT-A	HARMIER	.042	.042	10.	0.	124.	104.1	122.3	88.	177.	1.72	.786	.531-04
MMH 243 VOT-A	HARMIER	.042	.042	10.	0.	125.	103.6	126.2	88.	214.	1.71	.829	.847-04
MMH 244 VOT-A	HARMIER	.042	.042	10.	0.	125.	107.7	126.1	117.	223.	1.73	.816	.931-04
MMH 245 VOT-A	HARMIER	.042	.042	10.	0.	151.	130.6	160.8	124.	279.	1.73	.831	.132-03
MMH 246 VOT-A	HARMIER	.042	.042	10.	0.	124.	104.6	118.6	85.	65.	1.66	.790	.108-03
MMH 247 VOT-A	HARMIER	.042	.042	10.	0.	124.	107.9	116.9	85.	86.	1.74	.713	.172-04
MMH 248 VOT-A	HARMIER	.042	.042	10.	0.	101.	86.6	95.2	85.	86.	1.72	.735	.140-04
MMH 249 VOT-A	HARMIER	.042	.042	10.	0.	82.	68.4	75.3	84.	66.	1.72	.735	.109-04
MMH 250 VOT-A	HARMIER	.042	.042	10.	0.	151.	129.1	141.5	84.	85.	1.72	.729	.203-04
MMH 251 VOT-A	HARMIER	.042	.042	10.	0.	124.	107.9	124.2	87.	173.	1.73	.765	.836-04
MMH 252 VOT-A	HARMIER	.042	.042	10.	0.	151.	129.1	141.4	87.	94.	1.74	.724	.234-04
MMH 253 VOT-A	HARMIER	.042	.042	10.	0.	122.	109.3	132.6	87.	240.	1.72	.808	.140-03
MMH 254 VOT-A	HARMIER	.042	.042	10.	0.	123.	109.1	129.9	88.	229.	1.74	.786	.127-03
MMH 255 VOT-A	HARMIER	.042	.042	10.	0.	123.	111.1	131.2	87.	328.	1.75	.773	.125-03
MMH 256 VOT-A	HARMIER	.042	.042	10.	0.	123.	113.2	131.1	124.	231.	1.73	.767	.146-03
MMH 257 VOT-A	HARMIER	.042	.042	10.	0.	152.	134.9	162.6	127.	280.	1.73	.798	.906-03
MMH 258 VOT-A	HARMIER	.042	.042	10.	0.	122.	133.4	162.0	125.	276.	1.72	.811	.383-03
MMH 259 VOT-A	HARMIER	.042	.042	10.	0.	127.	45.3	53.8	89.	90.	1.71	.876	.193-04
MMH 260 VOT-A	HARMIER	.042	.042	10.	0.	127.	45.3	53.8	89.	90.	1.74	.857	.196-04
MMH 261 VOT-A	HARMIER	.042	.042	10.	0.	80.	30.0	34.7	90.	90.	1.78	.815	.128-04
MMH 262 VOT-A	HARMIER	.042	.042	10.	0.	80.	29.8	35.1	85.	69.	1.74	.844	.126-04
MMH 263 VOT-A	HARMIER	.042	.042	10.	0.	80.	29.3	35.5	85.	90.	1.70	.891	.129-04
MMH 264 VOT-A	HARMIER	.042	.042	10.	0.	80.	29.3	35.3	88.	89.	1.71	.882	.126-04
MMH 265 VOT-A	HARMIER	.042	.042	10.	0.	124.	45.4	52.6	88.	89.	1.78	.817	.187-04
MMH 266 VOT-A	HARMIER	.042	.042	10.	0.	124.	45.4	53.0	88.	89.	1.76	.829	.189-04

LIST OF APPENDIX E

DATA SOURCES

1. Investigator - Lawver
Lawver, B. R., "High Performance N_2O_4 /Amine Elements
"Blowapart" Report 14186-DRL-3-1, Contract NAS 9-14186
November 1974
2. Investigator - RCKTD
Hines, W. S., "High Performance N_2O_4 /Amine Elements
Task III Definition of Governing Mechanisms Data Dump",
Report 75 RC1635, Contract NAS9-14126, Rocketdyne,
March 1975
3. Investigator - Zung - From Reference 27
Zung, L. B., and White, S. R., "Combustion Process
of Impinging Hypergolic Propellants", NASA CR 1704
Mardhall Industries, Irvine California, May 1971
4. Investigator - OMS-75
Ito, J. I., "Development of Test Report OMS Injector
Subscale Pattern Evaluation", Report No. 6673:207
PDRD TM05-25, Contract M4JXMA-483030H, ALRC, March
1976